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Canadä

Viscoelastic Properties of Paper in a Calender Nip

by
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Montreal

August 1994

A thesis submitted to the Faculty of Graduate Studies and Research in partial fulfillment of the requirements for the degree of Doctor of Philosophy

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ISBN 0-612-05681-3

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In memory of Lawrence and Roula

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Abstract

Paper in a calender nip is subjected to a compressive pressure pulse which reduces its thickness and roughness, thus improving the product quality. Strain recovery after the nip is time-dependent, and some permanent strain remains. The material behaviour is viscoelastic; recovery depends on the pulse magnitude and duration, and does not occur immediately on exiting the nip. An improved description of the viscoelastic response to a calender pulse would allow design of adaptive control systems using feedforward techniques.

Measurements of paper thickness in the nip, immediately after the nip and 24 hours after calendering were made with newsprint sheets running through an experimental calender operating at industrial conditions. The calculated strains were first related to the operating conditions using empirical curve-fitting methods, then using linear viscoelastic models. Empirical results describe the data well, and can be used to design improved control systems. Linear viscoelastic modeling was less successful since the material behaviour is not linear. Photomicrographs of sheet cross-sections were taken, and the observed non-linearities were discussed qualitatively in terms of paper and fibre properties.

Résumé

Dans une calandre, l'épaisseur et la rugosité d'une feuille de papier sont réduites par l'application d'une charge en compression. Ce procédé améliore la qualité du produit. La feuille subit une grande déformation dans la pince, puis recouvre une partie de l'épaisseur originale à la sortie. Cette récupération n'est pas instantanée, mais requiert un certain temps. Le papier se comporte donc de façon viscoélastique. Une description améliorée des caractéristiques viscoélastiques du papier servirait à la conception de systèmes de contrôle adaptifs pour l'épaisseur de la feuille à la sortie de la calandre.

Les déformations dans la pince, à deux endroits immédiatement après la pince et 24 heures après le calandrage ont étés mesurées sur une calandre échèlle laboratoire reproduisant des conditions industrielles typiques. Le papier en question était un papier journal du Québec. Les déformations sont décrites en premier lieu avec des relations empiriques, puis avec des modèles viscoélastiques linéaires. Les relations empiriques décrivent bien les déformations, et peuvent maintenant servir à la conception de systèmes de contrôle améliorés. Par contre, les modèles viscoélastiques se sont montrés moins utiles puisque le papier ne se prête pas facilement à une analyse linéaire. Des photomicrographies de sections de feuilles ont été prises, et l'aspect non-linéaire du papier est discuté en termes qualitatifs à partir des propriétés fondamentaux du papier et de la fibre.

Acknowledgements

It is now my pleasant duty to thank all the people who had a hand, however indirect, in the completion of this research project. Hopefully, I haven't forgotten anyone; if I have, please accept my apologies.

First, I wish to thank my advisors, Dr. R.H. Crotogino and Dr. W.J.M Douglas, for their continued technical and financial support, without which this thesis would not have been possible.

The calendering specialists at Paprican's Pointe Claire laboratory invariably had time for discussion. For their time and encouragement, I wish to extend my thanks to Dave McDonald, Michel Gratton, Jean Hamel, Larry Gee and Steve Smallwood.

James Drummond, of Paprican's Vancouver laboratory, prepared the samples and took the micro-photographs presented in Chapter 5.

The electronics and machine shop staffs at McGill University's Department of Chemical Engineering and at Paprican's Pointe Claire laboratory were unfailingly helpful in maintaining the complex experimental equipment necessary for this work.

The following people offered continuous moral support and encouragement, and their various contributions are gratefully acknowledged:

Anne Abbey, Sue Adams, Korkor Amarteifio, Jean-François Bond, Catherine Browne, Marie Carlson Browne, Michael Browne, Robert Browne, Anna Cicciarelli, Charles Dolan, Johanne Duchêne, Joseph Grassi, Dr. R.J. Kerekes, Bill Kreklewetz, Spiros Lazaris, Rebecca MacIntyre, Rob McCleave, Roula Najjar, Antoinette Paventi, Dr. I.I. Pikulik, Dr. I.T. Pye, Lawrence Ramsay, Patti Turner, Michael Weir, Domenico and Angela Zampini, Lucy Zampini, Phil Zampini, Rita Zampini.

Last but certainly not least, I wish to thank my family, Mary, Tara and Alexander, for their moral support and understanding. I would not have been able to finish this project without their help.

Nomenclature

The number in parentheses gives the Chapter where the variable is defined.

```
Ingoing nip length, mm (2.2)
a
                       Roughness compensated ingoing nip length, mm (5.3.2)
a_{\rm c}
                       Intermediate variable for Burger's model (5.2.2)
A
                        Calendering equation coefficients (2.1)
A_{ii}, a_{ii}
                               Subscripts:
                               i: o offset, L Load, M Moisture content, P Pressure,
                                       S Sheet speed, R Radius, t time, \theta temperature.
                               j: n in-nip, r recovered, p permanent.
                        Second order calendering equation coefficient (4.1.1)
A_2, a_{o2}, a_{L2},
                        Coefficients for density equation (4.1.1)
        a_{R2}, a_{S2}
B_i, B_p
                        Initial and permanent (fully recovered) paper bulk, cm<sup>3</sup>/g (2.1)
                        Outgoing nip length, mm (4.2.1)
b
                        Largest possible value of b, mm (4.2.1)
b_{
m max}
                        Roll circumference, m (3.2)
CD
                        Cross-machine direction in a paper machine (1)
                        Coefficient of variation (4.1.1)
\mathbf{C}_{\mathbf{V}}
                        Curve fitting coefficients (4.2.3)
c, c<sub>0</sub>, c<sub>1</sub>
                        Elastic modulus, Pa (2.1)
E_{xx}, E_{yy}, E_{zz}
                        Elastic moduli, GPa (2.2)
F(\varepsilon)
                        Shape factor (2.1)
G, G<sub>e</sub>
                        Spring moduli, Pa (2.2)
G_K,\,G_M
                        Spring moduli, Pa (5.2.2)
G_{xx}, G_{yy}, G_{zz}
                        Shear moduli, GPa (2.2)
                        Intermediate variables for Burger's model (5.2.2)
K_1, K_2
k_1, k_2, k_3
                        Curve fitting coefficients (5.2.3)
L
                        Nip load, kN/m (2.1)
                        Curve fitting constant (4.2.2)
L_{lim}
                        Paper moisture content, % (2.1)
M
M_{50}
                        Paper moisture content corrsponding to 50% R.H., % (3.2)
                        Machine direction in a paper machine (1)
MD
m, n
                        Curve fitting constants (4.2.1)
                        Curve fitting constants (4.2.2)
m_1, m_2
                        Intermediate variables for Burger's model (5.2.2)
p<sub>1</sub>, p<sub>2</sub>
                        Nip pressure, Pa (2.1)
P(x,z)
                        Point on the roll surface with coordinates x,z (5.2.1)
                        Intermediate variables for Burger's model (5.2.2)
q_1, q_2
                        Roll radius, m (2.1)
R
                        Intermediate variables for Burger's model (5.2.2)
r_1, r_2
                        Sheet speed, m/min (2.1)
```

```
S.E.
                            standard error (4.1.1)
T
                            Total data acquisition time, s (3.2)
                            Initial, in-nip and recovered paper thicknesses, \mu m (2.1)
T_o, T_n, T_r
                            Elapsed time, s (2.1)
                            Dwell times, s (2.2)
t_1, t_2
t<sub>e</sub>
                            Nip dwell time, s (5.2.2)
                            Variable of integration (2.2)
u, u_o
                            Upper roll position, \mu m (3.1.2)
                            Lower roll position, \mu m (3.1.2)
\nu, \nu_{o}
                            Volume of air or fibre in a paper sheet (5.3.3)
                            Sheet velocity, m/s (4.2.1)
W
                            Machine width, m (5.2.3)
W_{\Delta r}
                            Roll deformation width, mm (2.1)
                            Distance in the machine direction, m (3.2)
X
                            An arbitrary in-plane direction (5.3.2)
\mathbf{x}_1
                            Thickness direction in a paper machine (1)
z
z(t)
                            Paper surface position, \mu m (5.2.1)
                            Roughness-compessated sheet half-thickness, \mu m (5.3.2)
z_{\rm c}
                            Initial, in-nip, exit and permanent sheet half-thickness, \mum (4.2.1)
Z_i, Z_n, Z_e, Z_p
                            Roll indentation, \mu m (5.2.1)
z_{o}
                            Roughness height, \mu m (5.3.2)
Z,
\beta, \beta_{\text{max}}
                            Nip length ratios (5.2.2)
                            Machine drive torque, N m (5.2.3)
\delta(t)
                            Unit impulse function (5.2.2)
\Delta r_p \Delta T
                            Peak roll radius deformation, \mu m (2.1)
                            Change in paper thickness, \mu m (2.1)
                            Roughness compensated initial strain (5.3.2)
\varepsilon_{\rm c}
                            Corrected paper strain (3.2)
\varepsilon_{\rm corr}
                            Paper strain at the nip exit (4.2.1)
\varepsilon_{\rm e}
                            Instantaneous elastic strain recovery (5.2.3)
\varepsilon_{
m el}
                            Strain in a Kelvin element (5.2.3)
\epsilon_{\mathrm{K}}
                            In-nip, partially recovered and permanent paper strains (2.1)
\varepsilon_{\rm n}, \ \varepsilon_{\rm r}, \ \varepsilon_{\rm p}
                            Critical strain (2.1)
\varepsilon_{N}
                            Step strain input (2.2)
\varepsilon_0, \ \varepsilon_1
                            Curve fitting constants (4.2.2)
\varepsilon_0, \varepsilon_{\infty}
                            Recoverable strain (4.2.2)
\varepsilon_{\mathbf{R}}
\varepsilon_1(t)
                            In-nip strain history (5.2.1)
                            Post-nip strain history (5.2.1)
\varepsilon_2(t)
                            Viscosity, Pa s (2.2)
                            Viscosities, Pa s (5.2.2)
\eta_{K}, \eta_{M}
                            Sheet temperature, *C (2.1)
                            Nip intensity coefficient (2.1)
                            In-nip, recovered and permanent intensity coefficients (2.1)
\mu_{\rm n}, \ \mu_{\rm r}, \ \mu_{\rm p}
```

$\mu_{\rm pl},~\mu_{\rm p2}$	Second order calendering equation nip intensities (4.1.1)
	Nip intensity for density equation (4.1.1)
μ ₂ ξ	Devorah number (2.2)
ξ_1, ξ_2	Devorah numbers for Burger's model (5.2.2)
ξ_{K}, ξ_{M}	Devorah numbers for Burger's model (5.2.2)
ρ	Paper density, g/cm ³ (4.1.1)
$ ho_{ m f}, ho_{ m p}$	Fibre and paper paper density, g/cm ³ (2.1)
ρ_0, ρ^*	Initial and maximum paper density, g/cm ³ (2.1)
$(\rho_{\rm n})_{\rm max}, (\rho_{\rm p})_{\rm max}$	Maximum in-nip and permanent paper densities, g/cm ³ (4.1.2)
$\rho_{xy}, \rho_{xz}, \rho_{yx}, \rho_{yz},$	
$ ho_{\sf zx}, \ ho_{\sf zy}$	Poisson's ratios (2.2)
σ	Compressive stress, Pa (2.1)
$\sigma_0, \ \sigma_1$	Step stress input, Pa (2.2)
$\sigma_{\mathbf{K}}(t)$	Stress imposed on a Kelvin element, Pa (5.2.3)
$\sigma_{V}(t)$	Stress imposed on a viscous component, Pa (5.2.3)
$\phi(t), \ \phi_1(t), \ \phi_2(t),$	
$\phi_3(t)$	Strain retardation functions, Pa ⁻¹ (2.2)
$\psi(t), \psi_1(t), \psi_2(t),$	
$\psi_3(t)$	Stress relaxation functions, Pa (2.2)
τ	Time constant, s (2.2)
$ au_1, \ au_2$	Time constants for Burger's model (5.2.2)
$ au_{\mathrm{K}}, au_{\mathrm{M}}$	Time constants for Burger's model (5.2.2)

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1. Introduction

For most grades of paper, the last processing step is calendering. This operation reduces paper thickness and roughness by pressing the sheet between two or more large cast iron rolls. The high loads encountered in the nip between two smooth calender rolls flatten high spots in the rough sheet by permanently deforming wood fibres on the surface of the sheet, thus reducing the roughness of the sheet. Fibres inside the sheet are also deformed, reducing the thickness. The process is the papermaker's last chance to reduce thickness variations along the length and width of the finished sheet. A smoother sheet results in improved print quality, while more uniform thickness profiles improve the winding process and reduce sheet breaks in printing presses. The calendering operation thus improves the quality of the finished product.

In the Cartesian coordinate system defined for paper machines, the direction of paper travel is called the machine direction (MD), the direction across the width of the machine the cross-machine direction (CD), and the direction perpendicular to the paper plane the z-direction. Specific volume, referred to in the paper industry as paper bulk, cm³/g, is the ratio of thickness (or z-direction dimension, in μ m) to basis weight (mass per unit area, in g/m²), and is the inverse of apparent density. Thickness is typically reduced by calendering; since MD and CD dimensional changes due to calendering are typically several orders of magnitude smaller than the corresponding z-direction strain, basis weight changes are usually small and the thickness reduction is essentially equal to the bulk reduction.

While surface properties are mainly of interest to the printer, the papermaker can most easily measure thickness or bulk on-line. The relationships between thickness reduction and surface properties such as roughness and gloss have been well documented in the past [19, 24, 26]; thus any change in thickness reduction can be related to changes in paper properties by referring to published work. The present work therefore refers only to thickness reduction, bulk reduction and strain.

The amount of bulk reduction depends on several process variables. Load,

machine speed and roll radius define the magnitude and duration of the applied pressure pulse. These variables, along with the number of nips, determine the work done on the paper. Bulk reduction also depends on paper properties such as initial bulk, moisture content and in-nip temperature, which determine paper response to a calender pulse. The effects of these calendering variables on the CD average bulk reduction have been quantified and can be calculated by means of the calendering equation [18, 22, 23, 24, 76].

Paper bulk at the entrance to a calender may vary both in the machine and crossmachine directions. Machine direction variations can be reduced in the calender by altering the average cross-direction load. Hydraulic loading systems respond relatively quickly, so real-time control of long-wavelength machine direction thickness or bulk variations is straight-forward.

Cross-machine variations can be minimized by varying the local roll radius profile, thus changing the local pressure profile. Commonly, a larger local roll radius for more bulk reduction is obtained by heating the roll locally using impinging hot air jets, induction coils or infrared heaters. Higher local temperatures result in small radial deformations of the roll surface. While roll diameter is typically of the order of 1 meter, paper thickness entering the calender is of the order of 10⁻⁴ m; thus a radial roll deformation of a few microns, corresponding to a radial strain of the order of 10⁻⁶, is very small compared to the scale of the roll but enormous when measured on the scale of the paper, where the same roll deformation imposes a z-direction strain on the paper of the order of 10⁻². The cross-direction radius profile resulting from a given temperature profile can be calculated [41, 42, 68, 69] given details of the heating system and the roll geometry. Unlike hydraulic loading systems, thermal methods for CD control are slow due to the high thermal inertia of a typical calender roll, leading to significant delays in implementing a control action.

Web temperature and moisture content can vary across the width of a paper web as it enters the calender, often due to uneven conditions in the dryer section. These variations can cause roll temperature variations as the sheet wraps around the roll, in turn

causing further local roll radius changes. Along with variations in initial bulk and control actions taken by the papermaker, these thermal deformations result in cross-direction variations in the load distribution which cannot be taken into account using the calendering equation since local values of roll radius and nip load are not easily measured. In order to determine the local bulk reduction or the final bulk profile using the calendering equation, either the load distribution or the radius distribution and its effect on the load must be known.

The roll temperature profile required for a given radius profile can be calculated; the means of achieving the desired profile is also known. However it has not been possible to determine the radius profile required to create a desired bulk profile since the rheological properties of paper in a rolling nip have not been known. A relationship between local stress and strain for paper in a calender nip is the final element required for a complete explicit description of cross-machine calender control.

This stress-strain relationship cannot easily be determined using simple methods since the behaviour of paper under compression is neither elastic nor viscous, but viscoelastic. Both the strain under load and the permanent residual strain after removal of the load depend not only upon the magnitude of the load, but also on the dwell time during which it was applied. Since the calendering process applies a roughly parabolic pressure profile which reaches peaks of several megapascals and lasts for only a few hundred microseconds, equipment such as bench-top platen presses are inadequate for this purpose.

The present work describes part of an ongoing research project whose goal is the development of a cross-machine control system for bulk leaving a calender. The project consists of three distinct but complementary steps. The first step focused on the unsteady state and steady state response of a rotating calender roll to heat transfer from an array of impinging hot air jets, and on the radius profile resulting from thermal deformation of the roll [41, 42, 68, 69]. The second step focused on describing the paper response to local variations in calendering conditions, and was itself divided into two parts: the first part [8, 9] was the design and construction of a new experimental paper calender

allowing accurate and repeatable measurement of the in-nip paper thickness under industrially relevant calendering conditions; the second part, the topic of this thesis, was a complete experimental program to make these local, in-nip thickness measurements using the new equipment, followed by empirical and viscoelastic modeling of paper response to local calendering conditions. The final step, now underway as the topic of a future thesis, is the combination of the two previous steps, actuator design and material response to an actuator input, into a complete description of cross-direction calender control.

2. Literature review

2.1 Paper calendering

Paper is compressed by a pressure pulse in a calender nip, then exhibits time-dependent partial recovery of its initial thickness. The in-nip and permanent deformations and their dependence on this pulse are the subject of the current work. Peel [66, 67] compiled comprehensive reviews of calendering practice and theory, updating an earlier review by Baumgarten et al. [3]. This first section of the literature review examines some of the more important results described by Peel and subsequent researchers.

The effect of a pressure pulse on paper thickness was investigated by Chapman and Peel [15] using a platen press. They derived empirical relationships for compressed and recovered paper thicknesses as functions of maximum pressure and pulse duration, which they called master creep relationships:

$$\frac{T_o - T_n}{T_o} = A_n (1 + \tanh \mu_n)$$

$$\frac{T_o - T_r}{T_o} = A_r (1 + \tanh \mu_r)$$
[2.1]

where the nip intensity coefficients μ_n and μ_r are given by

$$\mu_{n} = a_{on} + a_{Pn} \log_{10} P + a_{tn} \log_{10} t$$

$$\mu_{r} = a_{or} + a_{Pr} \log_{10} P + a_{tr} \log_{10} t$$
[2.2]

and where T_0 , T_n , T_r are the initial, in-nip and recovered paper thicknesses. The constants A, a_0 , a_p and a_t were obtained by curve fitting, and describe the effects of maximum applied pressure P and dwell time t. The shortest pulse duration was about 6 ms, which is an order of magnitude or more greater than the dwell time in a commercial calender, and the pressures applied were fairly constant over that duration; the results therefore describe the creep characteristics of paper rather than the response to a

calendering pulse. Furthermore, the effect of successive compressions was not analyzed. The work was extended by Colley and Peel [17] to include the effect of web temperature and moisture content.

Kerekes et al. [44, 46] modified these relationships to predict directly the thickness reduction in a calender in terms of nip load, roll radius and machine speed rather than maximum pressure and pulse duration, an advantage to the papermaker since load, speed and radius are easily measured on-line while maximum pressure and dwell time are not. The predictions were based on assumptions regarding the viscoelastic behaviour of paper, which will be discussed subsequently, and were verified using a laboratory-scale calender at speeds approaching industrial values. The equipment thus produced a pressure pulse much more typical of a commercial calender than the platen press used by Peel et al. [15, 17].

Haglund and Robertson [34] used optical methods to measure paper thickness in the nip of a small laboratory calender run at low speeds. Their roll diameters and sheet speeds were both extremely small compared to industrial practice, which limits the applicability of their results; nonetheless they found (in agreement with Colley and Peel [17]) that while initial density had an effect on bulk reduction, initial thickness did not. They suggested rewriting Equations 2.1 and 2.2 to include the ratio of initial density ρ_0 to the maximum density obtainable in the nip ρ^* :

$$\frac{\Delta T}{T_o} = \frac{1}{2} \left(1 - \frac{\rho_o}{\rho^*} \right) (1 + \tanh \mu)$$
 [2.3]

They show that this expression along with measured values of the parameter A from Chapman and Peel's work consistently predicts ρ^* to be similar to accepted values for the density of cellulose, which is presumably the limit in paper densification.

Crotogino [18] extended Kerekes' work [44] to include the effects of initial bulk. His calendering equation is an empirical relationship giving the average cross-machine recovered bulk reduction in terms of web speed, roll radius, CD average nip load and

CD average initial paper properties. Since it accounts for initial bulk, this CD average calendering equation may be used successively for multiple nips.

Much of the work of preceding authors is summarized in the calendering equation. The present work extends the CD average calendering equation to an in-nip calendering equation. The nature of the calendering equation will therefore be described in greater detail.

Permanent bulk reduction ε_p is a measure of strain:

$$\varepsilon_{p} = \frac{B_{i} - B_{p}}{B_{i}}$$
 [2.4]

where B_i and B_p are initial and permanent recovered bulks. Baumgarten [4], Krenkel [47], and Gay et al. [29] have shown that CD paper dimensional changes in calendering are small, reaching at most 1% under the most severe conditions of load and speed; work described in Section 4.5 will show that MD changes in length are equally small. The surface area and therefore the basis weight of a sheet are thus essentially unchanged by calendering even at high loads and low speeds. Recalling that bulk is thickness divided by basis weight, and that basis weight is essentially constant, ε_p as defined by Equation 2.4 is an engineering strain, with positive values corresponding to a thickness reduction. The calendering equation, illustrated in Figure 2.1, relates bulk reduction in terms of either B_i or ε_p to the nip load L in kN/m, machine speed S in m/min, roll radius R in m, paper temperature θ in *C and moisture content M in %:

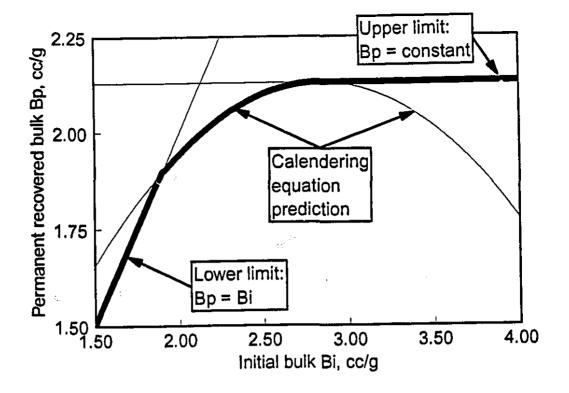
$$\varepsilon_{p} = A + \mu B_{i}$$
 [2.5]

where

$$\mu = a_0 + a_1 \log_{10} L + a_2 \log_{10} S + a_2 \log_{10} R + a_0 \theta + a_M M$$
 [2.6]

The constants are all furnish dependent and must be determined experimentally.

Equations 2.5 and 2.6 are valid for a range of B_i defined by



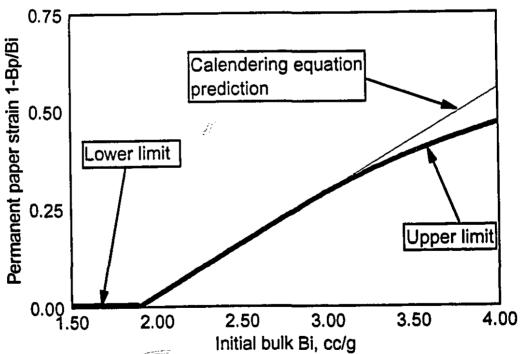


Figure 2.1 Calendering equation of Crotogino et al. at constant nip intensity μ .

$$-\frac{A}{\mu} \leq B_i \leq \frac{1-A}{2\mu}$$
 [2.7]

The lower limit is the point below which, for a given nip intensity μ , no bulk reduction will occur, and corresponds to a line of slope 1 on a plot of B_p vs. B_i :

$$B_{p} = B_{i} \quad when \quad B_{i} \leq -\frac{A}{\mu}$$
 [2.8]

Below this limit the calendering equation predicts $B_p > B_i$, or $\varepsilon_p < 0$, which is not applicable.

The upper limit of applicability of the calendering equation is the point beyond which an increase in initial bulk has no significant effect on bulk reduction, and arises from the fact that the calendering equation, which is the result of empirical curve fitting, predicts the recovered bulk B_p as a parabolic function of B_i . Above this limit, B_p is independent of B_i :

$$B_p = \frac{(1-A)^2}{4\mu}$$
 when $B_i \ge \frac{1-A}{2\mu}$ [2.9]

The calendering equation may be applied successively for multiple nips, with the final bulk from the previous nip serving as the new initial bulk. As well, once the coefficients are known it may be used to predict the new bulk reduction arising from a change in calendering conditions. The calendering equation has been used extensively to calculate the cross-direction average thickness reduction as a function of cross-direction average calendering variables [18, 22, 23, 24, 35, 76]. However, the bulk reduction predicted by the calendering equation is a cross-machine average, since local loads and strains across the width of the machine are unknown.

Timms [76] has reported successful use of the calendering equation in an industrial setting to optimize calender performance and troubleshoot problems using inexpensive laboratory results instead of more expensive machine trials. Hamel et al. [35] used the calendering equation to calculate the nip load distribution from the recovered

thickness profiles of paper calendered at low speeds. The procedure is useful for locating misaligned or poorly ground rolls; however, it does not provide any information about the nip shape or thickness of the paper in the nip, and thus cannot yield an in-nip stress-strain relationship or provide the missing link between roll radius profile and final thickness profile.

Browne et al. [8, 9] performed an initial set of measurements of in-nip paper thickness in a calender operating under industrial conditions using a laboratory calender with a web width of only 70 millimetres in order to maintain constant radius and pressure profiles across the width of the calender. Their calender, which was also used for this study and which is described in greater detail in Chapter 3, consists of a differential slice out of an industrial calender. It can therefore be used to measure local values of the calendering conditions and in-nip strains. From this data an in-nip calendering equation, complementing the CD average calendering equation, may be obtained. Their results confirmed the validity of the calendering equation, and showed that an empirical relation of the form of Equations 2.5 and 2.6 applies for in-nip strain ε_n :

$$\varepsilon_{n} = A_{n} + \mu_{n} B_{i} \qquad [2.10]$$

where

$$\mu_n = a_{on} + a_{Ln} \log_{10} L + a_{Sn} \log_{10} S$$
 [2.11]

Equations 2.10 and 2.11 do not include terms for radius, moisture content or temperature since these were not varied; the effect of these variables is included in the term a_{on} . As well, limits of the in-nip calendering equation were not evaluated since the range of calendering intensities was not extensive enough.

While previous work described so far concerns empirical descriptions of paper strain during and after calendering, a number of authors have proposed theoretical descriptions of paper response to a compressive stress. Kerekes [45] predicted the pressure pulse in a calender nip would be approximately parabolic with a certain amount of skew due to the time-dependent response of paper, as shown in Figure 2.2. The

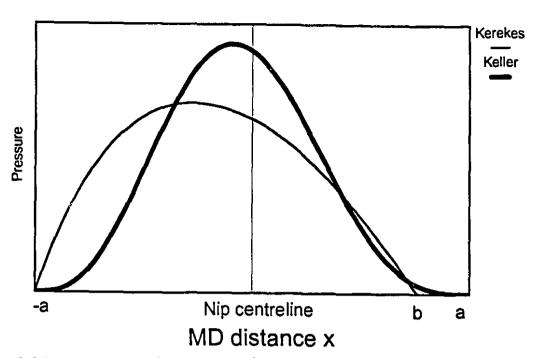


Figure 2.2 Pressure profiles in a calender nip for the same line load.

analysis was based on work by May et al. [53], Hunter [37] and Alblas and Kuipers [2], and modeled paper under compression as a standard linear solid. As will be detailed in Section 2.2, this model is based on the assumption that the permanent paper deformation of the sheet is small compared to the initial thickness. Since permanent compressive strains are often in the range of 10% to 30%, this is a poor approximation.

Keller [43] experimented with a calender using a soft roll and a pressure sensor embedded in a matching hard roll. In-nip pressure profiles (illustrated in Figure 2:2) show the skew predicted by Kerekes [45]. In spite of difficulties determining where the nip begins and ends, Keller's work also showed a nip contact area which is not symmetrical about a line through the roll centres. This asymmetry is expected given both the large permanent paper deformations encountered in calendering and the longer contact area due to the elastic nature of the soft roll. Keller's measured pressure profiles also deviate markedly from the skewed parabola predicted by Kerekes at the entrance and exit of the nip. At the entrance, the pressure increases much more slowly than predicted; at

the exit, the pressure drops quickly before tapering off. There are two possible causes for these deviations: first is elastic deformation of the soft roll, whose modulus of elasticity and Poisson's ratio are both large compared to those of paper [54], but small compared to the hard roll; second is the finite MD length of the pressure sensor (0.56 mm) compared to the MD nip length (2 to 7 mm).

Ionides et al. [38] used a Poisson model of fibre distribution together with an exponential relationship between stress and strain to model paper compression. The results are shown to reproduce published experimental results in a narrow range of loads and speeds. Their model predicts the in-nip strain as a linear function of the permanent strain; as will be seen in Chapter 4, an exponential fit describes the data over a wider range of calendering conditions.

Rodal [73] proposed separating the compressive stress-strain curve for paper into three more or less distinct phases. Under extremely low loads, Hooke's law applies and Young's modulus $E = \partial \sigma/\partial \varepsilon$ is a constant. At higher loads the fibre network begins to collapse, leading to a much lower modulus $\partial \sigma/\partial \varepsilon$. Finally, very large loads result in little additional strain as the fibres themselves collapse; the modulus E here approaches infinity as the curve of stress vs. strain becomes vertical. These three regimes are described using a modified version of Hooke's law, in which the proportionality between stress and strain is modified by a nonlinear term $F(\varepsilon)$:

$$\sigma = \mathbf{E} \varepsilon \, \mathbf{F}(\varepsilon) \tag{2.12}$$

The so-called "shape factor" $F(\varepsilon)$ is further decomposed into two additive parts, one due to buckling of the fibre network and one to collapse of the fibres themselves, which are then derived in terms of a critical strain ε_N at which buckling of the fibre network begins. The model is shown to fit data from the literature in the high stress regimes, but requires estimated values for critical strain and several other parameters. Finally, Rodal suggests the best results are obtained when calendering in the low modulus region.

Osaki et al. [61, 62] investigated compressive properties of paper. Based on the

assumption that there are two compressive regimes, one due to fibre network collapse and one due to fibre crushing, they derive separate stress-strain relationships for the two regimes using statistical descriptions of fibre distributions. Theoretical results were tested using a platen press. Loads and dwell times were not given, and the pulp type and basis weight were not specified. Compression rates were very low compared to those attained in commercial calenders, ranging up to 85 μ m/s where a typical industrial rate is 50,000 μ m/s, but good agreement with theory was found.

While paper response to a load input is a key element in building a complete CD caliper control system, another essential element is the ability to predict the roll radius deformation, and thus the calender nip profile, due to a given heating profile. Pelletier et al. [68, 69] and Journeaux et al. [41, 42] described how to vary the radius profile of a calender roll using heating or cooling impinging air jets. Arrays of impinging jets are often used as actuators in CD bulk control systems. Journeaux determined the relationships between the roll and jet geometry, jet Reynolds number and temperature, and roll dimensions and speeds on the one hand, and heat transfer rates to the roll on the other. The corresponding transient and steady-state thermal deformations of the roll radius were then estimated using finite element methods. At steady state, a peak radius deformation Δr_p and a measure of the width of the deformation $W_{\Delta r}$ were reported for various roll types and various thermal conditions.

In order to calculate a cross-direction thickness profile, CD local values of all calendering variables must be known. CD local initial bulk, moisture content and web and roll temperatures can all be measured online. Roll diameter does not vary on a level that would significantly affect the radius term $a_R log_{10}R$. However, as pointed out in Chapter 1, a roll deformation which is small on the scale of the roll can be enormous on the scale of the paper; the small radius changes described by Journeaux [41, 42] alter the CD nip shape significantly from the point of view of the paper, producing a non-uniform CD load profile. Since load is the variable having the greatest influence on bulk reduction, the result is a significant change in the cross-machine bulk profile. Roll diameter variations across the width of the machine can be estimated using Journeaux's

 $\overset{n}{>} \preceq$

work, but the local nip load cannot be calculated from the roll radius profile without a stress-strain relationship for paper in a rolling nip.

In conclusion, paper is pressed in a calender where it is deformed to an extent dependent on the process variables. Local paper deformation is controlled by local thermal deformation of the roll which alters the local radius and pressure profiles. The CD control actuator settings generating a desired radius profile are known, but the pressure profile and thus the CD bulk reduction profile resulting from that radius profile are not known since the viscoelastic properties of paper in a rolling nip are not known. The goal of the work described here was the measurement of local in-nip paper thicknesses at different loads, leading to quantitative and qualitative descriptions of paper response to a calendering pulse.





2.2 Linear viscoelastic models for paper compression

When a load is applied to a body, energy is either stored or dissipated (Tschoegl [79]). Dissipation may be modeled using viscous elements, while storage may be described in terms of elastic or inertial elements. In the case of paper under compression, the inertial terms are small and can be ignored, leaving elastic elements as the only storage media. Paper compression is not accompanied by significant expansion in the MD or CD; calendering essentially imposes a compressive load on paper. Therefore, the review of stress analysis presented here concentrates on the uniaxial case, as the more general treatment of stress and strain as tensor quantities is not relevant here.

The simplest form of linear viscoelastic model is a single elastic element described using Hertzian elastic theory [39, 40], as proposed for use with paper by Lyons and Thuren [52]. However, this model does not account for the time-dependent behaviour of paper, and further predicts no permanent deformation of the sheet. Thus Hertzian theory remains an approximation to true paper behaviour; its usefulness lies in predicting the new paper strain resulting from a small change in calendering conditions. In order to predict both time-dependent recovery and permanent deformation, a viscoelastic model using a combination of elastic and viscous elements is necessary. Lee [48], Lee and Radok [49], Štok and Kranjec [75], Radok [72] and Carini and de Donato [11], among others, point out that once a solution for an elastic material and a given geometry has been found using Hertzian theory, a viscoelastic solution may be found by substituting viscoelastic operators for the elastic material constants. A model made up of one or more elements may be described in terms of its response to a step input, whether stress σ_0 or strain ε_0 , cf. Findley et al. [27]:

$$\varepsilon(t) = \phi(t) \sigma_0$$

$$\sigma(t) = \psi(t) \varepsilon_0$$
[2.13]

where the retardation function $\phi(t)$ and the relaxation function $\psi(t)$ depend on the

: 12

particular model. If the material behaves in a linear fashion, the retardation and relaxation functions are material properties unaffected by the applied strain or strain rate, and the response to a second step input σ_1 or ε_1 added to the first at a later time t_1 is

$$\varepsilon(t) = \phi(t) \ \sigma_0 + \phi(t-t_1) \ \sigma_1$$

$$\sigma(t) = \psi(t) \ \varepsilon_0 + \psi(t-t_1) \ \varepsilon_1$$
[2.14]

The response at any time t to an arbitrary sequence of earlier step inputs beginning at time t=0 is then the sum of the responses to these individual inputs. In the limit Equations 2.14 become the Boltzmann superposition integrals:

$$\varepsilon(t) = \int_{0}^{t} \phi(t-t') \frac{\partial \sigma(t')}{\partial t'} dt'$$
 [2.15]

$$\sigma(t) = \int_{0}^{t} \psi(t-t') \frac{\partial \varepsilon(t')}{\partial t'} dt'$$
 [2.16]

The claim of Lee and Radok [49] that Equations 2.15 and 2.16 are viscoelastic operator forms of a Hertzian solution can be verified by setting $\psi(t)$ equal to the elastic modulus E and integrating Equation 2.16, which yields Hooke's law.

Two basic models may be assembled from ideal mechanical elements. The Maxwell model consists of a spring and damper in series. If the spring has modulus G and the damper viscosity η , the relaxation function is given by Tschoegl [79]:

$$\psi(t) = G e^{\left(\frac{-tG}{\eta}\right)}$$
 [2.17]

As illustrated in Figure 2.3, the stress in a Maxwell unit subjected to a step strain decays to zero at sufficiently long times. When the strain is removed at earlier times, a Maxwell element exhibits instantaneous partial recovery of the initial thickness.

The Kelvin unit, a parallel spring and damper, has retardance given by

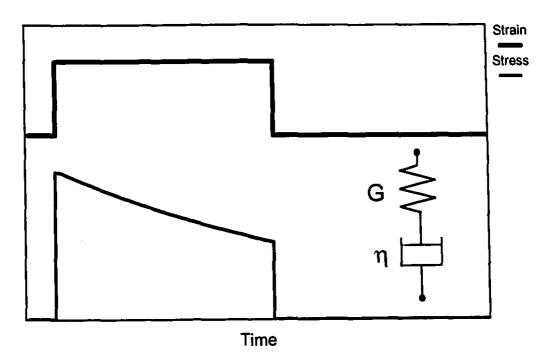


Figure 2.3 Maxwell unit response to a step strain.

$$\phi(t) = \frac{1 - e^{\left(\frac{-tG}{\eta}\right)}}{G}$$
 [2.18]

Strain in a Kelvin unit subjected to a step stress eventually reaches a constant value, as illustrated in Figure 2.4. There is complete time-delayed recovery once the stress is removed.

More complex behaviour can be approximated by combining three or more elements. Several authors [37, 45, 53] have used the standard linear solid, a spring in series with a Kelvin model, to describe a thin viscoelastic sheet pressed between two rolls. If the additional spring, which models instantaneous elastic response, has modulus G_e , the retardation function is

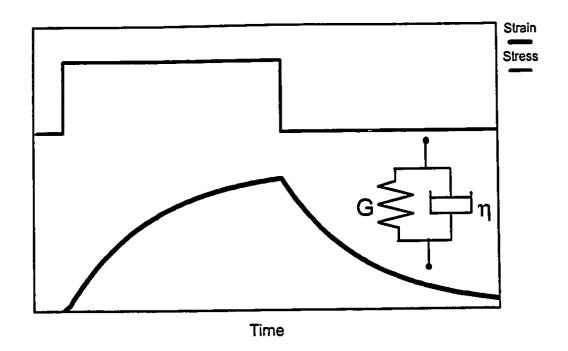


Figure 2.4 Kelvin unit response to a step stress.

$$\phi(t) = \frac{1}{G_e + G} \left\{ 1 + \frac{G \left(1 - e^{(-t/\tau)} \right)}{G_e} \right\}$$
 [2.19]

where $\tau = \eta(1/G + 1/G_e)$. If the standard solid is subject to a step stress σ_0 at time t=0 which is removed at time t_2 , the strain for times later than t_2 is found by substituting Equation 2.19 into Equation 2.15:

$$\varepsilon(t>t_2) = \sigma_0 \frac{G}{G_e} \left(e^{t_2/\tau} - 1 \right) e^{-t/\tau}$$
 [2.20]

At large enough times, Equation 2.20 (illustrated in Figure 2.5) predicts that permanent strain approaches zero. As this is not the case with paper, the model remains an approximation when applied to calendering.

Given the geometry of the nip, and based on the assumption that the initial thickness of the material is small compared to the roll radius, May et al. [53] and Kerekes [45] derived expressions (described in greater detail in Chapter 5) for the strain

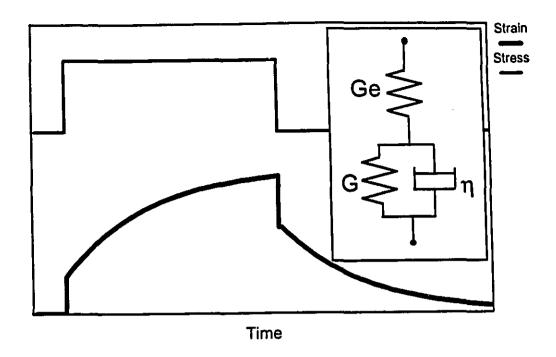


Figure 2.5 Standard linear solid response to a step stress.

 $\varepsilon(t)$ and strain rate $\partial \varepsilon(t)/\partial t$ in a calender nip. Kerekes also calculated the stress profile for a standard linear solid in a rolling nip using the Boltzmann superposition principle. This profile, illustrated in Figure 2.2, is parabolic with a certain amount of skewing towards the ingoing side of the nip. Kerekes showed that the factor controlling the skew is the ratio of the material relaxation time τ to the time a/S required for the web to cover the distance a from the nip entrance to the nip centreline. When this ratio, also known as the Deborah number ξ , is close to 1, the behaviour of the material is viscoelastic. When ξ is very much greater than 1, the processing time a/S is much shorter than the material property τ , and behaviour appears viscous since any recovery occurs over a time equal to many processing times. There will be eventual full elastic recovery of the initial thickness, but that recovery is significantly delayed compared to the dwell time in the nip. Similarly when ξ is very small the processing time is much longer than the material property, and what recovery there is occurs immediately since the rate at which the pressure is removed at the nip exit is slower than the rate at which the material can

recover. At both extremes the amount of skew is small, reaching a maximum only when $S\tau/a \approx 1$. It should be noted that viscoelastic behaviour will be observed for any standard linear solid (described by its time constant τ) and any nip geometry (described by the ingoing half-length a) at a sheet speed $S \approx a/\tau$.

Beginning with elastic descriptions of the stress-strain properties of paper, Mann et al. [54] have measured all nine orthotropic elastic constants for paper using ultrasonic methods. The paper was a thick (680 μ m), dense (520 g/m²) milk carton stock. Castagnede et al. [12, 13] extended the work to include a 240 g/m² kraft sheet. The experimental method is based on measurement of the speed of propagation of an ultrasonic pulse through the material. Since the speed of the pulse is very high, the stresses are applied for a very short time, the resulting strains are very small, and the measured values of elastic and shear moduli are thus independent of viscoelastic effects. The results of Mann et al. [54], listed in Table 2.1, show the strong anisotropy of paper; for example, the z-direction elastic modulus Ezz is two orders of magnitude less than the inplane moduli, and the Poisson's ratios for a load in the z-direction are one order of magnitude smaller than the in-plane ratios ρ_{xy} or ρ_{yx} . Castagnede et al. [13] point out that measurements of elastic constants obtained with mechanical methods tend to be significantly different from acoustical measurements: moduli are typically 20% to 30% higher, and Poisson's ratios are 20% to 40% lower, when acoustic methods are used. The discrepancy in moduli is explained by the short duration of the acoustic tests, which effectively eliminates any viscous contribution to the general stiffness tensor. The discrepancy in Poisson's ratios is attributed to propagation of error in calculating ρ from the measured quantities. They conclude that the assumptions of symmetry inherent in the definition of the orthotropic stiffness tensor are only approximations, especially for paper in the z-direction.

Watanabe and Amari [80] have proposed modeling paper compression using a four-element model, called a Burger's model by Tschoegl [79], made up of a Maxwell and a Kelvin model in series. The Maxwell model predicts instantaneous elastic deformation (due to the spring) and permanent deformation once the load is removed

TABLE 2.1: Orthotropic elastic constants, from Mann et al.

Young's modulus, GPa		Shear modulus, GPa		Poisson's ratio			
E _{xx}	7.44	G _{xy}	2.04	$ ho_{xy}$	0.32	$ ho_{yx}$	0.15
E _{yy}	3.47	G _{xz}	0.137	ρ _{xz}	1.52	$ ho_{zx}$	0.008
E _{zz}	0.039	G _{yz}	0.099	$ ho_{ m yz}$	1.84	$ ho_{zy}$	0.021

(due to the dashpot), while the Kelvin element predicts the viscoelastic behaviour, in particular the time-dependent recovery. They measured paper strain in the nip of a laboratory-scale printing press by pasting a sheet of paper between two sheets of aluminum foil; the foil acted as a capacitor whose dielectric constant was related to paper thickness. The method thus yields a continuous strain profile of the nip. Roll radius, sheet speed and line load were small, and typical dwell times were large compared to a commercial calender; nonetheless the strain profile resembles the stress profile of Keller [43], a skewed parabola with entrance and exit effects superimposed. From the strain vs. time profile they determined an approximate retardation time for the Kelvin element of 2.5 ms, and predicted the spring rate of the Maxwell element (which they link to the compressive modulus of a single wood fibre) to be about 217 MPa, independent of line load or moisture content. This value is higher than the value of 39 MPa reported by Mann et al. [54]; the difference may be due to the different paper types used. Equations governing the behaviour of a Burger's model are derived in Chapter 5.

More complex models may also be used to give a closer approximation to nonlinear materials. Among these, the short time approximation proposed by Huang and Lee [36] and modified by Lubliner [51] may be applied to calendering of paper at high speeds. The first step is to use a polynomial series to approximate nonlinear behaviour, replacing Equations 2.13 with new expressions for response to an input step stress σ_0 or

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strain ε_0 :

$$\varepsilon(t) = \phi_1(t) \sigma_0 + \phi_2(t) \sigma_0^2 + \phi_3(t) \sigma_0^3 + \dots$$

$$\sigma(t) = \psi_1(t) \varepsilon_0 + \psi_2(t) \varepsilon_0^2 + \psi_3(t) \varepsilon_0^3 + \dots$$
[2.21]

Additional time-dependent material functions $\phi_2(t)$, $\phi_3(t)$, etc., need to be determined from either experiment or theory. In principle, increasing accuracy is obtained by including more terms in Equation 2.21. In the limit, for a given stress or strain history, Equations 2.21 become integrals analogous to the Boltzmann integrals, illustrated here for strain:

$$\varepsilon(t) = \int_{0}^{t} \phi_{1}(t-t_{1}) \frac{\partial \sigma(t_{1})}{\partial t_{1}} dt_{1}$$

$$+ \int_{0}^{t} \int_{0}^{t} \phi_{2}(t-t_{1}, t-t_{2}) \frac{\partial \sigma(t_{1})}{\partial t_{1}} \frac{\partial \sigma(t_{2})}{\partial t_{2}} dt_{1} dt_{2}$$

$$+ \int_{0}^{t} \int_{0}^{t} \int_{0}^{t} \phi_{3}(t-t_{1}, t-t_{2}, t-t_{3}) \frac{\partial \sigma(t_{2})}{\partial t_{1}} \frac{\partial \sigma(t_{2})}{\partial t_{2}} \frac{\partial \sigma(t_{3})}{\partial t_{3}} dt_{1} dt_{2} dt_{3}$$

$$+ \int_{0}^{t} \int_{0}^{t} \int_{0}^{t} \phi_{3}(t-t_{1}, t-t_{2}, t-t_{3}) \frac{\partial \sigma(t_{2})}{\partial t_{1}} \frac{\partial \sigma(t_{2})}{\partial t_{2}} \frac{\partial \sigma(t_{3})}{\partial t_{3}} dt_{1} dt_{2} dt_{3}$$

Truncating the series after three terms and integrating by parts gives an expression containing terms which are negligibly small for short times (cf. Lubliner [51]); ignoring these gives

$$\varepsilon(t) = \phi_1(0) \, \sigma(t) + \phi_2(0,0) \, \sigma^2(t) + \phi_3(0,0,0) \, \sigma^3(t)$$

$$- \int_0^t \left[\frac{\partial \phi_1(t-t_1)}{\partial t_1} \, \sigma(t_1) + 2 \frac{\partial \phi_2(t-t_1,0)}{\partial t_1} \, \sigma(t) \, \sigma(t_1) \right] dt_1$$

$$- \int_0^t \left[3 \, \frac{\partial \phi_3(t-t_1,0,0)}{\partial t_1} \, \sigma^2(t) \, \sigma(t_1) \right] dt_1$$
[2.23]

The first three terms give the instantaneous nonlinear elastic deformation, while the integrals give the time-dependent history.

Finally, the various integral forms may be used with material functions ϕ or ψ which are dependent on the stress or strain rate as well as time; Equations 2.15 and 2.16 are then nonlinear. Determining the form of $\phi(t,\varepsilon,\partial\varepsilon/\partial t)$ or $\psi(t,\sigma,\partial\sigma/\partial t)$ requires stress-strain data for step stress or step strain inputs over a large range of times. In a calender nip, the strain input is not a step but a modified parabola, and the range of industrially-relevant dwell times is very small; thus the extra effort expended in determining $\phi(t,\varepsilon,\partial\varepsilon/\partial t)$ or $\psi(t,\sigma,\partial\sigma/\partial t)$ may not be worth the additional precision of the results.

As models become more complex, the constitutive equations can more closely approximate stress-strain behaviour over a larger range of dwell times, thus providing a more complete description of material behaviour for a broader range of processing conditions; however, this improvement comes at the cost of increasing complexity. More complex models require substantially increased computational effort and more extensive experimentation in order to determine numerical values or functional forms of the material response functions $\phi(t,\varepsilon,\partial\varepsilon/\partial t)$ or $\psi(t,\sigma,\partial\sigma/\partial t)$. The duration of the loading becomes a particularly important independent variable in models with significant viscous behaviour. In calendering of paper, the dwell times are very short and vary by at most an order of magnitude, so the additional work required to obtain a better description of the time dependency of ϕ or ψ is not justified by the added precision of the results. The simplest model exhibiting all the observed stress-strain properties of paper in a calender

nip, such as creep under load and time-dependent partial recovery, will be the most useful for purposes of improving CD control systems.

- 3. Experimental procedures
- 3.1 Equipment overview
- 3.1.1 Mechanical systems

The experimental equipment, a modified version of that used recently by Browne et al. [8, 9], is designed to reproduce industrial conditions of nip load, web speed and roll radius while allowing accurate and repeatable measurements of the separation of the rolls. To this end all sources of cross-direction thickness variation had to be minimized. This goal was attained by making the calender as narrow as possible, thus eliminating errors due to roll bending or bearing deflection. Figure 3.1 is an overall view of the calender, which has a face width of 75 mm and a maximum paper width of 70 mm.

The calender layout, Figure 3.2, is designed to take continuous webs from a reel of up to 1.0 m in diameter, and to feed them through a single calender nip to a rewinder at speeds of up to 1000 m/min. Nip loads are varied using a hydraulic pump and cylinder, and range up to 210 kN/m. The hydraulic system can be used for nip relief, allowing loads less than those due to the gravity loading applied by the mass of the upper roll alone. Three pairs of rolls are provided, with diameters of 404, 508 and 711 mm. The shoulder machined into the side of each roll, Figure 3.3, serves as the target for a displacement sensor recording the position of both rolls relative to a fixed stand. Signals from these sensors are used to calculate in-nip paper thicknesses.

A computerized data acquisition system records sheet tensions and speeds before and after the nip, nip load, upper and lower roll positions relative to a fixed point, and paper caliper at two positions immediately after the nip. The acquisition rate is keyed to the web speed and roll radius: 150 to 200 data points are acquired per roll revolution over a total of 10 to 15 roll revolutions, thus keeping the distance along the sheet between samples approximately constant. As an example, at 1000 m/min with a 404 mm diameter roll, 150 samples are acquired per roll revolution over about 14 revolutions; the acquisition rate is thus about 2000 Hz, with 2048 samples acquired roughly every 8.5 mm over a total distance of about 17 m along the sheet. At 90 m/min with a 711 mm

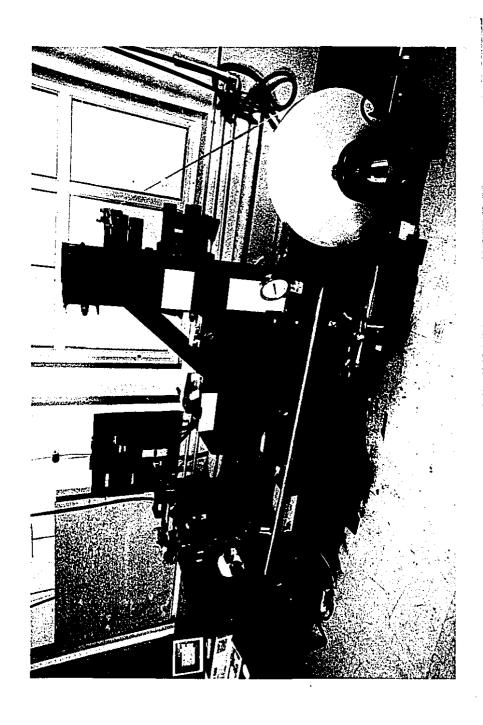


Figure 3.1 Overall view of calender; 202 mm radius rolls shown.

diameter roll, 300 samples are acquired per roll rotation, one every 7 mm over 15 m of paper for an acquisition rate of about 200 Hz.

In the present study the loading mechanism was revised from that described by Browne et al. [8, 9] so that the hydraulic cylinder now pulls down on the upper roll supporting arm, instead of pushing from above. As well, the frame was stiffened. The purpose of these modifications was to reduce the bending loads applied to the main calender stand and thus reduce the movement of the upper roll relative to the measurement sensors as nip loads are increased. Other modifications include the addition of computer control over the nip load and winder sheet tension to simplify operation of the equipment and to reduce the frequency of sheet breaks.

The most important modification was to the sensors which measure roll position, or nip gap, Figure 3.3. Previously these were contact devices riding on shoulders machined in the sides of the rolls. While these sensors had a high electrical natural frequency, allowing data acquisition at high rates, the mechanical natural frequency was low, causing the gauges to bounce at certain combinations of roll radius and web speed. As well the carbide contact tips were subject to wear, leading to frequent re-calibration.

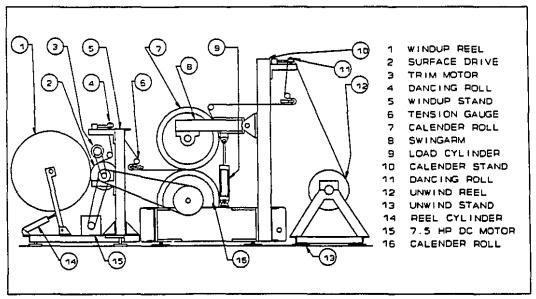


Figure 3.2 Layout of experimental calender. Length = 4 m, height = 2 m.

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Figure 3.3 Roll displacement sensors; 355 mm radius rolls shown.

The contact gauges were replaced with non-contact inductive type sensors which react to the presence of a metallic mass, in this case the surface of the roll shoulder. The new gauges have better resolution at all speeds and do not require frequent re-calibration. As well, since the new gauges are non-contact devices, the problem of bounce has been eliminated, resulting in less wasted experimental time. Roughly half the data presented here was acquired using the older setup.

Paper caliper is now measured at two positions, 296 mm and 1050 mm after the nip, using inductive sensors buried in a ceramic anvil; the target for the sensor is an aluminum disc carried in the floating head of a modified industrial caliper gauge. Illustrated in Figure 3.4, these sensors are more reliable than the type described previously [8], but since they still involve a contact element a certain amount of bounce can occur under certain conditions. The advantage of the new system is better dynamic control of the floating head.

Sheet speed is now measured in three ways. A tachometer on the main drive motor is used to calculate the surface speed of the lower (driven) calender roll. This is the value used in all calculations and plots, and is unchanged from the method used previously.

In order to measure machine-direction strain imposed by the calender, the change in sheet speed from the unwind stand to the winder at a constant set of calendering conditions was computed from two additional measurements of sheet speed. Two idler rolls, one before and the other after the nip, were covered with 10 alternating strips of matte black paint and shiny aluminum tape. One of these devices is illustrated in Figure 3.5. Facing each roll was an optical sensor which responded to the different optical properties of the paint and tape by generating a square wave. The diameters of the idlers were 63.35 mm before the nip and 63.34 mm after; each rotation of the idlers thus corresponded to a length along the sheet of about 199 mm, and generated 10 cycles of a square wave. The output was filtered using a Schmidt trigger, and the pulses were counted using the counter/timer function on the A/D board. The total number of pulses in a given time, divided by the number of pulses per rotation and multiplied by the idler

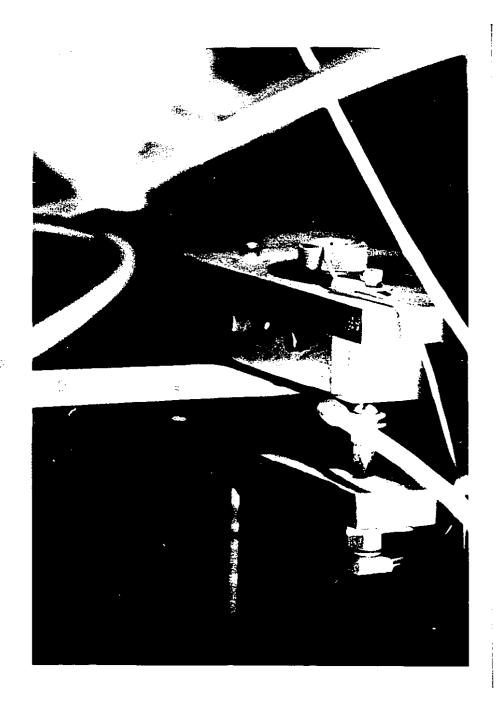


Figure 3.4 Online caliper gauge 296 mm after the nip.

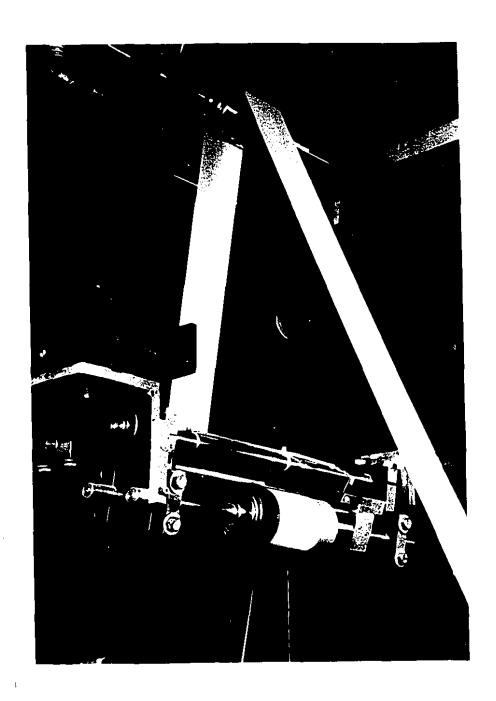


Figure 3.5 Sheet speed sensors.

circumference, is proportional to the sheet speed at that idler. Comparison of the speed before and after the nip yields an estimate of the percent speed increase, which is equal to the MD stretch.

To minimize the possibility of slip between the sheet and idler, the idlers were roughened by sandblasting and were installed so that the sheet wrap was a full 180 degrees. Pulse counting using the A/D counter function was initiated sequentially for the two counters just before acquisition of the in-nip thickness data was started; the counters were stopped in the same sequence immediately after the end of the in-nip data acquisition. The MD stretch data was thus typically computed over a distance of about 17 meters of paper, or about 1000 pulses from each sensor.

Sheet tension is another factor influencing sheet stretch; although the target tension was low at 600 N/m, the measured tension was often either lower or higher, and occasionally varied substantially within a set of data. Thus all sheet extension measurements made when either unwind or winder tension was excessively variable, or when the two tensions were not within 20% of one another, were discarded. In other words if the tension is sufficient to cause additional stretch, the effect is assumed to be similar on both sides of the nip when the tensions are similar.

3.1.2 Data acquisition and processing systems

All analog sensor outputs were filtered using a simple RC filter with a cutoff frequency of 34 Hz to eliminate interference from 60 Hz supply circuits. The filtered signals were then connected to a pair of 12 bit analog-to-digital conversion boards installed in an IBM-compatible personal computer. The boards also included digital input/output ports as well as digital-to-analog conversion, which were used to control the equipment. After acquisition, initial processing included taking a moving average of each digitized signal; in this process, each data point was replaced with the average of the four preceding points, the current point and the four following. Since every fourth point then included information about the three intervening points, only the fourth point was saved, reducing the data set from 2048 points to 512. Acquisition, control and initial processing were performed using a custom program written using Microsoft's QuickC programming language; a listing appears in Appendix A2.

The most important aspect of the data processing concerns the treatment of the output from the nip thickness gauges. The calibration of these gauges, whether the older contacting type or the new non-contact devices, is crucial to the acquisition of reliable in-nip paper thicknesses. Ideally, the lower roll remains fixed relative to the gauge, and the upper roll is deflected upwards by the presence of paper in the nip. Assuming a linear relationship between sensor output and roll position, the paper thickness in the nip T_n is then the difference between the upper roll position as seen by the sensor with paper in the nip, u, and the position without, u_0 . The relationship between roll position and sensor output was indeed linear; however, there are two problems with this simple approach.

First, the lower roll is not fixed relative to the gauges, but moves vertically in its bearings which have a small amount of radial play. As the load is increased at constant speed, grease is forced from the lower ball races in the bearings, allowing the lower roll to drop as much as 20 μ m when the load is increased from lowest to highest. Since the paper sheet and upper roll ride on the lower roll, both are thus displaced downwards. On the other hand, hydrodynamic forces between the ball bearing and race cause the lower

roll to lift over a range of about 10 μ m as speed is increased from lowest to highest at constant load. Since paper thicknesses in the nip are less than the initial thickness, and since that initial thickness was about 120 μ m for the paper used in this study, these errors are substantial.

The solution was to fit a second gauge facing the lower roll. The in-nip paper thickness is then the sum of the distances from the gauges to the two rolls with paper in the nip, compared with the same distances measured when there is no paper in the nip. If ν and ν_o are the position of the lower roll respectively with paper and at rest, the true in-nip paper thickness is then

$$T_{n} = (u - u_{o}) + (v - v_{o})$$

$$= (u + v) - (u_{o} + v_{o})$$
[3.1]

where the second form has been used to reduce the error in subtracting v_0 from v.

The second problem is due to the fact that both roll surfaces and shoulders are all slightly eccentric with respect to their bearing axes. The signals from the gauges are thus superimposed on a sinusoidal wave whose period is the rotation period of the roll, and whose amplitude is the magnitude of the eccentricity. (The eccentricities of the roll surfaces were measured relative to the bearing centerlines and found comparable in both amplitude and phase to those of the roll shoulders). Further, the signal from the gauge locating the upper roll is the sum of two sinusoidal waves, since the upper roll rides on top of the lower one and is thus deflected by an amount depending on the sum of the amplitudes and the relative angular position of the two rolls. Since the eccentricity ranges from 25 to 40 μ m for the different rolls, this effect is also significant, and has been eliminated by using averages (as described in the previous Section) of 2048 values of u, u_0 , v and v_0 taken over a minimum of 10 revolutions of the lower roll. At 1000 m/min with a 404 mm diameter roll, this works out to a minimum distance of about 17 m of paper.

The final problem involves a slight shift in the paperless reading $(u_0 + v_0)$

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depending on whether the hydraulic system is providing load or relief. This was ascribed to a slight twisting effect of the calender stand due to the difficulty of aligning the load cylinder perfectly with the machine-direction centerline of the paper, bringing a different circumference of the roll shoulders into position in front of the sensors as the frame twists. The difference between paperless readings varied by about 3 μ m from loading to relieving, and was constant over the rest of the range of loads; on the assumption that readings with paper would vary by the same amount, values of $(u_0 + v_0)$ were routinely obtained for both loading and relieving situations, the appropriate one being used to compute T_n . The assumption was a fair one since the position of the lower roll depended only on load and speed, and not on the presence or absence of paper.

The slight twist of the frame described above did not result in measureable deformation of the nip; this was verified by taking paper thickness measurements of calendered sheets and computing the average cross-machine variation in paper thickness. The measured variation did not vary from loading to relieving, but remained constant for all loading conditions.

Readings from the load cell giving the nip load also showed a small sinusoidal variation due to the eccentricities of the rolls adding and trying to lift the upper arm. Sheet speed was also affected, since the variable nip load causes a variable load to be put on the DC drive. In both these cases the amplitude of the oscillation was small, of the order of 1 kN/m and 5 m/min respectively, and was eliminated by averaging over the same distance as with the nip gauges.

3.2 Experimental conditions and procedures

The main experimental aspect of this work was the measurement of paper thickness before, during and after calendering under a variety of conditions, and the computation of in-nip and permanent strain. Modified versions of the calendering equation were then fitted to the strain data.

Two types of newsprint were used: a specialty newsprint from Boise-Cascade, Fort Frances, Ontario, and a newsprint from thermo-mechanical pulp (TMP) made by F. Soucy, Rivière-du-Loup, Québec. Both papers were obtained wound in 1.2 m wide rolls which were then cut into several narrow rolls, 70 mm wide and 1.0 m diameter. Properties of these papers before calendering are listed in Table 3.1. All properties were measured after conditioning for 24 hours at 23°C and 50% relative humidity.

The specialty newsprint was made from a furnish containing 71.5% semi-bleached groundwood pulp and 28.5% bleached kraft pulp. It had been lightly calendered at the mill, resulting in a low initial bulk of 1.97 cm³/g. This paper was calendered here to compare results with those published by Browne et al. [8, 9] using the same equipment before the modifications described in Section 3.1.1. The calendering conditions were extended beyond those reported earlier to include a larger range of loads and speeds, and experiments were performed using different roll diameters. Experiments were performed using most combinations of independent variables, with replicates performed only at a few points.

The largest part of the work involved calendering the TMP newsprint. The three basis weights had essentially the same initial bulks. Most experiments were performed using the uncalendered bulk as received of 2.65 to 2.90 cm³/g, but several lower bulks were obtained by precalendering a roll to a bulk in the range 2.0 to 2.25 cm³/g, then recalendering.

A full range of loads, speeds and roll diameters was investigated using a full factorial experimental design, and several replicates were performed at most conditions. Calendering conditions for both paper types are listed in Table 3.2.

TABLE 3.1: Paper properties as received.

Paper type	Boise- Cascade	F. Soucy		
Basis weight, g/m ² , standard deviations	52.0	42.3 0.42	44.4 0.39	46.6 0.44
Caliper, μm, standard deviations	102	117 3.11	122 2.71	131 3.12
Bulk, cm ³ /g, standard deviations	1.97	2.65 0.10	2.75 0.08	2.91 0.10
Moisture content, %	9.0	7.93	7.89	7.36
Parker Print Surf, μm, top/wire side, standard deviations	4.78, 5.17 0.14, 0.14	5.04, 5.91 0.14, 0.18	5.18, 6.27 0.17, 0.20	5.53, 6.53 0.16, 0.38

Experimental procedure was as follows. After allowing all sensors to warm up for a minimum of 2 hours, the calender was started up with no paper in the nip to obtain the calibration offset $(u_0 + v_0)$ as described in Section 3.1.2. Calibrating tests had shown that this reading was insensitive to sheet speed, but depended slightly on whether the loading system was operating in loading or relieving. This so-called zeroing run was therefore repeated at several loads but at only one speed, typically 90 m/min.

After taking an uncalendered sample, paper from a new roll was threaded through the calender. In some cases this paper had previously been calendered to a lower bulk. The computer was used to accelerate the main motor slowly up to 90 m/min under a light load while tracking and tension of the sheet were checked. The motor was then accelerated to the desired speed, the desired load was applied, and once speed and

TABLE 3.2: Calendering conditions.

Load, kN/m	10 to 210		
Roll Radius, mm	202, 254, 355		
Sheet Speed, m/min	90 to 950		
Bulk, cm ³ /g (specialty newsprint)	1.97		
Bulk, cm ³ /g (TMP newsprint)	2.01 to 2.25; 2.55 to 2.90		

tension were reasonably stable the complete set of sensors was scanned 2048 times at a rate determined using the reasoning described in Section 3.1.1. The location in the wound roll corresponding to the data set was marked using a slip of paper inserted in the winder nip, the data was saved to disk, and the speed and load were then adjusted to the next condition. The process was repeated until the experiment was ended either by the operator, by the end of the unwind roll, or by a sheet break.

Raw data were then smoothed using a moving average method. In the method selected, each data point was replaced with the average of 9 points: the current point and the 4 preceding and following. Every fourth point was saved, for a total of 512; the remainder was discarded. Since a minimum of 150 in-nip thickness readings were acquired for every roll revolution, the eccentric signature of the roll was still easily identifiable after this process.

Sufficient paper usually remained on the unwind roll for another uncalendered sample to be taken. Samples were never taken from the first or last few meters of paper on a roll, but always a minimum of several centimeters radially from the core or rim of the roll. Once the wind-up reel was full, the roll of calendered paper was cut up and samples corresponding to each data set were retrieved. All paper samples were cut to a length corresponding to one circumference of the calender roll in use at the time in order

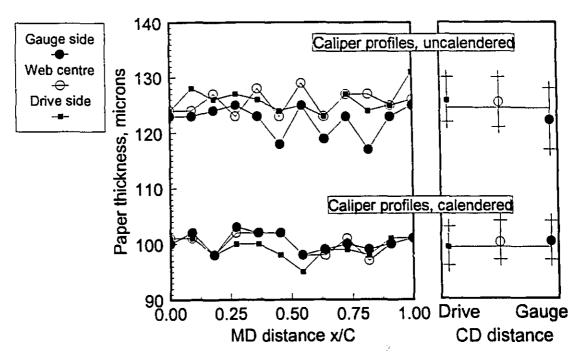


Figure 3.6 Machine-direction thickness profiles, TMP newsprint (run #895E)
Calendering conditions: 202 mm, 65 kN/m, 554 m/min, 2.54 cm³/g.

to average out any machine direction variations due to roll crowns or eccentricities. Calendered and uncalendered samples were conditioned for 18 to 24 hours at 23°C and 50% relative humidity. Paper thickness was determined according to TAPPI Standard 411 (CPPA standard D.4) using a standard electronic micrometer, at 36 points in three rows of 12 along the machine direction of each strip, one row in the center of the 70 mm wide sheet, and the other two 15 mm from each edge. Where the average cross-direction variation in the recovered thickness exceeded 2 μ m, the data were discarded and the roll alignment was checked; otherwise the average of all 36 thicknesses was used as the sample paper thickness. Typical machine direction thickness profiles and averages are shown in Figure 3.6 for uncalendered and calendered samples from the same roll; the machine direction x is normalized by dividing by the roll circumference C. Average paper thicknesses and standard deviations for the 36 measurements in this example were 124.6 μ m, 2.77 μ m (uncalendered) and 99.9 μ m, 1.76 μ m (calendered), showing that in

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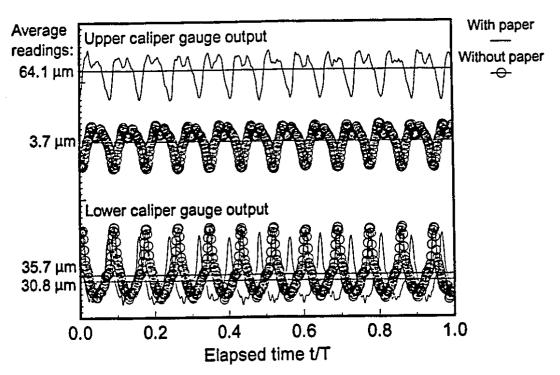


Figure 3.7 Typical caliper gauge output (runs #894B and #895E).

this case calendering improved the CD caliper profile. It appears from the individual averages that the thick side of the ingoing web was calendered heavily enough to become the thin side afterwards; however, the thickness measurements are for samples taken from different MD positions in the roll.

The in-nip thickness data were processed as described in Section 3.1.2. In particular, the reading with paper (u + v) was obtained by averaging the 512 values of u and v before adding. This was repeated for $(u_0 + v_0)$, and the in-nip paper thickness obtained by Equation 3.1. Figure 3.7 shows typical values of u, v, u_0 and v_0 , with the elapsed time t normalized by dividing by the total acquisition time T. Readings with no paper were obtained at a calendering speed of 90 m/min and a nip load of 65 kN/m, while those with paper were obtained at 554 m/min and 65 kN/m. The slight difference in the frequency of the roll eccentricities between the two sets of curves is due to the different data acquisition rates not quite correcting for the different rotational velocities.

The upper gauge output shows that the upper roll, after removing the roll eccentricities by averaging, was deflected upwards by the presence of paper in the nip from an average position $u_0 = 3.7 \,\mu\text{m}$ from a fixed reference point to another position $u = 64.1 \,\mu\text{m}$ from the same reference point, a displacement $(u - u_0) = 60.4 \,\mu\text{m}$, while the lower roll was also deflected upwards (due to the increased speed) from a position $v_0 = 35.7 \,\mu\text{m}$ from another reference point (fixed relative to the first one) to a new position $v = 30.8 \,\mu\text{m}$, a displacement $(v - v_0) = -4.9 \,\mu\text{m}$. The average in-nip gap, equal to the average in-nip paper thickness, was therefore $(u - u_0) + (v - v_0) = 55.5 \,\mu\text{m}$ in this case.

A second paperless reading was often obtained after one or more experimental runs to verify that no changes had occured in the position of the rolls relative to the gauges. A typical variation in the value of $(u_0 + v_0)$ was $\pm 1~\mu m$ with the newer inductive gauges, $\pm 3~\mu m$ with the older contact gauges. Where this was larger than 2 μm (4 μm for the older gauges), the cause was investigated and the data set discarded.

The experimental program was designed to minimize roll changes, as these required time-consuming alignment and re-calibration procedures. Furthermore, sheet breaks were more frequent at higher speeds; higher speeds were thus attempted first to ensure acquisition of sufficient data at these speeds.

The first set of measurements were made with the 202 mm radius set installed, using the contact type of in-nip thickness gauge described in Section 3.1.1. Rolls were then changed for the 254 mm pair and the acquisition process was repeated. Next, the contact-type in-nip caliper gauges were replaced with the inductive type at the same time the 355 mm rolls were installed. Once a full set of data had been acquired using all three sets of rolls, the 202 mm pair was reinstalled and a complete data set acquired again. No significant changes in in-nip or permanent strains were noted between the first and last runs using the 202 mm roll set, in spite of the fact that the earlier data set was acquired using a different type of in-nip displacement sensor. The new gauges resulted in less scatter and fewer data points being discarded, especially at high speeds.

The time required to change roll pairs and acquire a new data set was typically

several months. As the equipment was not installed in a controlled environment room, large seasonal changes in relative humidity and thus paper moisture content tended to mask the effect of roll radius on paper strain. After the statistical analysis described in Chapter 4 was performed, the effect of moisture content on the strain data was estimated, and the analysis repeated. The permanent moisture content coefficient was estimated from data published by Crotogino et al. [22] to be $a_{Mp} = 0.005$. The in-nip coefficient was estimated using data published by Colley and Peel [17], where it appears that $a_{Mn} = 0.85 \ a_{Mp}$. In-nip and permanent strain data were then adjusted by adding the following quantity, where $M_{50} = 7.69$ is the percent moisture content corresponding to 50% relative humidity:

$$\varepsilon_{corr} = B_i a_M (M - M_{so})$$
 [3.2]

This processing step was not applied to the specialty newsprint, since data were acquired with only one roll radius.

In-nip and recovered thickness were determined for the same strip of paper, since the location in the wound roll corresponding to the acquisition of a set of in-nip thickness data was marked and could be retrieved later by unwinding or cutting up the roll. Initial thickness, on the other hand, was measured at only a few points distributed over the 6 to 8 km of paper wound on each roll; these samples were thus not the same ones which were eventually calendered. Initial thickness was assumed to remain constant throughout the roll from sample to sample. Occasionally the initial thickness of a strip taken from the beginning of the roll differed from one taken towards the end, sometimes by as much as 5 μ m; when this occurred strains for the first half of the data acquired with that roll were computed using the earlier initial thickness, and for the last half using the later one. A linear interpolation method was not used because at least one of the rolls was found to have a thickness change of 5 μ m occurring over a machine-direction distance of only 1 m. Since the narrow rolls were all cut from a single 1.2 m wide roll, it was assumed that adjacent slices must also exhibit the same step change, presumably due to the action of a control mechanism at the mill at the time of manufacture.

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Finally, initial and final paper thickness were obtained using a standard paper micrometer and are thus subject to a smaller random error than in-nip thickness, which was obtained dynamically while the calender was running.

- 4. Experimental results
- 4.1 In-nip and permanent strain measurements
- 4.1.1 The calendering equation

All of the strain data described previously were processed using SYSTAT, a statistical analysis package running on IBM personal computers. In-nip and permanent versions of the calendering equation were fitted to the strain data:

$$\epsilon_{n} = A_{n} + \mu_{n} B_{i}$$

$$\epsilon_{p} = A_{p} + \mu_{p} B_{i}$$
[4.1]

where the nip intensity coefficients μ_n and μ_p are

$$\mu_{n} = A_{on} + a_{Ln} \log_{10} L + a_{Sn} \log_{10} S + a_{Rn} \log_{10} R + a_{\theta n} \theta + a_{Mn} M$$

$$\mu_{p} = A_{op} + a_{Lp} \log_{10} L + a_{Sp} \log_{10} S + a_{Rp} \log_{10} R + a_{\theta p} \theta + a_{Mp} M$$
[4.2]

and ε_n , ε_p are the in-nip and permanent strains, B_i is the initial bulk, L the nip load, S the machine speed, and R the roll radius R. The web temperature θ and moisture content M were not measured in this study, but corresponded to equilibrium conditions in the laboratory, which varied from 23°C and 75% R.H. in the summer to 18°C and 30% R.H. in the winter. The effect of these variables is therefore included with the coefficients A_{on} and A_{op} , and the modified calendering equation as given by Equation 4.3 was used in the curve fitting:

$$\mu_{n} = a_{on} + a_{Ln} \log_{10} L + a_{Sn} \log_{10} S + a_{Rn} \log_{10} R$$

$$\mu_{p} = a_{op} + a_{Lp} \log_{10} L + a_{Sp} \log_{10} S + a_{Rp} \log_{10} R$$
where
$$a_{on} = A_{on} + a_{\theta n} \theta + a_{Mn} M$$

$$a_{op} = A_{op} + a_{\theta p} \theta + a_{Mp} M$$
[4.3]

The variation of moisture content and temperature thus contributed to the random

error, as their effect was not accounted for explicitly. As described in Section 3.2, the effect of moisture content was estimated for the TMP data set, strains adjusted to a relative humidity of 50% were calculated, and the curve-fitting was repeated. (The effect of temperature was small due to the small range of ambient temperatures; the data set was thus not corrected for this variable.) These results are presented later in this Section.

The effect of initial bulk was not investigated in the case of the specialty newsprint from Boise-Cascade, so in this case the coefficients a_{on} and a_{op} are zero, and their effect is included in the coefficients A_n and A_p . As well, there was insufficient data to determine the radius coefficient for this paper. The intercepts for the two papers are thus not comparable, but the load and speed coefficients a_L and a_S are.

Equations 4.1, 4.2 and 4.3 are valid for a range of B_i defined by

$$-\frac{A_{n,p}}{\mu_{n,p}} \le B_i \le \frac{1 - A_{n,p}}{2\mu_{n,p}}$$
 [4.4]

The limits, described in Section 2.1, were also used as input to the analysis program, which decided on the basis of least squares which portion of the calendering equation applied for a given set of conditions. The lower limit applies only to the permanent strain, since at no time in this study was an in-nip strain of zero measured:

$$\begin{vmatrix}
B_p = B_i \\
\varepsilon_p = 0
\end{vmatrix}$$
when $B_i \le -\frac{A_p}{\mu_p}$ [4.5]

The upper limit was applied to both in-nip and permanent strain:

$$B_{n} = \frac{(1-A_{n})^{2}}{4\mu_{n}}$$

$$\varepsilon_{n} = 1 - \frac{(1-A_{n})^{2}}{4\mu_{n}B_{i}}$$
when $B_{i} \ge \frac{1-A_{n}}{2\mu_{n}}$
[4.6]

$$B_{p} = \frac{(1-A_{p})^{2}}{4\mu_{p}}$$

$$\varepsilon_{p} = 1 - \frac{(1-A_{p})^{2}}{4\mu_{p}B_{i}}$$
when $B_{i} \ge \frac{1-A_{p}}{2\mu_{p}}$
[4.7]

SYSTAT offers two methods for non-linear function minimization. The first method, known as a Quasi-Newton method, was described by Fletcher [28]. The method is an extension of Newton's method for root-finding, using the gradient of the function to decide the direction and magnitude of the change at each iteration. The second method, called the Simplex method, is described by O'Neill [59]. In this method, the function to be minimized, which depends on N independent variables, is evaluated at N+1 points in N dimensions, the points forming a simplex about an initial user-supplied approximation to the solution. In two dimensions, the simplex is a triangle; in three, a tetrahedron. Then the simplex is reflected, expanded or contracted by replacing the largest point with a new smaller one, either outside (reflection or expansion) or within the simplex (contraction). The method is slow, but it requires no estimate of the derivatives, as the Quasi-Newton method does, and is therefore more robust. As well it is more likely to find a global minimum to a function, rather than the closest local minimum, especially when the initial estimate is poor.

SYSTAT allows the user to specify the loss function to be minimized; the function selected here was the classical least-squares error estimate.

Both analysis methods yielded essentially similar results. Where results differed significantly, typically one method or the other converged to a solution whose 95%

confidence interval allowed the possibility that one or more coefficients were zero. These results were discarded.

The raw data, typical command scripts submitted to the program and typical program output are given in Appendix A1, and are summarized next.

Specialty newsprint results using the calendering equation

The coefficients determined from the Boise-Cascade specialty newsprint data are given in Table 4.1; typical data are plotted with the previously published data of Browne et al. [8, 9] along with the predicted curve in Figure 4.1. Figure 4.1 illustrates well the reduction in the scatter obtained with the new gauges. Raw data is given in Table A1.1, the command script in Table A1.2, and the program output in Table A1.3. It should be pointed out that the logarithm operator in Table A1.2 is the natural logarithm, so that the coefficients in Table A1.3 need to be changed to base 10 logarithm for comparison with previous work. This has been done in Table A1.7.

Data were obtained only for the purpose of comparing with previous work, so only certain combinations of load and speed were investigated, and few replicates were performed.

Compared to the data presented previously by Browne et al. [8, 9], the 95%

Table 4.1: Calendering coefficients, specialty newsprint (Boise-Cascade).

	In-nip	Permanent
A	-0.3874	-0.0740
S.E.	0.0362	0.1461
a _{Ln,Lp} (g/cm ³)	0.2597	0.0882
S.E.	0.0117	0.0097
a _{Sn,Sp} (g/cm ³)	-0.0111	-0.0272
S.E.	0.0023	0.0046
r²	0.995	0.94

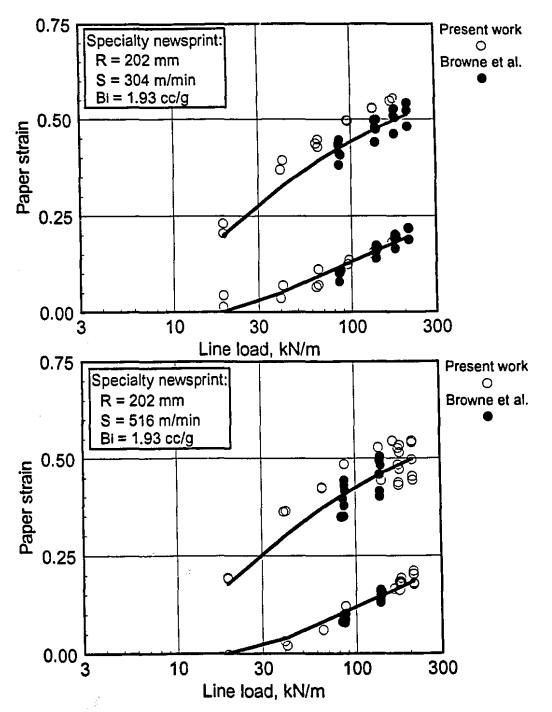


Figure 4.1 Strain vs. load, specialty newsprint: b) 516 m/min.

a) 304 m/min;

Curve fitting using the calendering equation.

confidence intervals of the permanent load coefficients a_{Lp} do not overlap, with the newer value being significantly lower. This is due to the acquisition of new data at lower loads, as seen in Figure 4.1, which displaces the limits of the calendering equation as determined by the statistical analysis program. Similarly in the in-nip case, the sum-of-squares is minimized if the linear portion of the curve corresponds to the steeper region at low loads, resulting in a substantially larger in-nip load coefficient a_{Ln} .

The permanent velocity coefficient a_{Sp} is slightly larger, for the same reason the load coefficients have changed; the in-nip coefficient a_{Sn} appears smaller but is statistically similar due to large standard errors in both coefficients.

TMP results using the calendering equation

The strain data set acquired with the paper made from TMP contains 1380 data points distributed over 312 unique combinations of load, speed, roll radius and initial bulk. Only one or two replicates exist for several of these combinations, while many replicates exist for the combinations which were easily attainable given the dynamics of the machinery. In other words, using all replicates in the curve-fitting process tends to weight those experimental conditions which imposed no extreme stresses on the equipment. In order to give equal weight to all replicates, combinations with three or more replicates were replaced by single average values of independent and dependent variables, while for combinations with only one or two replicates, all replicates were saved without averaging. The result was a summary data set of 359 data points, containing all the information in the larger set but in a more compact form. This set was used as input to the statistical analysis package.

Calendering coefficients for the TMP are presented in Tables 4.2 and 4.3, and typical data and curves are plotted in Figures 4.2 to 4.7. Along with the estimates, the tables give the standard error (S.E.), the coefficient of variation C_V and the 95% confidence limits. The overall standard error is the root mean square residual computed by comparing measured and predicted bulks in cm³/g.

Data points plotted in the figures come from the full set of all 1380 replicates, not the averaged summary set used to calculate the regression coefficients. In Figure 4.2, the complete data set for one roll radius has been divided into subsets of low and high initial bulk; the scatter in these curves is thus due to sheet speed. Conditions used to generate the curves with the calendering equations are given. Subsequently data subsets corresponding to unique combinations of roll radius, sheet speed and initial bulk are plotted along with the calendering equation prediction and the 95% confidence limits on that prediction. The upper limit of the calendering equation is indicated on each curve with an arrow; the lower limit occurs for ε_p when the predicted value of ε_p is less than zero.

Raw data, SYSTAT command scripts and SYSTAT output are given in Tables

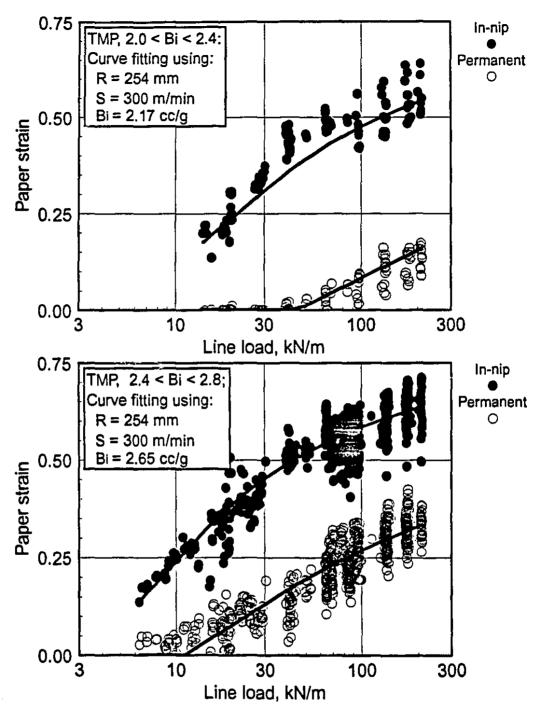


Figure 4.2 Strain vs. load, 202 mm, all data, TMP:

a) low initial bulks;
b) high initial bulks.

Curve fitting using the calendering equation.

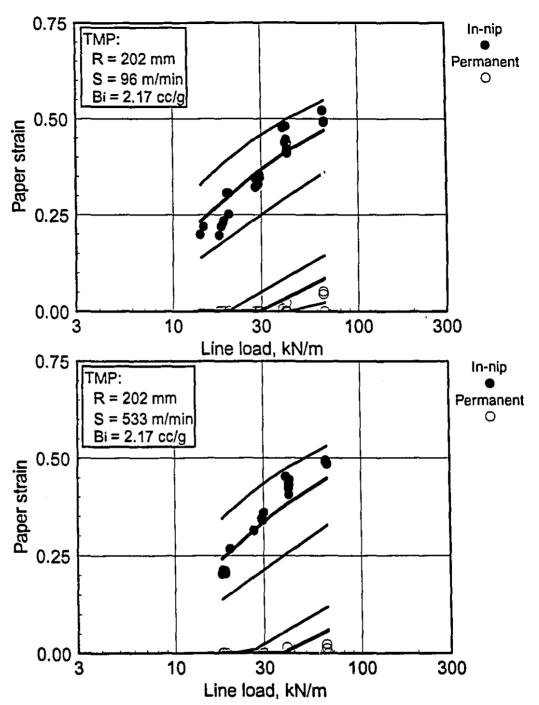


Figure 4.3 Strain vs. load, low initial bulks, TMP:

a) 96 m/min;
b) 533 m/min.

Curve fitting using the calendering equation.

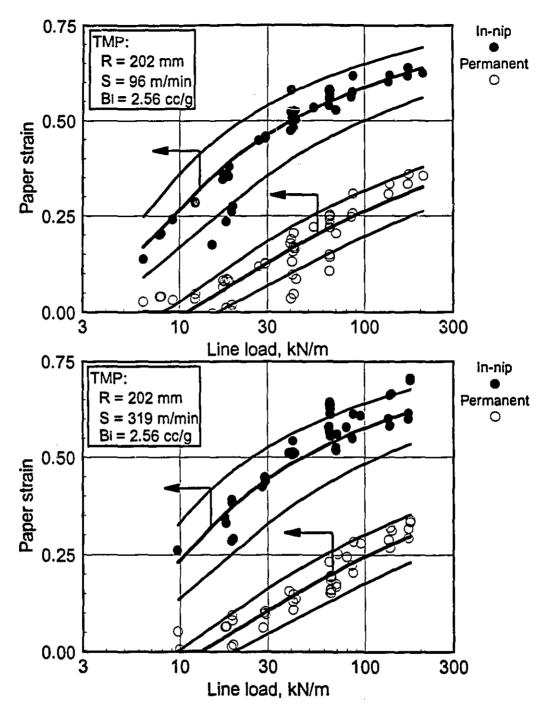


Figure 4.4 Strain vs. load, 202 mm, 2.56 cm³/g: a) 96 m/min; b) 319 m/min. Curve fitting using the calendering equation.

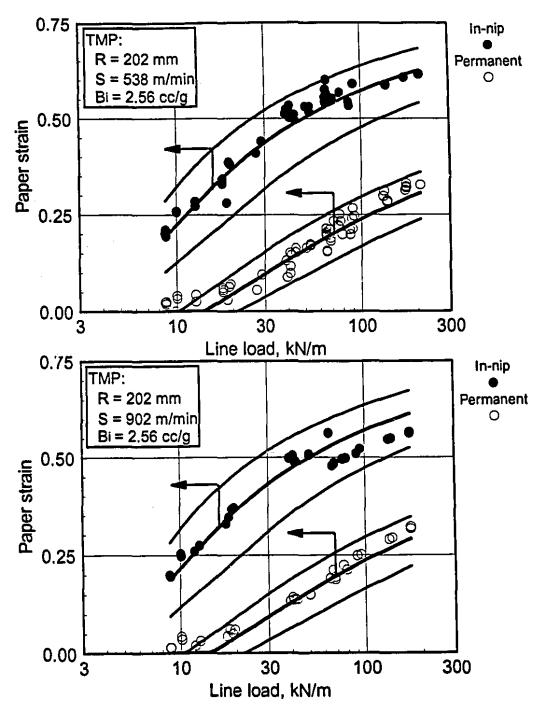


Figure 4.5 Strain vs. load, 202 mm, $2.56 \text{ cm}^3/\text{g}$:

1: -

a) 538 m/min;

b) 902 m/min.

Curve fitting using the calendering equation.

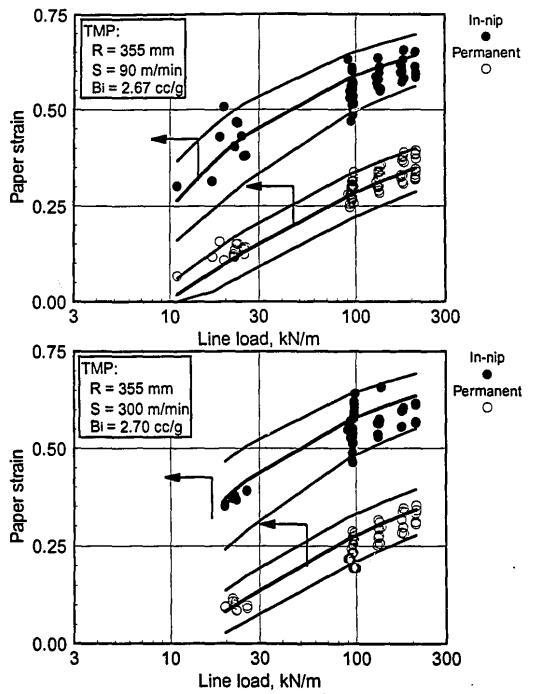


Figure 4.6 Strain vs. load, 355 mm, 2.69 cm³/g: a) 90 m/min; b) 300 m/min. Curve fitting using the calendering equation.

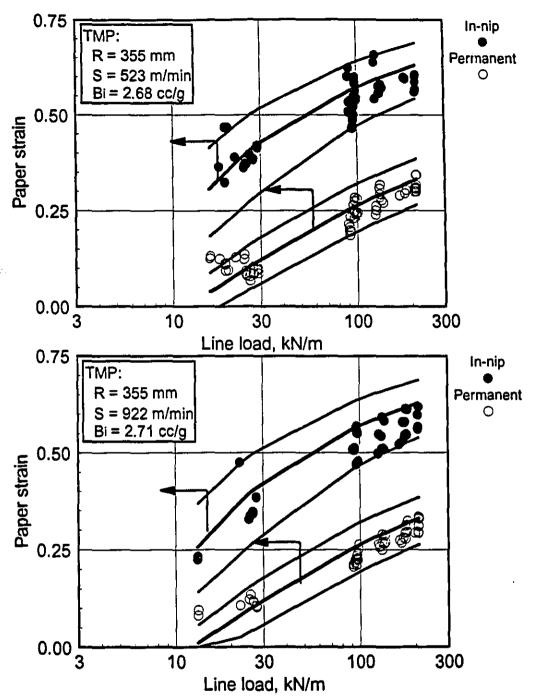


Figure 4.7 Strain vs. load, 355 mm, 2.69 cm³/g:

a) 523 m/min;
b) 922 m/min.

Curve fitting using the calendering equation.

TABLE 4.2: In-nip calendering coefficients, TMP (F. Soucy).

Parameter	Estimate	S.E. (C _V)	Lower 95%	Upper 95%
An	-0.3629	0.0764 (0.21)	-0.5131	-0.2127
a _{on} (g/cm ³)	0.0418	0.0202 (0.48)	0.0021	0.0815
a _{Ln} (g/cm ³)	0.1936	0.0193 (0.10)	0.1557	0.2319
a _{Rn} (g/cm ³)	-0.0771	0.0173 (0.22)	-0.1112	-0.0431
a _{Sn} (g/cm ³)	-0.0219	0.0058 (0.27)	-0.0332	-0.0108
r ²	0.8703			
S.E., cm ³ /g	0.119			

A1.4, A1.5 and A1.6. Results converted to base 10 logarithm are given in Table A1.8.

In-nip coefficients, presented in Table 4.2, have not been determined in the past. The values reported are statistically different from zero, and show that the magnitudes of the effect of the various independent variables on in-nip strain is the same as for the permanent strain, i.e. load has the greatest effect, foilowed by roll radius, then sheet speed.

Crotogino et al. [18, 22] proposed the calendering equation and measured permanent coefficients for a variety of newsprints. The permanent coefficients in Table 4.3 can thus be compared directly with their coefficients, which are given in Table 4.4.

Comparing the intercepts A_p and a_{op} first, it can be seen that the magnitude of A_p as reported here is slightly higher. This coefficient is a measure of the ease with which a permanent strain can be imposed. For given calendering conditions and initial bulk, the term μB_i will be constant; if the absolute value of A_p is larger the permanent deformation will be smaller. The coefficient A_p is thus a paper property, and the higher

TABLE 4.3: Permanent calendering coefficients, TMP (F. Soucy).

Parameter	Estimate	S.E.	Lower 95%	Upper 95%
A _p	-0.7017	0.0625 (0.09)	-0.8246	-0.5788
a _{op} (g/cm ³)	0.1743	0.0176 (0.10)	0.1398	0.2089
a _{Lp} (g/cm ³)	0.1135	0.0078 (0.07)	0.0981	0.1289
a _{Rp} (g/cm ³)	-0.0187	0.0092 (0.49)	-0.0368	-0.0005
a _{Sp} (g/cm ³)	-0.0158	0.0021 (0.13)	-0.0200	-0.0115
r ²	0.9141		·	
S.E., cm ³ /g	0.094		. <u></u>	

magnitude presented here means that this paper is less easily deformed than those used in the earlier work. This confirms previous work by Mitchell et al. [56], which showed that papers made from TMP pulps are more difficult to calender.

The coefficient a_{op} varies quite a lot in the previous study, to the point where the sign changes for different pulp types. No physical explanation for the significance of this coefficient has been found, although it can be suggested that it corrects for all the other effects not already included in the calendering equation, such as freeness, formation, pressing and drying conditions, etc. In this study, a_{op} also includes the terms $a_{\theta}\theta$ + $a_{M}M$.

The load coefficient a_{Lp} reported here is slightly higher than those given previously by Crotogino et al.; however this coefficient also appears to be a paper property, since Crotogino et al. [18, 22] have reported values ranging from 0.117 to 0.148. Thus the value of 0.1135 given here is not unreasonable.

The sheet speed coefficient a_{Sp} is 25% lower in the present study. The range of

TABLE 4.4: Permanent calendering coefficients, from Crotogino et al.

	Trial #1	Trial #2	Trial #3	Trial #4
A _p	-0.500	-0.334	-0.289	-0.417
a _o (g/cm ³)	0.0498	-0.00102	-0.00209	0.0568
a _L (g/cm ³)	0.0988	0.0922	0.0914	0.0912
a _R (g/cm ³)	-0.0390	-0.0374	-0.0353	-0.0354
a _S (g/cm ³)	-0.0208	-0.0175	-0.0208	-0.0204
a _T (g/cm ³ °C)	0.000943	0.000876	0.000567	0.000863
a _M (g/cm ³)	0.00545	0.00462	0.00529	0.00520
r ²	0.928	0.939	0.930	0.887
S.E., cm ³ /g	0.107	0.109	0.105	0.117

earlier results shows that a_{Sp} , like the other coefficients, is a paper property which varies from furnish to furnish.

Finally, the root mean square of the residual error reported by Crotogino et al. [22] is similar to the error reported here. Its magnitude can be attributed, firstly, to the random nature of paper structure which subjects strain measurements to a certain unavoidable variation, and secondly, to the uncertainty in the uncontrolled independent variables temperature and moisture content. Of the two, Crotogino et al. [22] have shown that moisture content has the larger effect; in this study its value depended on ambient relative humidity, which varied over a wide range. Using $a_{\rm M}=0.005$, the increase in predicted permanent strain for an initial bulk of 2.65 cm³/g, as moisture content increases from 5% to 10%, is 0.066.

The permanent data may also be compared with work published by Kerekes [44, 46], who measured a load coefficient of 0.090 and a speed coefficient of -0.018 after correction for the effect of initial bulk. Kerekes also published estimates of a_L and a_S based on the work of Colley and Peel [17]; these values are 0.102 and -0.012, respectively. The load coefficients are slightly smaller than the new value of 0.1135, but both speed coefficients are statistically similar to the value of -0.0158 measured here.

Radius and moisture effects

Radius effects have only been measured once in the past, as described below. Generally, radius coefficients have been calculated from the average of the load and speed coefficients, as suggested by Kerekes [45]:

$$a_{R} = -\left(\frac{a_{L} + a_{S}}{2}\right)$$
 [4.8]

Using the results from Table 4.3, Equation 4.8 predicts $a_{Rp} = -0.0489$, which is more than double the measured value reported. In the in-nip case, it can be seen from Table 4.2 that Kerekes' prediction gives $a_{\tilde{x}^{i}n} = -0.0859$, which is not statistically different from the measured value of -0.0771.

Baumgarten and Göttsching [5] measured permanent calendering coefficients for a variety of newsprints. Their load and speed coefficients agree with those presented here once the effect of bulk is removed. However, their roll radius coefficient agrees with Kerekes' prediction rather than the value in Table 4.3.

The standard errors for the two radius coefficients presented here are large, particularly for the permanent data where the 95% confidence interval just barely excludes the possibility of $a_{\rm Rp}=0$. As mentioned in Chapter 3, the equipment was not located in a controlled environment. As well, roll changes involved a time-consuming rebuild of the calender. For this reason, roll sets were not changed until a full data set had been acquired, a process often requiring several months. By the time the next roll pair had been installed, seasonal changes in relative humidity had caused significant changes in the paper moisture content, and thus its compressibility. This resulted in the possibility of a significant systematic error due to the combined effect of roll radius and sheet moisture content.

Strain data was therefore corrected to 50% relative humidity as follows. Equilibrium moisture content was estimated from the ambient relative humidity using published data for kraft, Bond et al. [6], and for newsprint, Crotogino [20]. Next, the

TABLE 4.5: In-nip calendering coefficients after moisture content corrections, TMP (F. Soucy).

Parameter	Estimate	S.E. (C _V)	% change in estimate
A _n	-0.3647	0.0753 (0.21)	0.5
a _{on} (g/cm ³)	0.0247	0.0194 (0.79)	-40.9
a _{Ln} (g/cm ³)	0.1920	0.0189 (0.10)	-0.8
a _{Rn} (g/cm ³)	-0.1068	0.0184 (0.17)	38.5
a _{Sn} (g/cm ³)	-0.0216	0.0058 (0.27)	-1.4
r ²	0.86		•
S.E., cm ³ /g	0.119		

moisture coefficients for the calendering equation were estimated from data published by Crotogino et al. [22], who measured $a_{Mp}=0.005$; this value was essentially the same for four different pulp types. Colley and Peel [17] measured the effect of moisture content on compressed thickness; from that work it can be deduced that $a_{Mn} \approx 0.85 a_{Mp}$. From these, the following amount was calculated and added to each strain measurement:

$$\varepsilon_{\text{corr}} = B_i a_M (M - M_{50})$$
 [4.9]

where M is the moisture content corresponding to ambient conditions, and $M_{50} = 7.69\%$ is the moisture content corresponding to 50% R.H. The regressions performed for the original data set were repeated on the new set; results are given in Tables 4.5 and 4.6. Along with the standard error and coefficient of variation, the tables give the percent change in the estimate from those given in Tables 4.2 and 4.3.

The only coefficients which are statistically different are the radius coefficients

TABLE 4.6: Permanent calendering coefficients after moisture content corrections, TMP (F. Soucy).

Parameter	Estimate	S.E. (C _V)	% change in estimate
A _p	-0.6095	0.0188 (0.03)	-13.1
a _{op} (g/cm ³)	0.1256	0.0101 (0.08)	-27.9
a _{Lp} (g/cm ³)	0.1085	0.0041 (0.04)	-4.4
a _{Rp} (g/cm ³)	-0.0530	0.0085 (0.16)	183.4
a _{Sp} (g/cm ³)	-0.0170	0.0018 (0.11)	7.6
r ²	0.90		
S.E., cm ³ /g	0.093		

 a_R and the intercepts A and a_o . Of the four intercepts, a_{on} is no longer statistically different from zero; however, published values of a_{op} range from -0.002 to +0.057 for different furnishes (Crotogino et al. [22] and Table 4.4). This coefficient accounts for all the pulp properties and paper machine parameters not explicitly included in the calendering equation, and so can vary substantially from case to case.

Using the new correlation, Kerekes' prediction is now verified for both in-nip and permanent strains at the 95% confidence level. As well, the coefficients of variation for the radius coefficients are reduced significantly, especially for $a_{\rm Rp}$. Thus it appears that the moisture content correction eliminates a large portion of the unexplained error in the first set of results.

The calendering equation revised

Residual bulks computed from the full set of all TMP replicates are plotted in terms of initial bulk in Figure 4.8 for both in-nip and permanent data. As can be seen, there is a slight tendency for the residuals to be positive at low and high bulks, and negative at medium values. (Plots of residuals versus other independent variables show no such trends). In the permanent case, this implies that the basic form of the calendering equation, which is linear in Bi, is not entirely correct. Assuming it is not the result of a systematic error in bulk measurements, several methods can be proposed for dealing with the trend in the residuals. One alternative is to add a second order term to the calendering equation:

TABLE 4.7: Permanent calendering coefficients for Equation 4.10.

Parameter	Estimate
\mathtt{A}_1	-0.5503
a _o (g/cm ³)	0.1145
a _L (g/cm ³)	0.1056
a _R (g/cm ³)	-0.0167
a _S (g/cm ³)	-0.0141
$\mu_{\rm p2}~({\rm g/cm^3})^2$	0.0014
r ²	0.9188
S.E., cm ³ /g	0.094

$$\varepsilon = A_1 + \mu_{p1}B_i + \mu_{p2}B_i^2$$
 [4.10]

The results of fitting this equation to the permanent data yields a value of μ_2 which is statistically different from zero, as shown in Table 4.7, but which is an order of magnitude smaller than the other coefficients. Given a large initial bulk of 3.2 cm³/g, the change in ε_p due to $\mu_{p2}B_i^2$ is small at 0.014 when typical permanent strain values of 0.100 and up are considered. Compared with the normal calendering equation, the root mean square of the residuals is unchanged and the value of r^2 is slightly lower. Finally, there is no obvious theoretical basis for stating that ε_p should be a parabolic function of

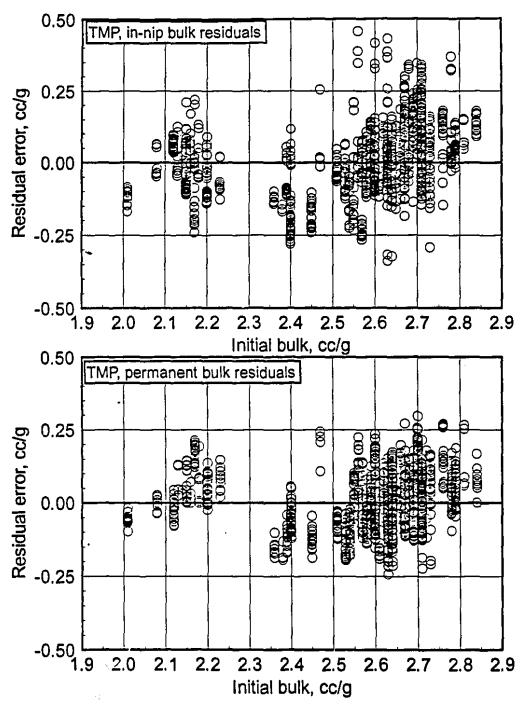


Figure 4.8 Bulk residuals vs. initial bulk,

- a) In-nip data;
- b) Permanent data

Calculated using the calendering equation.

B_i. The increased computational effort associated with the addition of a second-order term may not therefore be offset by any significant improvement.

It can also be suggested that one or more of the coefficients contained in the nip intensity factor μ_p may themselves be functions of B_i . Both in-nip and permanent load coefficients were found to change slightly when low bulk data is ignored. However, the range of initial bulks was too small and there was insufficient data at lower bulks to be able to extract statistically meaningful relationships between the coefficients and B_i . As well, there is no theoretical basis for suggesting any of the coefficients are functions of B_i since the calendering equation itself is not based on any theoretical considerations but on curve fitting results. The benefits resulting from the addition of a second curve-fitting exercise are therefore doubtful.

Another alternative to the calendering equation is suggested by classifying the data reported here according to which portion of the calendering equation applies. For each data point in the full TMP set, the in-nip and permanent nip intensity factors μ_n , μ_p were computed using the coefficients determined by SYSTAT. Regimes based on the limits of the calendering equation were defined next:

Regime 0:
$$B_i \leq -\frac{A}{\mu}$$
 $\varepsilon = 0$

Regime 1: $-\frac{A}{\mu} \leq B_i \leq \frac{1-A}{2\mu}$ $\varepsilon = A + \mu B_i$ [4.11]

Regime 2: $B_i \geq \frac{1-A}{2\mu}$ $\varepsilon = 1 - \frac{(1-A)^2}{4\mu B_i}$

Data in Regime 0 falls below the lower limit of the calendering equation, as described in Section 2.1; there is no permanent deformation. In Regime 1 the calendering equation applies, and in Regime 2 the upper limit has been exceeded. The number of data points in each Regime is given in Table 4.8, from which it will be seen that 74% of the permanent strain data, and 95% of the in-nip data fall above their respective upper limits.

TABLE 4.8: Distribution of data according to calendering equation limits.

	Regime 0	Regime 1	Regime 2
$\epsilon_{ m p}$	90 (6.5%)	270 (19.5%)	1021 (74%)
ε_{n}		75 (5.4%)	1305 (94.6%)

(Note that there is no in-nip data in Regime 0, since there were no experimental conditions yielding an in-nip strain of zero.)

Most of the permanent data and virtually all of the in-nip data falls above the upper limit, where strain is proportional to $1/\mu B_i$. The expression for strain in Regime 2 was thus rewritten in a linear form:

$$\frac{1}{1-\varepsilon} = \frac{4\mu B_i}{(1-A)^2}$$
 [4.12]

Recalling that apparent density ρ is the inverse of bulk, and defining new constants, Equation 4.12 can be rewritten:

$$\frac{\rho}{\rho_i} = A_2 + \frac{\mu_2}{\rho_i} \tag{4.13}$$

where

$$\mu_2 = a_{02} + a_{S2} \log_{10} S + a_{L2} \log_{10} L + a_{R2} \log_{10} R$$
 [4.14]

Equation 4.13 is linear in its coefficients, and can therefore be fitted using any linear regression package, such as the one included in most spreadsheets; iterative methods are not necessary. Comparing Equation 4.13 with Equation 4.12 gives an expression relating the coefficients of μ_2 to those of the calendering equation, illustrated with the load coefficient:

TABLE 4.9: Coefficients for Equations 4.13 and 4.14.

	A ₂ , S.E.	a _{O2} , S.E.	a _{1.2} , S.E.	a _{S2} , S.E.	a _{R2} , S.E.	r², S.E.
$\rho_{\rm n}/\rho_{\rm i}$	1.3371	-0.4742	0.4304	-0.0534	-0.2129	0.74
	0.1968	0.0765	0.0148	0.0129	0.0432	0.123 cc/g
$ ho_{ m p}/ ho_{ m i}$	1.1684	-0.2582	0.1870	-0.0334	-0.0612	0.82
	0.0397	0.0317	0.0063	0.0031	0.0110	0.177 cc/g

TABLE 4.10: Calendering coefficients given by Equation 4.15.

	a _L	a _S	a _R
$\varepsilon_{ m n}$	0.2004	-0.0249	-0.0991
$arepsilon_{ m p}$	0.1211	-0.0216	-0.0397

$$a_{L} = \frac{(1-A)^2}{4} a_{L2}$$
 [4.15]

Given $A_n = -0.36$ and $A_p = -0.60$, as generated by SYSTAT, Equation 4.13 can be fit to the data in Regime 2, then the calendering coefficients can be determined using Equation 4.15. This was done using the summary data set of 359 data points after data points in Regimes 0 and 1 were identified and removed; results are given in Tables 4.9 and 4.10 and are plotted in Figures 4.9 to 4.12.

All coefficients are significant at the 95% level, and the calendering equation coefficients are correctly predicted. However, Equation 4.13 fits the data less well than

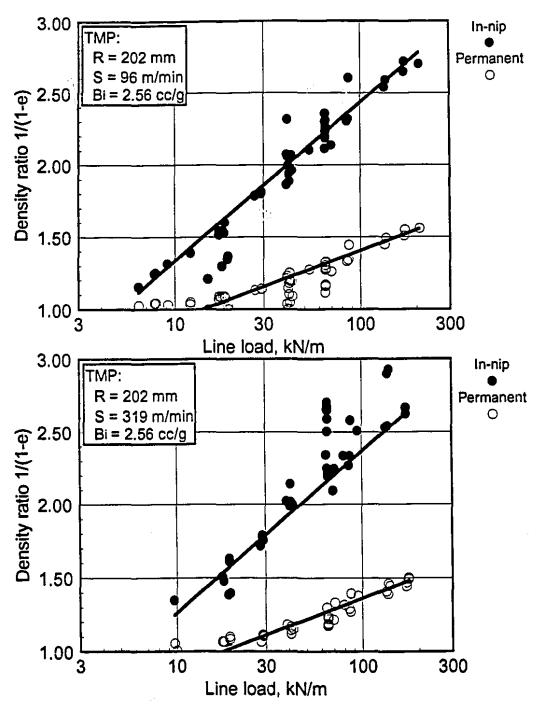


Figure 4.9 Density ratio vs. load, 202 mm, 2.56 cm³/g: a) 96 m/min; b) 319 m/min. Curve fitting using Equation 4.13.

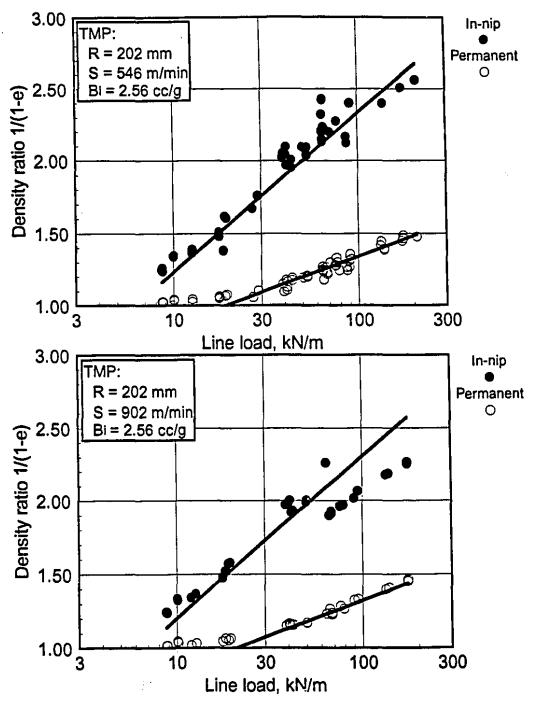


Figure 4.10 Density ratio vs. load, 202 mm, 2.56 cm³/g:

a) 546 m/min;b) 902 m/min

Curve fitting using Equation 4.13.

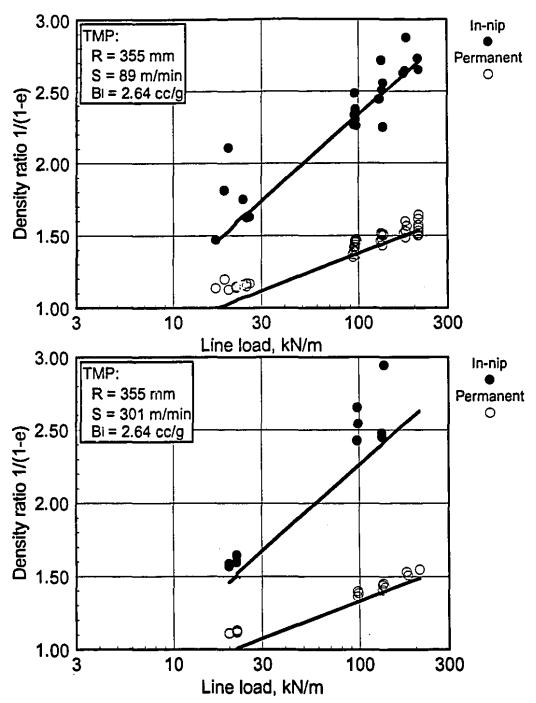


Figure 4.11 Density ratio vs. load, 355 mm, 2.64 cm³/g:

- a) 89 m/min;
- b) 301 m/min.

Curve fitting using Equation 4.13.

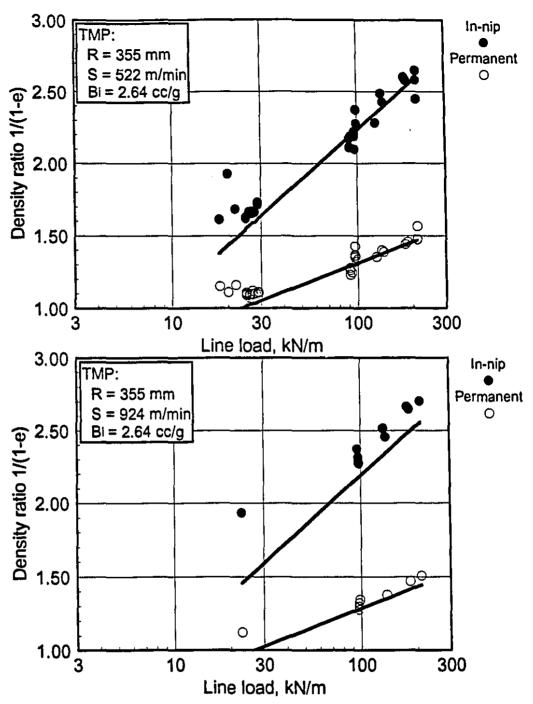


Figure 4.12 Density ratio vs. load, 355 mm, 2.64 cm³/g: a) 522 m/min; b) 924 m/min.

Curve fitting using Equation 4.13.

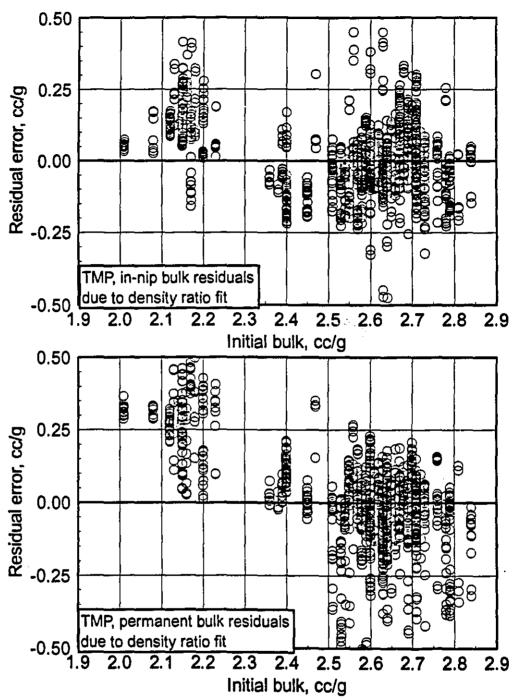


Figure 4.13 Bulk residuals vs. initial bulk,

- a) In-nip data,
- b) Permanent data

Curve fitting using Equation 4.13.

the calendering equation itself, with lower values of r^2 and higher standard errors, as seen in the plot of the residuals, Figure 4.13, and the numerical values in Table 4.9. The plot of the residual bulk also shows that the apparent effect of initial bulk has not been accounted for; this is sensible, since Equation 4.13 is simply an algebraic manipulation of the calendering equation.

The benefit of this method thus lies in the use of linear instead of non-linear regression methods, and it may be most useful in estimating in-nip strains since very little load is required to exceed the upper limit of the in-nip calendering equation.

High basis weight streaks in a light weight sheet

It has been suggested by Haglund [33] that a thick streak in a lightweight sheet might be deformed in a nip to the extent that it winds up thinner after calendering than the surrounding, initially thinner sheet. The reasoning proposed was that the thicker area, carrying most of the load in the nip, would see much higher pressures than the adjoining area and would thus be the only portion of the sheet to be calendered. This proposed behaviour can now be tested using the in-nip and permanent calendering equations. Haglund [33] took roll bending into account; the analysis here will first assume an infinite roll bending strength, then a qualitative assessment of the effect of roll bending will be made.

Two cases of a thick streak in a thin sheet are illustrated in Figure 4.14 and in Table 4.11. The in-nip and permanent calendering equations, complete with limits as described in Section 4.1.1 (Tables 4.5 and 4.6), were used to generate strain curves at an initial bulk of 2.4 cm³/g (Figure 4.14a) or 3.0 cm³/g (Figure 4.14b). Conditions used in the simulation were a sheet speed of 300 m/min and a roll radius of 254 mm. In Figure 4.14a, the variation from low to high weight was assumed small (45 to 55 g/m²), while a large variation was used in Figure 4.14b (30 to 60 g/m²). If the sheet is then compressed to a thickness of 60 μ m in the nip, the two areas see different strains (point A and B) due to the different initial calipers. Permanent strains are higher in the heavier spots (point D), but the lighter spots (point C) are still thinner.

This is illustrated again in Figure 4.15, where the same data has been re-plotted in terms of thickness rather than strain. Both spots are compressed to the same 60 μ m thickness in-nip, at points A and B. After the nip, point A recovers to point C, while B recovers to D. In both cases the initially thicker spot is still thicker after the nip, although the calendering process has been successful in reducing the variation. There is no obvious combination of two points A and B on a horizontal line which yield a thickness at D less than that at C, at least for this paper type. A paper with a permanent strain curve closer to the in-nip curve, i.e. a paper which recovers less after the nip,

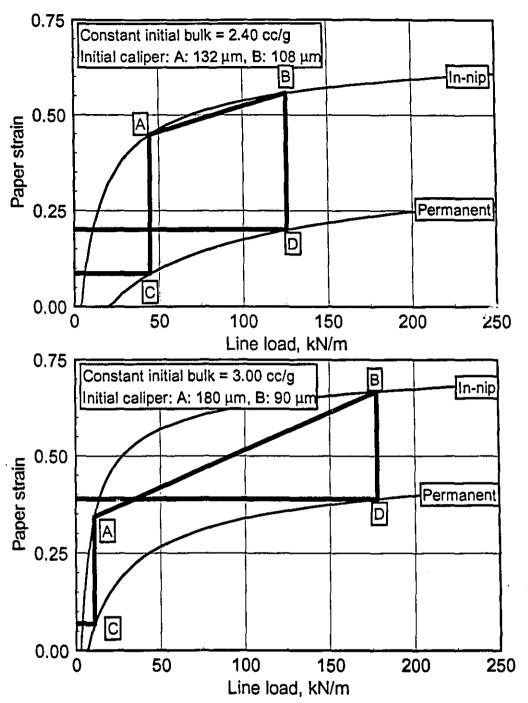


Figure 4.14 Strain in a calender for a thick streak in a thin sheet:
a) low variation; b) high variation.
Curve fitting using the calendering equations for TMP.

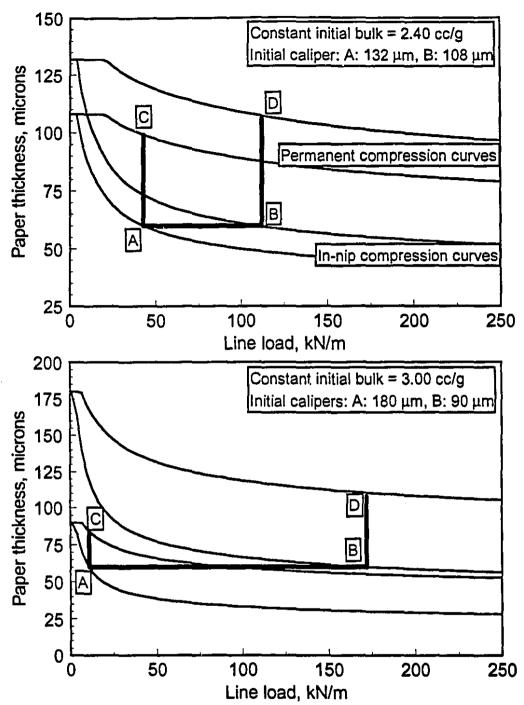


Figure 4.15 Thickness in a calender for a thick streak in a thin sheet:
a) low variation; b) high variation.
Curve fitting using the calendering equations for TMP.

might exhibit this behaviour when the in-nip strain at A is low enough for the permanent strain at C to be zero and the strain at D is high enough that the thickness there is lower than at C.

The effect of roll bending is to alter the nip profile so that nip thickness is not a constant across the width of a calender. This results in point A in Figure 4.15 being displaced slightly along the compression curve so that line AB is no longer horizontal. Line AC must remain vertical, however, so that point C is displaced an equal amount horizontally. There is still no possible set of points A and B yielding D less than C.

A "high basis weight streak" is defined in terms of bulk and caliper. A heavy

TABLE 4.11: Conditions in a calender for a thick streak in a thin sheet.

	Heavy spot (55 g/m ²) (Figures 4.14a, 4.15a)						-
	Bulk	Caliper	Strain	Bulk	Caliper	Strain	
Initial	2.40	132		2.40	108		
In-nip	1.08	60	0.55	1.33	60	0.45	
Final	1.92	106	0.20	2.21	99	0.09	
	1	y spot (60 g res 4.14b, 4		Light spot (30 g/m ²) (Figures 4.14b, 4.15b)			
	Bulk	Caliper	Strain	Bulk	Caliper	Strain	
Initial	3.00	180		3.00	90	***	
In-nip	1.00	60	0.67	2.00	60	0.33	
Final	1.80	108	0.39	2.76	83	0.07	

streak can thus be made up of a taller stack of identical fibres, as above, or it can consist of an equally thick stack of denser fibres. The case of a dense streak is illustrated in Table 4.12. When the sheet is compressed to $60 \mu m$ in the nip, both areas see the same strain (0.50) since the initial thicknesses were the same. The heavy spot now sees less permanent strain and is thus still thicker than the surrounding area after calendering.

For this particular type of TMP pulp, then, a heavy streak may see more strain, which is a relative measure of deformation, but in absolute terms will always be as thick or thicker than a surrounding lightweight area. In the most likely case, where the same type of fibre is deposited in a thicker stack due to variations at the headbox, the heavy spot sees greater permanent strain, but this is not enough to make it thinner than the surrounding, fighter area.

TABLE 4.12: Conditions in a calender for a dense streak.

- <u>'</u>	Heavy spot (55 g/m ²)			Ligh	t spot (45 g	/m ²)
	Bulk	Caliper	Strain	Bulk	Caliper	Strain
Initial	2.18	120	••	2.67	120	
In-nip	1.09	60	0.50	1.34	60	0.50
Final	1.96	108	0.10	1.65	74	0.38
	Heav	y spot (60 g	g/m ²)	Light spot (30 g/m ²)		
	Bulk	Caliper	Strain	Bulk	Caliper	Strain
Initial	2.00	120		4.00	120	
In-nip	1.00	60	0.50	2.00	60	0.50
Final	1.94	110	0.08	2.80	84	0.30

4.1.2 Master creep relationship

The summary data set extracted from the TMP raw data set as described in the previous section was also fitted to a modified version of the master creep equation proposed by Colley and Peel [17] for data obtained in a platen press:

$$\varepsilon_{n,p} = A \left[1 + \tanh (a_p \log_{10} P + a_t \log_{10} t + a_M M + a_\theta \theta + a_o) \right]$$
 [4.16]

where P is the maximum applied pressure in MPa, t the dwell time in seconds, M the sheet moisture content in percent, and θ the temperature in degrees Celsius. Kerekes [45] proposed a relationship between the coefficients of the variables P, t and L, V, R:

$$a_{L} = a_{P}$$

$$a_{S} = -a_{t}$$

$$a_{R} = \frac{a_{t} - a_{P}}{2}$$
[4.17]

Values for a_L and a_S as computed from data of Colley and Peel [17] and

Table 4.13: Master creep coefficients, from Kerekes and Colley and Peel.

	Colley &	Kerekes	
	In-nip	Permanent	Permanent
A	0.33	0.33	
a_L	1.090	0.900	0.890
a _S	-0.063	-0.130	-0.165
a _R predicted by Equation 4.17	-0.514	-0.385	-0.363

TABLE 4.14 In-nip coefficients, master creep equation, TMP (F.Soucy).

Parameter	Estimate	S.E.	Lower 95%	Upper 95%
$(\rho_{\rm n})_{\rm max} ({\rm g/cm}^3)$	1.2807	0.0668	1.1471	1.4143
a _{on}	-1.4197	0.1043	-1.6248	-1.2147
a_{L_D}	1.0076	0.0659	0.9344	1.1937
$a_{ m Rn}$	-0.4340	0.0905	-0.6120	-0.2560
a _{Sn}	-0.0976	0.0286	-0.1536	-0.0417
r ²	0.87		***************************************	<u> </u>
S.E., cm ³ /g	0.116			

measured by Kerekes [44, 46] are given in Table 4.13.

Haglund and Robertson [34] proposed that since $A(1 + \tanh x)$ is asymptotic to 2A for large x, there should be a relationship between A and the maximum density obtainable either in the nip, $(\rho_n)_{max}$, or after, $(\rho_p)_{max}$. In the current study, Equation 4.16 was modified by substituting Kerekes' relationship, Equation 4.17, and Haglund and Robertson's relationship for limiting densities:

$$\varepsilon_{n,p} = \frac{1}{2} \left(1 - \frac{\rho_i}{\rho_{max}} \right)_{n,p} (1 + \tanh \mu_{n,p})$$
 [4.18]

where the nip intensity factors μ_n , μ_p are defined as for the calendaring equation.

Curve fitting results were obtained for the TMP data using SYSTAT as described earlier for the calendering equation, with the limitation on initial bulk just given, and are listed in Tables 4.14 and 4.15.

TABLE 4.15: Permanent coefficients, master creep equation, TMP (F. Soucy).

Parameter	Estimate	S.E.	Lower 95%	Upper 95%
(ρ _{p)} max (g/cm ³)	0.5930	0.0074	0.5784	0.6077
a _{op}	-3.1502	0.2553	-3.6524	-2.6481
a _{Lp}	1.8863	0.3339	1.6876	2.0850
a _{Rp}	-0.8294	0.1932	-1.2095	-0.4490
a_{Sp}	-0.2729	0.0541	-0.3792	-0.1665
r ²	0.88			
S.E., cm ³ /g	0.107			

Since SYSTAT lacks a built-in hyperbolic function, the exponential definition was used:

$$\tanh \mu = \frac{e^{\mu} - e^{-\mu}}{e^{\mu} + e^{-\mu}}$$
 [4.19]

Figures 4.16 to 4.19 show typical TMP data with the master creep equation superimposed; it can be seen that the predicted strain curves are smoother than those predicted by the calendering equation, especially for permanent strains at low loads. This is due to the continuous nature of the function.

The limiting density in the nip is 1.28 g/cm³, slightly more than Haglund and Robertson's value of 1.20 but less than the accepted value for the density of cellulose of 1.50 g/cm³, which is presumably a limiting factor in compression. For a sheet with a certain number of large pores, the limiting density in the nip would be less than the density of fibre, so limiting values of 1.2 to 1.3 make sense.

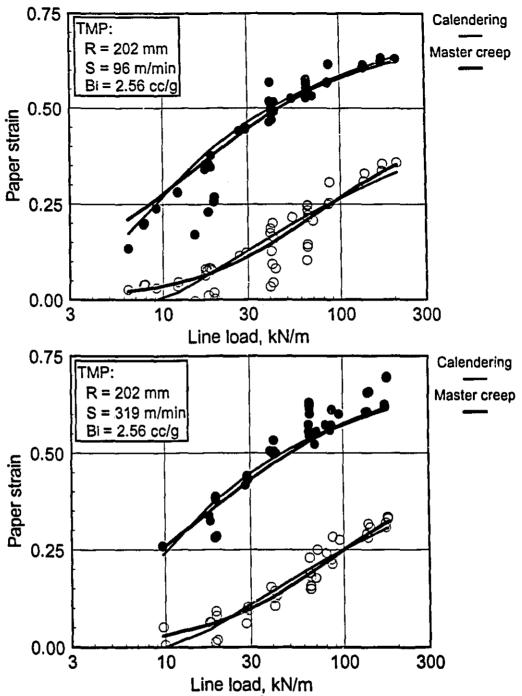


Figure 4.16 Strain vs. load, 202 mm, 2.56 cm³/g,

a) 96 m/min;

b) 319 m/min

Curve fitting using the master creep equation.

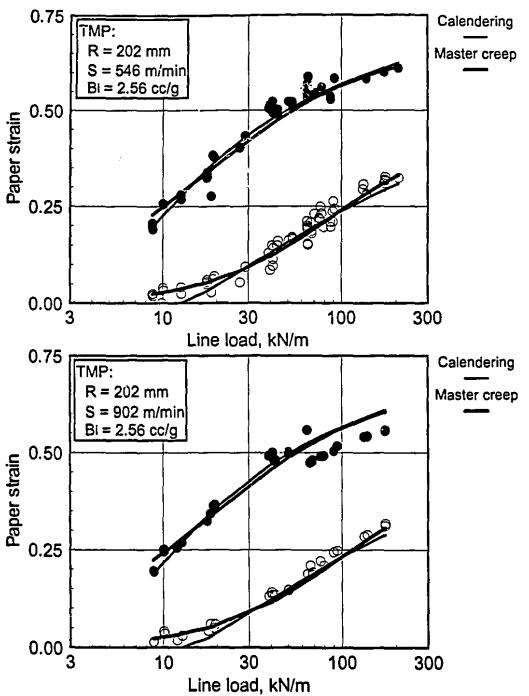


Figure 4.17 Strain vs. load, 202 mm, 2.56 cm³/g:

a) 546 m/min;
b) 902 m/min.

Curve fitting using the master creep equation.

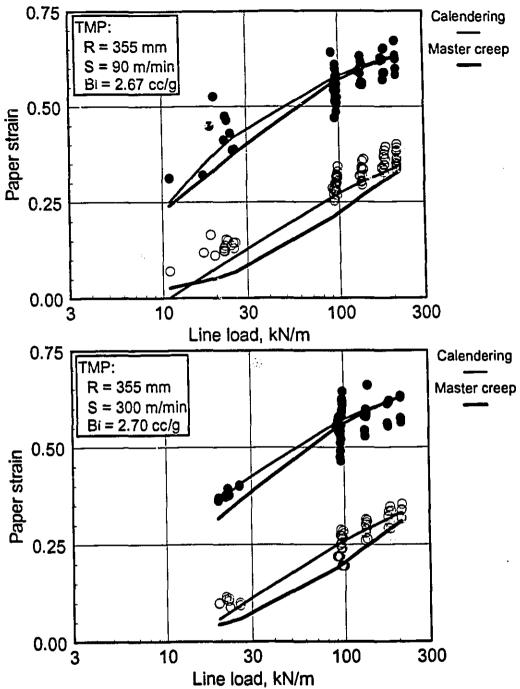


Figure 4.18 Strain vs. load, 355 mm, 2.68 cm³/g, a) 90 m/min, b) 300 m/min Curve fitting using the master creep equation.

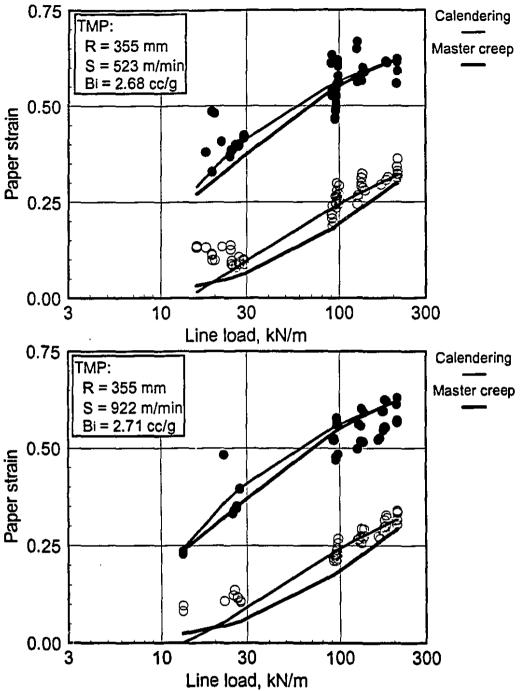


Figure 4.19 Strain vs. load, 355 mm, 2.69 cm³/g:

a) 523 m/min,
b) 922 m/min

Curve fitting using the master creep equation.

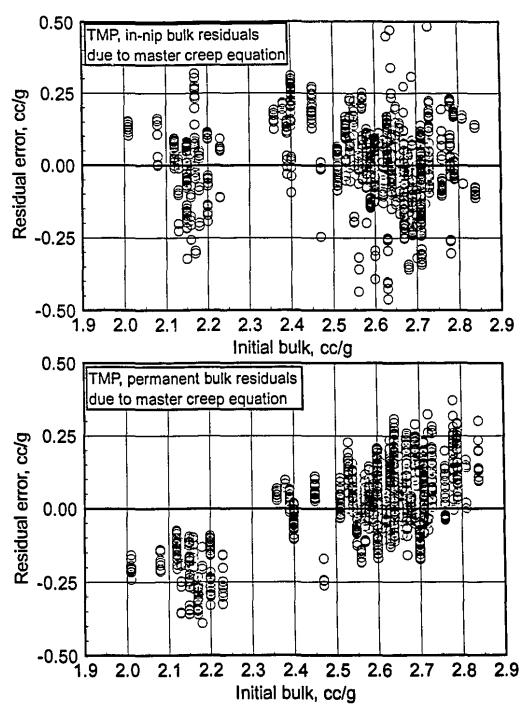


Figure 4.20 Bulk residuals vs. initial bulk,

- a) In-nip data,
- b) Permanent data

Curve fitting using the master creep equation.

In the permanent case, the limiting density is 0.59 g/cm³, corresponding to a minimum bulk of 1.70 cm³/g. This is comfortably lower than the lowest bulk attained after two nips, which was 1.85 cm³/g. After many more heavily loaded nips, a final bulk approaching 1.70 makes sense.

Residuals are plotted in Figure 4.20, where it can again be seen that the equation does not adequately predict the effect of initial bulk on permanent strain. Equation 4.18 predicts strain to be inversely proportional to the initial bulk where the calendering equation was linear with bulk; comparing with the permanent bulk residuals in Figure 4.8b it can be seen that where the calendering equation consistently predicts a lower permanent strain at low initial bulks, the master creep equation consistently predicts a higher strain.

Thus if strain is not a linear function of initial bulk, neither is it inversely proportional. There was insufficient data obtained in the present work to explore the relationship between initial bulk and strain any further.

Overall the fit of the master creep equation is similar to the calendering equation, with similar values of r^2 and RMS residuals. However, non-linear regression methods are necessary if a large range of nip intensities is to be investigated, and computational effort is greater than for the calendering equation. With these difficulties in mind, the continuous nature of the function, compared with the pieced-together nature of the calendering equation, make it an interesting alternative for industrial users wishing to investigate the effect of machine changes on paper caliper, especially at low nip intensities.

Comparing Table 4.14 with the in-nip data reported by Colley and Peel in Table 4.13, all coefficients are statistically similar. Comparing Equations 4.16 and 4.18, and using a typical initial bulk of 2.5 cm³/g, the value of A can be estimated at 0.34, identical to Colley and Peel's.

In the permanent case, however, as illustrated in Table 4.15, there is no agreement between the current predictions and published work. This can be attributed to the value of A_p . Using Equations 4.16 and 4.18, with $B_i = 2.5$ as for the in-nip data,

TABLE 4.16: Permanent coefficients, master creep equation, $(\rho_p)_{\text{max}} = 1.28$.

Parameter	Estimate	S.E.	Lower 95%	Upper 95%
(ρ _{p)} max (g/cm ³)	1.28	0.0 (fixed)	1.28	1.28
a _{op}	-2.4757	0.1461	-2.7631	-2.1883
a_{Lp}	1.0341	0.0440	0.9475	1.1207
a _{Rp}	-0.5395	0.1059	-0.7481	-0.3311
a _{Sp}	-0.1140	0.0315	-0.1759	-0.0518
r ²	0.76			
S.E., cm ³ /g	0.137			

 A_p is 0.16 compared in the current work compared with 0.33 previously; the low value of A_p here is then offset by higher values of the other coefficients. Effectively, previous work was based on the assumption that the limit is the same for both in-nip and permanent densities, at 1.20 to 1.30 g/cm³. This implies that it is possible to press a sheet so hard that it doesn't recover at all. Further, the coefficients in Table 4.15 are not directly comparable to those in Table 4.13. In order to make a comparison, Equation 4.18 was fitted to the permanent data with $(\rho_p)_{max}$ fixed at 1.28 g/cm³. The results are given in Table 4.16. Comparing these results with Table 4.13, it can be seen that the load coefficient is slightly higher, as it was for the calendering equation; the other coefficients are not different at the 95% level from Colley and Peel's data. Since the standard error and r^2 values are worse in Table 4.16, Table 4.15 contains the better fit, and the limiting density for permanent compression is substantially lower than the density of pure fibre.

4.2 Strain recovery after the nip

4.2.1 Strain recovery at short times after the nip

As mentioned in Section 3.1.1, the calender was equipped with modified industrial caliper gauges mounted 0.296 m and 1.050 m after the nip. Data from these gauges was frequently unuseable due to bounce of the floating head; however sufficient data was acquired at industrially relevant speeds to draw some useful conclusions.

Defining $\varepsilon_r(t)$ the partially recovered strain at a distance x after the nip, the recovery time t required for the sheet to cover a distance x is then x/V, where V is the sheet speed in m/s: V = S/60. Recovery times thus range from 20 to 200 ms for the first gauge at x = 0.296, and 60 to 600 ms for the second gauge at x = 1.050 m.

In general, the scatter in the strain data at x = 1.050 made it impossible to detect any statistically significant difference with those measured 24 hours after calendering. The first conclusion is therefore that within the accuracy of these online gauges, paper recovers from a calender pulse extremely quickly. As well, this was true even at higher speeds, when the recovery time was shorter. The recovery therefore occurs more quickly at higher speeds, implying the dwell time in the nip has an effect on the recovery time.

To quantify this statement, the strains measured in the nip, at x = 0.296 m and after 24 hours were used to estimate the recovery time constant τ for paper leaving a calender nip:

$$e^{-t/\tau} = \frac{\varepsilon_p - \varepsilon_r(t)}{\varepsilon_p - \varepsilon_e}$$
 [4.20]

where $\varepsilon_{\rm e}$ is the strain at the nip exit. This strain is unknown since it has not been measured, but must be in the range $\varepsilon_{\rm n} < \varepsilon_{\rm e} < \varepsilon_{\rm r}(t=0.296/{\rm V})$. Since $\varepsilon_{\rm r}(t=0.296/{\rm V})$ is still significantly greater than $\varepsilon_{\rm r}(t=1.050/{\rm V})$, the time constant must be similar to 0.296/V, implying that $\varepsilon_{\rm e}$ is closer to $\varepsilon_{\rm n}$ than to $\varepsilon_{\rm r}(t=0.296/{\rm V})$. Equation 4.20 was thus used to estimate τ using $\varepsilon_{\rm e} = \varepsilon_{\rm n}$.

The value of τ obtained using Equation 4.20 was not a constant for all conditions,

TABLE 4.17: Curve fitting constants for Equation 4.21.

m,	21.612
S.E.	1.655
n,	-1.100
S.E.	0.029
r ²	0.69

but was a strong function of sheet speed. Using the speed S in m/min, the time constant (in seconds) was found to fit Equation 4.21:

$$\tau = mS^n \qquad [4.21]$$

with values of m and n given in Table 4.17.

The possibility of a systematic error related to speed was considered and rejected. Errors occured when the floating gauge head, excited by small amplitude thickness changes in the moving sheet, ceased to contact the sheet due

to bounce, or when paper dust or other material became lodged between the sheet and the head. The result of either of these errors was a false high thickness, causing a false low strain; the true strain $\varepsilon_r(t)$ (if either of these errors had occured) would thus be higher than the estimate used in Equations 4.20 and 4.21. Bounce did occur more frequently at higher speeds, but was immediately obvious from a plot of the sensor output signal versus time; these data points were discarded. Accumulated material in the gauge was also obvious since it caused a step change in the thickness reading from the sensor; compressed air was used to clean the head when this occured, and as a matter of routine after every few runs.

The time constant τ and the curve given by Equation 4.21 are plotted in Figure 4.21a, from which it can be seen that at industrially relevant speeds the time constant approaches a value in the vicinity of 10 ms.

It will be seen in Chapter 5 that the ingoing nip half-length a (the distance from the point of first contact between paper and roll to the nip centre line) depends on the innip strain ε_n , roll radius R and initial sheet half-thickness z_i :

$$a = \sqrt{2 \, \varepsilon_{\rm a} R \, z_{\rm i}} \tag{4.22}$$

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The nip dwell time can thus characterized by the ingoing dwell time a/V, which

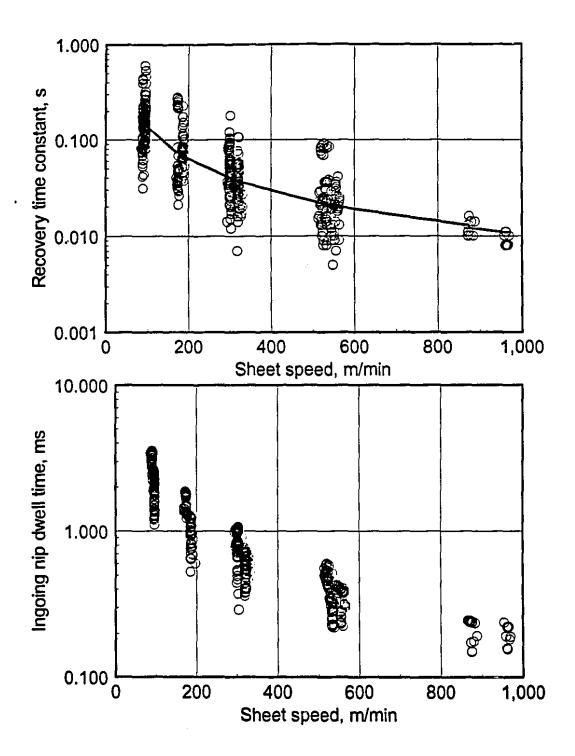


Figure 4.21 a) Strain recovery time constant τ , computed using Equation 4.21 b) Ingoing nip dwell time a/V, computed using Equation 4.22.

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becomes a sensible choice for normalizing the strain recovery time t. While load does not appear explicitly in Equation 4.22, its effect is present in the form of the measured in-nip strain ε_n . As seen in Figure 4.21b, typical values of a/V at industrially relevant speeds are in the range 0.100 to 0.300 ms.

In Chapter 5 it will also be seen that if the ingoing sheet thickness is small compared to the roll radius, the strain while the paper is in contact with the roll surface is parabolic:

$$\varepsilon(t) = \varepsilon_n \left(\frac{Vt}{a}\right) \left(2 - \frac{Vt}{a}\right)$$
 [4.23]

Paper strain is thus zero when Vt/a = 0 at the nip entrance, and again (for an elastic material) when Vt/a = 2. Paper is not elastic, and so contact with the roll ends somewhere in the range $1 \le Vt/a \le 2$. Analogous to the ingoing half-length a is the outgoing half-length b, which is equal to a for an elastic material, and must be in the range 0 to b_{max} for paper:

$$b_{\text{max}} = \sqrt{2Rz_{i}(\varepsilon_{n} - \varepsilon_{p})}$$

$$= a \sqrt{1 - \frac{\varepsilon_{p}}{\varepsilon_{n}}}$$
[4.24]

where b_{max} is the outgoing half-length if paper recovers immediately to its final thickness.

Figures 4.22 and 4.23 show the parabola defined by Equation 4.23, along with typical recovery strain data and strain recovery curves predicted using τ from Equation 4.21 substituted into Equation 4.20. The curves have been plotted in terms of normalized strain $\varepsilon/\varepsilon_n$ as a function of normalized time Vt/a.

Equation 4.20 with $\varepsilon_{\rm e}=\varepsilon_{\rm n}$ predicts the paper ceases to contact the roll surface almost immediately after passing the nip centerline; in effect, b tends to zero. This appears to be contradicted by at least two published articles.

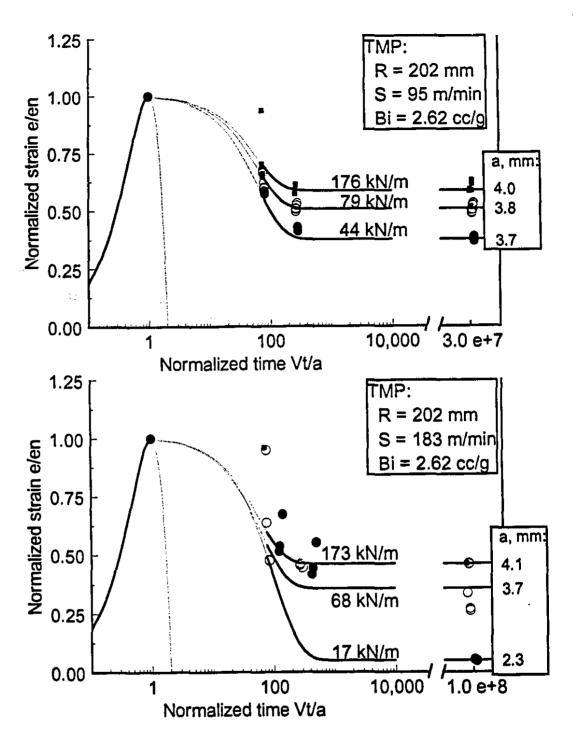


Figure 4.22 Strain recovery, 202 mm, 2.61 cm³/g:

- a) 95 m/min ($\tau = 145$ ms),
- b) 183 m/min ($\tau = 70$ ms).

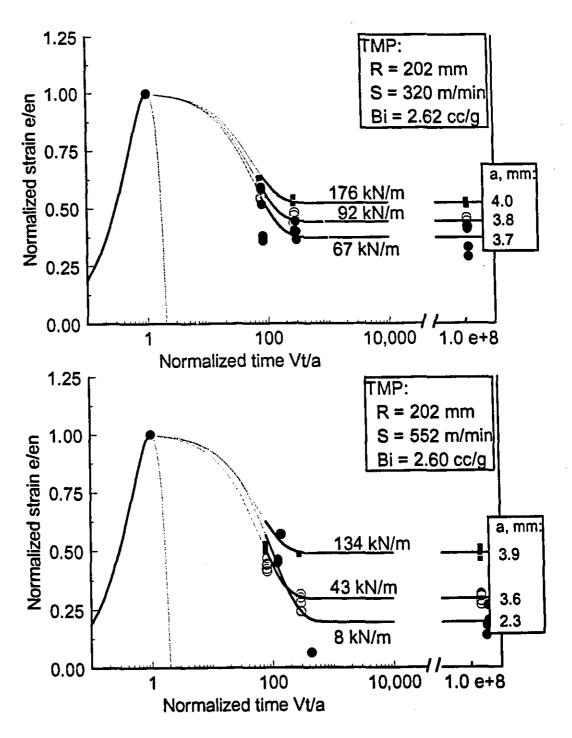


Figure 4.23 Strain recovery, 202 mm, 2.61 cm³/g:

a) 320 m/min ($\tau = 38$ ms),

b) 552 m/min ($\tau = 21$ ms).

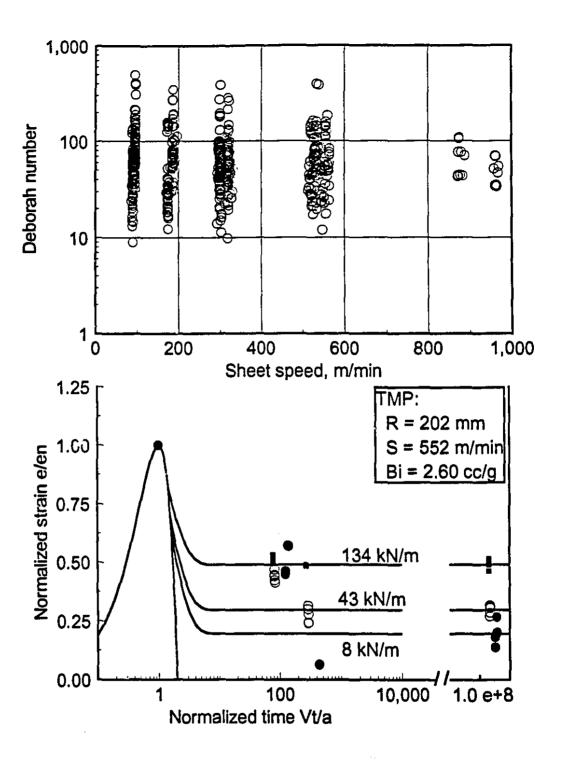


Figure 4.24 a) Deborah number $\xi = V\tau/a$; b) Strain recovery for a material with $\xi = 1$.

Keller [43] has published graphs of pressure vs. MD distance in the nip which clearly show paper remaining in contact with the roll surface for a substantial portion of the range $1 \le Vt/a \le 2$; b is smaller than but of similar magnitude to a. The roll radius, sheet speed and nip load are all similar to the results presented here. However, one of the rolls in Keller's work was soft, with an elastic cover, which tends to keep the sheet in contact with the hard roll; the apparent large value of b is due to instantaneous elastic recovery of the roll surface, not viscoelastic recovery of the paper.

Watanabe et al. [80] measured paper strain at several points in the nip of a small bench scale printing tester; their results also show a significant b. However, reported nip dwell times are long, with $a/V \approx 5$ ms, and they estimate the recovery time τ at about 2.5 ms. From these numbers, the Deborah number for their conditions is about 0.5, compared with typical values of 30 to 100 found in the present work, cf. Figure 4.24a. As pointed out in Section 2.2, viscoelastic behaviour may be obtained with a sheet speed $V \approx a/\tau$, equivalent to a Deborah number of about 1. Under these conditions the strain recovery would appear as in Figure 4.24b, where it can be seen that the value of b is now significantly larger than zero. Thus the large values of b which can be deduced from their work are not inconsistent with the essentially negligible value reported here.

4.2.2 Recoverable strain at long times after the nip

Recoverable strain is defined as the difference between in-nip and permanent strain, and is the amount of strain which is recovered when a load is removed:

$$\varepsilon_{\rm R} = \varepsilon_{\rm n} - \varepsilon_{\rm p}$$
 [4.25]

Figures 4.25 to 4.28 illustrate the recoverable strain ε_R computed from measured values of ε_n and ε_p (given by filled squares) and from the calendering equation estimates for ε_n and ε_p presented in Tables 4.5 and 4.6 (solid line). The calendering equation estimates were obtained using the full equations, including the limits as described by Equations 4.4, 4.5, 4.6 and 4.7. In-nip and permanent strains are also plotted for comparison.

At low loads behaviour is essentially elastic; the permanent strain becomes very small and virtually all the in-nip strain is recoverable. At moderate loads, the permanent strain begins to increase, but the recoverable strain remains essentially constant, possibly decreasing slightly at higher loads.

In order to quantify the possibility of a decreasing ε_R at high loads, the recoverable strain data was fitted to a pair of line segments. The first segment, at loads less than some limiting load L_{lim} , has a steep positive slope and a zero intercept; the second segment has a small, possibly negative slope and an intercept of about 0.35:

$$\epsilon_{R} = m_{1}L$$
when $L < L_{lim}$

$$= m_{2}L + \epsilon_{o}$$
when $L \ge L_{lim}$
[4.26]

Since iterative methods were necessary to locate the unknown intersection L_{lim} , and since the small magnitude of m_2 caused instability in the numerical methods, Equation 4.26 was rewritten in a more robust form, with the slope m_2 defined explicitly:

TABLE 4.18: Parameters for recoverable strain ε_R as given by Equation 4.26.

Parameter	Estimate	Lower 95%	Upper 95%	
m ₁ , m/kN	0.0152	0.0140	0.0164	
L _{lim} , kN/m	20.00	19.94	20.06	
$\varepsilon_{ m o}$	0.3461	0.3320	0.3603	
€∞	0.3007	0.2886	0.3129	
m_1L_{lim}	0.3040	0.2687	0.3290	
m ₂ , m/kN	-0.000216	-0.000341	-0.000091	
r ²		0.241		

$$\mathbf{m_2} = \frac{\boldsymbol{\varepsilon_n} - \boldsymbol{\varepsilon_o}}{\mathbf{L_n}}$$
 [4.27]

where $L_{\infty}=210$ kN/m is the largest load obtained and ε_{∞} , the recoverable strain at a load L_{∞} , is the parameter controlling the slope. The parameters are thus m_1 , L_{\lim} , ε_0 and ε_{∞} , from which m_2 can be calculated; estimates of the parameters are given in Table 4.18.

While the value of r^2 is poor, due to the scatter inherent in subtracting two quantities which themselves are subject to substantial scatter, the 95% confidence intervals for all the parameters exclude the possibility of a value of zero. The root mean square of the residual strain error computed using Equations 4.26 and 4.27 is 0.060, compared with the value of 0.044 computed when the measured data is compared to the difference between the calendering equation predictions. Thus the value of Equation 4.26 is in the predictions of the slope m_1 and the maximum recoverable strain m_1L_{lim} ; the

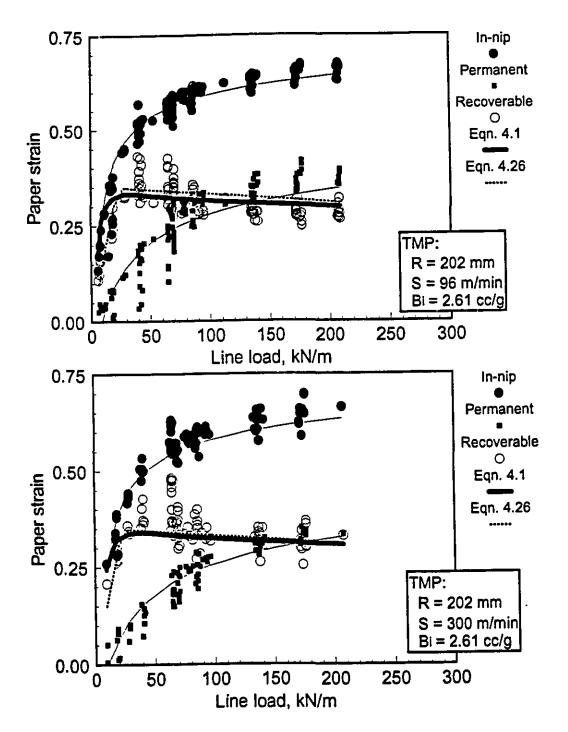


Figure 4.25 Recoverable strain $\varepsilon_{\rm n}\text{-}\varepsilon_{\rm p}$, 202 mm, 2.62 cm³/g:

- a) 96 m/min,
- b) 316 m/min.

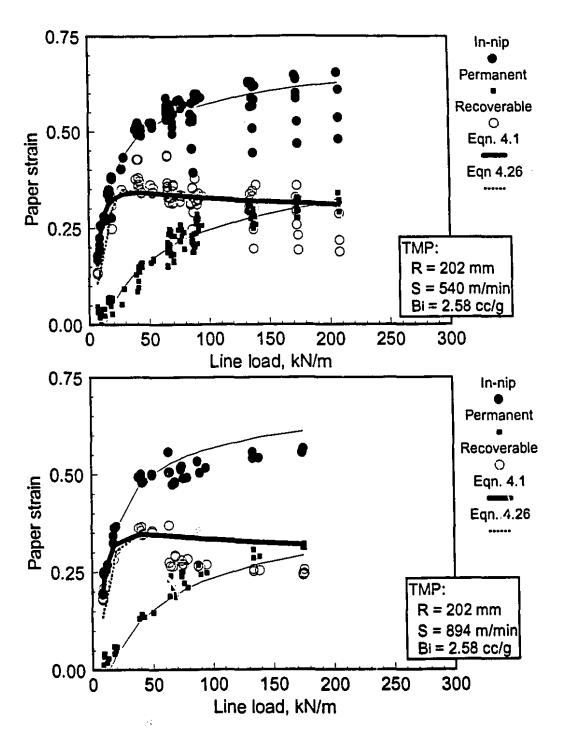


Figure 4.26 Recoverable strain ε_n - ε_p , 202 mm, 2.61 cm³/g:

- a) 542 m/min,
- b) 881 m/min.

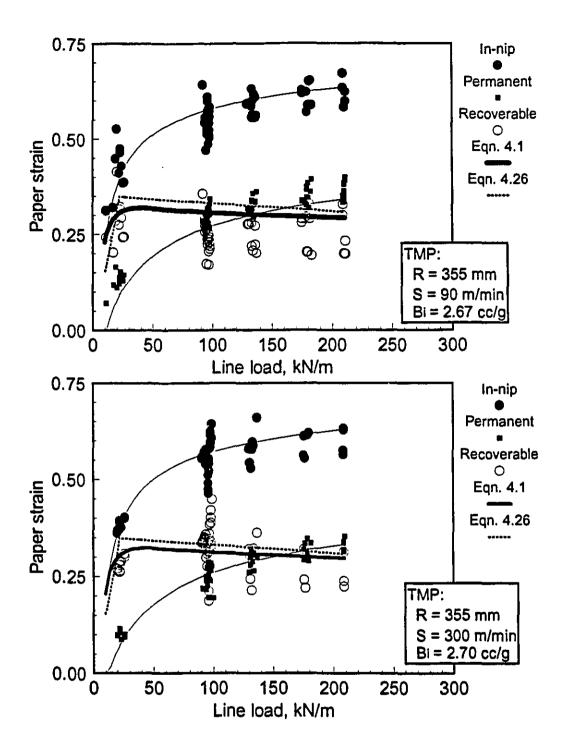


Figure 4.27 Recoverable strain $\varepsilon_{\rm n}\text{-}\varepsilon_{\rm p},~355~{\rm mm},~2.65~{\rm cm}^3/{\rm g}$:

a) 90 m/min,b) 301 m/min.

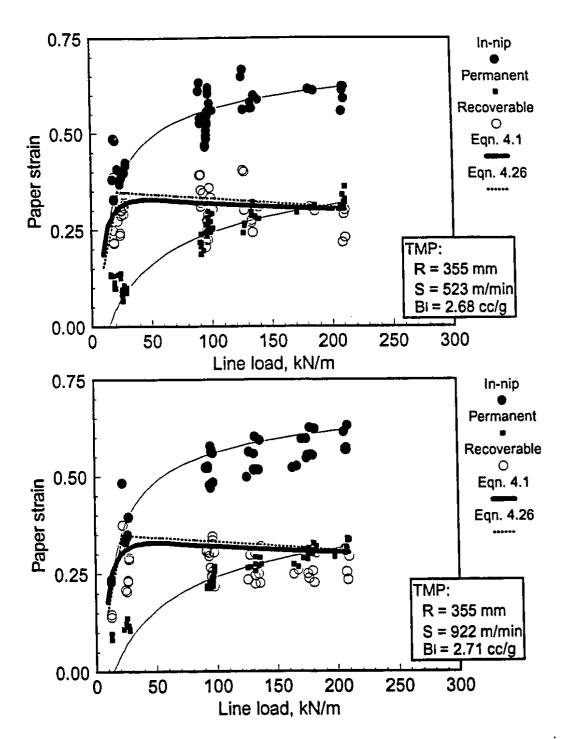


Figure 4.28 Recoverable strain $\varepsilon_{\rm n}\text{-}\varepsilon_{\rm p}$, 355 mm, 2.65 cm³/g:

a) 523 m/min,

b) 923 m/min.

calendering equations provide an adequate description of recoverable strain as long as the calendering limits are respected.

The shape of the recoverable strain curves suggests that there are two stages to paper compression in a calender nip. At low loads, the strain is essentially elastic (whether instantaneous or time-delayed); an apparent modulus may be obtained from the slope m_1 , which predicts 0.66 kN/m per 0.01 strain. Recalling $(a + b) \approx a$ from Section 4.3.1, an average nip length a for loads $L < L_{lim}$ was estimated at 3.121 mm. Converting line load to average nip pressure using this nip length gives an apparent elastic modulus of about 22 MPa for loads up to a few megapascals. This value is only approximate since it is based on four assumptions, first that the elastic recovery is instantaneous rather than time-delayed, that the parabolic stress profile in the nip can be approximated with a single average value L/a, that $b \approx 0$, and that the load vs. recoverable strain is indeed a straight line up to 20 kN/m. However, it is similar to the value of 39 MPa obtained by Mann et al. [54] for a different furnish; the similarity is improved when their statement that their methods tend to overestimate moduli by about 25%, and the probable curvature of the load vs. strain curve at very low loads are taken into account.

At higher loads, permanent deformation becomes substantial, but the recoverable strain remains essentially constant, ranging from 0.269 $< m_1 L_{\rm lim} < 0.329$ at 20 kN/m to 0.289 $< \epsilon_{\infty} < 0.313$ at 210 kN/m.

It can thus be suggested that the recoverable behaviour at very low loads is due to fibres which are stressed to some point well below the yield point, while permanent strain is due to irreversible fibre deformation; the yield point is 20 kN/m, or about 7 MPa when a = 3.121 mm. In other words, the results imply that fibres behave elastically up to a point (with either instantaneous or time-delayed full recovery), while all further strain beyond the yield point is permanent.

4.2.3 Relationship between in-nip and permanent strain

Ionides et al. [38] have suggested, in the course of an analysis of paper compression based on a statistical description of fibre distribution in the plane of a sheet, that a simple linear relation might exist between ε_n and ε_p . They suggest a relationship of the form

$$\varepsilon_{n} = c + m \varepsilon_{p} \qquad [4.28]$$

Using published data obtained with a platen press, they predict m = 0.75 for the approximate range $0.2 < \varepsilon_p < 0.4$.

Using all TMP data with initial bulk greater than 2.50 cm³/g, data in the range $0.2 < \varepsilon_p < 0.4$ was fitted to Equation 4.28. Data for the entire range of permanent strains, also with $B_i > 2.50$ cm³/g, was also fitted to Equation 4.28. The results of both fittings are given in Table 4.19, where it can be seen that all the coefficients are statistically non-zero. The subset $0.2 < \varepsilon_p < 0.4$ generated m = 0.67, statistically smaller than 0.75; however the full set generated m = 0.93, implying that moving the cutoff point from $\varepsilon_p = 0.2$ to $\varepsilon_p \approx 0.15$ would generate m ≈ 0.75 .

Both values of r^2 are poor, however, and in Figure 4.28 it can be seen that while it may be possible to fit a straight line to data in the range $0.2 < \varepsilon_p < 0.4$, a nonlinear relationship is necessary for the full range of strains investigated. Accordingly, a logarithmic fit was tried:

TABLE 4.19: Coefficients for Equation 4.28.

Range of ε_p :	m, S.E.	c, S.E.	r ²
$0.20 < \varepsilon_{\rm p} < 0.40$	0.666, 0.032	0.392, 0.040	0.37 (712 data points)
$0.00 < \varepsilon_{\rm p} < 0.45$	0.930, 0.020	0.319, 0.056	0.69 (1000 data points)

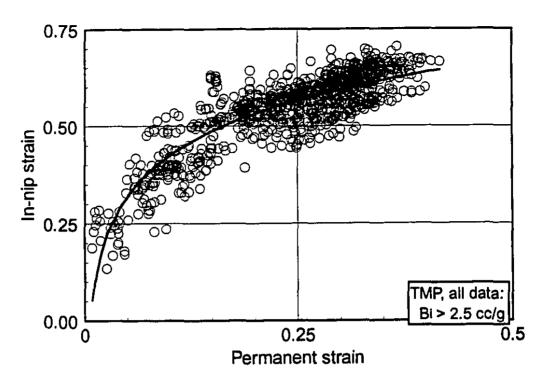


Figure 4.29 Relationship between $\varepsilon_{\rm n}$ and $\varepsilon_{\rm p}$; curve fitting using Equation 4.29.

$$\varepsilon_{\mathbf{n}} = c_0 + c_1 \log_{10} \varepsilon_{\mathbf{p}} \tag{4.29}$$

The coefficients defined in Equation 4.29 are given in Table 4.20 and the curve is plotted in Figure 4.29. The coefficients are all statistically non-zero, and the predicted curve passes (within 95% confidence limits) through the origin. If paper exhibits elastic behaviour at low loads (as suggested in Section 4.3.2), the in-nip strain at those low loads would be accompanied by a zero permanent strain, which the curve in Figure 4.29

TABLE 4.20: Coefficients for Equation 4.29.

c ₀ , S.E.	c ₁ , S.E.	r ²
0.778, 0.049	0.354, 0.007	0.74

predicts.

There is considerable scatter in the data, and the value of r^2 is low. The scatter is due in large measure to the effect of initial bulk. Strain data at low bulk fall below the curve in Figure 4.29; unfortunately there is insufficient data at low bulk to determine a statistically useful relationship. For the curve fitting, bulk ranged from 2.5 to 2.8 cm³/g, a substantial variation. The coefficients are also likely to be furnish dependent.

Nonetheless, this relationship fulfills one of the goals of this research project, namely facilitating design of a complete cross-direction control system for paper strain leaving a calender. Once the bulk and furnish dependencies have been elucidated, the curve-fitting results or a graph similar to Figure 4.29 can be used to estimate the in-nip strain existing at a given CD position, from the permanent strain measured at that position. If a lower or higher permanent strain is required locally, the required new local in-nip strain can also be estimated. The difference between the current and required in-nip strains gives explicitly the local roll deformation which is required. With knowledge of the roll response to an actuator input, the energy required to effect the required change in permanent strain can be calculated.

4.3 Machine-direction extension of paper in a rolling nip

Extension of paper in a calender nip was measured as described in Section 3.1. Results are described next and plotted in Figure 4.30. Figure 4.30a shows the measured MD sheet speed increase, or tensile strain, in terms of the change in sheet tension before and after the nip. (Data acquired when the magnitude of the sheet tension change was greater than 20% were discarded). Average sheet tension for the data given here was 627 N/m on the unwind side, with a standard deviation of 65 N/m, and 566 N/m on the wind-up side with a standard deviation of

13 N/m. The lower variability in the wind-up tension was due to computer control on this side; unwind tension was under manual control. The average tension change through the nip was -9.7%.

The speed increase was measured by counting pulses from two optical triggers over a set period of time. The resolution of the system corresponds to a strain of about 0.12%, which is the strain resul-

Table 4.21: Curve fitting constants for Figure 4.30b.

Slope (% draw per 1000	
kN/m) at 96 m/min	1.766
Standard error	0.335
Intercept	-0.025
Standard error	0.105
r ²	0.40

ting from a difference of 1 in the number of pulses counted. An error of ± 1 count therefore results in a strain error of $\pm 0.24\%$, a large amount considering the small overall range of strains measured. Nonetheless, it can be seen that the MD strain does not exceed 0.5%, or 0.005 in absolute terms; average MD strain was 0.106%, with a standard deviation of 0.144%. Compared to z-direction permanent strains ranging up to 30%, these strains are extremely small.

Next, the effect of line load can be seen in Figure 4.30b, where data at constant speed, radius and bulk have been plotted in terms of line load. A straight line has been fitted to the data; the constants defining the line are given in Table 4.21. The slope is non-zero at the 95% confidence level, while the intercept is not significantly different

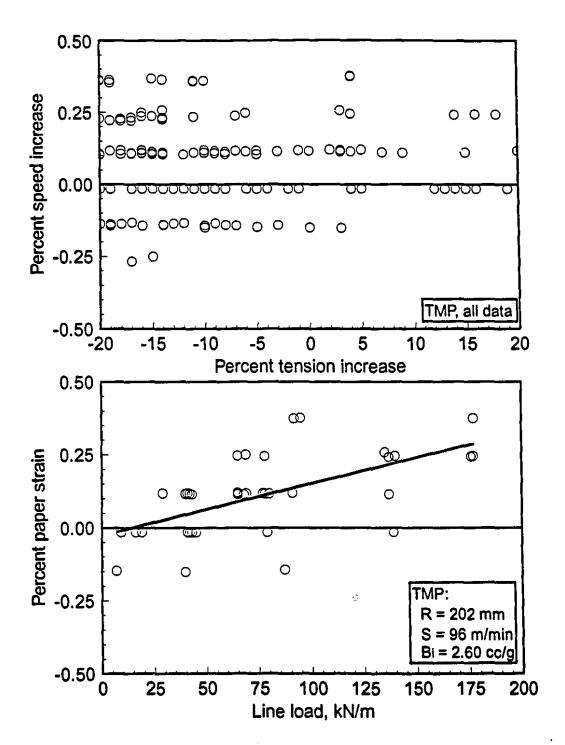


Figure 4.30 Machine direction strain,

a) as a function of sheet tension change;

31

b) as a function of line load.

from zero. The value of r^2 is low, reflecting large scatter in the data; however the zero intercept and positive slope are consistent with the expected behaviour of paper, i.e. that the strain vs. load curve should pass through the origin, and that paper will stretch as the compressive load is increased.

The small values of MD strains measured here are consistent with CD strain data published by Krenkel [47], Baumgarten [4], Gay et al. [29] and Mann et al. [54], and can be explained by considering the structured nature of paper. Since fibres are in contact at localized crossing points with large voids between them, there is ample space for the structure to deform in the z-direction before all the pores and gaps are filled. A typical fibre lying in the plane of the sheet may be a few millimetres long; as it enters the nip one end of that fibre is deflected downwards by only a few tens of microns. The fibre is bent more than it is stretched by this action, and there is very little in-plane stress imposed despite the large z-direction stresses. This is also confirmed by the Poisson's ratio measurements of Mann et al. [54]. Once the density of the sheet approaches that of individual fibres, further compression will result in greater MD and CD strains, as described by Gay et al. [29] and elaborated here.

Data at higher speed show smaller slopes with larger standard errors. This is partly due to the increasing scatter, and partly to the fact that at higher speed the paper is less highly stressed, leading to lower stretch. At industrial speeds, MD stretch becomes too small to measure using the equipment described here.

In conclusion, the quality of the MD strain measurements make it difficult to extract statistically meaningful results; however, it can be deduced that typical MD strains under industrial conditions are two orders of magnitude smaller than the corresponding z-direction strain, and that the MD strain increases with load and decreases with speed. This implies that compressive stresses applied at a point in a sheet dissipate quickly in the fibre network and are thus not felt by unstressed fibres at distances more than several fibre lengths away. Approximating paper compression with one-dimensional models ignoring shear stresses, as will be done in Chapter 5, is thus not unreasonable.

4.4 Conclusions

The experimental part of this research project leads to the following conclusions regarding calendering practice and theory:

The calendering equation still provides the best description of permanent paper deformation in a calender nip, except at low nip intensity where the master creep equation fits better. The dependency of both equations on bulk needs to be elucidated.

As in-nip paper deformation is generally above the upper limit of the calendering equation, describing deformation in terms of the density ratio ρ_n/ρ_i may be more useful in obtaining coefficients from laboratory in-nip measurements since that relationship fits as well as the calendering equation but only requires a multiple linear regression package rather than non-linear iterative methods.

Kerekes' prediction for the radius coefficient is verified.

A convenient relationship between in-nip and permanent strain can be found, and is probably a function (at least) of initial bulk and pulp properties. This relationship will permit estimating in-nip strain at a local CD position given local permanent strain.

Strain recovery immediately after an industrial nip is governed by a time constant whose magnitude is in the range 10 to 30 ms. At high speeds with small rolls, significant recoverable strain remains from the first nip at the entrance to the second nip. However, the effect is small.

Recoverable strain is essentially constant above a certain point, implying that fibres are essentially elastic (with time-dependent complete recovery) until some plastic limit is exceeded.

Machine direction stretch is two orders of magnitude less than compressive strain. Cross direction stretch was too small to measure by simple scaling, implying that it is less than 1 part in 70, or 1.5%. One-dimensional approximations ignoring shear stresses in the nip are therefore not unreasonable.

5. Modeling paper compression in a rolling nip

5.1 Introduction

The data presented in Chapter 4 show the viscoelastic behaviour of paper when compressed in a calender nip: there is partial, time-delayed strain recovery after the nip with very little instantaneous elastic recovery. The relationship between in-nip and permanent strains proposed in Section 4.2.3 and plotted in Figure 4.29 illustrates this behaviour. Data for a material exhibiting elastic recovery would fall on the ordinate since there would be no permanent strain regardless of in-nip strain; paper approaches this limit for loads at or below the lower limit attainable with the equipment described in Chapter 3. Similarly data for a viscous material which has no strain recovery ($\varepsilon_n = \varepsilon_p$) would fall on a line of slope 1; this is true of paper under moderate calendering conditions, although a portion of the total strain applied is not recoverable as it would be in a purely viscous material. Finally, under extreme conditions of high load and low speed in-nip paper strain levels off as sheet density in the nip approaches a limiting density constrained by the density of pure fibre. Paper thus exhibits viscoelastic behaviour when z-direction compressive stresses are in the range encountered in calendering.

The calendering equations fitted to the data in Chapter 4 describe this behaviour in purely empirical terms. The various coefficients, obtained by curve-fitting methods, were not related to more fundamental parameters describing fibre or paper properties. Several authors [45, 61, 62, 80] have used viscoelastic methods to provide more insight into the calendering process. These methods are now examined in the search for a relationship between the empirical results of Chapter 4 and the physical properties of fibres and paper.

Models for compression of a viscoelastic sheet due to an indenting object have been described by many authors [2, 27, 37, 45, 49, 50, 53, 57, 58, 76, 77, 78], either with or without rolling motion of the indenter. In the most general case, models must allow for significant shear stresses in the material as the indenting object (whether flat,

cylindrical, spherical or random) is pressed into it. In the case of paper in a calender nip, shear stresses may be neglected, as shown next.

As machine direction extension of paper in a calender nip was shown in Section 4.3 to be small, and others have shown the magnitude of cross-direction extension to be small [4, 29, 47, 54], paper compression can be approximated as a one-dimensional process. This can be demonstrated by considering a fibre lying in the MD on the surface of a sheet 120 μ m thick, as illustrated in Figure 5.1. Typical fibre dimensions (exaggerated for clarity in Figure 5.1) are 2 mm long and 10 μ m in diameter. As it enters the calender nip one end of the fibre is both flattened and displaced downward from the surface towards the sheet centerline 60 μ m below. If the in-nip strain is 0.50, the downward displacement is about 30 μ m. The new length of a fibre with one end at the nip centerline and one end just entering the nip, as deduced from geometry, is 2.0002 mm, which represents a negligibly small lengthwise strain of 0.01%. A fibre closer to the sheet midplane will see even less MD strain since the downward displacement goes to zero at the sheet centreline. The larger MD strains reported in Section 4.3 can be explained by fibre flattening, as described next.

The change in width of fibres lying in the CD is likely to contribute more to the MD strain than the lengthwise strain described above. When a fibre with diameter of 10 μ m is compressed, the width of the fibre will increase to 12 to 14 μ m in the nip depending on the wall thickness and how much open lumen remains. The increase in width corresponds to an in-plane strain on the scale of the fibre of 0.2 to 0.4, a large amount; however, the strain on the scale of the sheet is much smaller since fibres expand into inter-fibre voids as they compress and do not necessarily force adjacent fibres to displace. A change in the width of a fibre is likely to cause strain in the fibres crossing it; the extension of a fibre caused by 5 fibre crossings per mm of fibre length, each crossing extending 3 μ m, is 15 μ m, corresponding to a total lengthwise strain of 1.5%. This amount is consistent with measured MD and CD strains reported here and elswhere.

Overall, MD strain in a calender is due to the large deformation of CD oriented fibres being constrained by small deformations and by unkinking of the MD oriented

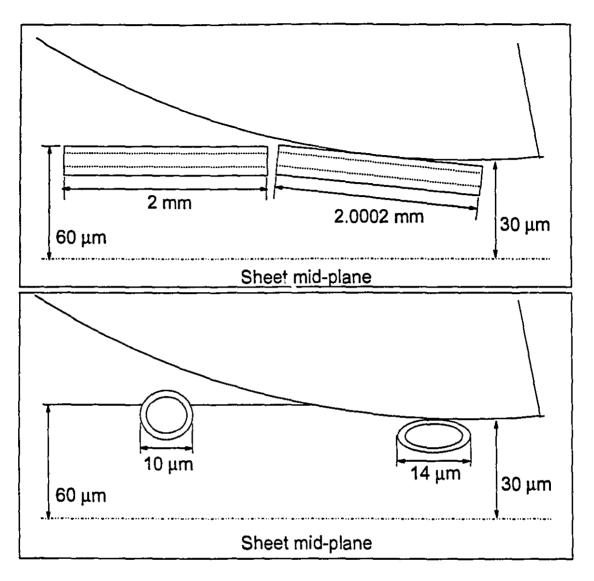


Figure 5.1 A typical fibre entering a calender nip. Fibre orientation: a) in the MD; b) in the CD.

fibres to which they are bonded. The three-dimensional structure of the sheet shifts and stretches, dissipating the large radial deformation of individual fibres. These conclusions are supported by experimental work of Mann et al. [54], as discussed in Sections 2.2 and 4.2.2. Their measurements of in-plane Poisson's ratios show that large z-direction strains result in very small CD or MD strains.

Thus modeling the viscoelastic behaviour of paper using one-dimensional models which neglect shear forces is reasonable. This approach will be taken in the discussion which follows. Quantitative results will then be compared with a qualitative description of fibre and paper structure.

5.2 Quantitative results: linear viscoelastic models

5.2.1 Geometry of a smooth, thin strip in a rolling nip

The analysis of May et al. [53] of the geometry of a thin strip in a rolling nip can be used to describe the strain history imposed on a sheet as it passes through the calender nip. Since strain history is a required element for viscoelastic modeling, the nip geometry is described next.

In the cartesian coordinate system shown in Figure 5.2, the origin is at the intersection of the plane of the undeformed sheet and the centreline joining the rolls. The x-axis is positive in the direction of sheet displacement, or machine direction, while the z-axis is positive downwards. The sheet mid-plane is assumed to be a plane of symmetry. Time is measured from the point of first contact between sheet and roll; thus t = 0 at the entrance to the nip when x = -a. Distances z_i , z_n and z_p are the initial, in-nip and permanent recovered sheet half-thicknesses, z(t) is the downward deflection of a point P(x,z) on the surface of the paper from the plane defining the initial paper surface, $z_0 = z_i - z_n$ is the maximum value of z(t) at the point of minimum paper thickness, and z_e is the half-thickness at the nip exit. Distances a and b are the ingoing and outgoing nip lengths; b_{max} is the value of b if the material recovers to its final thickness immediately on leaving the nip. Calendering variables are the roll radius R, sheet velocity V and line load L. The symbol V is used for sheet velocity in meters per second in order to distinguish it from sheet speed S in meters per minute, as used in previous chapters.

The expression for z(t), the deflection of a point with coordinates P(x,z) from the initial paper surface due to intrusion of the roll, is derived from the right-angle triangle with hypotenuse R and sides x and $R-z_0$. As the roll radius R is much larger than $z_0-z(t)$, terms in $[z_0-z(t)]^2$ can be ignored, and

$$z(t) = \frac{Vt}{2R}(2a - Vt)$$
 [5.1]

where

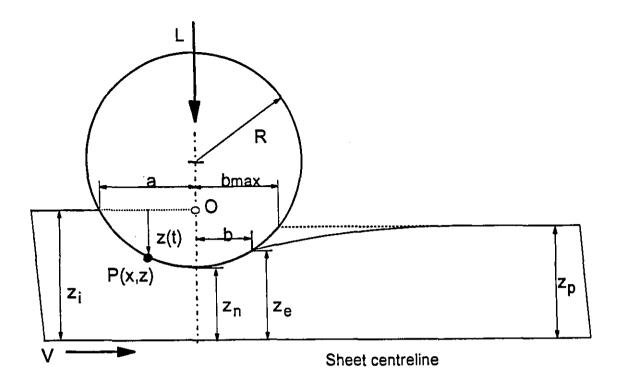


Figure 5.2 Geometry of a thin strip in a rolling nip.

$$a^2 = 2R(z_i - z_n)$$
 [5.2]

The deflection of a point in contact with the roll is therefore parabolic when second-order terms are neglected. By definition, strain is the change in thickness divided by the initial thickness:

$$\varepsilon_1(t) = \frac{Vt}{2Rz_i} (2a - Vt)$$
 [5.3]

In-nip and permanent strains are defined as follows:

$$\varepsilon_{n} = \frac{z_{i} - z_{n}}{z_{i}}$$
 [5.4]

$$\varepsilon_{\rm p} = \frac{z_{\rm i} - z_{\rm p}}{z_{\rm i}}$$
 [5.5]

The ingoing and maximum outgoing nip lengths a and b_{\max} are then given by

$$a^2 = 2Rz_i \varepsilon_n$$
 [5.6]

$$b_{\text{max}}^2 = 2Rz_i (\varepsilon_n - \varepsilon_p)$$
 [5.7]

The strain and strain rate while paper remains in contact with the rolls are

$$\varepsilon_1(t) = \varepsilon_n \left(\frac{Vt}{a}\right) \left(2 - \frac{Vt}{a}\right)$$
 [5.8]

$$\frac{\partial \varepsilon_1(t)}{\partial t} = \frac{2 \varepsilon_n V}{a} \left(1 - \frac{Vt}{a} \right)$$
 [5.9]

The largest strain rate is thus the initial strain rate at the nip entrance, when t=0:

$$\left(\frac{\partial \varepsilon_1(t)}{\partial t}\right)_{\text{max}} = \frac{2\varepsilon_n V}{a}$$
 [5.10]

In the experimental program described in Chapter 4, ingoing nip lengths, as computed using Equation 5.2, ranged from 2 to 5 mm, and so initial strain rates varied from 2000 to 6000 s⁻¹.

5.2.2 Constitutive equations for linear viscoelastic models

Mathematical descriptions of stress and strain for a thin, smooth continuous web in a rolling nip are derived next. The web is modeled using three linear viscoelastic models, each made up of ideal elastic elements with moduli G_i and ideal viscous elements with viscosities η_j . The models are the standard linear solid, the standard linear liquid, and Burger's model. In the derivations, the following assumptions are made.

- 1. The deformation of the rolls under the largest load applied is assumed to be small compared to the resulting deformation of the material in the nip.
- Roughness of the strip is assumed small compared to z_i, and full contact between the sheet and roll is assumed to occur immediately upon entering the nip.
- 3. The material is assumed to be a continuum in the z-direction.
- 4. The material is assumed linear under typical calendering conditions, i.e. the moduli G_i and viscosities η_j are assumed to be material constants unaffected by applied stress, the resulting strain, or stress or strain rates.
- 5. A compressive pulse in the z-direction is assumed to result in negligible deformations in either the machine or cross directions, thus limiting the problem to one dimension. Another way of stating this assumption is to specify that Poisson's ratio is negligibly small. The tensor form of the constitutive equation can therefore be replaced with a scalar form.
- 6. The thickness of the web is assumed small compared to the roll radius.

Assumption #1 is valid when both calender rolls are steel, since the roll elastic modulus is then about 200 GPa compared to effective paper moduli of the order of 40 MPa (Mann et al. [54]). In the case of a soft nip, where the roll cover modulus may be 1 GPa, deformation of the soft roll is significant and the analysis presented here does not apply. Assumptions #2 and #3 do not apply for paper; the rough, structured nature of paper will be discussed in Section 5.3. The validity of assumption #4 has been questioned by Popil [70], and will be discussed in Section 5.3. Assumption #5 was discussed in

Section 5.1, and is consistent with published data (Mann et al. [54]). The validity of assumption #6 is confirmed when typical roll radii (0.1 to 0.5 m) are compared to typical paper thicknesses (80 to 150 μ m).

The Deborah number ξ , the ratio of a typical material response time τ and a time characteristic of a process, occurs frequently in the derivation of pressure profiles caused by a given strain history. In a rolling nip, the Deborah number is the ratio of a time constant τ to the ingoing nip dwell time a/V:

$$\xi = \frac{\tau}{a/V}$$
 [5.11]

Depending on the type of time constant, there are two different interpretations of the Deborah number. First, the time constant may correspond to a stress relaxation time, in which case when τ is large compared to a/V, ξ is much larger than 1 and the material appears elastic since relaxation of the applied stress does not begin to occur in the short time the load is applied. If the stress relaxation time is very short compared to a/V, ξ is much smaller than 1 and the material appears viscous since there is little residual stress in material when the load is removed. Only when the relaxation and process times are similar does viscoelastic behaviour occur.

On the other hand the time constant may be a strain recovery time, as discussed in Section 2.2, in which case the situation is reversed: a long recovery time (corresponding to a large Deborah number) leads to behaviour which appears viscous since no apparent recovery occurs after removal of a load until many processing times have elapsed. Similarly, a short recovery time leads to apparently elastic behaviour since the strain recovery occurs immediately on removal of a load. As pointed out in Section 2.2, material behaviour in a rolling nip will be viscoelastic for a given material time τ , whether stress relaxation or strain recovery, if the sheet speed $V \approx a/\tau$.

A second non-dimensional number is the nip length ratio β , defined by



$$\beta = \frac{a+b}{a}$$
 [5.12]

When a material shows large permanent deformations (ε_p approaching ε_n), or recovers slowly due to a long strain recovery time, the outgoing nip length b approaches 0, and β approaches 1. On the other hand a material which recovers quickly due to a short recovery time will exhibit a longer nip length, with b approaching b_{max} and β approaching β_{max} :

$$\beta_{\max} = 1 + \sqrt{1 - \frac{\varepsilon_p}{\varepsilon_n}}$$
 [5.13]

When a material recovers quickly and shows no permanent deformation, z_p approaches z_i , β_{\max} and β both approach 2, ε_p approaches zero and b approaches a.

In linear viscoelastic behaviour, discussed in Section 2.2, the stress response to an arbitrary strain input is given by the Boltzmann integral:

$$\sigma(t) = -\int_{0}^{t} \psi(t-t') \frac{\partial \varepsilon_{1}(t')}{\partial t'} dt'$$
 [5.14]

For a rolling nip, the strain rate $\partial \varepsilon_1(t)/\partial t$ is given by Equation 5.9. The relaxation function $\psi(t)$ describes the material response to the imposed strain. In the case of a pure elastic material, Equation 5.14 reduces to Hooke's law. Once the stress history is known and can be differentiated for $0 \le t \le t_e$, where t_e is the elapsed time at the nip exit, the strain recovery for $t > t_e$ is given by

$$\varepsilon_2(t) = \int_0^{t_e} \phi(t-t') \frac{\partial \sigma(t')}{\partial t'} dt'$$
 [5.15]

Values of the various moduli and viscosities can be determined if strains or stresses are known at certain critical points. As knowledge of the maximum strain attained in the nip is crucial to solving Equations 5.14 and 5.15, the experimental measurements of Chapter 4 are essential in determining viscoelastic behaviour.

Two linear models, each made up of several springs and dampers and with specific relaxation and retardation functions, will be discussed in detail next. The standard linear solid, described briefly in Section 2.2, does not predict permanent deformation and is therefore a poor approximation for paper in a calender. Constitutive equations for this model in a rolling nip were derived by Kerekes [45], and are not repeated here.

6

The standard linear liquid

The standard linear liquid, illustrated in Figure 5.3, combines a single viscous element in series with a Kelvin model. The viscous element $\eta_{\rm M}$ models stress relaxation under load and permanent deformation once the load is removed, while the Kelvin model $\eta_{\rm K}$, $G_{\rm K}$ reproduces creep under load and time-delayed strain recovery. There is no instantaneous elastic response, either under load or afterward. If the material remains in contact with the roll surface after the nip centerline, it is due to the recovery of the Kelvin element occurring faster than the rate of removal of the load. Application of a compressive stress for a long period results in the predicted strain becoming very large, eventually exceeding 1. As this result implies negative thicknesses, the model is limited to cases where the duration of the load is short compared to material response times. Dwell times in industrial papermachine calenders meet this requirement, as illustrated by Figures 4.22 and 4.23, Section 4.2.1.

The stress relaxation and strain recovery functions describe the model's behaviour:

$$\psi(t) = \eta \left[\frac{\eta_M}{\eta_K} \frac{e^{-t/\tau}}{\tau} + \delta(t) \right]$$

$$\phi(t) = \frac{t}{\eta_M} + \frac{1 - e^{-t/\tau_K}}{G_K}$$
[5.16]

where $\delta(t)$ is the unit impulse function, and the equivalent viscosity η and overall time constant τ are given by

$$\eta = \frac{\eta_M \eta_K}{\eta_M + \eta_K} \qquad \tau = \frac{\eta_M + \eta_K}{G_K} \qquad [5.17]$$

The Kelvin time constant τ_K is equal to the ratio η_K/G_K . The unit impulse function arises because imposition of a step strain on the viscous element η_M requires an infinite stress of infinitely short duration. Substituting Equations 5.9 and 5.16 into the Boltzmannn

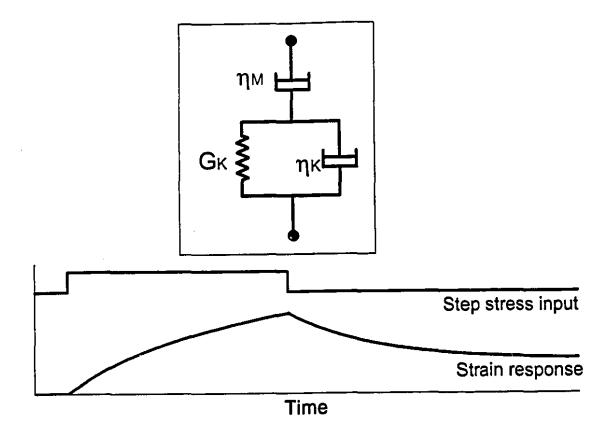


Figure 5.3 A linear liquid showing partial time-delayed recovery.

integral gives the stress profile in a rolling nip:

$$\sigma(t) = \frac{2\varepsilon_{n}V}{a} \frac{\eta_{M}^{2}}{\eta_{M} + \eta_{K}} \left[(1+\xi)(1-e^{-t/\tau}) - \frac{Vt}{a} \right] + \frac{2\varepsilon_{n}V}{a} \left(1 - \frac{Vt}{a}\right) \frac{\eta_{M}\eta_{K}}{\eta_{M} + \eta_{K}}.$$
[5.18]

where ξ is the Devorah number associated with τ :

4

$$\xi = \frac{V \tau}{a}$$
 [5.19]

The last term in Equation 5.18 is the strain rate times the apparent viscosity of the model. Equation 5.18 does not predict a zero stress when $t = 0^+$, as predicted by

Kerekes [45] and measured by Keller [43], but the pure Newtonian stress $\eta \dot{\epsilon}$ which is non-zero since the initial strain rate is non-zero. The stress at the nip exit should also be zero. Equation 5.18 correctly predicts the stress at the nip exit when $t = t_e$ and when the nip length is a:

$$(1+\xi)(1-e^{-1/\xi}) - 1 = 0$$
 [5.20]

When t_e is small compared to τ , as it is in a calender, ξ is large and Equation 5.20 reduces to the first two terms of the series expansion for e^{-x} , $x = 1/\xi$; when ξ is sufficiently large, as it is here, further terms in the expansion can be neglected, and Equation 5.20 reduces to an identity. Thus Equations 5.18 and 5.20 cannot be used to predict model parameters. The three unknowns in Equations 5.18 and 5.20 are the three model parameters implicit in the time constant τ . Three equations are necessary to solve for the unknowns: the first is obtained by integrating the stress over the length of the nip and equating to the line load L:

$$L = \int_{0}^{t_{a}} \sigma(t) V dt$$
 [5.21]

Two more equations can be obtained using the retardation function, the response to a step stress input, which is given by Findley et al. [27]:

$$\phi(t) = \frac{t}{\eta_M} + \frac{1 - e^{-t/\tau_k}}{G_K}$$
 [5.22]

where $\tau_K = \eta_K/G_K$ is the relaxation time for the Kelvin element. Differentiating Equation 5.18 with respect to time and substituting the resulting stress rate and the retardation function $\phi(t)$ into the Boltzmann integral for strain, Equation 5.15, gives an expression for the strain recovery curve after the nip. This expression consists of two terms, one constant and one an exponential decay. Equating the constant term to the permanent deformation gives a second relationship for the three unknowns. The decay term,

controlled by τ_K , gives a third relationship when τ_K is known, as it is here.

Equation 5.21, along with expressions for strain recovery and permanent strain, make up a set of three equations in three unknowns. In the absence of knowledge of the recovery time constant $\tau_{\rm K}$, a fourth equation may be obtained by comparing the strain at the nip exit computed using the strain recovery curve, $\varepsilon_2(t_{\rm e})$ in Equation 5.15, with the exit strain computed from geometry, $\varepsilon_1(t_{\rm e})$ in Equation 5.8. However, as the outgoing nip length b approaches 0, as it does for the standard linear liquid, this comparison yields the identity $\varepsilon_{\rm n} = \varepsilon_{\rm n}$, which provides no additional information.

Finally, strain rates at the nip exit may also be compared:

$$\frac{\partial \varepsilon_1(t)}{\partial t} = \frac{\partial \varepsilon_2(t)}{\partial t}$$
 [5.23]

As b approaches 0, however, Equation 5.23 predicts $\varepsilon_{\rm n}=\varepsilon_{\rm p}$. Thus knowledge of $\tau_{\rm k}$ and $\varepsilon_{\rm n}$ is essential for a solution. An estimate of the time constant is available; numerical solutions will be discussed in Section 5.2.3.

Burger's 4-element model

The Burger's model, shown in Figure 5.4, is composed of a Maxwell model in series with a Kelvin model. The viscous element $\eta_{\rm M}$ models permanent deformation of the material, the elastic element $G_{\rm M}$ models initial instantaneous elastic compression and partial instantaneous elastic recovery once a load is removed, and the Kelvin model exhibits time-delayed thickness recovery controlled by the time constant $\eta_{\rm K}/G_{\rm K}$. This model is therefore the minimum required to describe viscoelastic solids which show instantaneous elasticity, time-dependent partial recovery and creep under load.

The relaxation function $\psi(t)$ is given by Findley et al. [27] for a Burger's model:

$$\psi(t) = K_1 e^{-V\tau_1} - K_2 e^{-V\tau_2}$$
 [5.24]

where K_1 and K_2 , functions of the model parameters G_M , G_K , η_M , and η_K , have units of pressure. Similarly the time constants τ_1 and τ_2 also depend on the model parameters:

$$\begin{aligned} p_1 &= \frac{\eta_M}{G_M} + \frac{\eta_K}{G_K} + \frac{\eta_M}{G_K} \\ p_2 &= \frac{\eta_M \eta_K}{G_M G_K} \\ q_1 &= \eta_M \\ q_2 &= \frac{\eta_M \eta_K}{G_K} \end{aligned} \qquad \begin{aligned} (\tau_1, \tau_2) &= \frac{p_1 \pm A}{2p_2} \\ (\tau_1, \tau_2) &= \left(\frac{1}{r_1}, \frac{1}{r_2}\right) \\ (\xi_1, \xi_2) &= \left(\frac{V \tau_1}{a}, \frac{V \tau_2}{a}\right) \\ K_1 &= \frac{q_1 - q_2 \tau_1}{A} \\ K_2 &= \frac{q_1 - q_2 \tau_2}{A} \end{aligned}$$
 [5.25]

The retardation function $\phi(t)$ is also given by Findley et al. [27]:

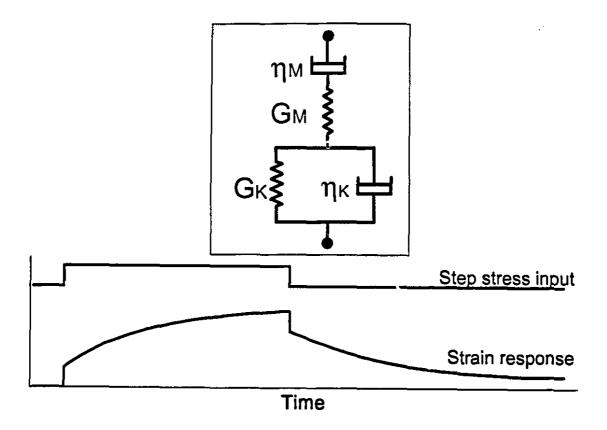


Figure 5.4 Burger's model showing instantaneous elasticity and delayed partial recovery.

$$\phi(t) = \frac{1}{G_{M}} + \frac{t}{\eta_{M}} + \frac{1 - e^{-t/\tau_{K}}}{G_{K}}$$

$$= \frac{1}{G_{M}} \left[1 + \frac{t}{\tau_{M}} + \frac{G_{M}}{G_{K}} (1 - e^{-t/\tau_{K}}) \right]$$
[5.26]

where the Maxwell and Kelvin time constants au_{M} and au_{K} are given by

$$\tau_{\rm M} = \frac{\eta_{\rm M}}{G_{\rm M}} \qquad \qquad \tau_{\rm K} = \frac{\eta_{\rm K}}{G_{\rm K}} \qquad [5.27]$$

 $\phi(t)$ is the sum of the responses of the Maxwell spring G_M , Maxwell damper η_M and the Kelvin element G_K , η_K , and has units of reciprocal pressure. Substituting

Equations 5.9 and 5.24 into Boltzmann's integral, Equation 5.14, gives an explicit expression for stress in the nip:

$$\sigma(t) = -\frac{2\varepsilon_n V}{a} \int_0^t \left(K_1 e^{-(t-t')/\tau_1} - K_2 e^{-(t-t')/\tau_2} \right) \left(1 - \frac{Vt'}{a} \right) dt' \qquad [5.28]$$

$$= -2\varepsilon_{n} \left[K_{1}\xi_{1}(1+\xi_{1})(1-e^{-t/\tau_{1}}) - K_{2}\xi_{2}(1+\xi_{2})(1-e^{-t/\tau_{2}}) - (K_{1}\xi_{1}-K_{2}\xi_{2})\frac{Vt}{a} \right]$$
[5.29]

Unlike the standard linear liquid, $\sigma(0) = 0$. Evaluating $\sigma(t_e)$ and equating to zero gives

$$K_1\xi_1(1+\xi_1)(1-e^{-\beta/\xi_1}) - K_2\xi_2(1-\xi_2)(1-e^{-\beta/\xi_2}) - (K_1\xi_1 - K_2\xi_2) = 0$$
 [5.30]

Equation 5.30 gives one relationship between the five unknowns G_M , η_M , G_K , η_K and b (implicitly in the variables K_1 , K_2 , ξ_1 , ξ_2 and β) in terms of the processing parameters L, V and R and the in-nip strain ε_n .

Integrating the expression for $\sigma(t)$ from t=0 to $t=t_e$ gives a second relationship:

$$L = \int_{0}^{t_{\bullet}} \sigma(t) \ V \ dt \qquad [5.31]$$

-

$$\frac{L}{2\varepsilon_{n}a} = K_{1}\xi_{1}^{2}(1+\xi_{1})(1-e^{-\beta/\xi_{1}}) - K_{2}\xi_{2}^{2}(1+\xi_{2})(1-e^{-\beta/\xi_{2}})
- (K_{1}\xi_{1}(1+\xi_{1}) - K_{2}\xi_{2}(1+\xi_{2}))\beta + (K_{1}\xi_{1}-K_{2}\xi_{2})\frac{\beta^{2}}{2}$$
[5.32]

Strain $\varepsilon_2(t)$ at times $t \ge t_e$ is given by

$$\varepsilon_{2}(t) = \frac{1}{G_{M}} \int_{0}^{t_{A}} \left[1 + \frac{t - t'}{\tau_{M}} + \frac{G_{M}}{G_{K}} \left(1 - e^{-(t - t')/\tau_{K}} \right) \right] \frac{\partial \sigma(t')}{\partial t'} dt' \qquad [5.33]$$

After equating the sum of constant terms to the permanent strain $\varepsilon_{\rm p}$, strain $\varepsilon_{\rm 2}(t)$ becomes

$$\varepsilon_{2}(t) = \varepsilon_{p} - \frac{2\varepsilon_{n}}{G_{K}} \left[\frac{K_{1}\xi_{1}(1+\xi_{1})}{1-\xi_{1}/\xi_{K}} \left(e^{-\frac{\beta}{\xi_{1}}\left(1-\frac{\xi_{1}}{\xi_{k}}\right)} - 1 \right) - \frac{K_{2}\xi_{2}(1+\xi_{2})}{1-\xi_{2}/\xi_{K}} \left(e^{-\frac{\beta}{\xi_{2}}\left(1-\frac{\xi_{2}}{\xi_{k}}\right)} - 1 \right) - \xi_{k} \left(K_{1}\xi_{1} - K_{2}\xi_{2} \right) \left(1 - e^{\beta/\xi_{k}} \right) \right] e^{-t/\tau_{k}}$$
[5.34]

The second term decays exponentially at a rate depending on the time constant τ_K of the Kelvin model.

The third of the five equations defining the Burger's model is obtained by equating ε_p to the constant terms obtained in the derivation of $\varepsilon_2(t)$:

$$\epsilon_{p} = \frac{Lim}{t - \infty} \epsilon_{2}(t)$$

$$= \frac{2\epsilon_{n}a}{V\eta_{M}} \left[K_{2}\xi_{2}(1 + \xi_{2}) (\beta + \xi_{2}) e^{-\beta/\xi_{2}} - K_{1}\xi_{1}(1 + \xi_{1}) (\beta + \xi_{1}) e^{-\beta/\xi_{1}} + \left(K_{1}\xi_{1}^{2}(1 + \xi_{1}) - K_{2}\xi_{2}^{2}(1 + \xi_{2}) \right) - \left(K_{1}\xi_{1} - K_{2}\xi_{2} \right) \frac{\beta^{2}}{2} \right]$$
(5.35)

As for the standard linear liquid, two more equations for the five unknowns can be obtained by evaluating the two expressions for strain and their derivatives at $t=t_e$. These equations become progressively less useful as β approaches 1. Equating $\varepsilon_1(t_e)$ to $\varepsilon_2(t_e)$ gives

$$\beta(2 - \beta) - \frac{\varepsilon_{p}}{\varepsilon_{n}} = \frac{2}{G_{K}} \left[\frac{K_{1}\xi_{1}(1+\xi_{1})}{1-\xi_{1}/\xi_{K}} (e^{-\beta/\xi_{1}} - e^{-\beta/\xi_{K}}) - \frac{K_{2}\xi_{2}(1+\xi_{2})}{1-\xi_{2}/\xi_{K}} (e^{-\beta/\xi_{2}} - e^{-\beta/\xi_{K}}) \right]$$

$$- \xi_{K}(K_{1}\xi_{1}-K_{2}\xi_{2})(1-e^{-\beta/\xi_{K}})$$
[5.36]

Equating the strain rates when $t=t_e$ gives a fifth equation:

$$\frac{2\varepsilon_{n}V(1-\beta)}{a} = \frac{\varepsilon_{p} - \varepsilon_{n}\beta(2-\beta)}{\tau_{k}}$$
 [5.37]

Equations 5.30, 5.32, 5.35, 5.36 and 5.37, along with the definitions of the various intermediate variables, make up a set of five equations in the five unknowns G_M , G_K , η_M , η_K and b. The required calendering parameters are the line load L, machine speed V, equivalent roll radius R, initial sheet half-thickness z_i , in-nip half-thickness z_n and fully-recovered half-thickness z_p . The number of unknowns and equations is reduced when estimates of τ_K and β are available; however two of these, Equations 5.36 and 5.37, are only useful when β is not too close to 1.

Estimates of the values of the parameters are made in the next section, using methods that do not require solving sets of non-linear equations.

5.2.3 Initial estimates of parameters

As shown in the previous section, there are sufficient equations to solve for the unknown model parameters and the outgoing nip length b when modeling either a standard linear liquid or a Burger's model. Since the sets of equations are highly non-linear for both models, numerical methods are necessary; the literature on iterative solutions for sets of non-linear equations is large [1, 28, 59, 60, 71]. However, a full numerical solution may not be possible since there are not enough equations when β approaches 1, which appears to be the case for paper. As well, the equations are extremely nonlinear, requiring a set of initial estimates for the parameters which is extremely close to the solution set. On the other hand, model parameters for the standard linear liquid and Burger's model can be estimated from the data presented in Chapter 4 without resorting to iterative methods.

As described in Section 4.2.1, the partially recovered strain ε_r can be used to estimate the strain recovery time τ_K , the Kelvin time constant. This value appears to be a fairly strong function of sheet speed, implying that the assumption of linearity (Assumption #4, Section 5.2.2) is false. However, as much of the present data was obtained at speeds which are low compared to industrial practice, useful information may still be obtained when data at low speeds is ignored. Restricting the analysis in this fashion, Figure 4.21a gives $\tau_K \approx 15$ ms when $V \approx 17$ m/s (S = 1000 m/min). From Figure 4.21b, a typical value of the dwell time a/V at these speeds is 0.2 ms. The Deborah number associated with the Kelvin element is therefore $\xi_K \approx 75$. Of the two components of the Kelvin element, G_K and η_K , only one need now be determined, the other being obtained from the definition of τ_K .

Both models predict that the permanent deformation is due to a single viscous element $\eta_{\rm M}$. The value of this parameter can be obtained by integrating Equation 5.38, relating viscous stress to permanent strain in the viscous element:

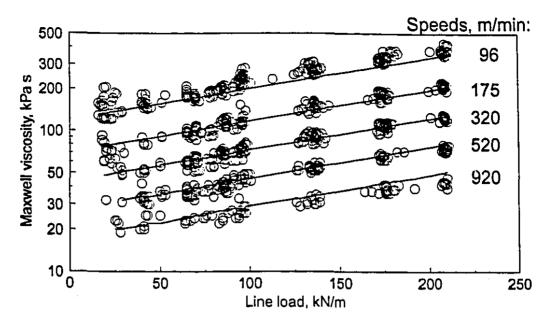


Figure 5.5 Maxwell viscosity η_M , for the standard linear liquid or the Burger's model.

$$\sigma_{\rm V}(t) = \eta_{\rm M} \frac{\partial \varepsilon_{\rm p}}{\partial t}$$
 [5.38]

$$\int_{0}^{t_{k}} \sigma_{V}(t) dt = \eta_{M} \epsilon_{p}$$
 [5.39]

By definition, the left hand side of Equation 5.39 is equal to the impulse L/V, as seen in Equation 5.31. Then Equation 5.39 gives

$$\eta_{\rm M} \approx \frac{L}{{\rm V}\,\varepsilon_{\rm p}} \tag{5.40}$$

Equation 5.40 (with ε_p computed using the calendering equation) is plotted in Figure 5.5, where it can be seen that η_M is a function of both load and speed. Nonlinear curve fitting methods were used to quantify this function; good agreement was obtained with Equation 5.41:

$$\eta_{M} = k_{1} S^{k_{2}} e^{k_{3} L}$$
 [5.41]

Values for the constants (with S in m/min and L in kN/m) are given in Table 5.1, and the curves are plotted with the data in Figure 5.5. Rearranging equation 5.40 gives

$$\frac{L}{a} = \frac{\varepsilon_p V}{a} \eta_M$$
 [5.42]

The ratio a/V is the time during which the average pressure L/a is applied, causing the deformation ε_p . The material property relating this average strain rate $\varepsilon_p V/a$ to the average pressure L/a is the viscosity η_M . This viscosity is not a constant, as seen in Figure 5.5, since the permanent strain ε_p is not proportional to the ratio L/V.

A similar analysis can be performed for the Kelvin element, where the strain imposed is all recovered:

$$\sigma_{K}(t) = \eta_{K} \frac{\partial \varepsilon_{K}}{\partial t} + G_{K} \varepsilon_{K}$$
 [5.43]

Substituting $G_K \tau_k$ for η_K , an expression for G_K can be obtained:

$$G_{K} = \frac{L}{V(\tau_{K} + t_{e})\varepsilon_{K}}$$
 [5.44]

TABLE 5.1: Coefficients for Equation 5.41. Sheet speed in m/min, load in kN/m.

	Parameter	S.E.
k ₁	5741	1.065
k ₂	-0.852	0.035
k ₃	0.00512	0.00057

Two terms must be defined: the total nip dwell time t_e , and the strain in the Kelvin unit ε_K . The nip dwell time depends on b, which is not known:

$$\frac{a}{V} \le \left[t_e = \frac{a+b}{V} \right] \le \frac{a+b_{\text{max}}}{V}$$
 [5.45]

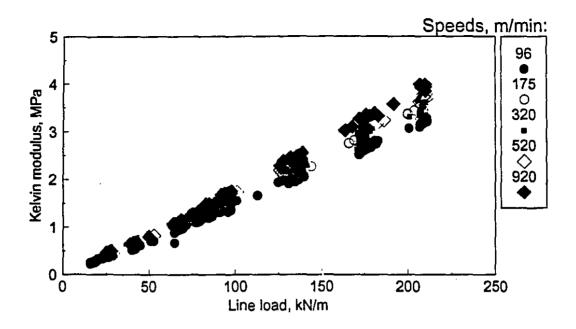
A non-zero value of b occurs when there is contact after the nip centreline, which in turn occurs either when there is instantaneous elastic recovery, or when the load is removed slowly enough that the recovery of the Kelvin element can keep up with the rate of load removal. The time constant governing the recovery, estimated in Section 4.2.1 at 10 to 30 ms, is far too high compared to typical nip dwell times of 100 to 300 μ s for the second case to apply in either model. Only the Burger's model can exhibit any instantaneous elastic recovery; for the standard linear liquid, therefore, the best estimate is b = 0, from which $t_e = a/V$.

For Burger's model, the length b depends on the unknown amount of instantaneous elastic recovery. As reported in Section 4.2, there was complete recovery of innip strain when $\varepsilon_{\rm n} < 0.3$; most of this recovery was time delayed but a small portion may be elastic. In Figure 4.29 the elastic portion of the curve is very short, and an estimate for the instantaneous elastic recovery $\varepsilon_{\rm el}$ of 0.01 to 0.03 seems reasonable; Rodal [73] has proposed $\varepsilon_{\rm el} < 0.05$. Using the estimated elastic modulus $G_{\rm M} = 22$ MPa and an ingoing nip length of 3 mm, the elastic limit for $\varepsilon_{\rm el}$ of 0.01 to 0.03 occurs at nip loads of 0.67 to 2.0 kN/m, and the magnitude of b can be estimated:

$$b = \sqrt{2Rz_i \varepsilon_{el}}$$
 [5.46]

Using $z_i = 60 \ \mu \text{m}$ and R of 202 to 355 mm, Equation 5.46 predicts b in the range 0.5 to 1.1 mm, which gives β in the range 1.17 to 1.37.

The maximum strain in the Kelvin element, ε_K , also needs to be defined. In the standard linear liquid, this is the total recoverable strain $\varepsilon_R = \varepsilon_n - \varepsilon_p$, since the Kelvin element is the only element predicting recovery after the nip. In Burger's model, the strain ε_K is the total recoverable strain ε_R less the instantaneous elastic recovery of the



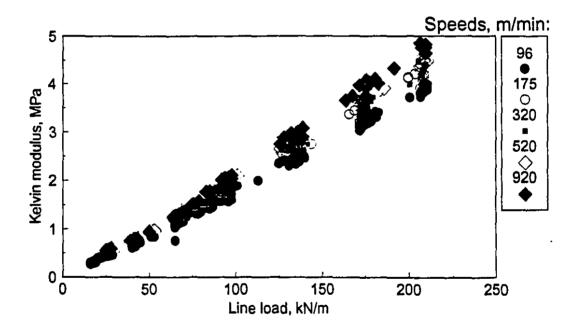


Figure 5.6 Kelvin modulus G_K:

- a) for the standard linear liquid;b) for Burger's model.

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Maxwell spring G_M , estimated above at $0.01 < \varepsilon_{el} < 0.03$. Results for G_K for both models are plotted in Figure 5.6; the values of in-nip and permanent strains were calculated using the calendering equations to avoid excessive scatter. The estimated elastic strain used to calculate data for the Burger's model in Figure 5.6b was $\varepsilon_{el} = 0.017$. For both models the modulus is a strong function of load, and a weak function of speed.

Finally, the viscous component of the Kelvin element is the product of G_K and τ_k ; since τ_k is approximately proportional to 1/V and G_K is proportional to L, η_K , like η_M , is approximately proportional to L/V.

Thus the viscous and time-dependent components in both models are strong functions of either load alone, or of both load and velocity. Table 5.2 summarizes the results for industrially relevant conditions, i.e. a speed of 920 m/min and loads of 10 to 50 kN/m. The Deborah number describing stress relaxation in the nip, $\xi_{\rm M}$, is in the range 2 to 10, while $\xi_{\rm K}$, which describes strain recovery after the nip, is an order of magnitude larger at 50 to 100. The behaviour under load thus appears viscoelastic or mildly elastic, while the recovery after the nip appears strongly viscous.

Figure 5.7 gives the predicted pressure profiles for the two models at a load of 25 kN/m. The pressure profile for the Burger's model shows the expected parabolic shape with skew to the ingoing side of the nip. Pressure is zero at the nip entrance (VUa = 0) and again at a distance b after the nip centerline, b < a. For the standard linear liquid, however, the pressure profile starts off high due to the high strain rate at $t = 0^+$, and decreases essentially linearly until it reaches zero at the nip centreline, where the strain rate is also zero. The exponential component in Equation 5.18 is small when τ is much larger than the dwell time t_e , as it is here; when $(1-e^{-t/\tau})$ is exactly zero, the stress predicted by Equation 5.18 is exactly linear, decreasing to zero at time $t_e = a/V$. An estimate of the validity of this model can be obtained by considering the torque Γ required to pull the sheet through the nip against this pressure profile. Integrating the stress field over the nip length gives the drive torque required to compress the sheet in the nip:

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TABLE 5.2: Summary of modeling results for L = 10 to 50 kN/m, S = 920 m/min.

	Standard linear liquid	Burger's model
$ au_{ m K}$, ms	15	15
ξ _K	75	75
G _K , kPa	19.3 kPa per kN/m	23.5 kPa per kN/m
η _K , Pa s	230 Pa s per kN/m	280 Pa s per kN/m
τ_{M} , ms		0.9
ξ _M		4.5
G _M , MPa		22
η _M , kPa s	25	25

$$\Gamma = \int_{0}^{t_a} W \sigma(t) (a - Vt) V dt$$
 [5.47]

where the product of machine width W and the pressure at a point in the nip $\sigma(t)$ is the portion of the total line load applied at the MD position defined by the elapsed time t; multiplied by an element dx = V dt, this becomes the force applied at that MD position. The distance (a - Vt) is the moment of that force about the roll axis. Summing the force times the moment arm gives the torque Γ required to impose a stress field on the ingoing nip while maintaining a steady speed. Using Equation 5.47 with typical values of S = 920 m/min, L = 20 kN/m, R = 202 mm, with the model parameters estimated as described above, and estimating ε_n using the calendering equation, the torque to pull a linear liquid through a single 70 mm wide calender nip at steady state is 4.8 Nm

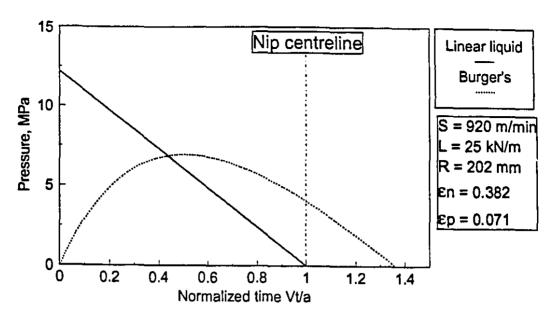


Figure 5.7 Predicted pressure profiles in a calender nip for two linear models. Conditions: S = 920 m/min, L = 25 kN/m, R = 202 mm.

(exclusive of bearing friction, web tension differential, etc.).

An empirical method for estimating the power requirements for a calender nip, Bentley and Derrick [6], predicts power requirements of 0.0009 HP per inch of machine width and per 100 fpm of machine speed to overcome bearing friction. Converting horsepower to torque using the angular velocity of the roll, this method produces a torque estimate of 0.75 Nm per bearing, or 3.0 Nm for a two roll stack, for the above machine conditions. They state that the nip work load due to compression of the sheet is "so small in magnitude in comparison to the major factors [governing drive requirements] that it may be neglected when selecting drive ratings". The authors caution that this assumption needs to be verified, especially for heavier grades. While it may therefore be accepted that the torque requirement to compress the light grade of newsprint tested here is much less than the torque to drive the bearings, Equation 5.47 predicts a substantially higher torque (4.8 Nm) for paper compression than to drive the bearings (3.0 Nm). Thus the linear liquid, while correctly predicting time-dependent

partial recovery and permanent deformation, does not adequately account for other aspects of nip mechanics.

The standard linear solid described by Kerekes [45] can be compared with the standard linear liquid and Burger's model. As shown in Figures 2.2 and 2.5, it predicts a skewed parabolic stress profile similar to the prediction of Burger's model. This is consistent with industrially realistic drive torques. Unlike Burger's model, however, the standard linear solid is unrealistic in predicting full recovery of initial thickness at a long enough time after removal of the load.

Of the three models, the Burger's model provides the best fit with observed behaviour. The standard linear solid is a poorer approximation, but is still superior to the standard linear liquid. Each standard model predicts just one aspect of paper response to a calender pulse, i.e. creep under load or permanent deformation, while Burger's model is the simplest linear model describing both. For all models, however, the parameters are not material properties but are strong functions of the processing parameters. The assumption of linearity does therefore not apply when considering paper in a calender nip. In light of the nonlinear results, the assumptions made at the beginning of this Section will be re-examined.

5.3 Qualitative results: paper as a rough, structured material

5.3.1 Introduction

Linear viscoelastic models were not successful in describing the strongly viscoelastic behaviour of paper in a calender nip since the modeling parameters turn out to be strong functions of processing parameters. Several reasons for the modeling difficulties can be proposed. Paper is a rough, heterogenous, structured material, contradicting Assumptions 2 and 3, Section 5.2.2. Furthermore, the linear modeling is based on the implicit assumption that there are no irreversible changes in paper or fibre response to a stress or strain input. As there is evidence of fibre fracture at high loads, this assumption may also be violated in calendering. The validity of these assumptions is now discussed qualitatively.

5.3.2 Surface roughness

The assumption of surface smoothness is examined first, with the structured nature of a sheet of paper considered subsequently.

A full description of the surface roughness of an anisotropic material such as paper is extremely complex [10, 30]. For an isotropic material, the probability distributions of three variables along a particular direction x_1 in the plane are necessary. These variables are the height z_r of the surface above a reference plane, the average slope of the surface dz_r/dx_1 , and the average surface curvature d^2z_r/dx_1^2 . For an anisotropic material, the variation of these distributions with the angle between the direction x_1 and some reference direction, such as the MD, is also necessary. As well, the problem of contact between smooth and rough surfaces is generally extremely difficult to solve analytically unless the distributions are assumed exponential, which is not necessarily the case with paper. Finally, there is an additional complication in the case of calendering, which is the parabolic nature of the imposed deformation. Thus no attempt was made to evaluate roll-to-paper contact on the scale of the roughness using formal mathematical methods. Rather, the contact area in the nip as described by the ingoing nip length a will be modified for a rough surface by using a common method of estimating roughness height for paper.

The roughness height of paper can be measured using an air-leak test instrument such as the one described by Parker [65]. The device measures the rate of air leakage at a constant supply pressure past a ring pressed into the sheet with a specified force. The average roughness height is then defined as the gap between two smooth plates giving the same volumetric flow rate for flow analogous to Hagen-Poiseuille flow in a cylinder; this gap is proportional to the cube root of the flow rate. The effect of in-plane permeability of the sheet, allowing additional leakage which is not due to roughness, is assumed small. This approximation is reasonable except in the extreme cases of a very smooth or very porous sheet. If the paper surface is modeled as a sinusoidal wave, the roughness obtained this way corresponds to the average amplitude.

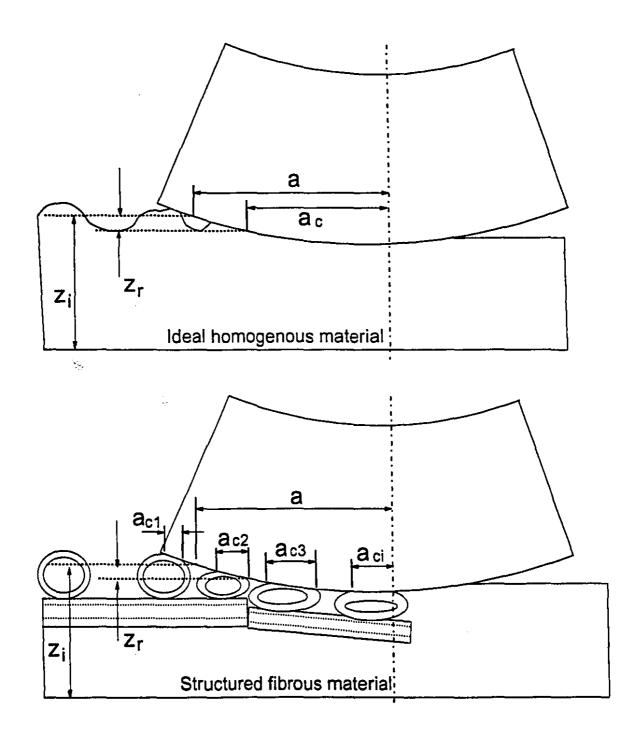


Figure 5.8 Geometry of a rough thin strip in a rolling nip:

a) for an ideal, homogenous material;
b) for paper.

Typical roughness values for the paper used in this study were given in Section 4.1, Table 4.2, where it was seen that the roughness height ranged from 5 to 6.5 μ m. These values are about 10% of the typical ingoing half-thickness z_i , which ranged from 55 to 65 μ m. From the point of view of roughness alone, therefore, full contact between a calender roll and this particular type of paper cannot be expected until strain in the nip exceeds 0.1.

It is useful to consider what happens as a peak is pushed into the body of the sheet. In Figure 5.8a, peaks are shown being compressed by the calender roll while low spots remain unaffected until the depth of penetration of the roll exceeds the typical roughness height. In fact, when the structured nature of paper is considered in the next section it will be seen that the low spots are likely to be deflected early in the nip due to the fibre network spreading the load, as seen in Figure 5.8b. Some low spots may never contact the roll if there are sufficiently dense areas immediately adjacent. This has been investigated by Chapman [14], who pressed sheets against a transparent prism. Photographic methods were then used to estimate the percentage of the sheet contacting the prism at various compression pressures. At the largest average load of 5.5 MPa, the fraction of the total prism area contacting a newsprint sheet was only 18%. This load compares with typical values of the average nip pressure L/a, which ranged from 7.5 to 40 MPa in the present study. Fractional contact area in a calender is thus more than 18%, but certainly considerably less than 100%.

The "true" contact area is thus smaller than the contact area for a smooth material as defined by the ingoing nip length a. Leaving the effect of the structured nature of paper for consideration in the next section, the effect of roughness on contact area can be described quantitatively. The roughness-compensated nip length, a_c , is illustrated in Figure 5.8a, and can be determined for a given roughness height z_r :

$$a_c^2 = 2R \varepsilon_n (z_c - z_n)$$
 [5.48]

where $z_c = z_i - z_r$. Defining ε_c , the strain correponding to a contact length a_c , as

(:

$$\varepsilon_{c} = \frac{z_{i}}{z_{i}}$$
 [5.49]

then the fractional contact area a_c/a comparing the estimates for rough and smooth ingoing nip lengths is

$$\frac{a_{\rm c}}{a} = \sqrt{1 - \frac{\varepsilon_{\rm c}}{\varepsilon_{\rm n}}}$$
 [5.50]

Equation 5.50 is illustrated in Figure 5.9 for various values of ε_c ranging from 0.01 to 0.30. The length a_c defined by Equation 5.48 remains an approximation to the "true" contact length for two reasons. First, Equation 5.48 ignores contact of peaks at the nip entrance, as shown in Figure 5.8, thus underestimating a_c . On the other hand, in a structured material such as paper, as a peak is deflected downwards low spots will be deflected as well by the surrounding structure, causing Equation 5.48 to overestimate a_c . The "true" contact length is in fact the sum of the individual contact lengths Σa_{ci} , which is less than a. Overall, a_c thus remains a better estimate of the contact length than the smooth estimate a, while still overestimating the true contact length.

Figure 4.4 showed that permanent strains are small unless in-nip strains exceed 0.25, while permanent strains of 0.15 are not possible without imposing in-nip strains exceeding 0.50. At $\varepsilon_n = 0.25$, the roughness-compensated area a_c for a typical roughness value equal to 10% of z_i is about 75% of the theoretical contact area a_c exceeding 90% only when ε_n exceeds 0.50. A similar argument applies to the outgoing area, with the smaller roughness after the nip reducing the difference between the smooth estimate b and the roughness compensated estimate b_c .

The assumption of full roll-to-paper contact immediately on entering the nip is thus not reasonable unless either the load exceeds industrially relevant values, or the sheet is already extremely smooth. The implication is that typical pressures encountered locally in the sheet must be higher than previously thought, since the line load is spread

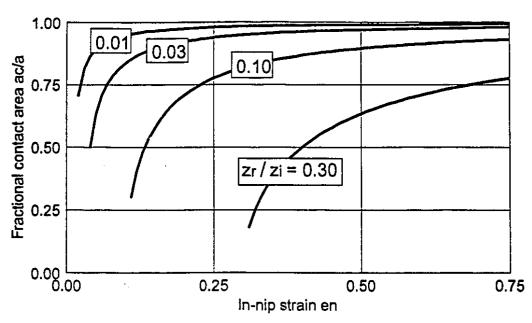


Figure 5.9 Fractional contact area in a calender nip for $\varepsilon_c = 0.01, 0.03, 0.10, 0.30$.

over a significantly smaller area. As well, as the strain is increased the contact area also increases. Since the contact area increases more quickly than the greater depth of indentation would imply, the pressure profile increases more slowly than predicted by the assumption of smoothness. The effect is particularly evident with rough sheets at low loads, where the actual maximum pressure encountered in the nip may be double the value predicted using the ratio L/a since the "true" value of the ingoing nip length a may be as little as half the value predicted by the assumption of minimal roughness. In Section 4.3.1 it was suggested that the outgoing nip length may be significantly shorter than the ingoing length; when these two effects are combined, peak pressures may easily reach values locally up to four times the estimates provided using Hertzian or other methods.

The effect of roughness is thus to distort the pressure profile predicted from the smooth model, which underestimates the maximum pressure by a significant amount; for a given ratio z_r/z_i the error is larger at lower in-nip strains. Any model for paper

compression in a calender nip must thus use an estimate of the ingoing nip length better than a. Since there exists no simple method for determining the true contact length Σa_{ci} at industrially relevant pressures, the roughness compensated length a_c is an obvious alternative.

Equation 5.42 defining $\eta_{\rm M}$ can be reconsidered in the light of the preceding discussion. The ratio a/V, the time from first contact to the nip centreline, remains unchanged even when the contact area is discontinuous. The average pressure L/a, on the other hand, is a poor estimate of the average nip pressure and should be replaced by $L/a_{\rm c}$. A better estimate for $\eta_{\rm M}$ is thus the roughness compensated viscosity $\eta_{\rm Mc}$:

$$\eta_{Mc} \approx \frac{L}{a_c} \frac{a/V}{\varepsilon_p}$$

$$= \frac{a}{a_c} \eta_M$$
[5.51]

where the ratio a/a_c is given by Equation 5.50.

From Table 3.1, the average roughness for the TMP newsprint is 5.74 μ m; ε_c thus ranges from 0.09 to 0.10. At low loads in-nip strain is low, and $a/a_c \approx 1.2$, while at high loads $a/a_c \approx 1.1$. When the estimate of η_{Mc} obtained from Equation 5.51 is fitted to Equation 5.41, the coefficients in Table 5.3 are obtained. Comparing these

TABLE 5.3: Coefficients for Equation 5.41 after roughness compensation. Sheet speed in m/min, load in kN/m.

	Parameter	S.E.
k ₁	6462	1.070
k ₂	-0.853	0.038
k ₃	0.00492	0.00056

results with those in Table 5.1, it can be seen that k_2 and k_3 are statistically unchanged at the 95% confidence level. Similar results are obtained when the estimates for η_K and G_K are revised using the same method; the correction does not account for the non-linearities but shifts the entire family of curves in Figure 5.5 upwards by about 12%.

An improved estimate for the true contact length Σa_{ci} might have a greater effect on the load dependency. The expression for strain rate could be modified using a statistical description of surface roughness. Paper could then be described using a large number of identical linear models in parallel, each with the same rheological behaviour but with different initial heights; as the sheet is compressed more of these elements would be activated and begin to contribute to carrying the applied load. Then the Burger's model could be modified to provide a stress-strain relationship consisting of a sum over all active elements. However, the pressure profile is also dependent on local density distributions in the plane of the sheet, as will be seen in the next section.

5.3.3 Paper and fibre structure

Paper is not a continuum, nor can it be described by a continuous density distribution function since it consists of two distinct phases, air and fibre; in effect paper could be considered fibre-reinforced air. Fibre and sheet properties are related by

$$\frac{\rho_{p}}{\rho_{f}} = \frac{v_{f}}{v_{f} + v_{a}}$$
 [5.52]

where ρ and v are density and volume, and the subscripts a,f,p indicate air, fibre and paper respectively. A typical initial bulk for uncalendered sheets in the present study was 2.70 cm³/g, corresponding to an apparent sheet density ρ_p of 0.370 g/cm³. The density of cellulose in wood fibres is $\rho_f = 1.5$ g/cm³; the uncalendered sheet is thus at most 25% fibre by volume, increasing to 35% fibre by volume when $\varepsilon_p = 0.30$, and reaching a maximum of over 80% fibre in the nip when $\varepsilon_p = 0.70$.

As strain is increased, void fraction decreases. Illustrated in Figure 5.10, thinwalled, flexible fibres deform first; the apparent elastic modulus is defined by the resistance of a cellulose sheet to bending. As strain is increased beyond 0.50, flexible fibres collapse completely, thus becoming more difficult to deform. Stiffer fibres then begin to deform, requiring a higher load for more deformation; the apparent elastic modulus increases. As the strain increases towards 0.70, the void fraction approaches 0.15, and the material becomes increasingly difficult to compress. At a sheet average strain of 0.70, the highest measured in this study, the sheet average void fraction has fallen below 0.15. Under these conditions, areas of higher than average local thickness experience local strains higher than 0.70, and locally the void fraction of the sheet approaches zero. Similarly in areas of high local density the local void fraction is also lower than in surrounding areas. An area which is thicker or denser may thus be compressed to a void fraction approaching zero even though the sheet average void fraction is significant. In these thick or dense spots, the apparent elastic modulus at high average strains is defined by the compressive properties of solid cellulose rather than by

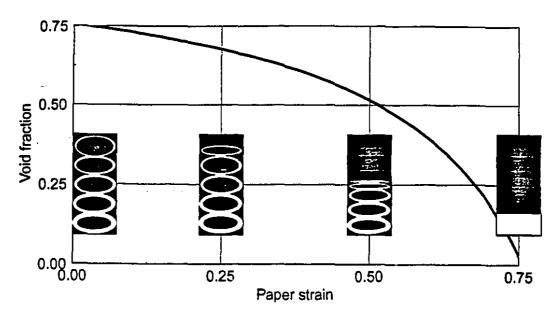


Figure 5.10 Void fraction as a function of average paper strain.

the open structure as before. The heterogenous, structured nature of paper thus results in large changes in the response to an applied stress for different levels of previously applied strain; the material behaviour is non-linear since the response function, whether stress relaxation or strain recovery, depends at the very least on the previous strain history. Furthermore, the non-linearities depend not only on the structure of paper, as defined by the initial apparent density and the spatial distribution of fibres of different flexibility, but on the structure of the fibre as it affects fibre compressibility.

The non-continuous nature of paper will now be examined using photo-micrographs of sheet cross-sections. The range of characteristic dimensions describing paper and machine geometry is large: roll radius may be 100 to 350 mm while nip lengths are about 2 to 5 mm; flocs are of the order of 1 cm in diameter while fibre dimensions are about 1 mm long and 0.01 mm in diameter. A 200 mm radius roll is shown reduced in Figure 5.11A, where the line width representing the paper web in the nip is not to scale. With the nip magnified by a factor of 50 in Figure 5.11B, roll curvature appears infinite and paper thickness is now apparent, but no details of paper

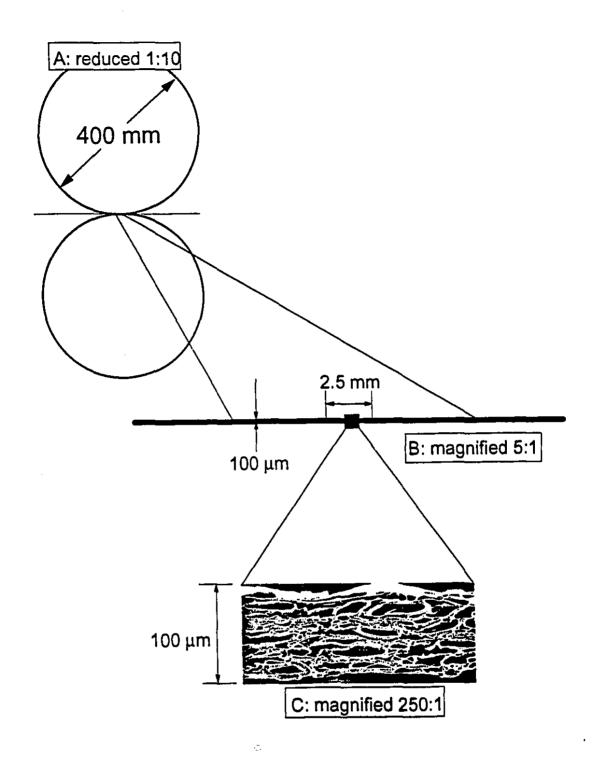


Figure 5.11 Relative dimensions of calender roll, nip, and paper. A: calender stack; B: nip length of 2.5 mm; C: sheet section 250 μ m long.

structure are visible. With an increase in magnification by a further factor of 50, Figure 5.11C, details of paper structure can be identified; there are large thickness and density variations on a scale of one-tenth the total nip length. From the point of view of a fibre, the radius of the roll is essentially infinite, and the contact length is similar to typical fibre lengths.

Cross-sections of TMP newsprint were prepared for microscopy by scanning electron microscope using methods developed by Williams et al. [81]. The section plane in the selected micrographs presented in Figures 5.12 to 5.20 is aligned with the cross-machine direction. As all sections were taken near the centre of the 70 mm wide web, they all come from the same CD position on the paper machine. All came from the same reel and were thus made within the space of a few minutes. An uncalendered sample was taken from the beginning of the reel; the remainder of the reel was calendered under a variety of conditions and samples were removed corresponding to two conditions. Each calendering condition is illustrated with a group of six micrographs taken from a series of ten spanning a continuous 2.8 mm wide CD section, a small distance compared to the machine CD width of 70 mm. After allowing for overlap between adjacent photographs, each covers a CD distance of 0.33 mm.

The micrographs illustrate several aspects of paper nonuniformities as they affect compression in a calender nip. On a large scale, of the order of a millimetre, significant variations are apparent from place to place in the sheet in both thickness and density, leading to non-uniformities in the CD and MD pressure profiles and thus in the local strain of the sheet. On the scale of the sheet thickness, about $100 \mu m$, there are variations in typical fibre dimensions. Earlywood fibres, which come from the spring and summer growth of the tree, have large diameters and thin walls, while latewood fibres are smaller and coarser with thicker walls. Every position in the sheet is made up of a stack of fibres with different distribution of fibre coarseness, leading to local variations in compressibility with applied strain. Finally, on the scale of the fibre diameter, about $10 \mu m$, the fibre wall is itself made up of fibrils wound in a spiral about the fibre axis; the fibril angle varies from the outside of the fibre to the inside, thus varying the strength properties as

the fibre is compressed.

Figures 5.12 to 5.17 show sections of uncalendered paper. The position of each micrograph in the continuous set of ten is shown schematically. The average thickness for these samples is 122 μ m when measured using a standard caliper gauge. It may be recalled that the platen on this gauge has a diameter of 16 mm, almost 50 times larger than the area shown in the micrographs. Basis weight of the samples is 45 g/m², and the average fibre content by volume in the uncalendered sheet is about 25%.

There is a large variety of fibre coarsenesses visible in the uncalendered samples, ranging from large diameter, thick-walled latewood (5.12, bottom centre) to small diameter fibres which are either hardwood fibres or ray cells (5.12, center left) to large diameter thin-walled earlywood (5.13, top right). At the upper left in 5.12 there is a fibre which has been split open and which then curled back on itself.

Density variations in Figures 5.12 and 5.13 appear small, but they are much larger in Figures 5.14 and 5.15. The lower edge of 5.14 shows a shive made up of earlywood fibres and ray cells, while to the lower left in 5.15, there is another large shive. In both cases these consist of fibres which were not separated during pulping. Since the shives are mainly thin-wall earlywood, the density in this area, about 1 mm from Figure 5.12, is extremely low. In the calender nip, low density areas such as this will not be deformed appreciably since adjacent high density areas can be expected to carry a larger portion of the load through the nip. Finally, there are still a variety of fibre types, including an extremely coarse fibre to the right of 5.14. This area shows how density can vary on a millimetre scale while caliper remains essentially constant.

If Figures 5.13 and 5.14 showed regions of variable density and constant caliper, Figures 5.16 and 5.17 show an area immediately adjacent of constant density and variable caliper which is composed of smaller and coarser fibres. The density is higher than in Figure 5.13, while the thickness is lower and much more variable over the total CD distance of 660 μ m illustrated. As well, there is another broken fibre to the upper left of 5.17.

These micrographs of uncalendered paper illustrate several aspects of the structure

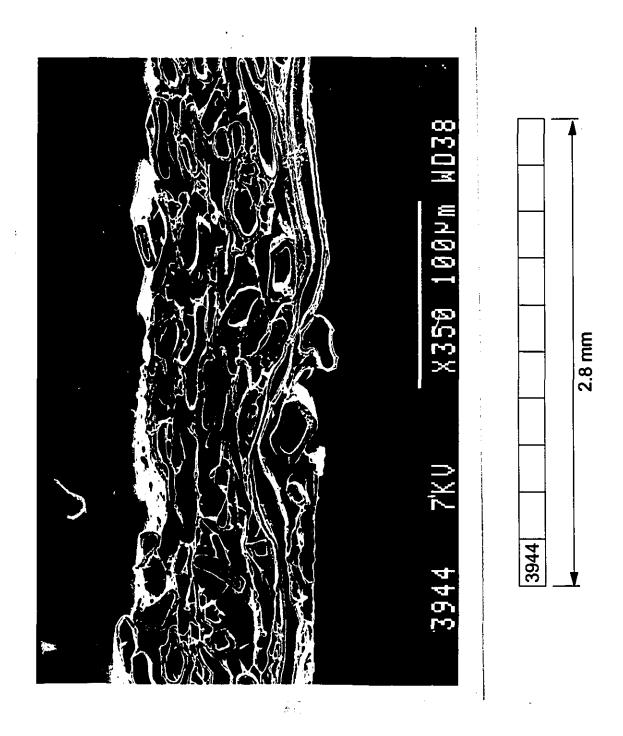


Figure 5.12 Uncalendered sheet. Section plane in CD; magnification 500X.

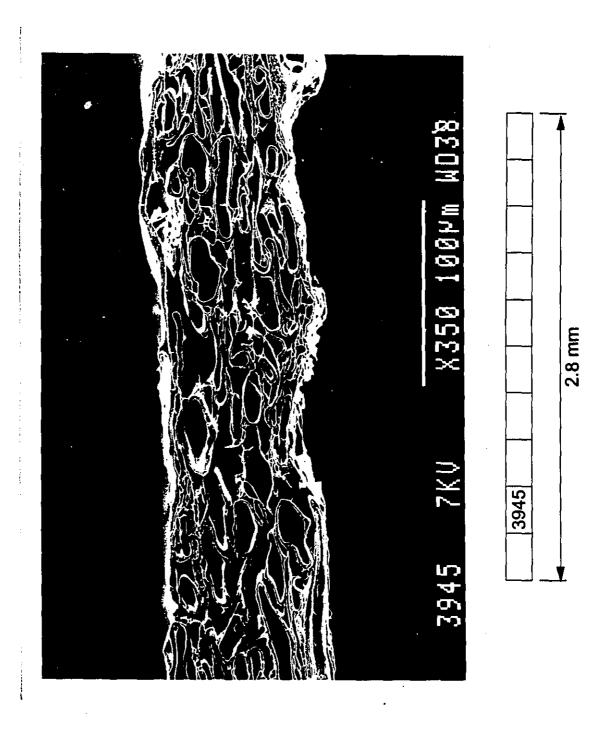


Figure 5.13 Uncalendered sheet. Section plane in CD; magnification 500X.

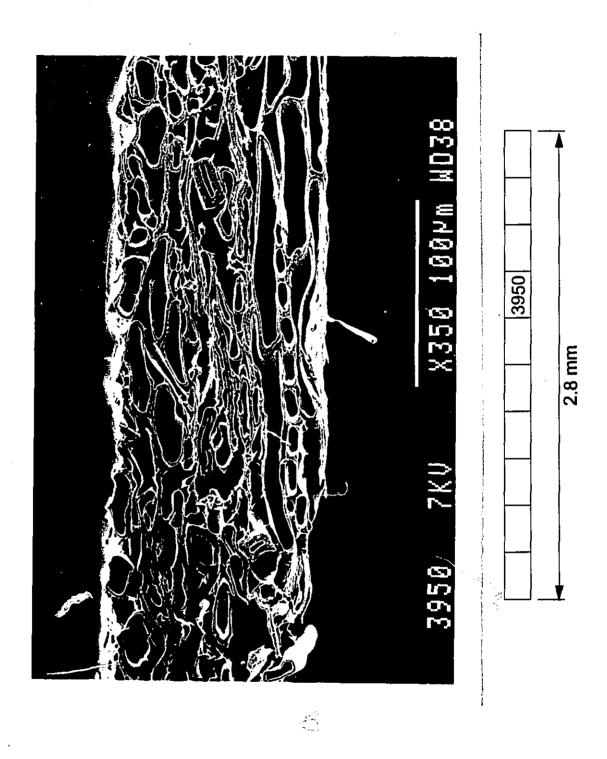


Figure 5.14 Uncalendered sheet. Section plane in CD; magnification 500X.

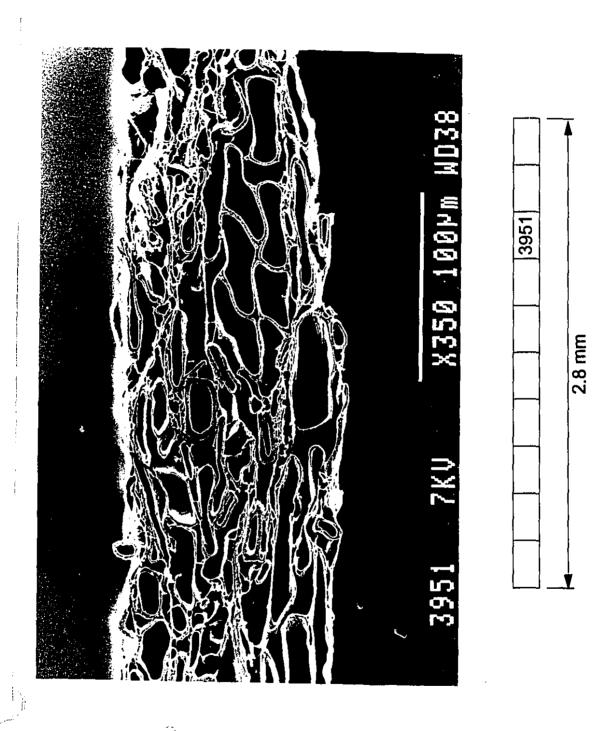


Figure 5.15 Uncalendered sheet. Section plane in CD; magnification 500X.

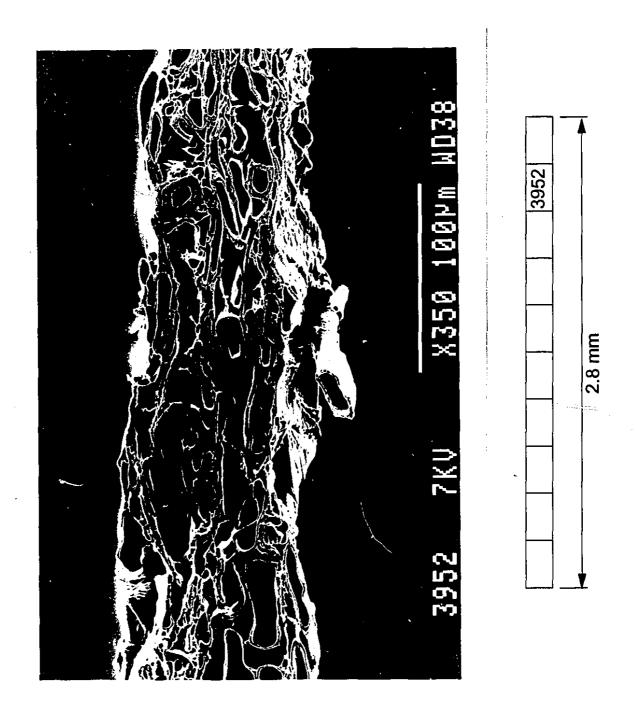


Figure 5.16 Uncalendered sheet. Section plane in CD; magnification 500X.

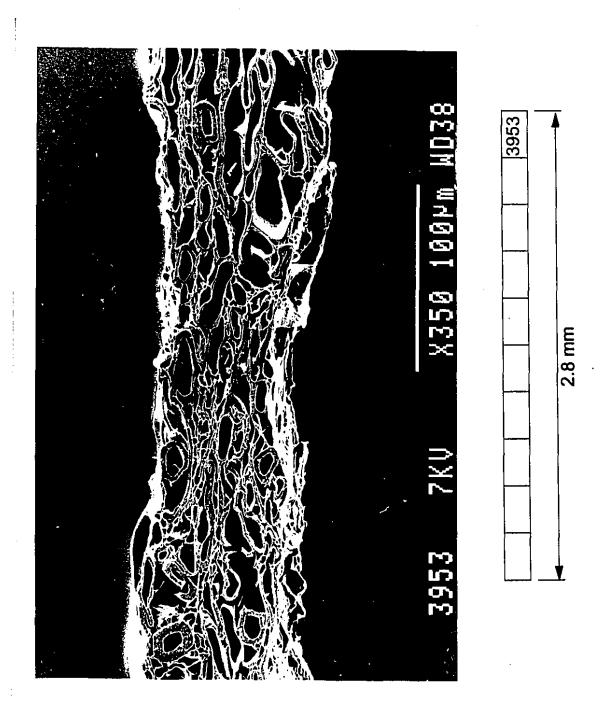


Figure 5.17 Uncalendered sheet. Section plane in CD; magnification 500X.

of a TMP sheet. First, both the thickness and density of the sheet vary significantly from place to place over distances which are much shorter than the nip length of 2 to 5 mm. Density variations, which are due to formation, will result in large local nonuniformity of pressure in the calender nip over small distances in both the machine and cross directions, with the scale of the pressure nonuniformities the same as those for density and thickness. There is also a wide range of fibre sizes and wall thicknesses which contribute to local small-scale variation in fibre density in all three directions. Finally, the sheet arriving in the calender contains a number of fibres damaged in the pulping process.

Some results of these large local pressure variations in the nip can be seen in Figures 5.18 to 5.23, which show sections of paper calendered at a sheet speed of 519 m/min and a line load of 28 kN/m. In the nip, average strain was about 0.35, average thickness was about 78 μ m while the average fibre content by volume was about 40%. After calendering, the average strain is about 0.10, average thickness is about 110 μ m, and average fibre content has increased slightly from 25% by volume in the uncalendered sheet to 27%.

The same small-scale nonuniformity in paper thickness and density seen in the uncalendered sheets are visible in these sections of moderately calendered samples. Fibres deformed by the action of the calender are not easily differentiated from those deformed earlier in the pulping process. However there is a new type of fibre damage, fracture of the fibre wall, which, if present, was not seen in any of the uncalendered samples. The first such fibre is to the left of Figure 5.18, at about the mid-plane of the sheet. This fibre has two internal cracks in the wall, located at the top and bottom of the fibre, and has a single external crack to the right. Another fibre with a single internal fracture may be seen in the centre of Figure 5.20. Numerous other fibres with smaller fractures are visible in these figures; this type of damage will be examined in the next section. There are also several broken fibres; an example may be seen lying on the surface of the sheet to the left in Figure 5.19. Finally, there is a fibre to the right of center in 5.23 which is delaminated: the inner wall has broken and separated from the

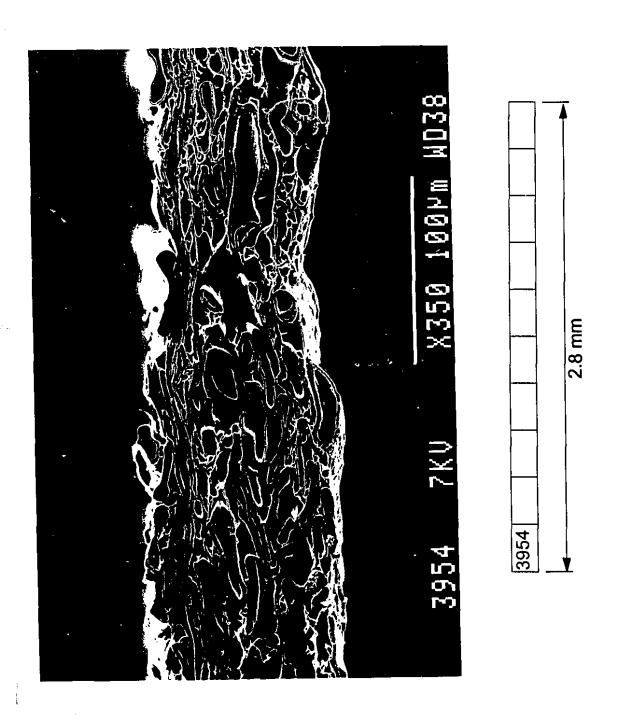


Figure 5.18 Lightly calendered sheet. Section plane in CD; magnification 500X.

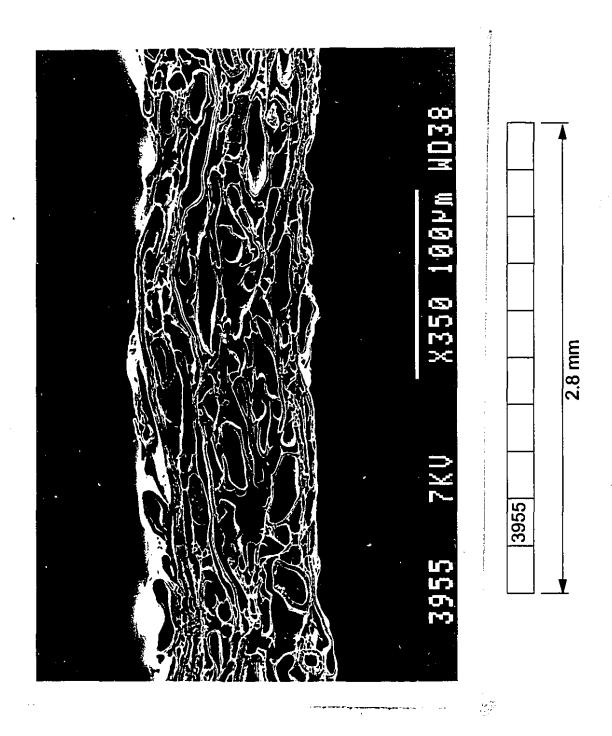


Figure 5.19 Lightly calendered sheet. Section plane in CD; magnification 500X.

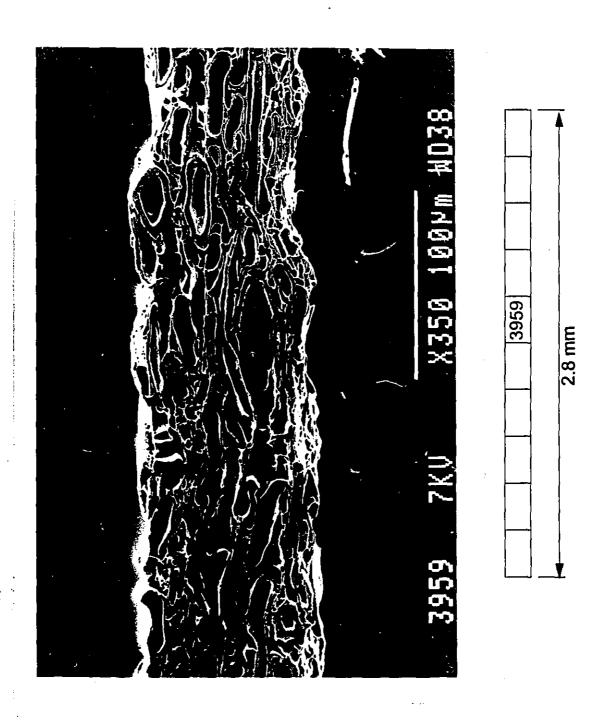


Figure 5.20 Lightly calendered sheet. Section plane in CD; magnification 500X.

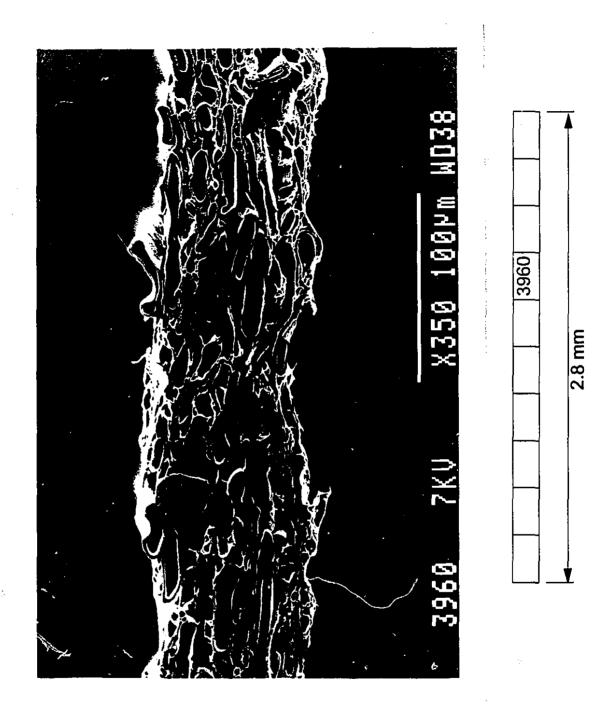


Figure 5.21 Lightly calendered sheet. Section plane in CD; magnification 500X.

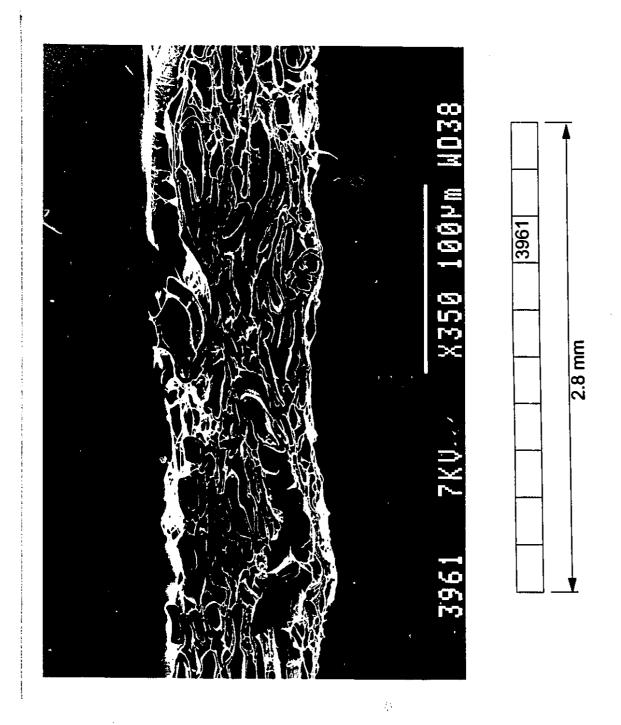


Figure 5.22 Lightly calendered sheet. Section plane in CD; magnification 500X.

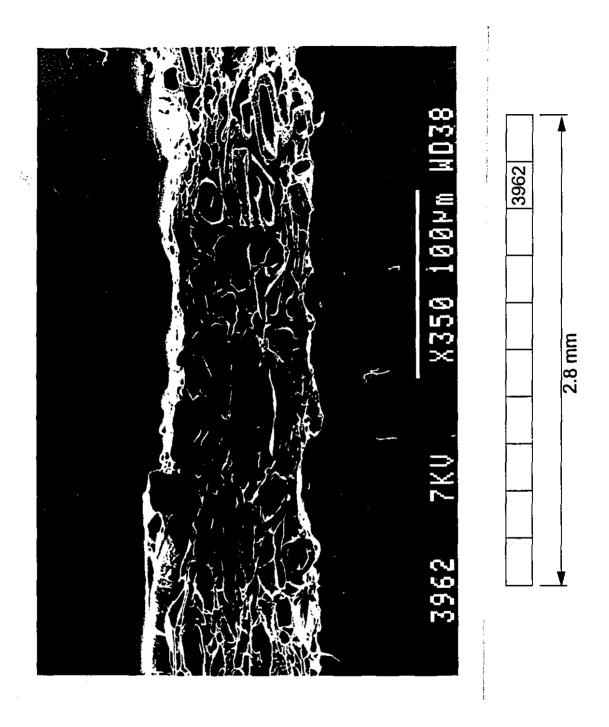


Figure 5.23 Lightly calendered sheet. Section plane in CD; magnification 500X.

outer wall. Taken to its extreme, this type of damage is similar to the broken fibres seen in Figures 5.12 and 5.19.

Results of local pressure variations are even more evident in Figures 5.24 to 5.29, which are sections of heavily calendered paper. Sheet speed was 304 m/min, and load was 206 kN/m. Average in-nip strain reached 0.65, at which point the average fibre content by volume was over 70% and the average thickness was 42 μ m. At this high average strain, which far exceeds industrially relevant conditions, all but the pores which were originally the widest would have been collapsed in the nip. Locally thick or dense areas would have seen much higher pressures, and in places the fibre content may have approached 100% by volume. Average permanent strain in this case is 0.33, and the average fibre content after calendering is 37% by volume.

Figures 5.24 and 5.25 illustrate the large number of damaged fibres in this heavily calendered sheet. In Figure 5.24, there are two large fibres to the lower left which are damaged: one fractured internally and one broken. To the right there are several fibres with internal fractures or breaks in the fibre wall, in particular a round one at the lower right which has several fractures on the inner side of the wall at the top. There are several other internal fractures in these six micrographs, but only a few external fractures; one may be seen in Figure 5.26 at the sheet mid-plane and to the right. Another fibre showing both internal and external fractures along with delamination of the inner wall may be seen in Figure 5.27, lying on top of the sheet to the right of centre.

The variability in paper density is well illustrated in Figures 5.28 and 5.29. A large dense area to the right of 5.28, which is continued at the left of 5.29, has supported the calendering load in the nip and the fibres have been severely deformed, while large coarse fibres in adjacent low-density areas to the left of 5.28 and right of 5.29 have seen much less deformation and thus little damage. The locally low in-nip pressure occurring in these bulky areas must be offset by correspondingly high local pressure in dense areas, implying that some fibres will have experienced extremely high pressure in the nip even when the CD average line load is low.

The conclusion to be drawn from these micrographs is that substantial local

g seeds

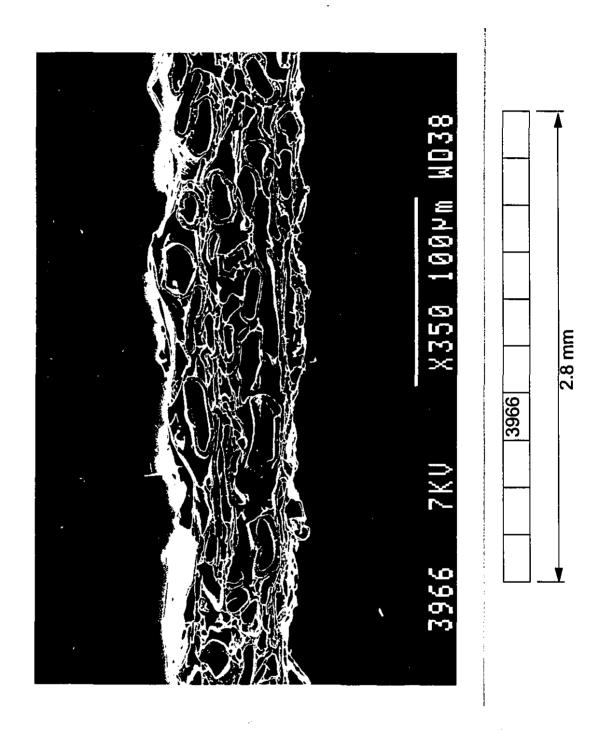


Figure 5.24 Heavily calendered sheet. Section plane in CD; magnification 500X.

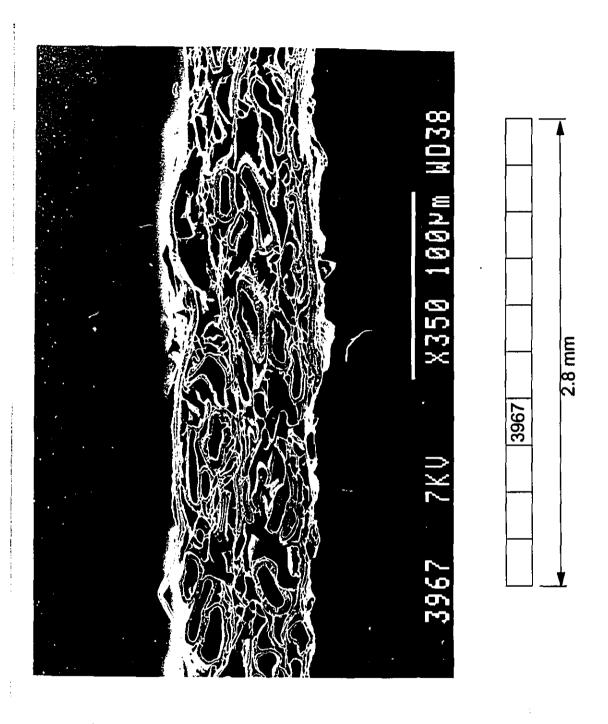


Figure 5.25 Heavily calendered sheet. Section plane in CD; magnification 500X.

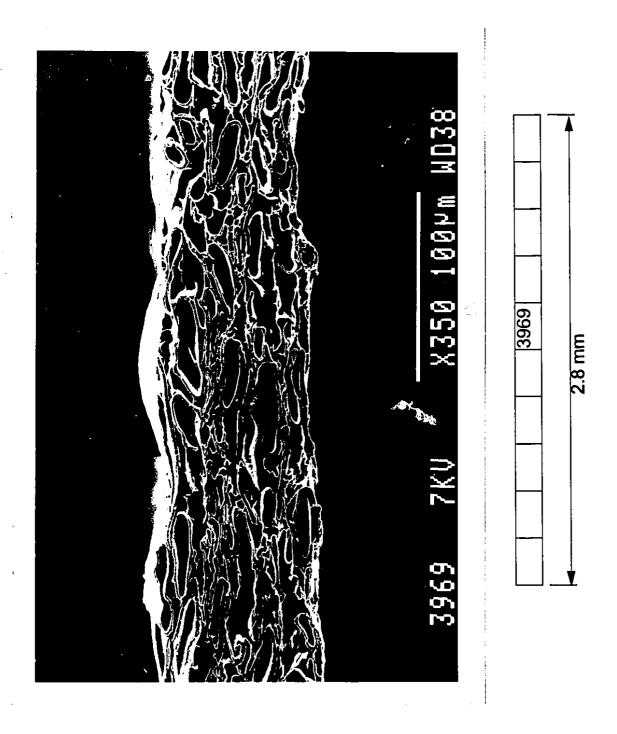


Figure 5.26 Heavily calendered sheet. Section plane in CD; magnification 500X.

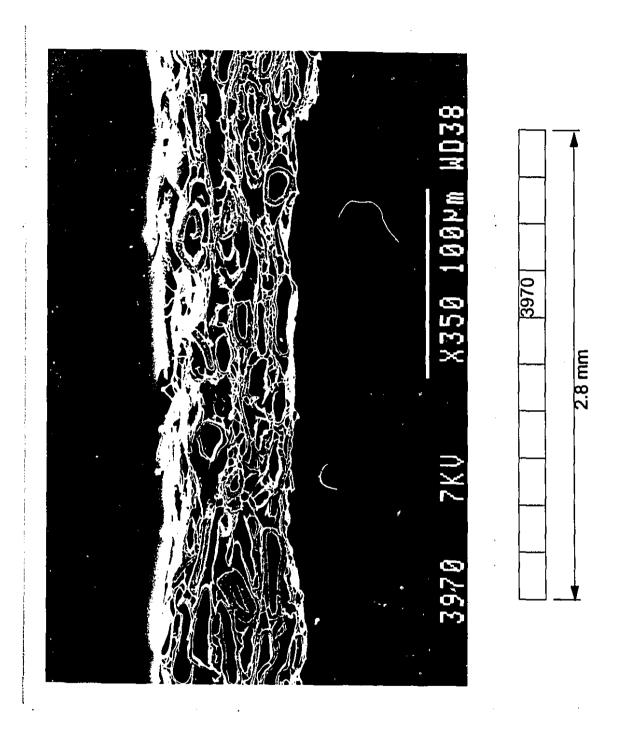


Figure 5.27 Heavily calendered sheet. Section plane in CD; magnification 500X.

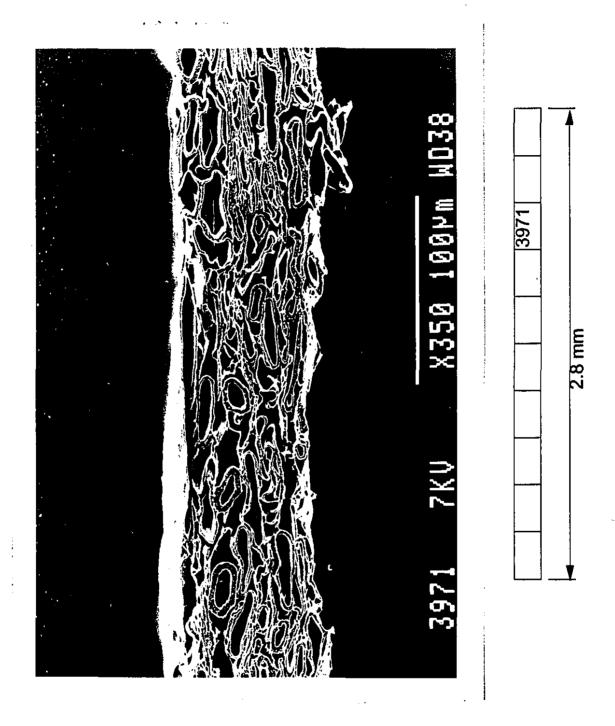


Figure 5.28 Heavily calendered sheet. Section plane in CD; magnification 500X.

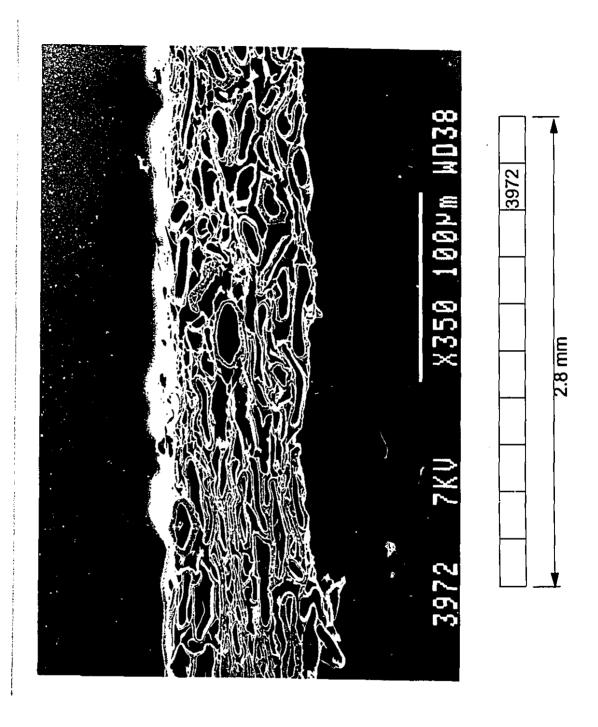


Figure 5.29 Heavily calendered sheet. Section plane in CD; magnification 500X.

nonuniformities in both thickness and density before calendering are a major factor in pressure distribution in both the MD and CD. Denser areas with in-plane dimensions in the range of a few fibre thicknesses will support the entire load through a hard nip, so that local in-nip pressure on the sheet varies substantially over distances as little as 100 μ m even when the roughness level is small. Added to the effect of roughness discussed in Section 5.3.2, and of a substantially shorter nip as suggested in Chapter 4, the result is that peak pressures locally can far exceed average values estimated from nip lengths and line loads.

In order to investigate the effect of earlier processing steps on fibre flexibility and paper response to a calender pulse, similar micrographs were taken of a machine-formed fully bleached kraft paper containing a mix of soft and hardwood fibres. The paper also contained a significant amount of filler. One sample was pressed in a hand sheet press to about 7 MPa, while a second sample was pressed to about 40 MPa. Pressing times, about 0.5 s, were long compared to typical calendering dwell times, about 0.5 ms. A third sample was not pressed. All three samples came from the same 300 mm square sheet. Details of the physical properties are given in Table 5.4.

As with the TMP samples, Figures 5.30 to 5.35 show selected micrographs taken from three continuous strips of ten photos, one strip from each pressing condition. Each strip covers a total distance of about 2.8 mm. The sheet is initially much denser, with less local thickness and density nonuniformities than for the TMP paper. Most fibres have already collapsed and show little or no lumen. Very few obviously damaged fibres are seen in Figures 5.30 and 5.31. At higher pressures, Figures 5.32 to 5.35, the sheet becomes significantly more compacted as the more flexible kraft fibres deform more readily, but there is still little sign of fibre damage such as delamination or fracture. Fibre processing as it affects flexibility is thus an important factor in paper behaviour in a calender.

Qualitatively, the micrographs show that compression of TMP paper is a two-step process, as suggested by Rodal [73]: there is an initial settling of the network and deformation of slender, flexible earlywood fibres, followed by permanent deformation

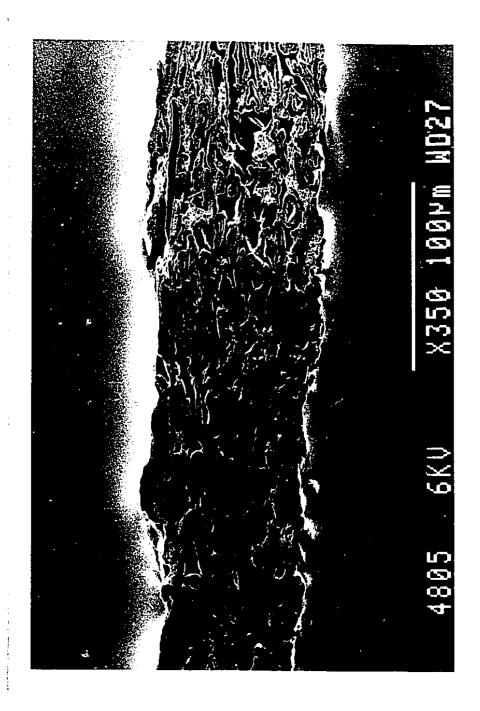


Figure 5.30 Kraft sheet, no pressing. Magnification 500X

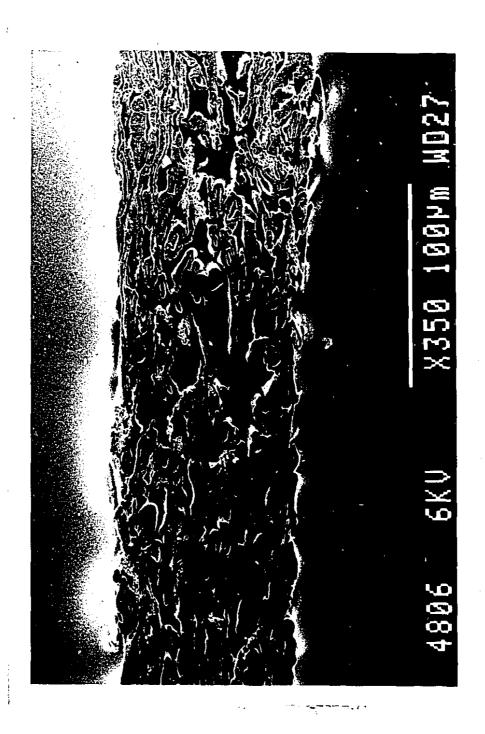


Figure 5.31 Kraft sheet, no pressing. Magnification 500X

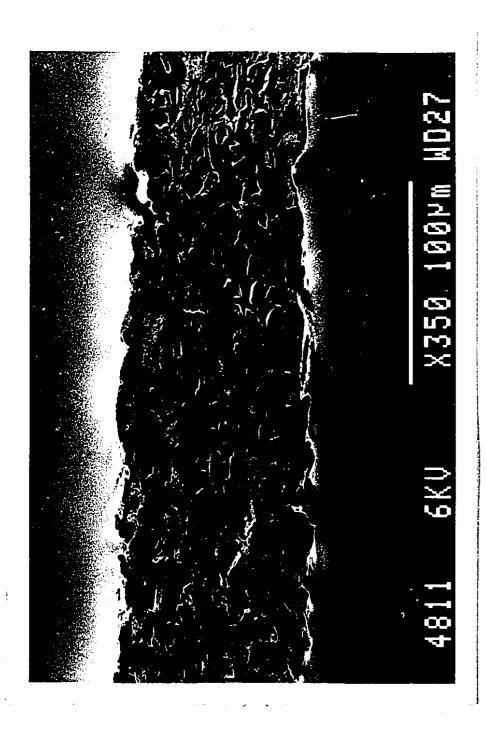


Figure 5.32 Kraft sheet, pressed at 7 MPa. Magnification 500X

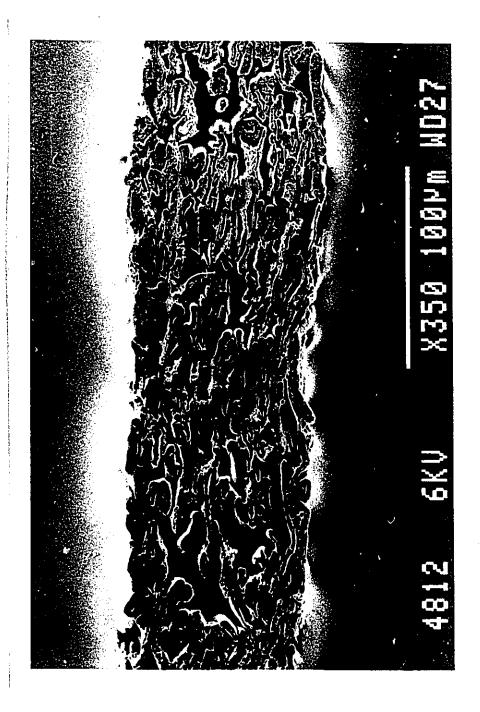


Figure 5.33 Kraft sheet, pressed at 7 MPa. Magnification 500X

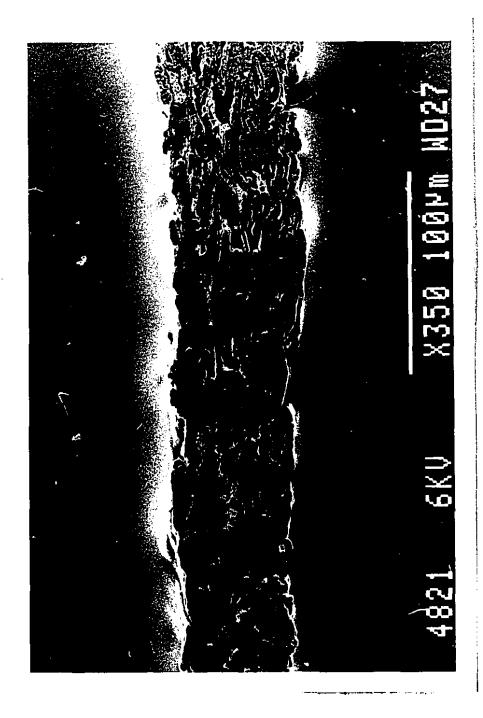


Figure 5.34 Kraft sheet, pressed at 40 MPa. Magnification 500X

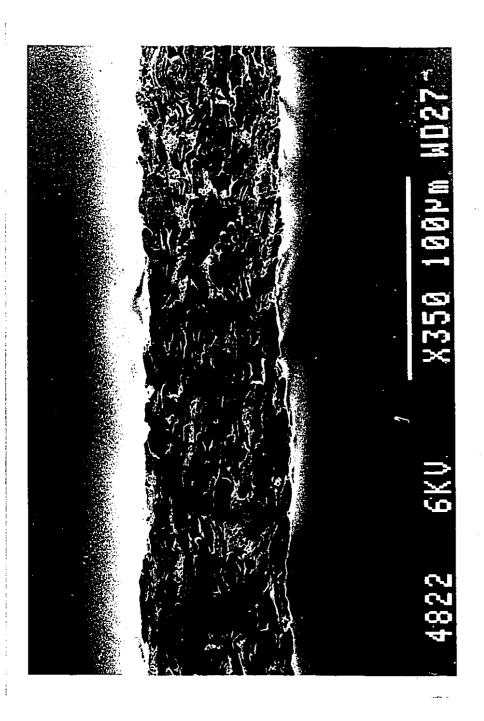


Figure 5.35 Kraft sheet, pressed at 40 MPa. Magnification 500X

of flexible fibres and crushing of stiffer latewood fibres. Kraft fibres are much more flexible and easily deformed, making the sheet easier to calender.

It was suggested in Section 5.3.2 that a linear viscoelastic model incorporating a statistical description of surface roughness might be adequate to describe paper behaviour in a calender nip. It has been shown in this section that while a

TABLE 5.4: Properties for kraft paper samples in Figures 5.30 to 5.35.

Basis weight	68.5 g/m ²	
Bulk	1.5g/cm ³	
Caliper	102 μm	
Filler	13.5%	

method of incorporating the effect of surface roughness may be neccessary, this is not sufficient for modeling paper in a calender nip. Additional probability functions describing local nonuniformity of density and thickness are also necessary. Thicker areas will experience higher strains, and dense spots will be harder to compress; at the very least the elastic moduli G_i will thus be functions of density. In the limit, each individual fibre could be modelled using a single Burger's model. However, several problems arise with this approach. First is the large variation in fibre wall thickness and flexibility, implying that each model would have its own viscoelastic parameters. Second, the fracture behaviour seen in the more brittle fibres represents a sudden change in behaviour which is not easily modeled using simple linear models. Finally, assembling fibres in a three-dimensional pattern does not create a sheet unless the bonds joining the fibres are also modeled. While there exists (or will soon exist) sufficient computing power to accomplish this modeling task using iterative numerical methods, there appears insufficient justification at this time to expend the effort.

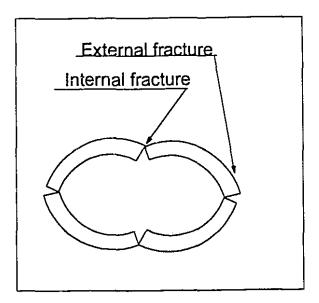
In conclusion, the assumptions of smoothness and homogeneity used in Section 5.2.2 are clearly invalid when the material is paper. The violation of these assumptions should be expected to cause non-linearities, as were indeed observed in the results of the linear viscoelastic modeling.

5.3.4 Fibre fracture and strength losses in calendering TMP

Calendering reduces the bulk, roughness and thickness variations in the sheet. These improvements are generally obtained at the expense of reduced strength properties. Charles and Waterhouse [16] measured the effect of supercalendering on strength properties such as breaking length and tear for several types of paper. They found that while densification by wet pressing improved these properties and reduced their anisotropic nature, densification by supercalendering had the reverse effect. They suggest that since wet pressing increases bonding, these results implicate bond failure as a major source of the observed strength loss. Similarly, Shallhorn [74] noted a reduction in effective fibre length when calendering kraft sheets. Effective fibre length is obtained by assuming the number of bonds per unit fibre length and the average strength of a bond are constants; since true fibre length is not decreased in calendering, the number or the quality of the bonds anchoring the fibre in the sheet must be reduced.

Gratton et al. [31] performed a series of strength measurements on food board made from a kraft pulp and calendered at various conditions. Their highest calender load was 100 kN/m, resulting in a permanent bulk reduction of about 0.26. At this loading condition, they reported decreases in MD and CD breaking strength, elastic modulus and in-plane compressive strength in the range of 10% to 15%. Stretch at break was increased by 10% in the CD and by 15% in the MD. Strength reductions when calendering recycled newsprint were also reported by Gratton [32]. Most strength properties were reduced slightly when permanent strains of up to about 0.25 were imposed, but dropped quickly at higher strains. For an uncalendered machine-made paper which was then recycled and made into handsheets, burst and tensile indices were reduced by only 3% and 5%, respectively, when the handsheet was calendered to $\varepsilon_p = 0.25$; however, the indices were reduced by 30% and 43%, respectively, at $\varepsilon_p = 0.36$. Similarly, tear index and specific Young's modulus were reduced by 15% and 33% at $\varepsilon_p = 0.25$, while both were reduced by 43% at $\varepsilon_p = 0.36$. Zero-span breaking length was unaffected by calendering when the paper entering the recycling stage was

};



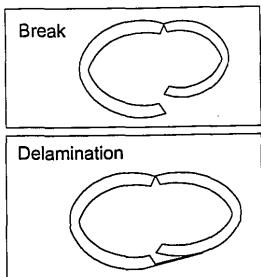


Figure 5.36 Types of fibre damage

uncalendered. Since zero-span is related to the axial tensile strength of an individual fibre, fibre fractures oriented along the axis of the fibre as seen in the micrographs of Figures 5.18 to 5.29 would not reduce the zero-span length unless the fibre were oriented at an angle to the direction of the applied stress.

Crotogino [21] published data showing conventionally calendered newsprint suffers strength losses of 20% to 25% when calendered to $\varepsilon_p = 0.23$, increasing to 30% to 48% at $\varepsilon_p = 0.34$. Strength losses for a given bulk reduction were lower when the newsprint was calendered using the temperature gradient technique, where the relatively cool sheet encounters a hot roll in the nip; the sheet is not wrapped around the roll and there is therefore no heat transfer to the sheet from the roll before the nip. The result is higher fibre flexibility and increased fibre deformation at the surface due to the elevated temperature, improving the smoothness and gloss with less bulk reduction. The cooler, stiffer, more brittle fibres in the centre of the sheet are not strained as highly since the more ductile surface fibres deform more readily, thus leading to less damage to the body of the sheet and increased strength properties.

The micrographs presented in the previous section illustrate two quite different

 s_{i+1}^*

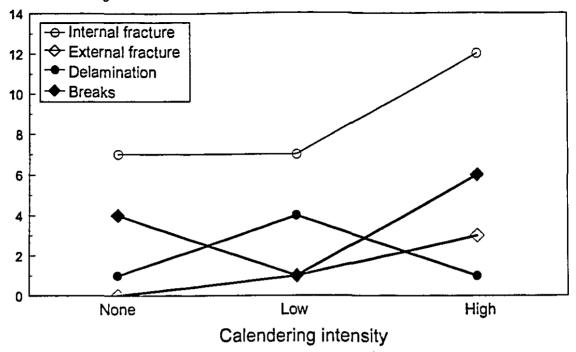
TABLE 5.5: Number of damaged fibres at different calendering conditions

	Uncalendered	Low intensity	High intensity
Internal fractures	7	7	12
External fractures	0	1	3
Delaminations	1	4	1
Breaks	4	1	6

deformation mechanisms: plastic, or ductile deformation of thinner-walled fibres, and fracture, or brittle deformation of coarser, stiffer fibres. Under more severe calendering conditions, the evidence of Figures 5.18 to 5.29 suggests that fracture is increasingly common. In order to quantify the fibre damage identified in Section 5.3.3, the complete set of micrographs was examined to obtain a count of fibres that had experienced different types of damage. Whole, damaged fibres (as opposed to small pieces of fibre) were subdivided into the four types illustrated in Figure 5.36, internal fracture, external fracture, delamination and breakage. The total number of micrographs taken for a given calendering condition covered a CD length of about 2.8 mm, or, given the initial thickness of about 120 μ m, a sectional area before calendering of about 0.33 mm². Results are given in Table 5.5 and Figure 5.37. As the sample size is small, the results should be interpreted with caution. Nonetheless, the four types of damage appear to be due to two different processing steps, each imposing a different type of load on individual fibres.

Page [63], Page and De Grâce [64] and McIntosh [55], among others, have shown that delamination occurs in refining and beating processes, where the fibre encounters shearing forces. In extreme cases of delamination, the type of axial break illustrated in Figure 5.36 can occur. As seen in Figure 5.37, the number of delaminated fibres in the

Number of damaged fibres



Flaws per square mm sectional area

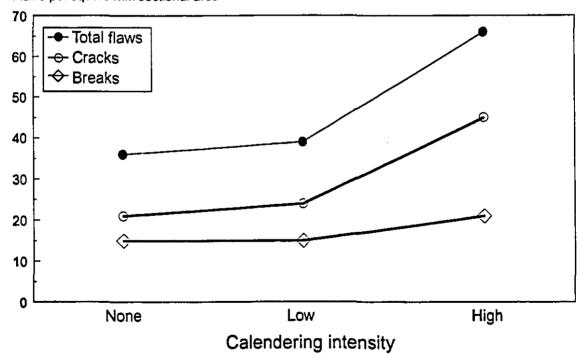


Figure 5.37 Damaged fibres as a function of calendering intensity.

TABLE 5.6: Effect of load condition on number of damaged fibres.

	Uncalendered	Low intensity	High intensity
Compression-damaged	7	8	15
Shear-damaged	5	5	7

samples obtained in this study increases from 1 to 4, then decreases to 1 as calendering intensity is increased. Similarly the number of broken fibres decreases at first from 4 to 1, then increases to 6. These values imply that the average number of fibres which are either delaminated or broken is essentially constant, at 2.0 and 3.7 per 0.33 mm² respectively, and that the standard deviations for these measurements are about 1.7 and 2.5. The sum of these shear-deformed fibres thus increases slightly from 5 to 7, as shown in Table 5.6. If this total is assumed unaffected by calendering then the micrographs display an average of 5.7 shear-deformed fibres with a standard deviation of 1.2. As the increase is not statistically significant, this type of damage is not attributable to the calendering process.

Fracture, on the other hand, can be attributed to tensile failure of the fibre wall caused by compressive loads. A wood fibre is itself a structure consisting of smaller fibrous strands called fibrils. The relatively low number of external fractures in all sheets can be explained by considering the fibril angle of a typical fibre. In the thicker, inner wall of the fibre, the fibrils are aligned at a small angle to the fibre centreline, while the thinner, external wall has the fibrils wound at a larger angle to the fibre centreline, as illustrated in Figure 5.38. A compressive stress on a cylindrical element, Figure 5.39, would be expected to generate tensile stresses perpendicular to the load at the inner wall at the top and bottom, and at the outer wall parallel to the load. An isotropic material would thus be as likely to fracture at the outer wall as at the inner. The data in Table 5.5 imply that this is not the case for paper, as the number of external fractures is con-

sistently smaller than the number of internal fractures. However, a tensile stress on the inside of the fibre tends to separate fibrils, while externally the stresses are much more closely aligned with the axis of the fibrils. Thus tensile stresses in the outer wall act to stretch individual fibrils. pulling with the grain of the outer wall, while internal fractures are caused by tensile stresses pulling at right angles to the grain of the inner wall. The result is a greater likelihood of internal fracture due to anisotropic strength properties. The positions of most of the fractures seen in the micrographs are consistent with this theory: most internal fractures are on the top or bottom of the fibre, as seen in both

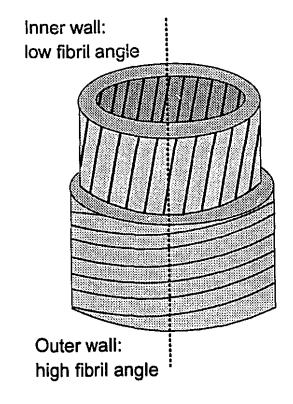


Figure 5.38 Typical fibril structure.

cases in Figure 5.39, while the few external fractures are in the plane of the sheet as seen in the anisotropic case. This effect is also seen in micro-photographs published by Williams et al. [81], where internal fibre fractures oriented in the plane of the sheet appear to be more frequent in TMP after calendering.

The standard deviation estimated for the total number of compression-damaged fibres may be assumed to be similar to that of the shear-damaged fibres, at 1.16. The total number of compression-damaged fibres increases from 7 to 15 as calendering intensity increases, a significant change given the estimated standard deviation. The type and frequency of the fractures is therefore consistent with the compressive loading experienced by the fibres. Compression fractures are much more numerous in a heavily calendered TMP paper than in an uncalendered sheet, and are extremely infrequent in kraft sheets due to the increased flexibility of the fibres.

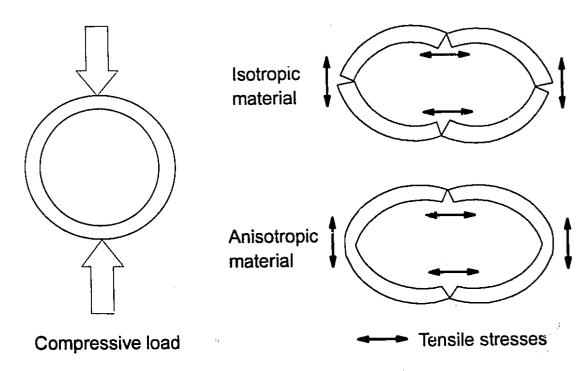


Figure 5.39 Tensile failure in isotropic and anisotropic cylinders loaded in compression.

The observed increase in fibre fracture in the calendered TMP paper is also consistent with observed strength losses. While internal fibre fracture contributes to a general weakening of the fibre network by providing a place for the initiation of a larger crack in the sheet, bond weakening or failure as a contributing factor is not eliminated. When a dense spot is compressed, fibres bordering a less dense area will tend to be displaced sideways in the plane of the sheet, putting shear forces on bonds. These forces in turn weaken or break bonds, creating further weak spots for the initiation of a crack when the sheet is loaded in tension. Unfortunately this type of failure is not visible in the photomicrographs presented here, and so the relative magnitude of the two effects cannot now be estimated. The fact that kraft sheets also lose strength but without the fibre damage seen in the TMP samples suggests that bond failure plays a significant part; nonetheless fibres damaged in the nip can only weaken the structure and must therefore play a part in the well-documented strength reduction in calendered paper.

5.4 Conclusions

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Paper behaviour in a calender nip is far more complex than can be adequately described in a linear viscoelastic model. Paper exhibits significant local nonuniformities in structure and material properties over widely different scales. Sheet formation problems result in the presence of flocs with typical dimensions of a centimetre. At a resolution in the range of a millimetre, there are local nonuniformities in both thickness and density. On a smaller scale, of the order of $100~\mu m$, there are variations in fibre coarseness which cause the effective compressive elastic modulus of the sheet to vary substantially as the sheet is stressed. Finally, at the micrometre level, the variation in fibril angle from the outside to the inside of the fibre, along with variations in fibre coarseness, cause different fibres to react differently to an applied stress: some will collapse and deform while others will fracture, mirroring the ductile vs. brittle failure of metals. All of these effects result in pressure profiles which are highly nonuniform over very small distances in a calender nip, resulting in material behaviour which also varies substantially over the same small distances.

Linear viscoelastic modeling is therefore inadequate for predicting the large deformations found in a calender nip. Paper behaviour in compression is related to fibre properties, which can vary substantially in mechanical pulps, and the distribution of the fibres within the sheet. As characteristic dimensions describing the fibre distribution may be similar to typical nip lengths, the objective of relating the coefficients of the empirical relations described in Chapter 4 to more fundamental parameters describing fibre properties and paper structure was not achieved.

The evidence obtained from photomicrographs of sheet cross-sections indicates that fibre fracture may play a significant role in the well-documented strength losses found in calendered papers. While it is likely that a large part of these strength losses are due to bond breakage, there is visual evidence that fibre fracture occurs more frequently as loads are increased, reaching a peak under extreme calendering conditions of several broken fibres every millimetre along the sheet. This damage to the fibres undoubtedly

contributes to the general strength losses noted in heavily calendered sheets, although the magnitude of the contribution is not known.

6. Conclusions

6.1 Summary

Paper is calendered to reduce its surface roughness, thus improving the printing properties, and to minimize thickness variations from place to place in the sheet, thus improving the printability of the sheet in modern high-speed printing presses: the process improves the quality of the finished product.

To reduce thickness variations in the cross-machine direction, improved control methods are necessary. One of the problems in CD calender control is the delay between the implemention of a control action and the material response. CD control is usually implemented using localized heating or cooling of the roll, altering the local roll radius and thus the local load; the delay is due to the large thermal inertia of a typical calender roll, resulting in a long response time. As a result of this response time, which may be as large as 10 to 20 minutes [41, 42], control systems using feedback methods may require excessively long times to settle if a reasonable estimate of paper response to a control action is not available. A typical 10 metre wide machine running at 1000 m/min will produce almost 30 tonnes per hour of 48 g/m² newsprint; if the control system requires 20 minutes to arrive at 95% of the target variation, almost 10 tonnes of substandard paper will have been made. To improve the response of the control system, there are two possibilities: reduce the response time by reducing roll thermal inertia, or improve the effectiveness of the control action by improving the estimate of the material response to a control input. Reducing the thermal inertia of a calender roll is limited by the necessity of maintaining roll resistance to bending and other mechanical considerations.

The present work shows that it is possible to improve the estimate of paper response to a control action. This improved estimate will allow the use of control algorithms using feedforward methods, thus eliminating the possibility of a control action overshooting the target. As described in Section 4.2.3, an empirical relationship between CD local permanent strain ε_p and CD local in-nip strain ε_p was obtained from the present

experimental data. This relationship will facilitate design of a complete cross-direction control system for paper strain leaving a calender. Once future work has elucidated the bulk and furnish dependencies, the in-nip strain existing at any CD position can be estimated given the permanent strain measured at that same position. Thick or thin streaks require higher or lower permanent strains; from this new permanent strain the required new local in-nip strain can be estimated. The difference between current and required in-nip strains gives explicitly the required local roll shape. With knowledge of the roll response to an actuator input, determined by a previous study in this laboratory, the thermal energy required to obtain the desired roll profile and thus effect the required change in permanent strain can be calculated.

The primary goal of this thesis, clarifying the material response to an actuator input, has thus been attained.

The relationship between load and strain has also been described qualitatively in terms of paper and fibre properties such as specific volume and fibre coarseness. This results in a better understanding of paper behaviour subjected to compressive loads. At very low loads the structure is deformed and thin-walled, flexible fibres collapse; if the load is removed there is very little permanent strain as most of the deformation is recovered. If the load is increased permanent deformation in both the structure and the more flexible fibres begins to occur, while coarser fibres tend to fracture. The limiting density in the nip, which was approached in this study, is close to the accepted value for fibre density while the limiting density after recovery is somewhat less, implying that within the range of loads examined here there is always some recoverable strain.

Thus the second goal of this project, an improved understanding of paper structure and properties, has also been attained.

Finally, viscoelastic modeling methods were used in an attempt to relate the coefficients of the empirical results to paper and fibre properties. Parameters for two linear viscoelastic models were estimated from the data and were found to be functions of the processing parameters. It is concluded that in calendering under typical industrial conditions, the stress-strain behaviour of paper cannot be described using linear

viscoelastic methods. The assumptions of continuum mechanics do not apply to paper, which is a rough, structured material made up of hollow fibres with a wide range of coarseness and flexibility. As well, irreversible damage to fibres and bonds occurs in calendering; modeling this type of behaviour is beyond the scope of simple stress-strain relationships. Thus linear viscoelastic methods are inadequate for describing paper compression in a calender nip, and no simple relationship exists between the coefficients of the various empirical results and more fundamental paper and fibre properties.

6.2 Contributions to knowledge

The present work made contributions in three distinct categories. In the experimental part of the work, empirical relationships describing unique in-nip strain measurements were presented which will be of use in the design of improved control systems. Secondly, the potential for the use of linear viscoelastic modeling methods to describe the behaviour of paper in a calender nip was determined. Finally, analysis of the microstructure of different types of paper subjected to different calendering conditions provided an improved understanding of the interaction between the calendering process and paper and fibre structure.

Contributions pertaining to improved control systems are:

- 1. Equations relating local CD paper strain in the nip of a calender to local CD load have been obtained from experimental measurements;
- 2. A relationship between in-nip and permanent strain which can be used to estimate local CD in-nip strain given local CD permanent strain has been obtained from experimental measurements; this relationship, describing the material response to an actuator input, is the missing link in a complete CD calender control model;
- 3. There is evidence that the outgoing nip length is small compared to the ingoing length, implying that average or peak pressures in the nip may be higher than current estimates;
- 4. The nip length corresponding to full contact is significantly shorter than the calculated length when paper roughness is as little as 5% of sheet thickness, again implying nip pressures are higher than current estimates;
- 5. The effects of points 3 and 4 being additive, current methods for estimating in-nip pressures appear to produce results which are unrealistically low;
 - 6. Machine direction stretch is small compared to typical z-

direction strain, which can be understood by considering fibre bending and flattening as the sheet enters the nip;

7. Kerekes' prediction for a_R has been verified.

The viscoelastic modeling of the experimental measurements led to the following contributions:

- 8. Paper thickness recovers exponentially on leaving a calender nip; the recovery can be described for typical industrial speeds using a time constant of 10 to 30 ms;
- 9. The instantaneous elastic response to a calendering pulse is small, verifying previous experimental work;
- 10. The total recoverable strain approaches a constant at higher loads, implying the existence of two mechanisms in paper compression: delayed elasticity and partial permanent deformation;
- 11. As a result of points 8, 9 and 10, paper behaviour in a calender nip may be described as viscoelastic;
- 12. Constitutive equations for two linear viscoelastic models were derived for the strain history imposed in a calender. Both models were fitted to the data; the Burger's model described observed behaviour significantly better than did the standard linear liquid;
- 13. Initial estimates of parameters for the Burger's model reproduce predicted and measured pressure profiles, but are functions of calendering conditions;
- 14. As a result of points 11 and 13, paper behaviour may be concluded to be non-linear viscoelastic; the most likely reason for the non-linear behaviour is that the assumptions of continuum mechanics are not applicable to paper.

The analysis of the effect of fibre and paper structure on the calendering process,

and of the process on the properties, produces the following contributions:

- 15. Paper compressibility is a strong function of void fraction;
- 16. Local nonuniformity of thickness, density and fibre coarseness can be substantial, causing correspondingly large local nonuniformity in the pressure pulse applied to the sheet;
- 17. Internal and external fibre fracture is more common in calendered than uncalendered sheets, and is seen mainly in sheets made from TMP. Sheets made from a fully bleached kraft exhibit very little fracture;
- 18. Delaminated and broken fibres are no more common in calendered sheets than in uncalendered sheets, and do not appear in significant numbers in kraft sheets;
- 19. As a result of points 17 and 18, delamination and breakage is probably attributable to shear stresses in the refiner, and fracture to compressive stresses in the calender;
- 20. There appear to be two types of fibre fracture, with internal fracture more frequent than external fracture; this finding was related to the variation of fibril angle from layer to layer in the fibre wall;
- 21. As a result of points 17, 18, 19 and 20, it is concluded that fibre fracture plays a role in strength losses when calendering paper made from TMP, although the relative magnitude of the mechanisms of fibre fracture and bond breakage in strength losses cannot yet be estimated.

6.3 Suggestions for future study

The present work has also raised new questions.

The various empirical stress-strain relationships, whether in-nip or permanent, and whether based on the master creep equation of Colley and Peel or the calendering equation of Crotogino, are functions of paper bulk. Linear relationships between strain and initial bulk are poor approximations to reality. An improved description of the bulk dependency of paper strain in a calender would improve the predictive ability of the various relationships when used successively for multiple nips. Experimentally, difficulties arise due to the narrow range of bulk (1.5 to 3.0 cm³/g) compared to the large range of load and roll radius commonly used in industry. In the case of the in-nip calendering equations, the dependency of strain on web temperature and moisture content also needs to be investigated.

Pulp type is another variable whose effect on in-nip strain has not been investigated. Kraft papers are more easily calendered than those from TMP, since kraft pulp fibres are more flexible; whether this effect will carry over to the in-nip case remains to be seen, although there is no obvious reason to suspect otherwise.

Fibre fracture and bond failure have both been suggested as causes for the strength reduction in calendered TMP paper. While the role of bond failure cannot be assessed from the data, the evidence presented here suggests that when TMP is calendered, fracture occurs frequently and is a likely cause of weaker tensile, tear and other strength measurements. However, the present evidence is based on a very small number of micrographs. The relationships between calendering intensity and number of fibre fractures, and number of fibre fractures and strength loss should be investigated in a systematic fashion.

Linear viscoelastic models are inadequate for describing paper compression in a calender nip. Alternative non-linear methods developed to describe shear in a polymer melt are not suited to the structured nature of paper. Other methods of describing paper strain in a calender might involve defining the pressure pulse in terms of an impulse

function, and representing paper and fibre structure using either statistical or fractal methods. The applicability of the modeling might be improved by developing methods for measuring the pressure pulse and nip length in a hard nip, much as Keller has done in a soft nip.

The development of better empirical or viscoelastic models for local CD strain both in the nip and after is of immediate use to industry only if these relationships are used to design better control systems. Development of a complete CD calender control algorithm using feedforward methods is the next step in improving the quality of newsprint made from TMP. In the longer term, a better understanding of paper structure and properties will have benefits which cannot presently be estimated.

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Appendix A1: Raw data and non linear curve fitting results

- Table A1.1: Summary of experimental conditions (Boise-Cascade)
- Table A1.2: Command log input to SYSTAT (Boise-Cascade)
- Table A1.3: SYSTAT output (Boise-Cascade)
- Table A1.4: Summary of experimental conditions (F. Soucy)
- Table A1.5: Command log input to SYSTAT (F. Soucy)
- Table A1.6: SYSTAT output (F. Soucy)

- Table A1.7: SYSTAT output, converted to log₁₀ (Boise-Cascade)
- Table A1.8: SYSTAT output, converted to log₁₀ (F. Soucy)
- Table A1.9: Sorted summary of experimental conditions (F. Soucy)

TABLE A1.1: Summary of experimental conditions (Boise-Cascade)

Num.	Rad	Vel	Load	d Bin	Ca!	en	ep
3258	0.202	96	40	2.00	104	0.405763	0.114020
3259	0.202	96	65	2.00	104	0.454856	0.146035
3260	0.202	96	42	2.00	104	0.408755	0.100358
3261	0.202	96	19	2.00	104	0.234250	0.058499
3262	0.202	96	16	2.00	104	0.198642	0.056037
3263	0.202	96	40	2.00	104	0.412289	0.088185
3264	0.202	96	65	2.00	104	0.457027	0.123691
3265	0.202	322	65	2.00	104	0.447458	0.110804
3266	0.202	324	41	2.00	104	0.394341	0.069169
3267	0.202	323	19	2.00	104	0.230015	0.044361
3268	0.202	321	19	1.94	101	0.204310	0.013059
3269	0.202		40	1.94	101	0.368864	0.035727
	0.202		65	1.94	101		
	0.202		65	1.94	101		
	0.202		41	1.94	101		
	0.202		19	1.94	101	0.192796	
	0.202		19	1.94	101	0.191099	
	0.202		40	1.94	101	0.362231	0.032177
	0.202		65	1.94	101		
3277		97	62	1.94	101	0.448408	0.084662
3278	0.202	97	95	1.94	101	0.499232	
3279		97	134	1.94	101		
3280		97	175	1.94	101		0.195458
3281	0.202	97	170	1.94	101		
	0.202	97	135	1.94	101		0.177048
3283		97	95	1.94	101	0.501250	
3284		97	64	1.94	101	0.445435	
3285			63	1.94	101	0.437326	
3286			95	1.97	103	0.455909	
3287		320	134	1.97	103	0.530040	
3288	0.202		174	1.97	103	0.555424	0.184834
3289		323	169		103		0.181250
3290	0.202		134	1.97	103	0.528227	
3291	0.202		96	1.97	103		
3292	0.202		87		103		
3293	0.202	530		1.97	103		
3294	0.202	530	164	1.97	103	0.545339	0.165230

TABLE A1.2: Command log input to SYSTAT (Boise-Cascade)

```
By
Select
Weight
Use 'c:\phd\calcs\systat\BCAL.SYS'
By
Select
Weight
Print Long
Format 4
Options
Monlin
Model EN=(AA+B*(AL*log(L)+AV*log(V))),
    *(B<0.5*(1-AA)/(AL*log(L)+AV*log(V))),
+(1-0.25*((1-AA)^2)/(B*(AL*log(L)+AV*log(V)))),
*(B>=0.5*(1-AA)/(AL*log(L)+AV*log(V)))
Estimate /Iter=100 / Start = -.4,.1,-.02, Quasi, Print
By
Select
Weight
Use 'c:\phd\calcs\systat\BCAL.SYS'
By
Select
Weight
Print Long
Format 4
Options
Monlin
Model EP=0+(AA+B*(AL*log(L)+AV*log(V))),
    *((B>=-AA/(AL*log(L)+AV*log(V))),
AND (B<0.5*(1-AA)/(AL*log(L)+AV*log(V)))),
+(1-0.25*((1-AA)^2)/(B*(AL*log(L)+AV*log(V)))),
     *(B>=0.5*(1-AA)/(AL*log(L)+AV*log(V)))
Estimate /Iter=100 / Start = -.4,.1,-.02, Quasi, Print
```

TABLE A1.3: SYSTAT output (Boise-Cascade)

DEPENDENT VARIABLE IS EN

SOURCE	SUM-OF-SQUARES	DF	MEAN-SQUARE
REGRESSION	7.2524	3	2.4175
RESIDUAL	0.0024	34	0.0001
TOTAL	7.2222	37	
CORRECTED	0.4588	36	

RAW R-SQUARED (1-RESIDUAL/TOTAL) = 0.9997

CORRECTED R-SQUARED (1-RESIDUAL/CORRECTED) = 0.9947

PARAMETER	ESTIMATE	A.S.E.	LOWER	<95% UPPER
AA	-0.3874	0.0362	-0.4609	-0.3138
AL	0.1128	0.0051	0.1025	0.1232
AV	-0.0048	0.0010	-0.0068	-0.0028

DEPENDENT VARIABLE IS EP

SOURCE	SUM-OF-SQUARES	DF	MEAN-SQUARE
REGRESSION	0.4909	4	0.1227
RESIDUAL	0.0069	33	0.0002
TOTAL	0.5037	37	
CORRECTED	0.1178	36	

RAW R-SQUARED (1-RESIDUAL/TOTAL) = CORRECTED R-SQUARED (1-RESIDUAL/CORRECTED) = 0.9863 0.9414

PARAMETER	ESTIMATE	A.S.E.	LOWER	<95%> UPPER
**	-0.3874	0.0362	-0.4598	-0.3150
AL.	0.0383	0.0042	0.0298	0.0468
AV	-0.0118	0.0020	-0.0159	-0.0077

N = 8-3 miles					
Num Rad Vel Load		. ₽ _р	Num Rad Vel Load	Bin Cal en	ŧ,
638 0.202 96 70	2.60 128 0.563049	0.214025	729 0.202 528 136	2.62 128 0.621094	0.278717
639 0.202 96 86	2.60 128 0.593847	0.246898	730 0.202 500 172	2.62 128 0.639931	
640 0.202 96 137	2.60 128 0.630362	0.312001	731 0.202 528 207	3 43 130 0 4/7/00	0.314261
641 0.202 96 173	2.60 128 0.644190	0.330506		2.62 128 0.647189	0.336368
642 0.202 96 208			749 0.202 97 70	2.64 129 0.560995	0.216158
	2.60 128 0.656385	0.344223	750 0.202 96 86	2.64 129 0.590962	0.252901
	2.60 128 0.542160	0.181364	751 0.202 96 113	2.64 129 0.612769	0.297593
644 0.202 169 87	2.60 128 0.568938	0.215326	752 0.202 96 172	2.64 129 0.639536	0.350021
645 0.202 169 136	2.60 128 0.606718	0.258005	753 0.202 97 208	2.64 129 0.651359	0.365277
646 0.202 169 173	2.60 128 0.620863	0.284779	754 0.202 173 90	2.64 129 0.574967	0.247959
656 0.202 299 70	2.60 128 0.555454	0.158935	755 0.202 171 70	2.64 129 0.532736	
657 0.202 297 70	2.60 128 0.543094	0.155887		2.64 129 0.565497	0.186936
658 0.202 301 84	2.60 128 0.577954	0.200738			0.242372
			757 0.202 169 136	2.64 129 0.603881	0.297808
	2.60 128 0.570027	0.181796	758 0.202 169 173	2.64 129 0.621645	0.334766
661 0.202 299 70	2.60 128 0.541640	0.155887	759 0.202 169 208	2.64 129 0.633040	0.355823
662 0.202 301 86	2.60 128 0.579123	0.188763	760 0.202 298 .71	2.64 129 0.512394	0.183713
667 0.202 173 70	2.59 127 0.555515	0.194344	761 0.202 97 70	2.63 118 0.559317	0.188502
668 0.202 172 87	2.59 127 0.585882	0.221966	767 0.202 173 70	2.64 119 0.558479	0.202098
669 0.202 173 136	2.59 127 0.608090	0.308999	768 0.202 173 69	2.64 119 0.551407	0.195922
670 0.202 173 172	2.59 127 0.627037	0.329606	769 0.202 173 86	2.64 119 0.577620	
671 0.202 301 70	2.59 127 0.527083	0.168914			0.241118
672 0.202 301 86	2.59 127 0.548908	0.203551		2.64 119 0.574753	0.236449
			771 0.202 99 70	2.64 119 0.553924	0.211443
673 0.202 301 137		0.268004	772 0.202 99 86	2.64 119 0.584932	0.263551
674 0.202 301 172	2.59 127 0.599881	0.292557	773 0.202 98 136	2.64 119 0.623497	0.321731
683 0.202 99 70	2.62 128 0.563017	0.231203	774 0.202 98 173	2.64 119 0.640319	0.350464
685 0.202 98 135	2.62 128 0.634868	0.328927	775 0.202 98 208	2.64 119 0.648387	0.567287
686 0.202 98 172	2.62 128 0.650343	0.365330	776 0.202 173 70	2.64 119 0.540978	0.180607
687 0.202 98 209	2.62 128 0.661066	0.382232	777 0.202 100 70	2.64 119 0.568294	0.218458
690 0.202 170 88	2.57 126 0.585108	0.251791	778 0.202 304 69		
691 0.202 170 136	2.57 126 0.618969	0.313110		2.64 119 0.550039	0.181305
692 0.202 169 172	2.57 126 0.638333			2.64 119 0.546150	0.183879
693 0.202 170 208		0.344874	780 0.202 304 70	2.64 119 0.529769	0.230604
	2.57 126 0.653670	0.367815	781 0.202 301 85	2.64 119 0.560793	0.288549
	2.57 126 0.518759	0.175248	782 0.202 301 136	2.64 119 0.603550	0.311915
695 0.202 292 85	2.57 126 0.555302	0.222679	783 0.202 99 70	2.64 119 0.513029	0.207010
696 0.202 296 135	2.57 126 0.600321	0.287970	784 0.202 98 86	2.64 119 0.554572	0.253735
697 0.202 296 172	2.57 126 0.614109	0.315985	785 0.202 98 136	2.64 119 0.599624	0.315423
698 0.202 97 70	2.57 126 0.528283	0.204368	786 0.202 98 172	2.64 119 0.619100	0.342987
699 0.202 97 85	2.57 126 0.561799	0.246502	787 0.202 98 208	2.64 119 0.631240	0.360280
700 0.202 96 135	2.57 126 0.601700	0.306718	788 0 202 500 70	2.64 119 0.527156	
701 0.202 96 172	2.57 126 0.617775	0.333407			0.163313
702 0.202 97 207	2.57 126 0.625020			2.64 119 0.513579	0.166585
703 0.202 171 208		0.355023	790 0.202 539 86	2.64 119 0.544298	0.189018
	2.57 126 0.615648	0.334734	791 0.202 535 136	2.64 119 0.591878	0.257705
704 0.202 174 70	2.57 126 0.551132	0.190916	797 0.202 535 69	2.70 122 0.538860	0.132620
705 0.202 171 B6	2.57 126 0.585286	0.236354	798 0.202 528 87	2.70 122 0.560903	0.171192
706 0.202 171 136	2.57 126 0.633274	0.306496	800 0.202 171 70	2.70 122 0.550509	0.182606
707 0.202 173 172	2.57 126 0.642876	0.337156	801 0.202 171 86	2.70 122 0.574741	0.225516
708 0.202 171 208	2.57 126 0.654114	0.353037	802 0.202 171 136	2.70 122 0.616066	0.274368
709 0.202 171 207	2.57 126 0.648544	0.336719	803 0.202 171 172	2.70 122 0.630581	0.300393
		0.195492			
713 0.202 97 87	2.62 128 0.587768	0.243606	804 0.202 171 208	2.70 122 0.641189	0.312259
71/ 0 202 9/ 6/	2.02 120 0.301100		805 0.202 323 70	2.70 122 0.516023	0.168685
714 0.202 95 137	2.62 128 0.631038		806 0.202 323 86	2.70 122 0.544499	0.197898
715 0.202 97 172	2.62 128 0.645194	0.331383	807 0.202 321 136	2.70 122 0.581121	0.245608
716 0.202 96 207	2.62 128 0.655435	0.353273	808 0.202 323 173	2.70 122 0.602672	0.286464
717 0.202 171 70	2.62 128 0.536223	0.182705	809 0.202 173 70	2.70 122 0.508567	0.166401
718 0.202 169 87	2.62 128 0.566758	0.228652	810 0.202 171 87	2.70 122 0.538076	0.201095
719 0.202 171 135	2.62 128 0.607111	0.285219	811 0.202 171 136	2.70 122 0.577099	0.254967
720 0.202 171 174	2.62 128 0.622991	0.312094	812 0.202 170 173	2.70 122 0.595233	
721 0.202 169 209	2.62 128 0.627212	0.331383		2 70 166 0.777633	0.285092
	2.62 128 0.488312		813 0.202 169 208	2.70 122 0.604640	0.295142
		0.178370	816 0.202 95 69	2.66 120 0.559722	0.239080
723 0.202 500 85	2.62 128 0.519865	0.211964	817 0.202 94 86	2.66 120 0.583629	0.279406
724 0.202 528 135	2.62 128 0.558498	0.270264	818 0.202 94 136	2.66 120 0.622318	0.337814
725 0.202 500 173	2.62 128 0.582433	0.289120	819 0.202 94 173	2.66 120 0.637761	0.370030
	2.62 128 0.534726	0.179454	820 0.202 94 209	2.66 120 0.643639	0.383013
728 0.202 528 86	2.62 128 0.567763	0.217382	825 0.202 173 137	2.68 121 0.617463	0.332797

N- 8-d N-1 1 - 1 81			M						
Num Rad Vel Load Bin		₽ _P				oad Bir			* _P
	121 0.601262	0.317847						0.641815	0.279518
828 0.202 175 69 2.68	121 0.572799	0.226311	9 37	0.254	97	85 2.63	126	0.634045	0.223812
829 0.202 304 69 2.68	121 0.560561	0.205152	938	0.254	97 1			0.660661	0.296216
	126 0.534181	0.198679		0.254				0.674926	0.335171
	126 0.574351	0.255916		0.254				0.688444	0.344195
	126 0.619357	0.333187		0.254				0.504274	0.214235
	126 0.637882	0.358943		0.254				0.573662	0.278367
	126 0.566930	0.255036	976	0.254	174			0.498936	0.232438
836 0.202 171 136 2.58	126 0.609316	0.313374	977	0.254	173 1	37 2.67	128	0.559132	0.294176
837 0.202 170 174 2.58	126 0.620074	0.342653	978	0.254	174 1	75 2.67	128	0.592821	0.308909
	126 0.543544	0.211888	979	0.254	173 2			0.620738	0.334256
	126 0.588458	0.282554		0.254				0.487170	0.217269
	126 0.607044	0.312713		0.254				0.547623	0.281612
	126 0.615471	0.326582		0.254				0.581061	0.305875
	129 0.470818	0.217053		0.254				0.604757	0.326247
	129 0.525262	0.272099	984	0.254	500	87 2.67	128	0.469111	0.199073
848 0.202 500 174 2.63	129 0.543784	0.306422	985	0.254	535 13			0.534103	0.269266
849 0.202 528 208 2.63	129 0.554969	0.331678	986	0.254	538 1	76 2.67	128	0.568570	0.296563
	129 0.406009	0.196330		0.254				0.592429	0.322129
	129 0.459531	0.258931		0.254				0.583661	0.256709
	129 0.483829	0.286778		0.254					
			1001	0.224	100 1	30 Z./I		0.644055	0.328132
	129 0.496051	0.303832		0.254				0.678713	0.351650
	122 0.522703	0.221489	1003	0.254	100 Z	09 2.71		0.705831	0.368116
855 0.202 94 87 2.71	122 0.545979	0.270658	1004	0.254	177	87 2.71	130	0.583360	0.241952
856 0.202 94 135 2.71	122 0.588904	0.334851	1005	0.254	176 1	37 2.71	130	0.639193	0.309310
857 0.202 93 173 2.71	122 0.605262	0.361484		0.254			130	0.671779	0.343313
	122 0.522927	0.220578		0.254		09 2.71		0.694471	0.362127
	122 0.595282	0.221489	1008	0.254	307	8/ 2 71		0.569361	0.219927
865 0.202 543 88 2.71	122 0.373602	0.235622	1000	0.254	347 4	75 7 74		0.631880	0.285361
		0.235397							
	121 0.520412			0.254				0.664227	0.319579
	121 0.557415	0.307826		0.254				0.687604	0.332620
	121 0.565349	0.325698		0.254		85 2.71	130	0.548868	0.237464
	121 0.512955	0.245480	1017	0.254			130	0.604691	0.308887
877 0.202 309 139 2.69	121 0.548759	0.299113	1018	0.254	98 1	75 2.71	130	0.637685	0.328771
878 D.202 306 175 2.69	121 0.556713	0.334865	1019	0.254	99 2	10 2.71	130	0.666477	0.345237
879 0.202 304 210 2.69	121 0.566888	0.349998	1020	0.254				0.544670	0.226771
	121 0.546060	0.265192		0.254				0.600645	0.289426
	121 0.578949	0.327076		0.254				0.629857	0.318509
	121 0.512103	0.255563	1023	0.254	172 2	00 2 71		0.654120	0.334760
	121 0.550649		1023	0.254	E00	07 E.FI			
		0.321573						0.526988	0.190877
885 0.202 178 173, 2.69		0.333949	1033	0.254	229 1	3/ 2.00		0.609286	0.285826
	121 0.509438	0.242733		0.254			128	0.640704	0.314939
	121 0.548014	0.294987		0.254			128	0.667568	0.336887
888 0.202 101 87 2.69	121 0.513951	0.255563	1038	0.254	303	87 2.66	128	0.524683	0.205652
898 0.202 299 69 2.62	129 0.572599	0.209251	1039	0.254	297 1	38 2.66		0.601445	0.294078
	129 0.605947	0.255725	1040	0.254	297 1	75 2.64		0.631080	0.320586
	129 0.639628	0.318791		0.254				0.653036	0.342965
	129 0.646453	0.351626		0.254				0.510331	
		0.231971							0.194569
	129 0.547154			0.254	96 1			0.583599	0.287562
	129 0.618054	0.301086		0.254				0.618183	0.319937
	129 0.641915	0.330026	1061	0.254	96 2	09 2.66	128	0.646688	0.335143
	129 0.5277 9 3	0.235859	1091	0.254	95	83 2.68	121	0.526842	0.232491
906 0.202 500 137 2.62	129 0.569124	0.295900	1092	0.254	90 1	38 2.68	121	0.613445	0.319225
	129 0.588211	0.330026		0.254	90 1			0.645004	0.350285
	129 0.537408	0.252700		0.254	94 2	11 2.68		0.664401	0.365931
	129 0.577325	0.314903		0.254	San			0.513445	
	129 0.589419	0.335205							0.203502
			1077	0.254	313	83 2.68		0.507713	0.200048
	118 0.602903	0.232776		0.254				0.598905	0.276430
	118 0.647662	0.294152		0.254				0.625577	0.305418
	118 0.662862	0.319172		0.254		85 2.68		0.542708	0.237783
	118 0.668074	0.335702		0.254			121	0.617079	0.302428
	118 0.583924	0.203974	1109	0.254	165 1	75 2.68		0.646445	0.337861
	118 0.624092	0.260625		0.254				0.664593	0.349821
	-		•						

TABLE A1.4: Summery of experimental conditions (F. Soucy), continued

TABLE A1.4: Summary of experimental conditions (F. Soucy), continued

Num Rad Vel Lo	ad Rin Cal		Non-Book State		
1111 0.254 280 20	8 2.68 121 0.665821	0.340619	Num Rad Vet Load	Bin Cal an	*p
1112 0.254 287 17	9 2.68 121 0.646387		1217 0.254 300 176	2.67 128 0.621695	0.291167
1113 0.254 287 13		0.279420	1236 0.254 96 84	2.39 115 0.571391	0.229932
1114 0.254 284 8		0.205572	1237 0.254 95 137	2.39 115 0.616862	0.298598
1115 0.254 93 8		0.233866	1238 0.254 96 175 1239 0.254 95 209	2.39 115 0.639116	0.329304
1116 0.254 93 13	7 2.68 121 0.615529	0.297136		2.39 115 0.645940	0.346954
1117 0.254 94 8		0.222138	1240 0.254 523 87 1241 0.254 500 139	2.39 115 0.555697	0.197534
1118 0.254 93 13		0.297832	1242 0.254 515 175	2.39 115 0.604606 2.39 115 0.625982	0.264990
1119 0.254 94 17		0.320832	1243 0.254 520 209	2.39 115 0.640874	0.294971
1120 0.254 94 21	1 2.68 121 0.665592	0.342002	1244 0.254 939 86	2.39 115 0.488595	0.310203 0.172389
1133 0.254 520 8	2 2.70 122 0.441306	0.173595	1245 0.254 900 138	2.39 115 0.542354	0.253385
1134 0.254 500 13	7 2.70 122 0.531948	0.254911		2.39 115 0.569594	0.269101
1136 0.254 95 13	2 2.71 122 0.618573	0.302408	1247 0.254 900 209	2.39 115 0.586214	0.296180
1137 0.254 95 17		0.342886	1256 0.254 167 85	2.38 114 0.589156	0.232824
1138 0.254 95 20		0.360395	1257 0.254 204 137	2.38 114 0.635158	0.302015
1143 0.254 520 8		0.198730	1258 0.254 160 176	2.38 114 0.659698	0.329692
1144 0.254 512 13		0.280127	1259 0.254 167 208	2.38 114 0.670996	0.347414
1145 0.254 520 17		0.306730	1268 0.254 292 87	2.36 113 D.561626	0.207256
1146 0.254 518 20		0.333563	1269 0.254 289 138	2.36 113 0.612348	0.274421
1151 0.254 280 8		0.224416	1270 0.254 290 176	2.36 113 0.638569	0.308739
1152 0.254 169 8 1153 0.254 166 13		0.227600	1271 0.254 290 209	2.36 113 0.653418	0.331045
1154 0.254 166 17		0.294910		2.36 113 0.645985	0.310945
1155 0.254 166 20		0.324239	1273 0.254 513 179	2.36 113 0.633210	0.300650
1160 0.254 167 8		0.341748 0.216009	1274 0.254 518 139 1279 0.254 288 84		0.265841
1161 0.254 166 13		0.295360	1280 0.254 290 137	2.45 118 0.612257	0.231546
1162 0.254 166 17	5 2.71 122 0.640592	0.322872	1281 0.254 280 176	2.45 118 0.669947 2.45 118 0.693668	0.295917
1163 0.254 165 21		0.337195	1282 0.254 292 210	2.43 110 0.093000	0.330343 0.356988
1164 0.254 296 21		0.316049		2.45 118 0.617212	0.238148
1165 0.254 296 17		0.293766		2.45 118 0.663836	0.298275
1166 0.254 290 13	9 2.71 122 0.610314	0.273529	1285 0.254 520 177	2.45 118 0.680778	0.307000
1167 0.254 293 8	6 2.71 122 0.537078	0.192356	1290 0.254 95 85	2.45 118 0.589739	0.241920
1172 0.254 93 8		0.207146	1291 0.254 95 136	2.45 118 0.641342	0.312659
1173 0.254 94 13		0.287762	1292 0.254 95 175	2.45 118 0.662502	0.331522
1174 0.254 92 17		0.310658	1293 0.254 95 209	2.45 118 0.678536	0.344255
1177 0.254 921 8		0.202433	1294 0.254 169 209	2.45 118 0.680078	0.345199
1178 0.254 939 13		0.265283	1295 0.254 167 179	2.45 118 0.665475	0.318554
1179 0.254 912 19	·	0.298546	1296 0.254 168 139	2.45 118 0.648007	0.287900
1180 0.254 900 21		0.307035	1297 0.254 167 85	2.45 118 0.592413	0.219992
1185 0.254 900 8		0.190731	1390 0.254 921 85	2.40 115 0.515527	0.174741
1186 0.254 948 13		0.258859	1391 0.254 900 137	2.40 115 0.558038	0.239817
1187 0.254 900 17 1188 0.254 900 20	, , _ , _ , _ , _ , _ , , , , , ,	0.287538	1392 0.254 900 176	2.40 115 0.572891	0.264401
1189 0.254 285 8		0.299231	1393 0.254 900 210	2.40 115 0.581239	0.289467
1190 0.254 288 13		0.196702 0.273087	1394 0.254 900 179 1395 0.254 900 140	2.40 115 0.562187	0.272114
1191 0.254 280 17		0.273087		2.40 115 0.532023	0.236443
1192 0.254 285 20		0.329975	1396 0.254 900 87 1404 0.254 939 87	2.40 115 0.466175	0.174982
1201 0.254 500 13		0.292649	1405 0.254 900 137	2.40 115 0.573567 2.40 115 0.634432	0.184382 0.248012
1202 0.254 500 17	7 2.72 122 0.651851	0.326746		2.40 115 0.598622	0.183659
1203 0.254 500 21	0 2.72 122 0.674169	0.339241	1411 0.254 500 134	2.40 115 0.370022	0.163639
1204 0.254 958 21	0 2.72 122 0.680664	0.316060	1412 0.254 512 176	2.40 115 0.695885	0.289226
1205 0.254 930 17		0.292649	1413 0.254 509 210	2.40 115 0.711991	0.307785
1206 0.254 921 13	9 2.72 122 0.617590	0.254010	1414 0.254 500 178	2.40 115 0.697608	0.284647
1207 0.254 939 8	3 2.72 122 0.540347	0.186043	1415 0.254 500 140	2.40 115 0.674074	0.260063
1208 0.254 529 8	5 2.72 122 0.522746	0.196043	1416 0.254 939 136	2.40 115 0.649671	0.218848
1209 0.254 500 13		0.269917	1417 0.254 948 175	2.40 115 0.676997	0.249217
1210 0.254 500 17		0.302648	1418 0.254 939 208	2.40 115 0.696433	0.265124
1211 0.254 515 21		0.310831	1419 0.254 900 177	2.40 115 0.670445	0.235478
1212 0.254 300 17		0.291167	1420 0.254 930 139	2.40 115 0.641709	0.204387
1213 0.254 300 17		0.291167	1421 0.254 939 88	2.40 115 0.571575	0.137383
1214 0.254 300 17		0.291167	1426 0.254 93 86	2.40 115 0.585942	0.193059
1215 0.254 300 17		0.291167	1427 0.254 93 136	2.40 115 0.641293	0.255001
1216 0.254 300 17	6 2.67 128 0.625055	0.291167	1428 0.254 93 175	2.40 115 0.673869	0.298385

TABLE A1.4: Summary of experimental conditions (F. Soucy), continued TABLE A1.4: Summary of experimental conditions (F. Soucy), continued

Non Bod Val Load F	eta cal	_	W	بدو	Na.1	1	n:-	E-1	_	_
Num Rad Vel Load E 1429 0.254 93 209 2.		0.321282		Rad					8 _D	\$ _p
				0.355					0.374781	0.114990
	.40 115 0.675511	0.294047 0.183659		0.355					0.586077	0.264203
_ -	.40 115 0.592737			0.355					0.600237	0.296085
	.40 115 0.653490	0.247289		0.355		133			0.634001	0.337532
	.40 115 0.682284	0.276211		0.355		179	2.64			0.376002
	.40 115 0.702277	0.306339		0.355		210	2.64		0.654769	0.382872
	.40 115 0.609147	0.161726	_							
	.40 115 0.670991 .40 115 0.695293	0.243673 0.276211		0.355		136 96			0.611470	0.338906 0.304328
		0.291396		0.355	90				0.431259	0.304328
	.40 115 0.709458 .40 115 0.695749	0.279103		0.355		95			0.561206	0.13/374
	40 115 0.670816	0.239576		0.355		97			0.537569	0.300430
	.40 115 0.612584	0.164859		0.355		133			0.558911	0.339370
		0.192804		0.355		179			0.571860	0.368123
	.59 124 0.544752	0.220060		0.355		210			0.584333	0.385103
		0.250445		0.355					0.518945	0.313335
		0.311660		0.355	88	23			0.465220	0.313333
		0.331991	-	0.355					0.512181	0.149875
		0.343832		0.355		. –			0.530107	0.315372
		0.162866		0.355					0.556300	0.335748
		0.207549		0.355					0.565139	0.333746
		0.250574		0.355					0.559776	0.342087
		0.256699		0.355		96			0.496168	0.246547
		0.259324		0.355		99			0.561625	0.235739
-		0.234168		0.355					0.580376	0.275830
		0.188888	_	0.355					0.597563	0.295304
		0.191075		0.355					0.604124	0.310195
		0.255824		0.355					0.594542	0.302864
		0.242699		0.355					0.571043	0.269874
		0.209450		0.355					0.544337	0.243070
		0.179700		0.355		20			0.467069	0.094387
		0.236137		0.355		96	2 43	121	0.568740	0.074307
		0.236356		0.355					0.592173	
		0.209231		0.355					0.614280	
		0.196544		0.355					0.619485	0.329963
		0.154107	_	0.355		_			0.611922	0.313649
		0.192169		0.355					0.583016	0.268612
		0.224543		0.355		98			0.550875	0.251149
		0.248824		0.355					0.474533	0.106158
		0.298698		0.355		97			0.558804	0.238741
		0.326479		0.355		97			0.555189	0.225414
		0.335448		0.355		97			0.551735	0.213006
		0.232110		0.355		97			0.549402	0.274901
		0.257815		0.355					0.573854	0.307407
		0.016149		0.355		98			0.534426	0.323891
		0.042981		0.355		134			0.557009	0.351985
		D.047886		0.355		180			0.579720	0.378686
		0.111649		0.355		211			0.590434	0.393778
		0.142520		0.355		183			0.581011	0.388670
		0.157523		0.355					0.553723	0.355700
		0.144252		0.355					0.516488	0.299512
	01 96 0.530588	0.112803		0.355		23			0.468261	0.149756
		0.066640		0.355		97			0.517003	0.289296
		0.068371		0.355					0.540810	0.311818
		0.029421		0.355					0.564463	0.345020
	65 122 0.621444	0.282683	2128	0.355	173	209			0.576481	0.357093
		0.310062	2129	0.355	306	97			0.514459	0.266543
	.65 122	0.345426		0.355					0.536197	0.295798
	.65 122	0.352042		0.355					0.555444	0.316229
2008 0.355 303 182 2.		0.335159		0.355				_	0.567811	0.332250
2009 0.355 303 137 2.	65 122 0.658340	0.297513	2133	0.355	89	96			0.512871	0.282331
2010 0.355 303 99 2.	.65 122 0.604707	0.275838		0.355	89	96			0.510453	0.273973
		0.108145		0.355					0.501552	0.230323

								-	• •	•					
Num	Rad	Vel	Load	Bin	Cal	₽ _n	ž _p	Num	Rad	Vel	Load	Rin	Cal		
2137	0.355	294	100	2.76	123	••	0.193841		0.355					8 _n 0.599346	e _p
	0.355		99	2.76			0.193841		0.355	89	209	2.03	117	0.377346	0.312441
	0.355		99			0.641061			0.355					0.610612	0.318851
	0.355		98			0.612961	0.193841		0.355	89	9 7	4.03	117	0.551505	0.255698
	0.355		98	2 74	122	0.593902	0.193841			88	20	2.03	117	0.506713	0.106600
	0.355		98			0.580532			0.355		97			0.541181	0.304659
	0.355		97				0.193841		0.355		211			0.596074	0.371565
	0.355					0.576255	0.193841		0.355	89	98	2.64	116	0.533526	0.299403
			97			0.551357	0.193841		0.355	89	19			0.429163	0.155556
	0.355		97	2.10	123	0.539003	0.193841		0.355		97	2.64	116	0.520063	0.253524
	0.355		96			0.529394	0.193841		0.355		97			0.525649	0.247312
	0.355		97	2.17		0.474783	0.014973	2303	0.355	521	210	2.64	116	0.586296	0.305854
	0.355		97	2.17		0.481610	0.016124		0.355		97			0.501278	0.283393
	0.355			2.17	96	0.501241	0.045206	2305	0.355	522	211	2.64	116	0.565889	0.343369
	0.355			2.17	96	0.525286	0.064498	2306	0.355	513	18			0.363834	0.124253
2163	0.355	301	209	2.17	96	0.536999	0.087532	2307	0.355	520	22	2.64	116	0.389270	0.127599
	0.355			2.17	96	0.547087	0.089548	2308	0.355	89	96	2.64		0.0070.0	0.296057
2165	0.355	301	180	2.17	96	0.533075	0.073999		0.355	90	96	2.64			0.299881
2166	0.355	306	135	2.17		0.495661	0.038871		0.355		133	2.64			0.323297
	0.355		98	2.17		0.477179	0.016124		0.355		179	2.64			0.329032
	0.355			2.17		0.503981	0.041463		0.355		210	_			
	0.355			2.17		0.558898	0.060466		0.355		210	2.64			0.345520
	0.355			2.17		0.635660	0.096458					2.64			0.338590
2171	0.355	510	208	2.17		0.639587	0.105672		0.355		180	2.64			0.328315
2172	0.355	525	172	2.17		0.616627	_		0.355		134	2.64			0.302031
	0.355						0.080910	2316	0.355	89	95	2.64			0.265233
				2.17		0.592665	0.066513		0.355		25	2.154	_		0.136679
	0.355		96	2.17		0.561182	0.031385		0.355		22	2.64			0.117085
21/2	0.333	21/	129	2.17		0.578472	0.061330		0.355		22	2.64			0.122103
21/0	0.355	210	1/3	2.17		0.595387	0.088684		0.355	89	25	2.64	116		0.122820
	0.355			2.17		0.609772	0.109128		0.355	90	94	2.64	116		0.245639
	0.355	90	94	2.73	123	0.547055	0.278388	2322	0.355	177	94	2.64	116		0.223417
_	0.355		130			0.584141	0.310901	2324	0.355	174	174	2.64	116		0.299164
	0.355	89	175	2.73	123	0.617271	0.329867	2325	0.355	174	208	2.64			0.301075
2232	0.355	88	209	2.73	123		0.344995		0.355			2.64			0.304421
2233	0.355	90	178	2.73	123	0.612676	0.326254		0.355		94	2.64			0.219833
2234	0.355	92	133	2.73	123	0.575895	0.298934		0.355		23	2.64			0.098447
2235	0.355	92	95			0.538586	0.270937		0.355		23	2.64			0.094385
2236	0.355	95	22			0.404307	0.133659		0.355		94	2.64			0.213859
	0.355	86	94			0.533366	0.280194		0.355	89	93	2.64			
_==:	0.355		94			0.512871	0.222844		0.355		210	2.64			0.255436
		90	92			0.631007	0.279955		0.355						0.329988
	0.355		91			0.621968	0.234202			89	22	2.64			0.117085
	0.355			2 77	177	0.656164			0.355	-	22	2.64			0.114695
2247	0.355	E10	171	2.13	122	0.030104	0.260702		0.355		210	2.64			0.322581
22/0	0.355	217 E40	207	2.73			0.289015		0.355		210	2.64			0.318041
				2.73			0.307135		0.355	89	97			0.551538	0.298914
	0.355			2.73			0.298528		0.355	90	97	2.73	120	0.551351	
2200	0.355	527	90			0.601242			0.355	296	97	2.73	120	0.551157	
	0.355					0.637901		2346	0.355	89	96	2.67	120	0.530217	
	0.355		98			0.588782	D.2406B2	2347	0.355	89	96	2.67	120	0.609063	0.313698
2264	0.355	507	211	2.73	123		0.316094	2348	0.355	89	97	2.67	120	0.485690	0.315546
2265	0.355	526	101	2.73	123	0.538581	0.244068		0.355		97	2.67	120	0.480320	0.0.0040
2266	0.355	541				0.467233			0.355		96	2.67	120	0.473453	0.286902
2267	0.355	505				0.542753	0.242262		0.355		95			0.468698	0.264264
	0.355			2.73	123	0.599456	0.298709		0.355					0.470426	
	0.355			2.73		-,5,,,,,,	0.279743		0.355						0.298683
	0.355			2.73			0.129595				95			0.465275	0.242550
	0.355			2.73			0.125335		0.355		96			0.463740	0.252021
	0.355		16					4336	0.355	90	96			0.512365	0.289902
			99	2.73		0 500345	0.259873	2357	0.355	175	96			0.498379	0.267003
	0.355					0.581312		2358	0.355	297	96			0.489984	0.254179
	0.355					0.597676		2359	0.355	518	96			0.485866	0.247309
	0.355					0.585496		2360	0.355	926	96	2.70	121	0.470014	0.225784
	0.355		98			0.586365		2361	0.355	918	126	2.70	121	0.497834	0.264026
	0.355					0.557967		2362	0.355	915	164	2.70	121	0.521782	0.274330
2287	0.355	90	134	2.63	117	0.579609	0.288699	2363	0.355	916	198	2.70	121		0.291962
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	Rad					ŧ _n	*p			Rad						* p
						0.525715	0.266545			0.355		94			0.512191	0.216488
	0.355	89				0.566008	0.338182			0.355		94			0.507368	
	0.355	89				0.562777	0.313864			0.355		28			0.384178	0.105228
	0.355					0.600273	0.343864			0.355		28			0.384605	0.102100
	0.355		97	2.78	122	0.589295	0.313409			0.355		92			0.509260	0.211573
	0.355			2.78	122	0.616129	0.329773	244	4	0.355	923	92	2.76	124	0.508374	0.205764
	0.355					0.633215	0.348864	244	7	0.355	90	98	2.79	126	0.562613	0.296444
	0.355					0.644544	0.352955	244	8	0.355	90	22	2.79	126	0.412962	0.142699
	0.355			2.78	122	0.642865	0.348409	244	9	0.355	89	19			0.393838	0.138723
	0.355					0.619694	0.329318			0.355	90	20			0.395493	0.150210
	0.355			2.78	122	0.589109	0.299773	245	1	0.355	90	23	2.79	126	0.420627	0.159267
	0.355		97			0.565517	0.277500	245	2	0.355	176	23			0.409976	0.138502
	0.355					0.425684	0.142955			0.355		20			0.389774	0.127457
	0.355					0.401973	0.127273			0.355		20	2.79	126	0.393925	0.130108
	0.355					0.400869	0.123864	245	5	0.355	175	23	2.79	126	0.414624	0.142258
	0.355		23			0.423879	0.150227			0.355	176	96			0.556605	0.253590
	0.355			2.78	122	0.553740	0.266818			0.355	90	97	2.79	126	0.554757	0.272366
	0.355					0.631912	0.333636			0.355					0.558953	0.293290
2388	0.355	173				0.555355	0.247955	245	9	0.355	175	23			0.385525	0.103238
2389	0.355	173	21	2.78	122	0.401214	0.105455	246	0	0.355	174	20	2.63	118	0.363413	0.101361
	0.355		95			0.553609	0.237273	246	1	0.355	175	20	2.63	118	0.361386	0.094322
2392	0.355	301	97	2.71	119	0.540321	0.232851	246	2	0.355	175	23	2.63	118	0.385024	0.095260
	0.355					0.538150	0.234484	246	3	0.355	301	133	2.63	118	0.575137	0.293290
2394	0.355	300	134	2.71	119	0.571614	0.256416	246	4	0.355	298	22	2.63	118	0.376444	0.106757
	0.355	301	180	2.71	119	0.604609	0.281148			0.355		20	2.63	118	0.349605	0.093853
2396		89	96	2.71	119	0.538641	0.268782	246	6	0.355	300	20	2.63	118	0.356387	0.095260
2397	0.355	301	95	2.71	119	0.532725	0.231218	246	7	0.355	300	22	2.63	118	0.378824	0.100422
	0.355					0.539639	0.267615	246	8	0.355	300	133	2.63	118	0.570959	0.276161
	0.355					0.532885	0.241717	246	9	0.355	90	131	2.63	118	0.570718	0.301736
2400	0.355	301	131	2.71	119	0.567763	0.270649	247	0	0.355	89	175	2.63	118	0.597294	0.325434
2401	0.355	89	94	2.71	119	0.529975	0.245684	247	3	0.355	527	127	2.63	118	0.542236	0,250117
	0.355			2.71	119	0.564977	0.252217	247	4	0.355	519	28	2.63	118	0.384827	0.088691
2403	0.355	297	176	2.71	119	0.596766	0.283248	247	3	0.355	524	25	2.63	118	0.372732	0.084702
	0.355			2.71	119	0.616061	0.310079	247	6	0.355	519	25	2.63	118	0.369227	0.091976
2405	0.355	300	209	2.71	119	0.612218	0.305413	247	7	0.355	520	27	2.63	118	0.383194	0.102065
	0.355			2.71	119	0.599203	0.297014	247	8	0.355	521	91	2.63	118	0.507345	0.207180
	0.355		133	2.71	119	0.563569	0.255716	247	9	0.355	521	27	2.63	118	0.386992	0.084702
	0.355		95	2.71	119	0.526036	0.217919	248	0	0.355	520	25	2.63	118	0.371037	0.080009
2409	0.355	298	26	2.71	119	0.390103	0.097060	248	1	0.355	520	25	2.63	118	0.371042	0.080948
	0.355		23	2.71	119	0.366795	0.084928			0.355		27	2.63	118	0.381516	0.084233
2411	0.355	299	23	2.71	119	0.366941	0.085628	248	3	0.355	521	91	2.63	118	0.508743	0.198498
	0.355		26	2.71	119	0.391990	0.091227	248	5	0.355	90	101	2.81	126	0.555232	0.225594
2413	0.355	303	94	2.71	119	0.526752	0.209986	248	6	0.355	90	18	2.81	126	0.380533	0.103342
2419	0.355	90	96	2.84	125	0.545353	0.323699	248	7	0.355	90	16	2.81	126	0.358827	
2420	0.355	174				0.542257	0.289684	248	8	0.355	89	16			0.359838	0.088391
2421	0.355	176	132	2.84	125	0.575806	0.325033			0.355	90	19			0.383580	0.098065
2422				2.84	125	0.601215	0.344153			0.355	90	98			0.551464	0.227573
2423						0.616869	0.358604	249	1	0.355	90	97			0.552529	0.266150
2424	0.355	175	210	2.84	125	0.617926	0.353713	249	2	0.355	89	23	2.79	126	0.426889	0.142257
2425	0.355	175	178	2.84	125	0.598162	0.333482	249	3	0.355	90	50			0.403859	
2426	0.355	174	134	2.84	125	0.568847	0.303246	249	4	0.355	90	20	2.79	126	0.405175	0.128540
2427	0.355	175	96	2.84	125	0.541224	0.255892	249	5	0.355	90	24			0.425200	0.131858
2428			25	2.84	125	0.417548	0.141618			0.355		95			0.553242	0.242920
2429 (0.355	174		2.84	125	0.397186	0.120053			0.355					0.604033	
2430						0.535020	0.264969			0.355					0.599415	0.286726
2431 (0.542516	0.286193			0.355					0.602531	0.323230
2432						0.561773	0.296693			0.355					0.595270	0.309735
2433						0.548031	0.257819	250	Ĭ.	0.355	300	127			0.596010	0.293805
2434						0.580001	0.279714	250	3	0.355	89	128			0.606981	0.314381
2435						0.597855	0.295353	250	14	0.355	175	127			0.601122	0.292035
2436			= - :			0.597855	0.292449			0.355					0.603418	0.279204
2437						0.580480	0.278374			0.355					0.632177	0.298451
2438						0.543327	0.248883			0.355					0.652136	0.328761
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TABLE A1.4: Summary of experimental conditions (F. Soucy), continued

TABLE A1.4: Summary of experimental conditions (F. Soucy), continued

Num Rad Vel Load	l Bin Cal		Non-Barrelland		
2508 0.355 303 206	2.79 126 0.647633	e, 0.333850	Num Rad Vel Load		* _D
2509 0.355 304 173	2.79 126 0.628747	0.303319	2614 0.355 522 208 2615 0.355 523 181	2.12 96 0.511418	
2510 0.355 303 129	2.79 126 0.597016	0.280973	2616 0.355 524 135	2.12 96 0.492280	0.160268
2511 0.355 92 125	2.79 126 0.598219	0.292920	2617 0.355 519 135	2.12 96 0.450217 2.12 96 0.450974	0.143106
2512 0.355 93 129	2.66 120 0.585237	016,5,50	2618 0.355 514 208	2.12 96 0.512806	0.133799
2513 0.355 536 93	2.66 120 0.537991	0.194799	2619 0.355 521 208	2.12 96 0.506935	0.165212 0.155323
2514 0.355 526 29	2.66 120 0.418857	0.086139	2622 0.355 91 98	2.71 122 0.504045	0. 133323
2515 0.355 522 26	2.66 120 0.396315	0.065707	2623 0.355 91 98	2.71 122 0.497271	
2516 0.355 524 26	2.66 120 0.395005	0.078245	2624 0.355 928 132	2.71 122 0.508639	0.287927
2517 0.355 523 29	2.66 120 D.416125	0.098444	2625 0.355 920 177	2.71 122 0.545805	0.310706
2518 0.355 526 26	2.66 120 0.395779	0.085674	2626 0.355 918 209	2.71 122 0.565571	0.334169
2519 0.355 522 29	2.66 120 0.412649	0.095194	2627 0.355 923 209	2.71 122 0.559973	0.329157
2520 0.355 516 91	2.66 120 0.534281	0.184583	2628 0.355 927 181	2.71 122 0.546367	0.322551
2522 0.355 92 136	2.60 117 0.547506	0.328199	2629 0.355 925 136	2.71 122 0.509285	0.286788
2523 0.355 92 26	2.60 117 0.381205	0.140995	2630 0.355 926 98	2.71 122 0.478187	0.263326
2524 0.355 90 17 2525 0.355 91 25	2.60 117 0.315467		2631 0.355 924 26	2.71 122 0.335076	0.133713
2525 0.355 91 25 2526 0.355 177 98	2.60 117 0.379246	0.140521	2632 0.355 922 13	2.71 122 0.231659	0.094305
2527 0.355 175 25	2.60 117 0.509253	0.267299	2633 0.355 920 13	2.71 122 0.224900	0.080410
2528 0.355 176 18	2.60 117 0.367641 2.60 117 0.305883	0.122275	2634 0.355 921 25	2.71 122 0.326790	0.121185
2529 0.355 177 25	2.60 117 0.366055	0.088389 0.114929	2635 0.355 916 96	2.71 122 0.472584	0.234396
2530 0.355 529 134	2.69 121 0.557153	0.316648	2636 0.355 922 132	2.71 122 0.510497	0.269021
2531 0.355 534 24	2.69 121 0.367712		2637 0.355 918 176 2638 0.355 921 208	2.71 122 0.539755	0.293166
2532 0.355 521 19	2.69 121 0.324311	0.111825	2639 0.355 922 208	2.71 122 0.562689	0.312073
2533 0.355 525 19	2.69 121 0.322795	0.108840	2640 0.355 926 180	2.71 122 0.560902 2.71 122 0.547778	0.310251
2534 0.355 523 24	2.69 121 0.361971	0.123077	2641 0.355 926 135	2.71 122 0.510636	0.294533
2535 0.355 514 97	2.69 121 0.519659	0.247072	2642 0.355 927 97	2.71 122 0.476187	0.264692 0.224146
2536 0.355 531 133	2.69 121 0.558175	0.292767	2643 0.355 927 27	2.71 122 0.341094	0.113895
2537 0.355 523 133	2.69 121 0.555860	0.284271	2644 0.355 921 27	2.71 122 0.345664	0.116629
2574 0.355 302 94	2.73 123 0.570715	0.216486	2645 0.355 914 95	2.71 122 0.471335	0.209339
2575 0.355 302 94	2.73 123 0.570227		2647 0.355 92 98	2.73 123 0.565264	0.256793
2576 0.355 302 93	2.73 123 0.563916	0.216486	2648 0.355 91 135	2.73 123 0.596192	0.283062
2577 0.355 301 93	2.73 123 0.561766	0.216486	2649 0.355 91 181	2.73 123 0.629765	0.307745
2578 0.355 301 93	2.73 123 0.558052	0.216486	2650 0.355 92 209	2.73 123 0.649963	0.330163
2579 0.355 301 92	2.73 123 0.554893	0.216486	2651 0.355 91 11	2.73 123 0.301786	0.067482
2580 0.355 301 92	2.73 123 0.551323	0.216486	2653 0.355 91 98	2.53 114 0.536672	0.268620
2581 0.355 301 91	2.73 123 0.550974	0.216486	2654 0.355 91 11	2.53 114 0.271546	0.074237
2582 0.355 301 91 2583 0.355 301 91	2.73 123 0.548583	0.216486	2655 0.355 91 11	2.53 114 0.279820	0.073993
2583 0.355 301 91 2586 0.355 91 97	2.73 123 0.548332 2.14 96 0.462947	0.216486	2656 0.355 92 24	2.53 114 0.390901	0.125763
2587 0.355 303 97		0.084104	2657 0.355 91 24	2.53 114 0.397934	0.147253
2588 0.355 301 137	2.14 96 0.453850 2.08 94 0.497812	0.071098 0.102047	2658 0.355 303 25	2.53 114 0.385370	0.100611
2589 0.355 303 185	2.08 94 0.532756	0.137051	2659 0.355 302 12	2.53 114 0.279421	0.076190
2590 0.355 302 211	2.08 94 0.549263	0.135865	2660 0.355 301 13 2661 0.355 301 25	2.53 114 0.270914	0.062515
2591 0.355 303 210	2.08 94 0.546534	0.132898	2661 0.355 301 25 2663 0.355 92 96	2.53 114 0.382488	0.087912
2592 0.355 300 135	2.08 94 0.460261	0.132878	2664 0.355 174 95	2.53 114 0.533737 2.53 114 0.527848	0.276190
2593 0.355 304 97	2.08 94 0.420572	0.082171	2665 0.355 927 94	2.53 114 0.508046	0.250305 0.218803
2594 0.355 302 97	2.08 94 0.422185	0.084545	2666 0.355 922 28	2.53 114 0.388264	
2599 0.355 297 135	2.12 96 0.460578	0.122455		2.53 114 0.292477	0.072039
2600 0.355 296 98	2.12 96 0.420144	0.091041	2668 0.355 921 16	2.53 114 0.207111	0.076679
2601 0.355 304 99	2.12 96 0.423679	0.093950	2669 0.355 919 28	2.53 114 0.391806	0.097680
2602 0.355 301 134	2.12 96 0.458333	0.109075	2670 0.355 923 28	2.53 114 0.396568	3107.000
2603 0.355 300 181	2.12 96 0.497436	0.140780	2673 0.202 176 95	2.67 119 0.622866	0.308538
2604 0.355 302 135	2.12 96 0.458376		2674 0.202 175 140	2.67 119 0.635665	
2605 0.355 302 181	2.12 96 0.496403		2675 0.202 174 178	2.67 119 0.639193	0.372865
2606 0.355 313 135	2.12 96 0.461745	0.123037	2676 0.202 171 178	2.67 119 0.638790	0.376842
2607 0.355 303 210		0.156486	2677 0.202 177 144	2.67 119 0.634163	0.356491
		0.140780	2678 0.202 178 98	2.67 119 0.622207	0.311579
2609 0.355 302 185	2.12 96 0.501147		2679 0.202 174 82	2.67 119 0.616804	0.283275
	2.12 96 0.461520	0.160849	2680 0.202 179 73		0.259415
2611 0.355 530 131	2.12 96 0.452165		2681 0.202 174 73	2.67 119 0.606028	0.250760
2612 0.355 516 177 2613 0.355 522 209	2 12 04 0 EADED	V.148924	2682 0.202 177 140	2.54 113	0.344311
01333 366 EUY	4.16 70 U.JUY6JU	0.171902	2683 0.202 176 178	2.34 113	0.370853

TABLE A1.4: Summary of experimental conditions (F. Soucy), continued

TABLE A1.4: Summary of experimental conditions (F. Soucy), continued

		,	14 11 112					****	,,	,		Lu			
No.	Rad	V-1	Lond	Pi-	Cal			Maren	Rad	V-I	Lond	Die	Cal		
						٤n	0.349963		0.202					8 ₀	0.319668
_	0.202			2.54						96	77			0.604513	
	0.202		98	2.54			0.307201	-	0.202	96	91			0.617498	0.337204
_	0.202		82	2.54			0.293930		0.202		90			0.605881	0.290758
	0.202		74	2.54			0.259845		0.202					0.634597	0.326066
	0.202					0.662029	0.319980		0.202		66			0.571502	0.238187
	0.202					0.704563	0.333743		0.202		74	_		0.586937	0.248981
	0.202					0.701325	0.337921		0.202		89			0.600785	0.277285
	0.202					0.665303	0.312362		0.202		89			0.605285	0.266011
	0.202		95			0.607628	0.278693	_	0.202		76			0.590524	0.251379
	0.202		80			0.577915	0.244532		0.202		68			0.577779	0.236508
2694	0.202	319	71			0.560784	0.252642		0.202		64			0.511876	0.234828
	0.202		71	2.54	113	0.551750	0.232735		0.202		74			0.518519	0.253538
2696	0.202	552	77	2.54	113	0.566507	0.233473	2769	0.202	868	88	2.60	116	0.539478	0.270808
2697	0.202	552	91	2.54	113	0.589772	0.246498	2770	0.202	871	88	2.60	116	0.539214	0.272487
2700	0.202	96	137	2.71	121	0.646607	0.376295	2771	0.202	878	75	2.60	116	0.527891	0.255217
2701	0.202	96	176	2.71	121	0.673189	0.397468	2772	0.202	876	65	2.60	116	0.512419	0.244903
2702	0.202	96	176	2.71	121	0.669974	0.396548	2773	0.202	870	74	2.60	116	0.523109	0.241545
2703	0.202	96	139	2.71	121	0.623301	0.371231	2774	0.202	867	88	2.60	116	0.539741	0.271768
2704	0.202	95	95	2.71	121	0.572018	0.334407	2775	0.202	862	134			0.564087	0.310386
2705	0.202	95	80			0.546100	0.32 43	2776	0.202	861	176	2.60	116	0.573560	0.325498
2706	0.202	96	72			0.529092	0.307020	2777	0.202	870	176	2.60	116	0.572602	0.315423
	0.202	96	72			0.524091	0.303567		0.202	95	68			0.590433	0.296191
	0.202	95	78			0.534026	0.314154	2783	0.202	95	78			0.606478	0.314644
	0.202	95	91			0.552605	0.326122		0.202	95	92			0.621809	0.336645
	0.202	186	91			0.537360	0.310242	2785	0.202	95	137			0.653806	0.373788
	0.202			2.71	121	0.591589	0.348216		0.202		177			0.680455	0.398155
	0.202					0.632235	0.379747		0.202		177			0.679784	0.397918
2713	0.202	187				0.631459	0.385040		0.202		140			0.652459	0.365507
	0.202					0.595461	0.359033	2789	0.202	95	95			0.621183	0.330968
	0.202		93			0.540886	0.313003		0.202	95	79			0.607773	0.305891
	0.202		79			0.519503	0.293671		0.202	95	68			0.589639	0.290750
	0.202					0.503927	0.284235		0.202		68			0.578472	0.246984
	0.202		70			0.500698	0.277330		0.202		77			0.586547	0.248876
_	0.202					0.513249	0.288147		0.202		91			0.605707	0.277265
	0.202		90			0.521460	0.290909		0.202					0.637614	0.312042
2721	0.202	320	173			0.607299	0.338780		0.202		67			0.586890	0.288349
	0.202		69		_	0.479619	0.238205		0.202		77			0.601103	0.304226
	0.202		69			0.481496	0.249022		0.202		92			0.619091	0.318702
	0.202		172			0.607962	0.330495		0.202			= -		0.652432	0.353958
2725	0.202	95	93			0.497093	0.319678		0.202					0.684734	0.386411
2726	0.202	324	93	-		0.491981	0.301036		0.202					0.682870	0.388980
-	0.202		-	-		0.548752	0.342232	-	0.202					0.658348	0.353724
	0.202					0.588453	0.375604		0.202		95			0.624268	0.317067
	0.202			_	-	0.585459	0.350978		0.202		80			0.610004	0.296521
	0.202					0.548280	0.322900		0.202		68			0.595984	0.276442
-	0.202					0.488426	0.271577		0.202		30			0.496855	0.190754
	0.202		80			0.465116	0.258918		0.202		68			0.585328	0.265351
	0.202		72			0.448771	0.249482		0.202		67			0.569043	0.250119
	0.202		71			0.445759	0.242808		0.202		76			0.583245	0.261984
	0.202		78			0.458546	0.245109		0.202		91			0.605657	0.280256
	0.202					0.482763	0.255696		0.202					0.638655	0.314665
	0.202					0.545316	0.312773		0.202		67			0.571851	0.238728
	0.202	95	87			0.616958	0.308045		0.202		77			0.586348	0.249881
	0.202		87			0.613592	0.293066		0.202					0.602288	0.260085
	0.202		87			0.613484	0.283402		0.202					0.635720	0.294020
	0.202		138			0.649899	0.386730		0.202					0.574449	0.240437
	0.202		177			0.674007	0.413507		0.202					0.593724	0.254455
	0.202		177			0.673938	0.423697		0.202					0.612282	0.272036
	0.202					0.648891	0.387678		0.202					0.643496	0.272036
2751	0.202	96	95			0.619107	0.339573		0.202					0.663284	0.342837
2752	0.202	96				0.607856	0.319668	2828	0.202	317	175			0.662601	0.340223
	0.202	96				0.596901	0.309479		0.202					0.642623	0.305298
	0.202	95	70			0.599260	0.320379	DRIN	0.202	324	94			0.620543	
	4.545	,,					4.7EVJ(7	-050	O.LVE	264	74	2.00	11/	V.02U343	0.265859



Num I	Rad	Vel	Load	Rin	Cal		•	No en	Bad	Wal		n:-	0-1		
2831 0.			78	2.60	117	0.600662	0 263763		Rad 0.202						e _p
2834 0.			67	2.59	117	0.479315	0.212467		0.202					0.587515 0.584897	
2835 0	.202	875	76	2.59	117	0.496304	0.224364		0.202		42			0.503062	0.284667 0.163505
2836 D.	.202	867	90			0.510136	0.247680		0.202		17			0.346396	0.163303
2837 0.	.202	860	134	2.59	117	0.547277	0.288603		0.202	96	12			0.284430	0.047200
2838 0.	.202	858	175	2.59	117	0.562591	0.315965		0.202					0.287193	0.035230
2839 0.	.202	866	175	2.59	117	0.565044	0.321199		0.202	96	65			0.561742	0.25225B
2840 0.	.202	883	139	2.59	117	0.548871	0.293362	2920	0.202	96	65			0.560783	0.237805
2841 0.			95	2.59	117	0.521918	0.251487	2921	0.202	97				0.534151	0.217512
2842 0.			79	2.59	117	0.498262	0.213419	2922	0.202	564	53	2.51	123	0.517453	0.169603
2843 0.			69	2.59	117	0.486469			0.202		65			0.540043	0.195122
2844 0. 2845 0.	.202	6/3	69	2.59	117	0.482652	0.188199		0.202		65			9.544083	0.195122
2846 0.	202	70 <i>)</i>		2.57			0.251441		0.202					0.519445	0.173442
2847 0.	202	5/2	90	2.57 2.57			D.266330	2926	0.202	559	44			0.496740	0.151536
2848 0.				2.57			0.311720 0.330932		0.202					0.329673	0.055104
2849 0.				2.55			0.330732		0.202		13			0.272479	
2850 0.			67	2.55			0.210794		0.202		9			0.194350	0.024842
2851 0.			67	2.55			0.197725		0.202					0.200546 0.487232	0.023261
2852 0.			75	2.55			0.222168		0.202		50			0.509300	0.137534 0.148600
2853 0.			89	2.55			0.239835		0.202		50			0.506271	0.149503
2854 0.				2.55	115		0.298403		0.202					0.490258	0.137760
2855 0.	.202	541	175	2.55	115		0.322362		0.202					0.329236	0.042683
2856 0.	.202	549	175	2.55	115		0.320426		0.202		13			0.273196	0.030714
2857 0.				2.55			0.285818		0.202		9			0.196122	0.014228
2858 0.			94	2.55					0.202		9	2.51	123	0.198965	0.014228
2859 0.				2.55			0.198693		0.202		12	2.51	123	0.259186	0.018970
2860 0.				2.55			0.181994		0.202		42			0.518747	0.204059
2861 0.				2.55			0.188529		0.202		42			0.501800	0.144960
2862 0. 2863 0.			90	2.55		0 //4/00	0.213214		0.202		51	_		0.531398	0.163916
2864 0.			80 80			0.461489			0.202		65			0.556608	0.203390
2865 0.						0.457093 0.454665	0.253680		0.202		65			0.555275	0.198037
2866 0.			80	2.60	121	0.449016	0.235511 0.223551		0.202		53			0.531075	0.171053
2871.0.						0.566403	0.233616		0.202		44			0.510408	0.163470
2872 0.						0.577217			0.202		18 13			0.343136	0.060437
2873 0.			94			0.596029		2956	0.202	555	9			0.209629	9.041704 9.019402
2874 0.	.202	323	69			0.566379	0.238842		0.202		43			0.528014	0.193548
2875 0.	.202	322	79			0.581571	0.247986		0.202		65			0.577442	0.244987
2876 0.			93	2.60	128	0.600174	0.276072	2962	0.202		65			0.580651	0.253705
2877 0.				2.60	128	0.636218	0.315698	2963	0.202	96	45			0.537054	0.207280
2878 0.						0.653032	0.350316	2964	0.202	96	16			0.360687	0.082607
2879 0.	.202	320	177		_	0.654733	0.349445		0.202		7			0.173813	0.048169
2880 0.	.202	325					0.329632		0.202		43			0.533827	0.200305
2881 0.						0.599873	0.279991		0.202		43			0.513577	0.158675
2882 0. 2883 0.							0.256042		0.202		65			0.569251	0.207280
2884 O.			71 68			0.560581	0.230351		0.202		65			0.569528	0.209024
2885 0.			79	2.03	120	0.570069 0.591843	0.263101		0.202			2.60	127	0.517495	0.162816
2886 0.						0.605833		2972			16			0.343291	0.069529
2887 0.						0.638684	0.347854		0.202					0.170461	
2888 0.						0.574371	0.263964	2074	0.202	552	7 16			0.181406	0.048169
2889 0.			78	2.63	129	0.589539	0.283157		0.202		42			0.514922	0.059939 0.138622
2890 0.		95				0.591912	0.296528		0.202		65			0.583937	0.251526
2891 0.	.202	96	69	2.63	129	0.575300	0.280354		0.202		65			0.568729	0.216652
2892 0.		95				0.574054	0.273237		0.202		43			0.510399	0.155841
2893 0.		95	77			0.588964	0.289196		0.202		17			0.353180	0.071927
2894 0.		94	91			0.606579	0.316800		0.202		9			0.226308	0.031168
2895 0.		93				0.640651	0.355834	2981	0.202	555	9			0.226940	0.040759
2896 0.		90				0.662454	0.386888		0.202		17			0.353851	0.072581
2897 0.	_					0.661213	0.390554		0.202		41			0.505744	0.143636
2898 0.			139			0.644249	0.357343		0.202		65			0.564273	0.191805
2899 0.	. 202	95	94	2.63	129	0.605201	0.308820	2985	0.202	96	40	2.55	125	0.520198	0.188444

									_			
Num Rad Vel			F1	z p		Rad				Cal	₽ _n _	E _p
2986 0.202 964	41		25 0.500652	0.142444		0.202	95	18			0.361594	0.084543
2987 0.202 967	18		25 0.346193	0.061111		0.202	96	65			0.576106	0.219091
2988 0.202 96	40		4 0.523060	0.176629		0.202		65			0.566159	0.193924
2989 0.202 96	65		24 0.581478	0.247817		0.202		40			0.522844	0.087993
2990 0.202 98	19		4 0.379494	0.078128		0.202		65			0.598552	0.156166
2991 0.202 96	9		4 0.241254	0.031789		0.202		65			0.599675	0.153959
2992 0.202 187	40	-	4 0.516499	0.158943	_	0.202		41			0.533046	0.097802
2993 0.202 187	65		4 0.574193	0.221849	_	0.202		27			0.410066	0.054642
2994 0.202 190 2995 0.202 188	19 9		24 D.380930	0.086411		0.202		19	2.55		0.281421	0.028892
2996 0.202 325	39		24 0.239656	0.039624		0.202		10			0.612048	0.000201 0.160581
2997 0.202 320	64		6 0.510849 6 0.579034	0.155604 0.232308		0.202		65 41			0.543239	0.108347
2998 0.202 328	19		6 0.383449	0.079560		0.202		20			0.289515	0.016876
2999 0.202 322	10		6 0.261457	0.077368	_	0.202		28			0.425329	0.061508
3000 0.202 319	19		6 0.390867	0.092527		0.202		19			0.284936	0.012462
3001 0.202 541	65		6 0.575390	0.213626		0.202		10	2.55		0.204730	0.005841
3002 0.202 543	41		6 0.516334	0.150989		0.202		65			0.625616	0.159109
3003 0.202 545	19		6 0.381283	0.070110	_	0.202	-	65			0.623241	
3004 0.202 538	10		6 0.256860	0.033187		0.202		41			0.572054	
3005 0.202 535	10		6 0.258591	0.039560		0.202		41			0.579590	
3006 0.202 531	19		6 0.385853	0.063077		0.202		66			0.636406	0.152194
3007 0.202 528	39	2.58 12	6 0.510667	0.130769	3083	0.202	321	65			0.638487	0.152194
3008 0.202 961	39	2.58 12	6 0.498288	0.132527	3084	0.202	321	65	2.57	118	0.636567	0.152194
3009 0.202 954	64	2.58 12	6 0.563467	0.190769	3085	0.202	321	65	2.57	118	0.635114	0.152194
3010 0.202 965	41	2.58 12	6 0.505228	0.136264		0.202		65	2.57	118	0.637127	0.152194
3011 0.202 969	20		6 0.370167	0.059560		0.202		65			0.637376	0.152194
3012 0.202 963	10		6 0.247499	0.034945		0.202		65			0.642703	0.152194
3013 0.202 961	10		6 0.253213	0.041978		0.202		65			0.635960	0.152194
3014 0.202 958	19		6 0.367332	D.050330	_	0.202		65			0.641262	0.158625
3023 0.202 97	41		7 0.522343	0.156137		0.202		65			0.634000	0.158625
3024 0.202 96	66		7 0.578038	0.225765		0.202		66			0.627162	0.158625
3025 0.202 100 3026 0.202 97	43		7 0.526328	0.164676		0.202		65			0.634061	0.158625
3026 0.202 97 3027 0.202 97	29 18		7 0.458456 7 0.354428	0.125702 0.078626		0.202		65 65			0.627657	0.158625
3028 0.202 97	8		7 0.334428	0.039870	_	0.202		65	2.37	110	0.629650 0.626481	0.158625
3029 0.202 96			7 0.204014	0.037870		0.202	96	41	2.16		0.411972	0.158625
3030 0.202 96	17		7 0.359453	0.082348		0.202	97	29	2.16		0.328274	0.000000
3031 0.202 96	27		7 0.448479	0.118038			96	18	2.16		0.195950	0.000000
3032 0.202 187			7 0.443593	0.108185		0.202		18	2.16		0.205550	0.000000
3033 0.202 188	28		7 0.445318	D.110813		0.202		28	2.16		0.313059	0.000000
3034 0.202 189			7 0.349528	0.070743		0.202		41	2.16		0.408202	0.000000
3035 0.202 184	41		7 0.518977	0.151758		0.202		41	2.16		0.405575	0.000000
3036 0.202 326	29		7 0.449423	0.099865		0.202		18	2.16		0.212347	0.000000
3037 0.202 321	29		7 0.446781	0.104244		0.202		27	2.16		0.314115	0.000000
3038 0.202 322	18	2.59 12	7 0.344293	0.064393	3110	0.202	534	18	2.16		0.203441	0.000000
3039 0.202 317	41		7 0.514384	0.147816	3111	0.202	98	42	2.13		0.420530	0.000000
3040 0.202 547			7 0.440195	0.095267		0.202	96	29	2.13	98	0.328531	0.000000
3041 0.202 535	18		7 0.334889	0.049942		0.202	96	18	2.13	98	0.218781	0.000000
3042 0.202 524	41		7 0.501970	0.116068		0.202	95	28	2.13	98	0.321760	0.00000
3043 0.202 523	66		7 0.563030			0.202		42	2.13		0.410386	0.000000
3048 0.202 97			4 0.508575	0.130092		0.202		42	2.13		0.413767	0.000000
3049 0.202 327			4 0.507033	0.128874		0.202		66	2.13		0.464916	0.000000
3050 0.202 318			4 0.558610	0.196334		0.202		66	2.13		0.464599	0.000000
3051 0.202 320	66		4 0.554536	0.192194		0.202		19	2.13		0.201164	0.000000
3052 0.202 326			4 0.509533	0.135937		0.202		28	2.13		0.320529	0.000000
3053 0.202 324	29		4 0.438808	0.096241		0.202		42	2.13		0.409848	0.000000
3054 0.202 325 3055 0.202 322	18 66		4 0.330682 4 0.554991	0.064581		0.202		66 61	2.13		0.462965	0.000000
3056 0.202 96			5 0.569439	0.219091		0.202	95 95	41			0.437968	0.000000
3057 0.202 96			5 0.521611	0.168757		0.202	95	66 66			0.494399	0.000000
3058 0.202 96			5 0.453692	0.124714		0.202	96	42			0.491293	0.000000
3059 0.202 96			5 0.353222	0.083091		0.202	95	30			0.344990	0.000000
3060 0.202 96			5 0.137108	0.026949			95	19			0.228599	0.000000
	-	''		/-/	3.50			.,	10		4.450377	3.00000

Num Rad Vel Load	Dia Cal	_	Programme and the second
	Bin Cal En 2.18 100 0.233956	0.000000	Num Rad Vel Load Bin Cal sn . sp
	2.18 100 0.344209		3190 0.202 187 40 2.17 92 0.391275
3131 0.202 93 41	2.18 100 0.439213	0.000000	3197 0.202 187 65 2.17 92 0.461257
	2.15 98 0.436309	0.000000	3198 0.202 187 65 2.17 92 0.469558 3199 0.202 188 42 2.17 92 0.405748
3133 0.202 186 41	2.15 98 0.432681		3199 0.202 188 42 2.17 92 0.405748 3200 0.202 187 20 2.17 92 0.178933
3134 0.202 186 66	2.15 98 0.492220		3201 0.202 325 20 2.17 92 0.175618
	2.15 98 0.490328	0.000000	3202 0.202 319 40 2.17 92 0.389139
3136 0.202 187 42	2.15 98 0.442647	0.000000	3203 0.202 320 65 2.17 92 0.464888
3137 0.202 186 29	2.15 98 0.337891	0.000000	3204 0.202 321 41 2.23 94 0.450509 -0.006628
3138 0.202 186 18	2.15 98 0.206397	0.000000	3205 0.202 538 41 2.23 94 0.437478 -0.021034
3139 0.202 185 18	2.15 98 0.205866	0.000000	3206 0.202 533 41 2.23 94 0.440422 -0.003394
3140 0.202 185 16	2.15 98 0.134814	G.000000	3207 0.202 527 65 2.23 94 0.490530 0.011599
3141 0.202 186 27	2.15 98 0.323205	0.000000	3208 0.202 532 65 2.23 94 0.494033 0.022771
3142 0.202 544 41	2.15 98 0.422085	0.000000	3209 0.202 539 41 2.23 94 0.443958 -0.008980
3143 0.202 528 66	2.15 98 0.486125	0.000000	3210 0.202 536 20 2.23 94 0.267414 -0.019270
3144 0.202 533 66		0.000000	3211 0.202 533 20 2.23 94 0.266562 -0.018388
3145 0.202 534 30	2.15 98 0.359144	0.000000	3212 0.202 529 40 2.23 94 0.452631 0.014539
3146 0.202 533 30 3147 0.202 532 30	2.13 95 0.340437	0.000000	3213 0.202 96 39 2.20 93 0.476105 0.007019
3148 0.202 531 41	2.15 98 0.344577 2.15 98 0.430948	0.000000	3214 0.202 96 65 2.20 93 0.522470 0.045176
3149 0.202 534 19	2.15 98 0.204259	0.000000	3215 0.202 96 65 2.20 93 0.521359 0.051138
	2.15 98 0.210205	0.000000	3216 0.202 96 41 2.20 93 0.479721 0.020732
3151 0.202 332 41	2.15 99 0.434311	0.000000	3217 0.202 96 20 2.20 93 0.305539 -0.000136
3152 0.202 320 66	2.15 99 0.491548	0.000000	3218 0.202 96 15 2.20 93 0.218725 -0.006098 3219 0.202 96 14 2.20 93 0.199041 -0.005799
3153 0.202 322 66	2.15 99 0.492202	0.000000	3219 0.202 96 14 2.20 93 0.199041 -0.005799 3220 0.202 95 20 2.20 93 0.305896 -0.006396
3154 0.202 325 42	2.15 99 0.441172	0.000000	3221 0.202 187 20 2.20 93 0.299639 -0.012954
3155 0.202 323 30	2.15 99 0.372607	0.000000	3222 0.202 186 39 2.20 93 0.467665 0.016558
	2.15 99 0.246788		3223 0.202 186 65 2.20 93 0.516403 0.033848
	2.15 99 0.242044	0.000000	3224 0.202 187 65 2.20 93 0.511299 0.020136
3158 0.202 320 29	2.15 99 0.341485	0.000000	3225 0.202 188 41 2.20 93 0.461815 -0.004011
3159 0.202 322 20	2.15 99 0.234830	0.000000	3226 0.202 188 20 2.20 93 0.307312 0.001653
3160 0.202 319 41		0.000000	3227 0.202 186 14 2.20 93 0.219480 -0.000732
3161 0.202 97 41	2.15 99 0.444946	0.000000	3228 0.202 187 14 2.20 93 0.217681 -D.D05799
3162 0.202 98 29			3229 0.202 186 20 2.20 93 0.303488 0.002846
	2.15 99 0.251160		3230 0.202 322 20 2.20 93 0.304665 0.010894
	2.15 99 0.246408	0.000000	3231 0.202 319 39 2.20 93 0.456589 -0.001626
3165 0.202 187 28	2.15 99 0.338011	0.000000	3232 0.202 317 65 2.20 93 0.514245 0.029377
	2.15 99 0.341573	0.000000	3233 0.202 320 65 2.20 93 0.510232 0.029377
3167 0.202 187 20	2.15 99 0.244433	0.000000	3234 0.202 324 41 2.20 93 0.462081 0.003144
3168 0.202 185 41 3169 0.202 185 66	2.15 99 0.435070	0.000000	3235 0.202 321 20 2.20 93 0.301777 -0.005799
	2.15 99 0.491694 2.15 99 0.488846	0.000000	3236 0.202 320 15 2.20 93 0.199160 -0.01384B
	2.15 99 0.488846 2.15 99 0.435952	0.000000	3238 0.202 96 41 2.56 108 0.498585
3172 0.202 323 20	2 15 00 0 227000	0.000000	3239 0.202 95 65 2.56 108 0.555536 0.141959 3240 0.202 95 65 2.56 108 0.557403 0.148130
3173 0.202 319 20	2.15 99 0.233076	0.000000	
3174 0 202 314 41	2 15 00 0 /27421	0.000000	3241 0.202 96 43 2.56 108 0.502062 0.084619 3242 0.202 95 18 2.56 108 0.234615 0.011337
3179 0.202 321 40	2.78 118 0.508706 2.78 118 0.295643 2.78 118 0.583589 2.78 118 0.513482		3243 0.202 95 15 2.56 108 0.175301 -0.003062
3180 0.202 320 20	2.78 118 D.295643		3244 0.202 95 19 2.56 108 0.262478 0.018280
3181 0.202 322 65	2.78 118 0.583589		3245 0.202 96 41 2.56 108 0.495394 0.095675
3182 0.202 324 41	2.78 118 0.513482		3246 0.202 186 41 2.60 110 0.496347 0.077761
	2.78 118 0.530872		3247 0.202 188 65 2.60 110 0.557319 0.146910
3184 0.202 96 39	2.78 118 0.528489		3248 0.202 187 65 2.60 110 0.558265 0.150709
3185 0.202 321 39	2.78 118 0.515444		3249 0.202 187 43 2.60 110 0.503077 0.100811
	2.78 118 0.311095	0.070383	3250 0.202 187 18 2.60 110 0.251976 0.012918
	2.78 118 0.313010	0.084254	3251 0.202 187 15 2.60 110 0.191348 0.009119
	2.67 113 0.576928		3252 0.202 186 19 2.60 110 0.269652 0.014184
	2.67 113 0.574345	0.166533	3253 0.202 185 41 2.60 110 0.497699 0.084093
3190 0.202 320 65	2.67 113 0.563559	0.186581	
	2.67 113 0.486993	0.075114	
	2.47 105 0.474492	0.035381	
	2.47 105 0.538926	0.105588	
	2.47 105 0.481086 2.47 105 0.274308	0.046243	
3173 U.EUE 70 19	6.91 103 0.274308	0.001734	

```
TABLE A1.5: Command log input to SYSTAT (F. Soucy)
By
Select
Weight
Use 'c:\systat\data\SUM2REPL.SYS'
By
Select
Weight
Print Long
Format 4
Options
Montin
Model EN=(Aa+B*(ao+al*log(L)+ar*log(R)+av*log(V))),
  *(B<0.5*(1-Aa)/(ao+ai*log(L)+ar*log(R)+av*log(V))),
  +(1-0.25*((1-Aa)^2)/(8*(ao+al*log(L)+ar*log(R)+av*log(Y)))),
  *(B>=0.5*(1-Aa)/(ao+al*log(L)+ar*log(R)+av*log(V)))
Estimate /Iter=100 / Start = -.4,.1,0.1,-.02,-.02, Quasi, Print
Select
Weight
Use 'c:\systat\data\SUM2REPL.SYS'
By
Select
Weight
Print Long
Format 4
Options
Monlin
Model EP=0+(As+B*(ac+ai*log(L)+ar*log(R)+av*log(V)));
  *((B>=-Aa/(ao+al*log(L)+ar*log(R)+av*log(V))),
  AND (B<0.5^{\circ}(1-Aa)/(ao+al*log(L)+ar*log(R)+av*log(V))),
  +(1-0.25*((1-Aa)^2)/(8*(ao+al*log(L)+ar*log(R)+av*log(V)))),
  *(B>=0.5*(1-Aa)/(ao+al*log(L)+ar*log(R)+av*log(V)))
```

Estimate /Iter=100 / Start = -.4,.1,.1,-.02,-.02, Quasi, Print

TABLE A1.6: SYSTAT output (F. Soucy)

FILE IN USE IS c:\systat\data\SUM2REPL.SYS

MONLIN VERSION 5.03

```
1TER LOSS PARAMETER VALUES

0 .85917560+00 -.40000+00 .10000+00 .10000+00-.20000-01-.20000-01

1 .68553260+00 -.40030+00 .99580-01 .98270-01-.19430-01-.22230-01

2 .60165470+00 -.40200+00 .99580-01 .86140-01-.20320-01-.14580-01

3 .58574860+00 -.41720+00 .53630-01 .89390-01-.39230-01-.11830-01

4 .58433210+00 -.41920+00 .48680-01 .90150-01-.36310-01-.10530-01

5 .58191530+00 -.44460+00 .57070-01 .93360-01-.36120-01-.11270-01

6 .58121920+00 -.40980+00 .41980-01 .89420-01-.36900-01-.94790-02

7 .58071900+00 -.38230+00 .43540-01 .86170-01-.34610-01-.96950-02

8 .58060280+00 -.37370+00 .44340-01 .85290-01-.33740-01-.97960-02

9 .58057160+00 -.36680+00 .41780-01 .83870-01-.33320-01-.94940-02

10 .58056380+00 -.36300+00 .41810-01 .84220-01-.33320-01-.95140-02

12 .58056380+00 -.36290+00 .41820-01 .84140-01-.33470-01-.95120-02

13 .58056380+00 -.36290+00 .41820-01 .84140-01-.33470-01-.95120-02
```

DEPENDENT VARIABLE IS EN

SOURCE SUM-OF-SQUARES DF MEAN-SQUARE

REGRESSION 101.5741 5 20.3148 RESIDUAL 0.5806 347 0.0017 TOTAL 101.0736 352 CORRECTED 4.4770 351

RAW R-SQUARED (1-RESIDUAL/TOTAL) = 0.9943
CORRECTED R-SQUARED (1-RESIDUAL/CORRECTED) = 0.8703

PARAMETER ESTIMATE A.S.E. LOWER <95%> UPPER
AA -0.3629 0.0764 -0.5131 -0.2127
AO 0.0418 0.0202 0.0021 0.0815
AL 0.0841 0.0084 0.0676 0.1007
AR -0.0335 0.0075 -0.0483 -0.0187
AV -0.0095 0.0025 -0.0144 -0.0047

ASYMPTOTIC CORRELATION MATRIX OF PARAMETERS

AA AO AL AR AV

AA 1.0000 AO -0.5649 1.0000 AL -0.9779 0.4885 1.0000 AR 0.4576 0.1806 -0.4536 1.0000 AV 0.5095 -0.7936 -0.5077 0.2097 1.0000

TABLE A1.6: SYSTAT output (F. Soucy), continued

FILE IN USE IS c:\systat\data\SUM2REPL.SYS

```
ITER LOSS PARAMETER VALUES
     .43458680+02 -.40000+00 .10000+00 .10000+00-.20000-01-.20000-01 .22817930+02 -.40100+00 .98410-01 .93500-01-.17760-01-.28890-01
      .4662654D+01 -.3883D+00 .9770D-01 .5046D-01-.1705D-01-.2752D-01
      .43669660+01 -.42750+00 .78580-01 .39850-01 .22820-01-.25660-02 .41756840+01 -.49250+00 .78700-01 .20740-01-.16120-01 .87330-02
      .2251565D+01 -.5347D+00 .2037D+00 .12980-01 .1969D-01 .8033D-02 .7164177D+00 -.4487D+0D-.40680-01 .50960-01-.2042D-01 .79960-02
      .4448281D+00 -.4654D+00 .9113D-01 .4406D-01-.11480-01-.6317D-02 .4406292D+00 -.4743D+00 .1012D+00 .4189D-01-.8077D-02-.4806D-02
      .4397790D+00 -.4799D+00 .1167D+00 .4050D-01-.5226D-02-.5315D-02
10 .4385163D+00 -.4769D+00 .1095D+00 .4156D-01-.8149D-DZ-.5833D-02
11 .43760640+00 -.4807D+00 .1158D+00 .4103D-01-.6379D-02-.5822D-02
12 .4338900D+00 -.4923D+00 .1237D+00 .4148D-01-.3344D-02-.6020D-02
13 .4165416D+00 -.5803D+00 .1677D+00 .4255D-01 .1353D-02-.7219D-02
14 .4037120D+00 -.6556D+00 .1770D+00 .4564D-01-.3147D-02-.6885D-02
    .39806070+00 -.66790+00 .16280+00 .48030-01-.87820-02-.65920-02
.39770920+00 -.68750+00 .16840+00 .48750-01-.87730-02-.66760-02
.39746350+00 -.70260+00 .17520+00 .49190-01-.80400-02-.68130-02
.39743670+00 -.70170+00 .17420+00 .49290-01-.81970-02-.68260-02
15
16
17
18
.3974355D+00 -.7017D+00 .1743D+00 .4931D-01-.8133D-02-.6852D-02
21
       .3974355D+00 -.7017D+00 .1743D+00 .4931D-01-.8133D-02-.6852D-02
```

DEPENDENT VARIABLE IS EP

MISSING DATA OR ESTIMATES REDUCED DEGREES OF FREEDOM

SOURCE SUM-DF-SQUARES DF MEAN-SQUARE

REGRESSION 19.9625 5 3.9929 RESIDUAL 0.3974 354 0.0011 TOTAL 20.3392 359 CORRECTED 4.6293 358

RAW R-SQUARED (1-RESIDUAL/TOTAL) = 0.9805 COGRECTED R-SQUARED (1-RESIDUAL/CORRECTED) = 0.9141

PARAMETER ESTIMATE A.S.E. LOWER <95%> UPPER -0.7017 0.0625 -0.8246 -0.5788 AA 0.1743 0.0176 0.1398 AO 0.2089 AL 0.0493 0.0034 0.0426 0.0560 0.0040 -0.0160 -0.0002 -0.0081 AR -0.0069 0.0009 -0.0087 -0.0050

ASYMPTOTIC CORRELATION MATRIX OF PARAMETERS

AA AO AL AR AV

AA 1.0000 AO -0.8871 1.0000 AL -0.9008 0.6678 1.0000 AR 0.0800 0.2359 -0.1528 1.0000 AV 0.4154 -0.3796 -0.5525 0.2912 1.0000

TABLE A1.7: SYSTAT output, converted to log10 (Boise-Cascade)

DEPENDENT VARIABLE IS EN

PARAMETER

RAMETER ESTIMATE A.S.E. LOWER <95% UPPER AA -0.3874 0.0362 -0.4609 -0.3138 AL 0.2597 0.0117 0.2360 0.2837 AV -0.0111 0.0023 -0.0157 -0.0064

DEPENDENT VARIABLE IS EP

 PARAMETER
 ESTIMATE
 A.S.E.
 LOWER
 <95%> UPPER

 AA
 -0.3874
 0.0362
 -0.4598
 -0.3150

 AL
 0.0882
 0.0097
 0.0686
 0.1078

 AV
 -0.0272
 0.0046
 -0.0366
 -0.0177

TABLE A1.8: SYSTAT output, converted to log10 (F. Soucy)

DEPENDENT VARIABLE IS EN

PARAMETER

RAMETER ESTIMATE A.S.E. LOWER <95%> UPPER AA -0.3629 0.0764 -0.5131 -0.2127 AO 0.0418 0.0202 0.0021 0.0815 AL 0.1937 0.0193 0.1557 0.2319 AR -0.0771 0.0173 -0.1112 -0.0431 AV -0.0219 0.0058 -0.0332 -0.0108

DEPENDENT VARIABLE IS EP

PARAMETER ESTIMATE A.S.E. LOWER <95%> UPPER
AA -0.7017 0.0625 -0.8246 -0.5788
AD 0.1743 0.0176 0.1398 0.2089
AL 0.1135 0.0078 0.0981 0.1289
AR -0.0187 0.0092 -0.0368 -0.0005
AV -0.0158 0.0021 -0.0200 -0.0115

TABLE	A1.9: Emp	erimental r	resul ts					3104	99 45.8	2.16	320	18 0.202	0.206	0.039	0.038	0.000
		_ '						3156	99 45.8	2.15	325	20 0.202	0.247	0.045	0.045	0.000
(Sort	order: Re	idius, Bulk,	Velocity,	Load,	serial (number)		3157	99 45.8	2.15	321	20 0.202	0.242	0.039	0.040	0.000
								3159	99 45.8	2.15	322	20 0.202	0.235	0.042	0.043	0.000
Ser.	Cat B.W.	B _L Vell	Load Rad	In-nip	Rec1	RecZ	Perm.	3172	99 45.8	2.15	323	20 0.202	0.237	0.053	0.047	0.000
Mum.	µm g/m²	cm ³ /g m/m	kN/m m	strain	strain	strain	strain	3173	99 45.8	2.15	319	20 0,202	0.233	0.052	0.045	0.000
								3105	99 45.8	2.16	321	28 0.202	0.313	0.041	0.040	0.000
3103	99 45.8	2.16 96	18 0.202	0.196	0.052	0.040	0.000	3155	99 45.8	2.15	323	30 0.202	0.373	0.047	0.042	0.000
3113	98 45.8	2.13 96	18 0.202	0.219	0.058	0.035	0.000	3158	99 45.8	2.15	320	29 0.202	0.341	0.050	0.047	0.000
3128	100 45.8	2.18 95	19 0.202	0.229	0.054	0.038	0.000	3106	99 45.8	2.16	321	41 0.202	0.408	0.056	0.050	0.000
3129	100 45.8	2.18 95	19 0.202	0.234	0.057	0.040	0.000	3151	99 45.8	2.15	332	41 0.202	0.434	0.060	0.043	0.000
3163	99 45.8	2.15 99	20 0.202	0.251	0.076	0.053	0.000	3154	99 45.8	2.15	325	42 0.202	0.441	0.064	0.057	0.000
3102	99 45.8	2.16 97	29 0.202	0.328	0.066	0.051	0.000	3160	99 45.8	2.15	319	41 0.202	0.435	0.066	0.061	0.000
3112	98 45.8	2.13 96	29 0.202	0.329	0.069	0.040	0.000	3171	99 45.8	2.15	324	41 0.202	0.436	0.075	0.061	0.000
3114	98 45.8	2.13 95	28 0.202	0.322	0.067	0.045	0.000	3174		2.15	314	41 0.202	0.428	0.076	0.068	0.000
3127	100 45.8	2.18 95	30 0.202	0.345	0.064	0.044	0.000	3152	99 45.8	2.15	320	66 0.202		0.092	0.083	0.000
3130	100 45.8	2.18 95	28 0.202	0.344	0.071	0.053	0.000	3153	99 45.8	2.15	322	66 0.202	0.492	0.093	0.084	0.000
3162	99 45.8	2.15 98	29 0.202	0.355	0.091	0.062	0.000	3170		2.15	326	65 0.202	0.489	0.103	0.086	0.000
3101	99 45.8	2.16 96	41 0.202	0.412	0.073	0.050	0,000	3108	99 45.8	2.16	538	18 0.202	0.212	0.040	0.032	0.000
3111	98 45.8	2.13 98	42 0.202	0.421	0.090	0.056	0.000	3110	99 45.8	2.16	534	18 0.202	0.203	0.036	0.029	0.000
3115	98 45.8	2.13 97	42 0.202			0.061	0.000	3149	98 45.8	2.15	534	19 0.202	0.204	0.018	0.007	0.000
3123	100 45.8	2.18 95	41 0.202		0.074	0.049	0,000	3150	98 45.8	2.15	533	19 0.202	0.210	0.025	0.015	0.000
3126	100 45.8	2.18 96	42 0.202	0.436	0.077	0.056	0.000	3109	99 45.8	2.16	533	27 0.202	0.314	0.042		0.000
3131	100 45.8	2.18 93	41 0.202	0.439		0.071	0.000	3145	98 45.8	2.15	534	30 0.202	0.359		0.014	0.000
3132	98 45.8	2.15 95	41 0.202				0.000	3146	98 45.8	2.15	533	30 0.202	0.340	0.031	0.018	0.000
3161	99 45.8	2.15 97	41 0.202		0.104	0.068	0.000	3147	98 45.8	2.15	532	30 0.202	0.345	0.035	0.024	0.000
3124	100 45.8	2.18 95	66 0.202				0.000	3107	99 45.8	2.16	536 544	41 0.202	0.406	0.051	0.042	0.000
3125	100 45.8	2.18 95	66 0.202				0.000	3142 3148	98 45.8 98 45.8	2.15 2.15	531	41 0.202 41 0.202	0.422	0.053	0.039	0.000
3119	98 45.8	2.13 197	19 0.202				0.000	3143	98 45.8	2.15	528	66 0.202	0.431	0.048	0.036 0.067	0.000 0.000
3138	98 45.8	2.15 186	18 0.202				0,000	3144	98 45.8	2.15	533	66 0.202	0.485	0.079		0.000
3139	98 45.8	2.15 185	18 0.202				0.000	3219	93 42.4	2.20	96	14 0.202	0.199		0.003	
3140	98 45.8		16 0.202		0.038		0.000	3217	93 42.4	2.20	96	20 0.202	0.306	0.138	0.117	
3164	99 45.8	2.15 192	20 0.202				0.000	3218	93 42.4	2.20	96	15 0.202	0.219	0.133	0.123	
3167	99 45.8	2.15 187	20 0.202		0.070	0.046	0.000	3220	93 42.4	2.20	95	20 0.202	0.306		0.124	
3120	98 45.8	2.13 185	28 0.202		0.065	0.042	0.000	3213	93 42.4	2.20	96	39 0.202	0.476	0.162	0.128	0.007
3137	98 45.8	2.15 186	29 0.202			0.029	0.000	3216	93 42.4	2.20	96	41 0.202	0.480	0.163	0.140	0.021
3141	98 45.8	2.15 186	27 0.202				0.000	3214	93 42.4	2.20	96	65 0.202	0.522	0.194	0.165	0.045
3165	99 45.8	2.15 187	28 0.202			0.049	0.000	3215	93 42.4	2.20	96	65 0.202	0.521	0.196		0.051
3166	99 45.8	2.15 188	28 0.202			0.050	0.000	3227	93 42.4	2.20	186	14 0.202			0.121	
3116	98 45.8	2.13 192	42 0.202	_		0.044	0.000	3228	93 42.4	2.20	187	14 0.202			0.120	
3121	98 45.8	2.13 185	42 0.202	_		0.055	0.000	3200	92 42.4	2.17	187	20 0.202	0.179		0.129	0.000
3133	98 45.8	2.15 186	41 0.202			0.048	0.000	3221	93 42.4	2.20	187	20 0.202	0.300	0.124	0.114	-0.013
3136	98 45.8	2.15 187	42 0.202		0.080	0.056	0.000	3226	93 42.4	2.20	188	20 0.202	0.307	0.130	0.117	
3168	99 45.8	2.15 185	41 0.202		0.088	0.065	0.000	322 9	93 42.4	2.20	186	20 0.202	0.303	0.127	0.123	0.003
3117	98 45.8	2.13 187	66 0.202		0.107		0.000	3196	92 42.4	2.17	187	40 0.202	0.391	0.035	0.002	-1003
3118	98 45.8	2.13 187	66 0.202		0.110	0.081	0.000	3199	92 42.4	2.17	188	42 0.202	0.406	0.042	0.012	
3122	98 45.8	2.13 185	66 0.202		0.109	0.084	0.000	3222	93 42.4	2.20	186	39 0.202	0.468	0.141	0.129	0.017
3134	98 45.8	2.15 186	66 0.202			0.087	0.000	3225	93 42.4	2.20	188	41 0.202	0.462	0.132		
3135	98 45.8	2.15 186	66 0.202				0.000	3197	92 42.4	2.17	187			0.091		
3169	99 45.8	2.15 185	66 0.202	U-492	V.115	0.089	0.000	5.77	VL4-7							

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Ser.	Cal	8.W.	R.	Vel	Load	Rad	In-nip	Rec1	Rec2	Perm.	3026	127 49.0	2.59	97	29 0.202	0.458	0.220		
Num.		g/m²	~b/a		kH/m		strain	strain	strain	strain	3031	127 49.0		96	27 0.202				0.118
, and	-	3,	· / 3		,		44.4	•	•		3023	127 49.0	2.59	97	41 0.202				0.156
3198	02	42.4	2.17	187	65	0.202	0.470	0.095	0.072		3025	127 49.0	2.59	100	43 0.202		0.262		0.165
3223		42.4	2.20	186	_	0.202		0.172		0.034	3024	127 49.0	2.59	96	66 0.202				0.226
3224		42.4	2.20	187		0.202			0.149		3034	127 49.0	2.59	189	18 0,202	0.350	-	0.081	0.071
3201		42.4	2.17	325		0.202					3032	127 49.0		187	28 0.202	0.444		0.119	0.108
3230		42.4	2.20	322		0.202		0.122	0.119	0.011	3033	127 49.0	2.59	188	28 0.202			0.120	0.111
3235		42.4	2.20	321		0.202		0.114			3035	127 49.0		184	41 0.202		0.228		0.152
3236			2.20	320	_	0.202	0.199	0.089	0.097	-0.014	3080	118 45.8	2.57	187	41 0.202	0.572	0.201	0.110 0.180	
3202			2.17	319	40	0.202		0.011			3079	113 44.5	2.55	182	65 0.202 65 0.202	0.623			0.159
3204			2.23	321		0.202	0.451	0.104	0.091	-0.007	3090	118 45.8	2.57	188 186		0.634			0.159
3231		42.4	2.20	319	39	0.202	0.457	0.123	0.119	-0.002	3091	118 45.8 118 45.8		189	66 0.202	0.627			0.159
3234	93	42.4	2.20	324	41	0.202	0.462	0.124	0.124	0.003		118 45.8	2.57	186	65 0.202		0.210		0.159
3203	92	42.4	2.17	320	65	0.202	0.465	0.066	0.053		3093 3094	118 45.8		186	65 0.202		0.198		0.159
3232	93	42.4	2.20	317	65	0.202	0.514	0.155	0.155	0.029	3094 3095	118 45.8		189	65 0.202		0.196		0.159
3233	93	42.4	2.20	320	65	0.202				0.029	3096	118 45.8		183	65 0.202			0.170	0.159
3210	94	42.4	2.23	536	20	0.202				-0.019	3077	113 44.5		323	10 0.202				0.006
3211		42.4		533		0.202		0.096		-0.018	3038	127 49.0	2.59	322	18 0.202	0.344	0.134	0.070	0.064
3205		42.4		538		0.202		0.094		-0.021	3054	114 44.5		325	18 0.202		0.110		0.065
3206			2.23	533		0.202			0.081		3074	113 44.5	2.55	327	20 0 202	0.290	0.037	-0.003	0.017
3209		42.4	2.23	539		0.202			0.081		3076		2.55	321	19 0.202	0.285	0.033	0.001	0.012
3212		42.4	2.23	529	. :=	0.202				0.015	3036	127 49.0		326	29 0.202		0.173	0.116	0.100
3207		42.4	2.23	527		0.202		0.122	0.104 0.108	0.012	3037	127 49.0	2.59	321	29 0.202	0.447	0.173	0.111	0.104
3208		42.4	2.23	532		0.202		0.125 0.068		0.023	3053	114 44.5	2.56	324	29 0.2 02	0.439		0.113	0.096
3060		44.5	2.58	96		0.202	0.137	0.121	0.034	0.002	3075			320	28 0.202		0.073		0.062
3195			2.47	96 95		0.202	0.274	0.132	0.110		3039	127 49.0		317	41 0.202	0.514			
3242 3243		42.2 42.2		95		0.202		0.128		-0.003	3049	114 44.5	2.56	327		0.507			0.129 0.136
3244		42.2		95		0.202		0.139		0.018	3052			326	42 0.202			U. 130	0.138
3059		44.5	2.58	96		0.202				0.083	3073	113 44.5		328		0.543	0.124	0.198	0.196
			2.58	95		0.202				0.085	3050	114 44.5		318	66 0.202	0.559 0.555	0.233	0.199	
3058			2.58	96		0.202			0.155	0.125	3051	114 44.5		320	66 0.202 66 0.202	0.555			
		44.5	2.56	97		0.202	0.509			0.130	3055	114 44.5	2.58	322 330	65 0.202	0.333	0.209	0.216	0.194
3057		44.5	2.58	96		0.202	0.522	0.221	0.193	0.169	3063	115 44.5 113 44.5	2.55	327	65 0.202	0.612		0.149	
3081		45.8	2.57	97	41	0.202			0.125		3072	113 44.5	2.55	331	65 0.202	0.626			
3192		42.4	2.47	96		0.202		0.160		0.035	3078 3082	118 45.8	2.57		66 0.202	0.636	0.220		0.152
3194	105	42.4	2.47	96	42	0.202	0.481		_	0.046	3083	118 45.8			65 0.202		0.217	0.166	0.152
3238	108	42.2	2.56	96	41	0.202	0.499				3084	118 45.8	2.57		65 0.202	0.637	0.215	0.170	0.152
3241	108	42.2	2.56	96	43	0.202		_	0.208	0.085	3085	118 45.8			65 0.202	0.635		0.169	0.152
3245	108	42.2	2.56	96		0.202			0.200	0.096	3086		2.57	323	65 0.202	0.637	0.209	0.168	0.152
3056	115	44.5	2.58	96		0.202			0.243	0.219	3087	118 45.8	2.57	320	65 0.202	0.637		0.164	0.152
3062		44.5	2.58	96		0.202			0.251	0.219	3088			321	65 0.202	0.643		0.173	0.152
3193		42.4	2.47	96		0.202		_	0.264	0.106 0.142	3089	118 45.8	2.57	321	65 0.202	0.636	0,200	0.164	0.152
3239		42.2		95		0.202				0.142	3070	113 44.5	2.55	534	10 0.202		_ ^		0.000
3240		42.2		95		0.202		0.282		0.040	3041	127 49.0		535		0.335	0.120	0.059	U.U3U
3028		49.0	2.59	97	-	0.202			0.048	0.041	3069	113 44.5		536	19 0.202	0.281	0.034	0.011	0.029
3029		49.0		96		0.202 0.202				0.079	3040	127 49.0		547	29 0.202	0.440	0.130	0.106	0.055
3027			2.59	97 96		0.202	. U.J)~	0.173			3068	113 44.5	2.55	533	27 0.202	U.410	0.060	U.U43	U.U.J.
5030	127	49.0	c.74	70	, 17	U.EUE	. 5.337												

0.350 0.350 0.357 0.351 0.371 0.353 0.353	0.052 0.080 0.093 0.156 0.169	0.233 0.232 0.234 0.234 0.233 0.233	0.336 0.336 0.336 0.336 0.033 0.023 0.023 0.043 0.043 0.043 0.043 0.043 0.043 0.043 0.043 0.043	3
0.342	0.065 0.105 0.116 0.189	0.261 0.257 0.270 0.351 0.286	0.341 0.356 0.0356 0.0035 0.0035 0.0035 0.0036 0.0136 0.0147 0.0147 0.0148 0.0148	0
0.387	0.148 0.192 0.202 0.273	0.254	0.339 0.114 0.093 0.127 0.128 0.128 0.128 0.228 0.228 0.228 0.228 0.228 0.228 0.228 0.228 0.228 0.228 0.228	0.04
0.627 0.628 0.620 0.654 0.654 0.654	0.261 0.383 0.391 0.517 0.527	0.561 0.578 0.578 0.578 0.578 0.603 0.582	0.665 0.665 0.665 0.707 0.707 0.277 0.277 0.277 0.277 0.277 0.257 0.257 0.257 0.257 0.257 0.257	100.0
143 0.202 172 0.202 172 0.202 174 0.202 174 0.202 208 0.202 208 0.202 205 0.202 205 0.202	00000	0000000	157 0.202 175 0.202 177 0.202 177 0.202 177 0.202 17 0.202 18 0.202 18 0.202 19 0.202	202.0 16
			25	
		2.5.5.5.5.5.5.5.5.5.5.5.5.5.5.5.5.5.5.5		
		113 44.5 126 48.5 113 44.5 127 49.0 113 44.5 127 49.0	13 4 4 5 5 5 6 6 5 6 5 6 5 6 5 6 5 6 5 6 5	
266 265 265 265 265 265 265 265 265 265	2862 2862 2862 2866 2866 2866 2866 2866	2692 2692 2693 2693 2693 2693 2693 2693	2632 2632 2632 2632 2632 2632 2632 2632	3
	0.035 0.035 0.064 0.078 0.164		0.256 0.333 0.333 0.335 0.335 0.335 0.222 0.222 0.232 0.235 0.255 0.309 0.313	ö
strain 0.127 0.098 0.098 0.101 0.180	0.057 0.061 0.077 0.105 0.192	0.207 0.198 0.234 0.265 0.261	0.373 0.059 0.115 0.199 0.307 0.363 0.363	0.350 0.
	0.057 0.061 0.077 0.105 0.192	0.207 0.198 0.234 0.265 0.261	0.373 0.059 0.115 0.307 0.307 0.363	ö
strain 0.127 0.098 0.098 0.101 0.180	0.146 0.057 0.146 0.057 0.149 0.061 0.175 0.077 0.205 0.192 0.285 0.192	0.207 0.198 0.234 0.265 0.261	0.373 0.146 0.059 0.203 0.115 0.289 0.199 0.343 0.260 0.338 0.292 0.326 0.292	0.350 0.
Rad In-nip Rec1 Rec2 R 0.202 0.502 0.179 0.127 0.202 0.533 0.108 0.098 0.202 0.533 0.109 0.101 0.202 0.563 0.218 0.160 0.202 0.599 0.166 0.157 0.202 0.600 0.161 0.158	0.202 0.284 0.146 0.050 0.202 0.287 0.146 0.057 0.202 0.241 0.149 0.061 0.202 0.346 0.175 0.077 0.202 0.379 0.203 0.195 0.202 0.503 0.292 0.192	0.202 0.520 0.298 0.207 0.202 0.203 0.203 0.203 0.204 0.198 0.202 0.528 0.204 0.198 0.202 0.508 0.205	0.202 0.574 0.202 0.617 0.202 0.619 0.202 0.618 0.202 0.618 0.202 0.638 0.202 0.538 0.202 0.538 0.199 0.202 0.516 0.289 0.199 0.202 0.556 0.202 0.556 0.202 0.556 0.202 0.556 0.202 0.556 0.202 0.556 0.202 0.586 0.202 0.586 0.202 0.586 0.202 0.686 0.202 0.608 0.202 0.608 0.202 0.608 0.202 0.608 0.202 0.608	0.202 0.380 0.350 0.
Rad In-nip Rec1 Rec2 R 0.202 0.502 0.179 0.127 0.202 0.533 0.108 0.098 0.202 0.533 0.109 0.101 0.202 0.563 0.218 0.160 0.202 0.599 0.166 0.157 0.202 0.600 0.161 0.158	12 0.202 0.284 0.146 0.050 12 0.202 0.287 0.146 0.057 9 0.202 0.241 0.149 0.061 17 0.202 0.346 0.175 0.077 19 0.202 0.379 0.203 0.105 42 0.202 0.503 0.285 0.192 42 0.202 0.503 0.285 0.192	40 0.202 0.520 0.298 0.207 40 0.202 0.523 0.294 0.198 53 0.202 0.534 0.317 0.234 70 0.202 0.528 0.344 0.265 65 0.202 0.561 0.342 0.265 65 0.202 0.561 0.355 0.267 85 0.202 0.562	86 0.202 0.574 135 0.202 0.617 137 0.202 0.618 172 0.202 0.618 173 0.202 0.618 173 0.202 0.618 173 0.202 0.638 207 0.202 0.536 19 0.202 0.546 0.165 19 0.202 0.516 0.203 0.115 40 0.202 0.516 0.203 0.199 70 0.202 0.516 0.203 0.199 70 0.202 0.556 70 0.202 0.556 82 0.202 0.586 88 0.202 0.585 86 0.202 0.585 87 0.202 0.585 88 0.202 0.608 136 0.202 0.608 136 0.202 0.608 136 0.202 0.608	140 0.202 0.380 0.350 0.
Wel Load Rad In-nip Rec1 Rec2 Rec2 m/m kN/m m strain strain strain 524 41 0.202 0.502 0.177 0.127 546 40 0.202 0.523 0.108 0.098 547 41 0.202 0.533 0.109 0.101 523 66 0.202 0.563 0.218 0.160 526 65 0.202 0.599 0.166 0.157 534 65 0.202 0.600 0.161 0.158 534 65 0.202 0.600 0.161 0.158	96 12 0.202 0.284 0.146 0.050 96 12 0.202 0.287 0.146 0.057 96 17 0.202 0.346 0.175 0.077 98 19 0.202 0.379 0.203 0.105 96 42 0.202 0.503 0.285 0.192 97 42 0.202 0.503 0.285 0.192	96 40 0.202 0.529 0.298 0.207 96 40 0.202 0.535 0.294 0.198 97 53 0.202 0.534 0.317 0.234 97 70 0.202 0.528 96 65 0.202 0.562 0.344 0.265 96 65 0.202 0.561 0.355 0.267 97 85 0.202 0.562 0.365	95 86 0.202 0.574 95 87 0.202 0.617 96 135 0.202 0.617 96 137 0.202 0.618 95 177 0.202 0.618 97 207 0.202 0.638 97 207 0.202 0.638 190 19 0.202 0.516 0.203 0.115 197 40 0.202 0.516 0.203 0.109 173 70 0.202 0.516 0.203 0.199 174 70 0.202 0.551 177 82 0.202 0.551 177 82 0.202 0.586 177 86 0.202 0.586 171 87 0.202 0.586 171 87 0.202 0.585 171 87 0.202 0.608 171 136 0.202 0.608 171 136 0.202 0.608	177 140 0.202
E, Vel Load Rad In-nip Rec1 Rec2 Rec2 Rec2 Rec2 Rec2 Rec2 Rec2 Rec2	2.51 96 12 0.202 0.284 0.146 0.050 2.53 96 12 0.202 0.287 0.148 0.057 2.53 96 9 0.202 0.241 0.149 0.061 2.53 96 17 0.202 0.346 0.175 0.077 2.53 96 19 0.202 0.379 0.203 0.105 2.54 96 42 0.202 0.537 0.285 0.192 2.54 97 42 0.202 0.537 0.285 0.192	2.55 96 40 0.202 0.520 0.298 0.207 2.53 96 40 0.202 0.523 0.294 0.198 2.51 97 53 0.202 0.528 0.294 0.198 2.57 97 70 0.202 0.528 0.344 0.265 2.51 96 65 0.202 0.562 0.344 0.265 2.53 96 65 0.202 0.561 0.355 0.267 2.57 97 85 0.202 0.561 0.355 0.267	2.58 95 86 0.202 0.574 2.58 95 87 0.202 0.617 2.57 96 172 0.202 0.619 2.57 96 172 0.202 0.619 2.57 96 172 0.202 0.619 2.57 97 207 0.202 0.618 2.53 188 9 0.202 0.240 0.146 0.059 2.53 189 9 0.202 0.516 0.203 0.115 2.53 187 40 0.202 0.516 0.203 0.195 2.54 177 70 0.202 0.516 0.203 0.197 2.54 177 82 0.202 0.551 2.55 177 82 0.202 0.555 2.57 171 86 0.202 0.567 2.58 187 87 0.202 0.567 2.58 171 86 0.202 0.567 2.59 173 136 0.202 0.608 2.57 171 86 0.202 0.608 2.57 171 86 0.202 0.608 2.57 171 86 0.202 0.608 2.57 171 136 0.202 0.608 2.57 171 136 0.202 0.603	2.54 177 140 0.202 0.380 0.350 0.
8.W. B. Vel Load Rad In-nip Rec1 Rec2 Rg/m² cm³/g m/m kN/m m strain stra	48.5 2.51 96 12 0.202 0.284 0.146 0.050 4.8.5 2.51 96 12 0.202 0.287 0.146 0.057 4.8.5 2.53 96 9 0.202 0.287 0.148 0.057 4.8.5 2.51 96 17 0.202 0.346 0.175 0.077 4.8.5 2.53 98 19 0.202 0.379 0.203 0.105 4.8.5 2.54 97 4.2 0.202 0.379 0.285 0.192 4.8.5 2.54 97 4.2 0.202 0.519 0.285 0.192	46.5 2.55 96 40 0.202 0.520 0.298 0.207 46.5 2.53 96 40 0.202 0.523 0.294 0.198 46.5 2.51 97 53 0.202 0.534 0.317 0.234 49.0 2.57 97 70 0.202 0.528 0.344 0.265 48.5 2.51 96 65 0.202 0.562 0.344 0.265 48.5 2.53 96 65 0.202 0.561 0.355 0.267 49.0 2.57 97 85 0.202 0.561 0.355 0.267	49.0 2.58 95 86 0.202 0.574 44.5 2.58 95 87 0.202 0.617 46.0 2.58 95 137 0.202 0.618 49.0 2.57 96 172 0.202 0.618 49.0 2.57 96 172 0.202 0.618 49.0 2.57 96 172 0.202 0.638 49.0 2.57 97 207 0.202 0.638 48.5 2.53 190 19 0.202 0.536 0.195 48.5 2.53 197 70 0.202 0.536 0.195 48.5 2.53 197 70 0.202 0.536 0.289 0.197 48.5 2.54 177 70 0.202 0.556 49.0 2.57 174 82 0.202 0.586 49.0 2.57 177 86 0.202 0.586 49.0 2.57 178 86 0.202 0.586 44.5 2.58 186 87 0.202 0.586 44.5 2.58 178 86 0.202 0.586 44.5 2.58 178 87 0.202 0.586 44.5 2.58 178 88 0.202 0.586 44.5 2.58 178 87 0.202 0.586 44.5 2.58 178 87 0.202 0.586 44.5 2.58 178 87 0.202 0.586 44.5 2.58 178 87 0.202 0.688 44.5 2.58 178 87 0.202 0.688 44.5 2.58 178 186 0.202 0.688 44.5 2.58 178 186 0.202 0.688 44.5 2.58 178 186 0.202 0.608 44.5 2.58 178 136 0.202 0.608 45.0 2.57 171 136 0.202 0.603	44.5 2.54 177 140 0.202 0.380 0.350 0.
Cal 8.W. B. Vel Load Rad In-nip Rec1 Rec2 In gam g/m² cm³/g m/m kN/m m strain strain strain strain 127 49.0 2.59 524 41 0.202 0.502 0.179 0.127 113 44.5 2.55 546 40 0.202 0.533 0.109 0.101 127 49.0 2.59 523 66 0.202 0.533 0.109 0.101 127 49.0 2.59 523 66 0.202 0.503 0.216 0.150 113 44.5 2.55 526 65 0.202 0.500 0.161 0.158 113 44.5 2.55 534 65 0.202 0.000 0.161 0.158	123 48.5 2.51 % 12 0.202 0.284 0.146 0.050 123 48.5 2.51 % 12 0.202 0.287 0.146 0.057 124 48.5 2.53 % 9 0.202 0.287 0.149 0.067 123 48.5 2.53 % 17 0.202 0.346 0.175 0.077 124 48.5 2.53 % 19 0.202 0.379 0.203 0.105 124 48.5 2.51 % 49 0.202 0.503 0.285 0.195 125 48.5 2.51 % 49 0.202 0.503 0.285 0.195 125 48.5 2.54 % 7 4.2 0.202 0.503 0.285 0.195	125 48.5 2.55 96 40 0.202 0.520 0.298 0.207 124 48.5 2.53 96 40 0.202 0.523 0.294 0.198 124 48.5 2.51 97 53 0.202 0.534 0.317 0.234 126 49.0 2.57 97 70 0.202 0.538 0.317 0.234 123 48.5 2.51 96 65 0.202 0.562 0.344 0.265 123 48.5 2.51 96 65 0.202 0.561 0.342 0.265 124 48.5 2.53 96 65 0.202 0.581 0.355 0.267 126 49.0 2.57 97 85 0.202 0.562	2.58 95 86 0.202 0.574 2.58 95 87 0.202 0.617 2.57 96 172 0.202 0.619 2.57 96 172 0.202 0.619 2.57 96 172 0.202 0.619 2.57 97 207 0.202 0.618 2.53 188 9 0.202 0.240 0.146 0.059 2.53 189 9 0.202 0.516 0.203 0.115 2.53 187 40 0.202 0.516 0.203 0.195 2.54 177 70 0.202 0.516 0.203 0.197 2.54 177 82 0.202 0.551 2.55 177 82 0.202 0.555 2.57 171 86 0.202 0.567 2.58 187 87 0.202 0.567 2.58 171 86 0.202 0.567 2.59 173 136 0.202 0.608 2.57 171 86 0.202 0.608 2.57 171 86 0.202 0.608 2.57 171 86 0.202 0.608 2.57 171 136 0.202 0.608 2.57 171 136 0.202 0.603	113 44.5 2.54 177 140 0.202 0.380 0.350 0.

Ser.	Cal	R.V.	R.	Ve1	Load	Pad	tn-nip	Rec1	Rec2	Perm.	2932	123 48.5	2.51	673	50 0.202	0.509	0.201	0.129	0.149
Num.		g/m ²	ر فت	- m/m	kH/=		strain	strain	strain	Perm. strain	2933	123 48.5		872		0.506	0.199		0.150
HUM.		3/m	CW /8	, 49 14	,	•••	31.5	34.0		••••	2834	117 45.0	2.59	882	67 0.202	0.479		0.202	
2952	125		2.54				0.531				2843	117 45.0		877	69 0.202	0.486		0.204	
2695			2.54	563			0.552			0.233	2844	117 45.0		873		0.483		0,209	
	•		2.55	559	_	0.202			0.213	0.211	3009	126 48.5		954	64 0.202	0.563	0.185	A 270	0.191
		45.0		555		0.202			0.204	0.198	2835	117 45.0		875	76 0.202	0.496		0.239	
			2.55	557		0.202		0.167	0.204	0.182		117 45.0		881	79 0.202	0.498		0.215	-
2861		45.0	2.55	554		0.202			0.203	0.189	2836	117 45.0		867	90 0.202	0.510		0.237	
2923			2.51	553			0.540	0.272	0.184	0.195	2841	117 45.0		889	95 0.202	0.522 0.547			0.289
2924			2.51	555		0.202		0.273		0.195		117 45.0		860 883	134 0.202 139 0.202	0.549		0.296	
	125	48.5	2.54	551	65	0.202	0.557	0.272	0.197	0.203	2840	117 45.0		85B		0.563		0.317	
2951			2.54	554	65	0.202		0.268		0.198	2838	117 45.0		866	175 0.202	0.565		0.311	
3001	126	48.5	2.58	541		0.202	0.575	0.296	0.222	0.214	2839	117 45.0 127 48.5	2.60	96	7 0.202		0.137		
2696	113	44.5	2.54	552			0.567		0.285	0.233	2965 2964	127 48.5	2.60	96	16 0.202		0.193		0.083
2845	116	45.0	2.57	563		0.202			0.242 0.227	0.251	2960	127 48.5	2.60	96	43 0.202	0.528		0.216	0.194
2849	115	45.0	2.55	565	76	0.202			0.242	0.219	2963	127 48.5		96	45 0.202	0.537		0.230	0.207
		45.0		551	75	0.202			0.227	0.222	2966	127 48.5	2.60	95	43 0.202		0.311	0.229	0.200
		45.0		560	79	0.202		0.180	U. 224	0.199	638	128 49.0		96		0.563	0.522		0.214
831		49.0		500	87	0.202	0.534			0.199	660	128 49.0		97	70 0.202	0.570	0.498		0.182
		49.0		500	87	0.202	0.544			0.212	683	128 49.0		99	70 0.202	0.563			0.231
		45.0		550	90	0.202			0 274	0.200	712	128 49.0	2.62	97	70 0.202	0.554			0.195
				550	89	0.202			0.230	0.240	749	129 49.0	2.64	97	70 0.2 02	0.561			0.216
			2.55	546	90	0.202	A 500		0.234 0.280	0.213 0.266	761	118 45.0	2.63	97	70 0.202	0.559			0.189
			2.54	552 569	91	0.202	0.370		0.207	0.240	771	119 45.0	2.64	99	70 0.202	0.554			0.211
		45.0 49.0		535	170	0.202	n see		V. L.J.	0.222 0.199 0.199 0.212 0.266 0.240 0.213 0.246 0.283 0.312 0.298 0.286 0.313 0.322 0.320 0.327 0.031 0.014	777	119 45.0	2.64	100	70 0.202	0.568			0.218
839			2.57	542	133	0.202	0.300		0.314	0.202	779	119 45.0		99	70 0.202	0.546			0.184
2847 2854		45.0		542	132	B 202			0.287	0.298	783	119 45.0		99	70 0.202	0.513			0.207 0.239
			2.55	563	138	0.202			0.284	0.286	816	120 45.0		95	69 0.202	0.560	0.752		0.309
840		49.0	2.58	500	173	0.202	0.607			0.313	2753	117 44.5		96	70 0.202	0.597 0.599	0.332		0.320
2848			2.57		175	0.202			0.319	0.331	2754	117 44.5		95 ~~	70 0.202 68 0.202		0.337	A07 0	0.296
			2.55	541	175	0.202			0.329	0.322		117 45.0		95 95	68 0.202			0.296	0.291
		45.0		549	175	0.202			0.317	0.320	2791	117 45.0		96	69 0.202		0.378	0.276	0.280
841	126	49.0	2.58	500	208	0.202	0.615			0.327	2891 2892	129 49.0 129 49.0		95	69 0.202	0.574		0.275	0.273
2936	123	48.5	2.51	880	13	0.202	0.273	0.085		0.031	2961	127 48.5	2.60	95		0.577		0.270	0.245
2937	123	48.5	2.51	875	9	0.202	0.196	0.067		0.014	2962	127 48.5	2.60	96	65 0.202	0.581	0.351	0.275	0.254
2938		48.5	2.51	875	9	0.202	0.199	0.066		0.014	2976	127 48.5	2.60	96	65 0.202	0.584	0.349	0.266	0.252
		48.5		873	_							117 44.5	2.63	96	80 0.202	0.608	0.377		0.320
3012			2.58	963		0.202	_			0.035	2755	117 44.5		96	77 0.202		0.360		0.320
3013		48.5	2.58	961		0.202		0.082	0 031	0.042 0.043		117 45.0	2.61	95	78 0.202	0,606	0.355	0.322	
2935		48.5	2.51	860		0.202		0.108	0.021	0.043		117 45.0	2.61	95	79 0.202		0.364	0.317	0.306
2987			2.55	967		0.202		0.094	0.045		2890	129 49.0	2.63	95	78 0.202	0.592		0.294	0.297
3011		48.5	2.58	969	= = =	0.202				0.050	2893 2900 2901 639 713 750	129 49.0	2.63	95	77 0.202	0.589		0.292	0.289
3014		48.5	2.58	958	_	0.202		0.098		0.138	2900	129 49.0		94	80 0.202			0.299	0.295
2931		48.5	2.51	884	-	0.202		_		0.138	2901	129 49.0		94	80 0.202		0.393	U. <i>2</i> Y>	0.285
2934		48.5	2.51	878		0.202		0.175		0.142	639	128 49.0	2.60	96	86 0.202	0.594	0.535		0.247 0.244
2986		48.5	2.55	964 961		0.202		0.165	0.116	0.133	713	128 49.0	2.62	97	87 0.202	0.588			0.253
3008			2.58		37 41	0.202	0.505	0.168	0.122	0.136	7 50	129 49.0	2.64	96	86 0.202	0.591			Y.LJJ
5010	120	40. 3	2.58	70)	-1	J.EVE	0.,05	V											

Ser.	Cal	8.V.	8₁	Vel	Load	Rad	In-nip	Rec1	RecZ	Perm.	768	119 45.0	2.64	173	69 0.202	0.551			0.194
Num.		g/m²			kN/m		-			strain	776	119 45.0	2.64	173	70 0.202	0.541			0.181
		#• ····									2796	119 45.0	2.64	189	67 0.202	0.587			0.288
772	110	45.0	2.64	99	86	0.202	0.585			0.264	2805	119 45.0	2.64	186	68 0,202	0.596			0.276
784		45.0	2.64	98		0.202				0.254	2807	119 45.0	2.64	185	68 0.202	0.585			0.265
817		45.0	2.66	94		0.202				0.279	2884	129 49.0	2.63	188	68 0.202	0.570	0.363	0.263	0.263
2751		44.5	2.63	96		0.202		0.386		0.340	2888	129 49.0	2.63	189	68 0.202	0.574	0.366	0.262	0.264
2756			2.63	96		0.202		0.386		0.337	2797	119 45.0	2.64	185	77 0,202	0.601			0.304
2784		45.0	2.61	95		0.202			0.334	0.337	2804	119 45.0	2.64	186	80 0.202	0.610			0.297
2789		45.0	2.61	95		0.202		0.380	0.355	0.331	2885	129 49.0	2.63	184	79 0.202	0.592	0.381	0.278	0.293
2894	- •	49.0	2.63	94		0.202			0.310	0.317	2889	129 49.0	2.63	185	78 0.2 02	0.590	0.381	0.280	0.283
2899			2.63	95		0.202			0.316		644	128 49.0	2.60	169	87 0.202	0.569	0.574		0.215
751		49.0	2.64	96		0.202				0.298	718	128 49.0	2.62	169	87 0.202	0.567			0.229
640	_	49.0	2.60	96		0.202		0.588		0.312	754	129 49.0	2.64	173	90 0.202				0.248
685		49.0	2.62	98		0.202				0.329	756	129 49.0	2.64	169	87 0.202	0.565			0.242
714	_	49.0	2.62	96		0.202				0.310	769	119 45.0	2.64	173		0.578			0.241
773		45.0	2.64	98	136	0.202	0.623			0.322	770	119 45.0	2.64	173	86 0.202				0.236
785		45.0	2.64	98		0.202				0.315	2798	119 45.0	2.64	183	92 0.202	0.619			0.319
818		45.0	2.66	94	136	0.202	0.622			0.338	2803	119 45.0	2.64	186	95 0.202	0.624			0.317
2747		44.5	2.63	96	138	0.202	0.650	0.411		0.387	2886	129 49.0	2.63	185	92 0.202	0.606		0.296	
2750	117	44.5	2.63	96	140	0.202	0.649	0.428		0.388	645	128 49.0	2.60	169	136 0.202	0.607	0.597		0.258
2785	117	45.0	2.61	95	137	0.202	0.654	0.416	0.387	0.374	719	128 49.0	2.62	171	135 0.202	0.607			0.285
2788	117	45.0	2.61	95	140	0.202	0.652	0.416	0.394	0.366	757	129 49.0	2.64	169	136 0.202	0.604			0.298
2895	129	49.0	2.63	93	135	0.202	0.641	0.442	0.352	0.356	2799	119 45.0	2.64	183	138 0.202	0.652			0.354
2898	129	49.0	2.63	97	139	0.202	0.644	0.441	0.355	0.357	2802	119 45.0	2.64	186	141 0.202	0.658	0.404		0.354
641	128	49.0	2.60	96	173	0.202	0.644	0.601		0.331	2887	129 49.0	2.63	180	136 0.202		0.421	0.340	
686	128	49.0	2.62	98	172	0.202	0.650			0.365	646	128 49.0	2.60	169	173 0.202	0.621	0.595		0.285
715	128	49.0	2.62	97	172	0.202	0.645			0.331	720	128 49.0	2.62	171	174 0.202	0.623			0.312
752	129	49.0	2.64	96	172	0.202	0.640			0.350	758	129 49.0	2.64	169	173 0.202				0.335
774	119	45.0	2.64	98	173	0.202	0.640			0.350	2800	119 45.0	2.64	185		0.685			0.386
786	119	45.0	2.64	98	172	0.202	0.619			0.343	2801	119 45.0	2.64	185		0.683			0.389
819	120	45.0	2.66	94	173	0.202	0.638			0.370	721	128 49.0	2.62	169	209 0.202	0.627			0.331 0.356
2748	117	44.5	2.63	96	177	0.202	0.674	0.445		0.414	759	129 49.0	2.64	169		0.633			0.356
2749	117	44.5	2.63	96	177	0.202	0.674	0.445		0.424	656	128 49.0	2.60	299 297	70 0.202 70 0.202				0.156
2786	117	45.0	2.61	95		0.202			0.413		657	128 49.0 128 49.0	2.60 2.60	299	70 0.202				0.156
2787	117	45.0	2.61	95		0.202		0.440		•	661 760	129 49.0	2.64	298	71 0.202				0.184
2896	129	49.0	2.63	90		0.202				0.387	700 778	119 45.0	2.64	304	69 0.202				0.181
2897	129	49.0	2.63	92	175	0.202	0.661		0.378	0.391		119 45.0	2.64	304		0.530			0.231
642	128	49.0	2.60	96	208	0.202	0.656	0.621		0.344	780	129 49.0	2.62	299	69 0.202	0.573			0.209
687	128	49.0	2.62	98	209	0.202	0.661			0.382	898	117 45.0	2.60	328	66 0.202	0.574	0.298	0.255	0.240
716	128	49.0	2.62	96		0.202				0.353	2823	128 49.0	2.60	323	69 0.202	0.566	0.331	0.223	0.239
753	129	49.0	2.64	97	208	0.202	0.651			0.365	2874			323	71 0.202	0.561		0.226	0.230
775	119	45.0	2.64	98	208	0.202				0.367	2833	128 49.0	2.60	317	76 0.202	0.594	0.332	0.267	0.254
787	119	45.0	2.64	98	208	0.202	0.631			0.360	2824	117 45.0	2.60		78 0.202	U.601	0.316	0.268	0.244
820	120	45.0	2.66	94	209	0.202	0.644			0.383	2831	117 45.0	2.60	318		0.582		0.240	0.248
2806	119	45.0	2.64	185		0.202				0, 191	2875	128 49.0	2.60	322 324		0.579		0.244	0.256
643	128	49.0	2.60	169		0.202		0.513		0.181	2882	128 49.0	2.60	301		0.578	J.347	U.244	0.201
717	128	49.0	2.62	171		0.202				0.183	658	128 49.0	2.60	301		0.579	0.588		0.189
755	129	49.0	2.64	171		0.202				0.187	662	128 49.0			85 0.202		0.300		0.289
767	119	45.0	2.64	173	70	0.202	0.558			0.202	781	119 45.0	2.04	301	03 4.602	V. JO1			V.EU,

Ser,			BL	Vel	Load	Rad	In-nip	Rect	Rec2	Perm.	2969			2.60			-	0.570		0.208	0.209
Num,	FLIR	g/m²	cm ³ /9				strain	strain	strain	strain	2977	127 4			559		0.202		0.274	0.177	0.217
											2984	127 4		2.60	544 550		0.202	0.587	U.231	0.177	0.192
899	129	49.0	2.62	301	86	0.202	0.606			0.256	2760	116 4		2.60	556		0.202	0.591			0.251
908	129	49.0	2.62	299	88	0.202	0.537			0.253	2763	116 4 117 4		2.61	555		0.202	0.587	0.285		0.249
917	118	45.0	2.61	304		0.202				0.233	2793 2816	117 4		_	553		0.202	0.583			0.262
2825		45.0		316		0.202			0.295		2820	117 4		2.60	551		0.202			0.234	
2830		45.0		32%		0.202				0.266	2872	128 4			552		0.202	0.577		****	0.242
2876	128	49.0	2.60	319	_	0.202		0.362		0.276	723	128 4			500		0.202	0.520	0.5.5		0.212
2881	128	49.0		328		0.202		0.368	0.266	0.280	728	128 4			528		0.202	0.568			0.217
782			2.64	301		0.202				0.312	790	119 4			539		0.202	0.544			0.189
900		49.0		299		0.202				0.319	846	129 4			528		0.202	0.471			0.217
909		49.0	2.62	294		0.202				0.315	850	129 4			524		0.202	0.406			0.196
918		45.0	_	297		0.202				0.294		129 4		2.62	543		0.202	0.547			0.232
2826		45.0		311	_	0.202		0.372		0.314	905	129 4			531		0.202	0.528			0.236
2829		45.0		324		0.202				0.305		118 4		2.61	528		0.202	0.584			0.204
2877		49.0				0.202			0.313	0.316 0.330	2757	117 4		2.63	562		0.202	0.606	0.283		0.291
2880		49.0				0.202		0.400		0.352	2761	116 4		2.60	552	89	0.202	0.601			0,277
901		49.0		280		0.202				0.335	2762	116 4		2.60	555	89	0.202	0.605			0.266
910			2.62		_	0.202				0.333	2794	117 4			550	91	0.202	0.606	0.305	0.269	
		45.0				0.202		0.396	0.344	0.343	2817	117 4	5.0	2.60	552	91	0.202	0.606	0.289		0.280
2827		45.0	2.60	311		0.202		0.391		0.340	2821	117 4	5.0	2.60	552	91	0.202	0.602			0.260
2828		45.0	2.60	317		0.202		0.391	0.343	0.350	2873	128 4	9.0	2.60	551	94	0.202	0.596	0.332		0.261
2878		49.0	2.60	317 320		0.202		0.415		0.349	724	128 4	9.0	2.62	528		0.202	0.558			0.270
2879 920		49.0 45.0	2.61	299		0.202		0.415		0.336	7 29	128 4	9.0	2.62	528		0.202	0.621			0.279
2972			2.60	560		0.202		0.097		0.034	791	119 4					0.202	0.592			0.258
2973		48.5		556		0.202	:-			0.048	847	129 4		2.63			0.202	0.525			0.272
2980			2.60	557		0.202			0.014		851	129 4					0.202	0.460			0.259
2981			2.60	555		0.202				0.041	903	129 4					0.202	0.618			0.301 0.296
2971			2.60	564		0.202			0.054	0.070	906	129 4		2.62			0.202	0.569			0.261
2974		48.5		552		0.202			0.039	0.060	922			2.61			0.202	0.624	0.326		0.326
2979		48.5	2.60	565	17	0.202	0.353	0.143	0.054	0.072	2758	117 4		2.63			0.202			0.311	
2982		48.5	2.60	551		0.202		0,137		0.073	2795	117 4					0.202			0.309	
2967		48.5	2.60	561		0.202		0.241	0.161	0.159	2818	117 4					0.202			0.311	
2970	127	48.5	2.60	559	44	0.202	0.517	0.230	0.155	0.163	2822	117 4 128 4		2.62			0.202	0.582	0.515	0.511	0.289
2975		48.5	2.60	547	42	0.202	0.515	0.217	0.123	0.139	725 770	128 4		2.62				0.640			0.314
2978	127	48.5	2.60	562	43	0.202	0.510	0.224	0.142	0.156	730 848	129 4		2.63				0.544			0.306
2983	127	48.5	2.60	544			0.506	0.209	0.122	0.144	852	129 4		2.63				0.484			0.287
722	128	49.0	2.62	524	70	0.202				0.178	904			2.62			0.202	0.642			0.330
727	128	49.0	2.62	500		0.202	-			0.179	907	129 4		2.62			0.202	0.588			0.330
788	119	45.0	2.64	500		0.202				0.163	923	118 4		2.61				0.642			0.280
789	119	45.0	2.64	535		0.202				0.167	731	128 4		2.62	528			0.647			0.336
2759		44.5	2.60	566		0.202				0.238	849	129 4			528			0.555			0.332
2764		44.5	2.60	555		0.202	0.578			0.237	853	129 4		2.63	500			0.496			0.304
2792		45.0	2.61	564		0.202			0.242	U.24/	2767	116 4		2.60	684			0.512			0.235
2815		45.0		563		0.202		0.261		0.250 0.239	2772	116 4		2.60	876	65	0.202	0.512	0.231		0.245
2819		45.0	2.60	556		0.202				0.234	2768	116 4	_		874	74	0.202	0.519	0.240		0.254
2871		49.0	2.60			0.202	0.566	0.303	0.214	0.234 0.207		116 4			878	75	0.202	0.528	0.240		0.255
2968	127	48.5	2.60	225	65	U.2U2	U. 20Y	0.201	0.214	V. EU1											

Ser.			Bi	Vel	Load	Rad	In-nip	Rec1	Rec2	Perm.		122 45.0	2.70	171	86 0,202	0.575			0.226
Num.	JESÚ	g/m ²	cm ³ /g	m/ m	kN/m	-	strain	strain	strain	strain		122 45.0		171	87 0.202	0.538			0.201
											880	121 45.0		174	86 0.202	0.546			0.265
2773	116	44.5	2.60	870	74	0.202	0.523	0.240		0.242	883	121 45.0		177	87 0.202	0.512			0.256
		44.5	2.60	868	88	0.202	0.539	0.257		0.271	886	121 45.0		177	88 0.202	0.509			0.243
2770	116	44.5	2.60	871	88	0.202	0.539	0.257		0.272	2673	119 44.5	2.67	176	95 0.202	0.623	0.327		0.309
2774	116	44.5	2.60	867	88	0.202	0.540	0.257		0.272	2678	119 44.5	2.67	178	98 0.202	0.622		0.319	0.312
2775		44.5		862	–		0.564			0.310	2710	121 44.5		186	91 0.202	0.537	0.380	0.358	0.310
2776			2.60	861		0.202				0.325	2715	121 44.5		186	93 0.202		0.380	0.360	0.313
2777		44.5		870	-	0.202				0.315	802	122 45.0		171	136 0.202	0.616			0.274
3188			2.67	97		0.202		- 455			811 825	122 45.0 121 45.0			136 0.202				0.255
		42.2		187		0.202			0.112		827	121 45.0		173 173		0.617 0.601			0.333 0.318
		42.2		187		0.202			0.106	0.009	881	121 45.0		174	136 0.202	0.579			0.327
		42.2		186		0.202		0.139	-	0.014	884	121 45.0		177	136 0.202	0.551			0.322
		42.2		186		0.202			0.187		887	121 45.0			137 0.202	0.548			0.295
		42.2		187		0.202			0.189 0.185	0.101 0.084	2674	119 44.5	2.67		140 0.202		0 374	0.342	
		42.2 42.2		185 188		0.202				0.147	2677	119 44.5			144 0.202			0.370	0.356
3248		42.2		187		0.202		0.265	0.249	0.151	2711	121 44.5			133 0.202			0.413	0.348
3191			2.67	326		0.202		0.136		0.075	2714	121 44.5			137 0.202				
		42.4		321		0.202			0.230	0.167	803	122 45.0			172 0.202	0.631			0.300
			2.67	320		0.202			0.205	0.187	812	122 45.0	2.70	170	173 0.202	0.595			0.285
854			2.71	94		0.202		0.203	0.207	0.721	885	121 45.0	2.69	178	173 0.202	0.562			0.334
		44.5		96		0.202		0.387	0.359		2675	119 44.5	2.67	174	178 0.202	0.639	0.392	0.372	0.373
		_	2.71	96		0.202			0.351	0.304	2676	119 44.5	2.67	171	178 0.202	0.639	0.395	0.391	0.377
2705	_		2.71	95		0.202			0.378	0.327	2712	121 44.5				0.632	0.442		0.380
			2.71	95	78	0.202	0.534	0.390	0.386	0.314	2713	121 44.5			173 0.202		0.442	0.450	0.385
2863	121	45.0	2.68	95	80	0.202	0.461	0.265	0.281	0.269	804	122 45.0		171	208 0.202				0.312
855	122	45.0	2.71	94	87	0.202	0.546			0.271	813	122 45.0		169	208 0.202				0.295
888	121	45.0	2.69	101	87	0.202	0.514			0.256	805	122 45.0		323	70 0.202	0.516			0.169
2704	121	44.5	2.71	95	95	0.202	0.572	0.416	0.392		829	121 45.0		304	69 0.202	0.561		A 70/	0.205
2709	121	44.5	2.71	95	91	0.202	0.553	0.405	0.371	0.326	2722	121 44.5		320	69 0.202	0.480	0.293		0.238
			2.71	95		0.202			0.393	0.320	2723	121 44.5		320 320	69 0.202 72 0.202	0.481	0.303		0.249
		45.0	2.71	94		0.202				0.335	2733	121 44.5 121 44.5		319	71 0.202	0.449	0.262	0.292	0.249 0.243
		44.5		96					0.442		2734 2732	121 44.5		321	80 0.2 02	0.465	U.262	0.299	0.259
			2.71	96		0.202		0.448	0.428		2735	121 44.5		318	78 0.202	0.459	0.268	0.281	0.239
		45.0		93			0.605			0.361	2865	121 45.0		333	80 0.202			0.265	
		44.5	2.71	96			0.673		0.466		806	122 45.0		323	86 0.202		0.220	0.207	0.198
		44.5		96			0.670	0.477	0.466		872	121 45.0		306	86 0.202				0.175
800		45.0		171		0.202				0.183	876	121 45.0		309		0.513			0.245
809		45.0		173		0.202				0.166	2720	121 44.5		317		0.521	0.344	0.340	
828			2.68	175		0.202			0 270	0.226	2726	121 44.5		324	93 0.202	0.492	01244	0.331	0.301
2680		44.5	2.67	179		0.202			0.270 0.265	0.259	2731	121 44.5		324	96 0.202		0.292	0.308	0.272
2681		44.5	2.67	174		0.202			-	0.231	2736	121 44.5		317	92 0.202			0.308	
2717		44.5	2.71	187 186		0.202		0.353 0.349	0.337	0.277	807	122 45.0		321	136 0.202				
2718		44.5 44.5	2.71 2.67	174		0.202		0.349	0.337	0.277	873	121 45.0		301	136 0.202	0.557			0.308
2679 2716		44.5	2.71	187		0.202		0.365	0.243	0.294	877	121 45.0		309	139 0.202	0.549			0.299
2710 2719		44.5	2.71	186		0.202		0.361	0.344	0.288	2727	121 44.5		313	137 0.202	0.549		0.379	0.342
		45.0		194			0.457				2730	121 44.5	2.71	321	141 0.202	0.548		0.361	0.323
2004	16.1	77.0	£.UU	174		J.EVE	U. 731	V.E43	U, 1, 1 1	31634									

Ser.	Cat (B.U <u>.</u>	BL	Vel I	Load	Rad	In-nip			Perm.		115			93	175 0.254	0.682	0.276
Num.	pm (g/m²	cm³/g	m/m	kH/m	(1)	strain	strain	strain	strain		115		2.39	95	209 0.254	0.646	0.347
												118	-	2.45	95	209 0.254	0.679	0.344 0.321
2737	121	44.5	2.71	317	135	0.202	0.545		0.360			115		2.40	93	209 0.254	0.692	0.306
808	122	45.0	2.70	323	173	0.202	0.603			0.286	1434	115		2.40	93		0.702	0.233
874	121	45.0	2.69	304	173	0.202	0.565			0.326		114		2.38	167		0.592	0.220
878	121	45.0	2.69	306	175	0.202	0.557			0.335		118		2.45	167 166		0.609	0.162
2721	121	44.5	2.71	320	173	0.202	0.607			0.339		115			165	87 0.254	0.613	0.165
2724	121	44.5	2.71	320	172	0.202	0.608			0.330		115		2.40	204	137 0.254	0.635	0.302
2728	121	44.5	2.71	318	177	0.202	0.588			0.376		114		2.38			0.648	0.288
2729	121	44.5	2.71	320	177	0.202	0.585		0.397	0.351		118 4		2.45	168	136 0.254	0.671	0.244
879	121 4	45.0	2.69	304		0.202				0.350		115 4		2.40	163		0.671	0.240
797	122	45.0	2.70	535	69	0.202	0.539			0.133	1440	115		2.40	165 160		0.660	0.330
2866	121	45.0	2.68	562	80	0.202	0.449		0.246	0.224	1258	114		2.38 2.45	167		0.665	0.319
798	122	45.0	2.70	528	87	0.202	0.561			0.171	1295	118	-	2.40	164	175 0.254	0.695	0.276
858	122	45.0	2,71	528	87	0.202	0.523		0.246	0.221	1437	115 4 115 4		2.40	164	179 0.254	0.696	0.279
861	122	45.0	2.71	543	87	0.202	0.595			U.221	1439 1259	114		2.38	167	208 0.254	0.671	0.347
865	122	45.0	2.71	543	88	0.202				0.236	1294	118			169	209 0.254	0.680	0.345
3183	118		2.78	96		0.202		0.235			•	115		2.40	163	209 0.254	0.709	0.291
3184	118	42.4	2.78	96		0.202			0.23B		1268	113	-	2.36	292	87 0.254	0.562	0.207
3180	118		2.78	320		0.202			0.110			118		2.45	288	84 0.254	0.612	0.232
3186	118		2.78	322	_	0.202			0.116	0.070		113		2.36	289	138 0.254	0.612	0.274
3187	116		2.78	320		0.202			0.115	0.084	1280	118	-	2.45	290		0.670	0.296
3179	118		2.78	321		0.202			0.182		1270			2.36	290	176 0.254	0.639	0.309
3182	116	-		324		0.202			0.193		1281	118		2.45	280		0.694	0.330
3185	118		2.78	321		0.202		0.197			1271	113		2.36	290		0.653	0.331
3181	118		2.78	322			0.584	0.262	0.259	0.017	1282	118		2.45	292		0.707	0.357
1546		48.0		169	-	0.254				0.016	1240	115		2.39	523	87 0.254	0.556	0.198
1556		48.0	2.01	169		0.254				0.029	1283		48.0		515	86 0.254	0.617	0.238
1547		48.0	2.01	165		0.254				0.043 0.068	1410	115		2.40	520	85 0.254	0.599	0.184
1555		-	2.01	169		0.254				0.048	1241	115		2.39	500	139 0.254	0.605	0.265
1548		48.0	2.01	168		0.254				0.048	1274	113 4	48.0	2.36	518	139 0.254	0.606	0.266
1554			2.01	170		0.254				0.112	1284	118 4	48.0	2.45	500	138 0.254	0.664	0.298
1549		48.0	2.01	167		0.254				0.113	1411	115 4	48.0	2.40	500	134 0.254	0.665	0.255
1553			2.01	171		0.254				0.143	1415	115 4	48.0	2.40		140 0.254	0.674	0.260
1550			2.01	167 170			0.558			0.144	1242	115 4	48.0	2.39		175 0.254	0.626	0.295
1552		48.0 48.0	2.01 2.01	167		0.254				0.158	1273	113 4	48.0	2.36		179 0.254	0.633	0.301
1551	115		2.39	96		0.254				0.230	1285	118 4	48.0	2.45		177 0.254	0.681	0.307
1236	118		2.45	95		0.254				0.242		115 4				176 0.254	0.696	0.289
1290	115		2.40	93		0.254				0.193		115 4		2.40		178 0.254	0.698	0.285
1426	115		2.40	94		0.254				0.184	1243	115 4	48.0	2.39	520	209 0.254	0.641	0.310
1431			_	95		0.254				0.299	1272	113 4		2.36	518	210 0.254	0.646	0.311
:		48.0	2.45	95		0.254				0.313	1413	115 4		2.40	509	210 0.254	0.712	0.308
1291	118		2.40	93		0.254				0.255	1244	115 4		2.39	939	86 0.254	0.489	0.172
1427	115		2.40	93		0.254				0.247	1390	115 4	-	2.40	921	85 0.254	0.516	0.175
1432	115		2.39	96		0.254				0.329	1396	115 4		2.40	900	87 0.254	0.466	0.175
1238	115		2.45	95		0.254				0.332	1404	115 4		2.40	939	87 0.254	0.574	0.184
1292 1428	115		2.40	93		0.254				0.298	1421	115 4			939	88 0.254	0.572	0.137 0.253
		48.0		93			0.676			0.294	1245	115	48.0	2.39	A 00	138 0.254	0.542	0.233
1430	113	-0.0	2,70	,,,		3.234												

Ser.	Cal B.V.	e,	Vel 1	Load	Rad	In-nip	Rec1 Rec2	Perm.	1514	127	48.0	2.65	290	84 0.254	0.595	0.236
Num.	µm g/m²	cm ³ /g				•	strain strain		1539		48.0	2.66	298	83 0.254	0.616	0.232
	_								1540	_	48.0	2.66	296	84 0.254	0.595	0.258
1391	115 48.0	2.40	900	137 (0.254	0.558		0.240	1039	128	48.0	2.66	297	138 0.254	0.601	0.294
1395	115 48.0	2.40	900		D.254			0.236	1040		48.0	2.66	297	175 0.254	0.631	0.321
1405	115 48.0	2.40	900			0.634		0.248	1041	128	48.0	2.66	299	209 0.254	0.653	0.343
1416	115 48.0	2.40	939		0.254			0.219	1517	127	48.0	2.65	514	51 0.254	0.538	0.154
1420	115 48.0	2.40	930		0.254			0.204	1516	127	48.0	2.65	510	71 0.254	0.579	0.197
1246	115 48.0	2.39	900	_	0.254			0.269	1034	128	48.0	2.66	500	86 0,254	0.527	0.191
1392	115 48.0	2.40	900		0.254			0.264	1515	127	48.0	2.65	516	84 0.254	0.593	0.209
1394	115 48.0	2.40	900		0.254			0.272	1035	128	48.0	2.66	529	137 0.254	0.609	0.286
1417	115 48.0	2.40	948	175 (0.254	0.677		0.249	1036		48.0	2.66	500	175 0.254	0.641	0,315
1419	115 48.0	2.40	900	177 (0.254	0.670		0.235	1037		48.0	2.66	532	209 0.254	0.668	0.337
1247	115 48.0	2.39	900	209 (0.254	0.586		0.296	1000		48.0	2.71	101	85 0.254	0.584	0.257
1393	115 48.0	2.40	900	210 (D.254	0.581		0.289	1016		48.0	2.71	99		0.549	0.237
1418	115 48.0	2.40	939	208	0.254	0.696		0.265	1091		45.0		95		0.527	0.232
1482	124 48.0	2.59	308	50 (0.254	0.516		0.193	1115		45.0	2.68	93		0.533	0.234
1483	124 48.0	2.59	308	69 (0.254	0.545		0.220	1117		45.0	2.68	94		0.533	0.222
1484	124 48.0	2.59	307	84 (D.254	0.564		0.250	1172		45.0	2.70	93	84 0.254	0.544	0.207
1485	124 48.0	2.59	307	135 (0.254	0.606		0.312	1001		48.0	2.71	100		0.644	0.328
1486	124 48.0	2.59	306		D.254			0.332	1017		48.0	2.71	99	136 0.254	0.605	0.309
1487	124 48.0	2.59	307		D.254			0.344	1092		45.0	2.68	90	138 0.254	0.613	0.319
1488	124 48.0	2.59	513	-	D.254			0.163	1116		45.0	2.68	93	137 0.254	0.616	0.297
1489	124 48.0	2.59	514			0.572		0.208	1118		45.0	2.68	93	138 0.254	0.616	0.298
1518	127 48.0	2.65	93			0.525		0.192	1136		45.0	2.71	95	132 0.254	0.619	0.302
1499	127 48.0	2.65	93			0.514		0.251	1173		45.0	2.70 2.71	94	138 0.254	0.607	0.288
1519	127 48.0	2.65	93			0.558		0.225	1002 1018		48.0 48.0	2.71	100 98	175 0.254 175 0.254	0.679 0.638	0.352 0.329
1058	128 48.0	2.66	96	-		0.510		0.195	1093		45.0		90	177 0.254	0.645	0.350
937	126 48.0	2.63	97		0.254			0.224	1119	_	45.0	2.68	94	174 0.254	0.644	0.321
1500	127 48.0	2.65	93			0.531		0.257	1137		45.0		95	177 0.254	0.656	0.343
1520	127 48.0	2.65	93		D.254			0.249	1174		45.0	2.70	92	178 0.254	0.632	0.311
938	126 48.0	2.63	97		0.254			0.296	1003	_	48.0	2.71	100	209 0.254	0.706	0.368
1059	128 48.0	2.66	96		0.254			0.288	1019		48.0	2.71	99	210 0.254	0.666	0.345
1521	127 48.0	2.65	93		D.254			0.299 0.335	1094	_	45.0	2.68	94	211 0.254	0.664	0.366
939	126 48.0	2.63	97			0.675		0.333	1120		45.0	2.68	94	211 0.254	0.666	0.342
1060	128 48.0	2.66	96		0.254			0.326	1138		45.0	2.71	95	209 0.254	0.673	0.360
1522	127 48.0	2.65	94 97		0.254 0.254	-		0.344	976		48.0	2.67	174	87 0.254	0.499	0.232
940	126 48.0	2.63	96		-	0.647		0.344	1004		48.0	2.71	177	87 0.254	0.583	0.242
1061 1523	128 48.0	2.65	93	_	0.254 0.254			0.335	1020		48.0	2.71	172	87 0.254	0.545	0.227
	127 48.0 127 48.0		172	_	0.254 0.254			0.189	1107		45.0	2.68	160	85 0.254	0.543	0.238
1503 1508		2.65	173		D.254			0.191	1152		45.0	2.71	169	84 0.254	0.561	0.228
7 .	127 48.0	2.65	170	-	0.254 0.254			0.234	1160		45.0	2.71	167	84 0.254	0.535	0.216
1502	127 48.0	2.65			D. 254			0.259	977		48.0	2.67	173	137 0.254	0.559	0.294
1501	127 48.0	2.65	172 170		U. 254 D. 254			0.259	1005		48.0	2.71	176	137 0.254	0.639	0.309
1509	127 48.0	2.65	171		U.254 D.254			0.243	1021		48.0	2.71	172	138 0.254	0.601	0.289
1510	127 48.0	2.65	291		0.254 0.254			0.243	1108		45.0	2.68	164	136 0.254	0.617	0.302
1512	127 48.0		290		0. <i>2</i> 54 0.254			0.209	1153		45.0	2.71	166	138 0.254	0.634	0.295
1511	127 48.0	2.65	303		D.254			0.209	1161		45.0	2.71	166	137 0.254	0.613	0.295
1038	128 48.0 127 48.0	2.66	288			0.595		0.236	978			2,67		175 0.254		0.309
1513	127 40.0	2.03	200	04 (v.2.34	V.J7J		0.2.30								

Ser.	Cal B	. u .	BL	Vel !	Load	Red	In-nip		RecZ	Perm.	1100	121 45.0			139 0.254	0.599 0.532		0.276 0.255
Num.	pan g	/m²	cm ³ /g	m/m	kN/m	m	strain	strain	strain	strein	1134	122 45.0		500	137 0.254	0.532		0.250
											1144	122 45.0	2.71 2.72	512 500	138 0.254 139 0.254	0.618		0.293
1006	130 4	8.0	2.71	175	176	0.254	0.672			0.343	1201	122 45.0 122 45.0		500	138 0.254	0.593		0.270
1022	130 4	6.0	2.71	171	175	0.254	0.630			0.319	1209 986		2.67	538	176 0.254	0.569		0.297
1109	121 4	5.0	2.68	165	175	0.254	0.646			0.338	1101	128 48.0 121 45.0		500	175 0.254	0.626		0.305
1154	122 4	5.0	2.71	166		0.254				0.324	1145	122 45.0		520	175 0.254	0.632		0.307
1162	122 4	5.0	2.71	166		0.254				0.323	1202	122 45.0		500	177 0.254	0.652		0.327
979	128 4	8.0	2.67	173		0.254				0.334	1210	122 45.0	2.72	500	177 0,254	0.632		0.303
1007	138 4		2.71	176		0.254				0.362	987	128 48.0		541	209 0.254	0.592		0.322
1023	130 4		2.71	172		0.254				0.335	1146	122 45.0		518	209 0.254	0.655		0.334
1110	121 4		2.68	165	_	0.254				0.350	1203	122 45.0	2.72	500	210 0.254	0.674		0.339
1155	122 4		2.71	166		0.254	0.686			0.342	1211	122 45.0		515	210 0.254	0.645		0.311
1163	122 4		2.71	165		0.254				0.337	1177	121 45.0	2.69	921	85 0.254	0.475		0.202
980	128 4		2.67	305	_	0.254	-			0.217	1185	121 45.0		900	85 D.254	0.520		0.191
1008	130 4		2.71	307	_	0.254	_			0.220	1207	122 45.0		939	83 0.254	0.540		0.186
1114	121 4		2.68	284	-	0.254				0.206	1178	121 45.0	2.69	939	137 0.254	0.545		0.265
1151	122 4		2.71	280		0.254				0.224 0.192	1186	121 45.0	2.69	948	136 0.254	0.5B9		0.259
1167	122 4		2.71	293		0.254					1206	122 45.0		921	139 0.254	0.618		0.254
1189	121 4		2.69	285		0.254				0.197	1179	121 45.0		912	192 0.254	0.581		0.299
981	128 4		2.67	303		0.254				0.282 0.285	1187	121 45.0	2.69	900	176 0.254	0.623		0.288
1009	130 4		2.71	313		0.254				0.279	1205	122 45.0	2.72	930	178 0.254	0.653		0.293
1113	121 4		2.68	287		0.254				0.274	1180	121 45.0	2.69	900	210 0.254	0.591		0.307
1166	122 4		2.71	290		0.254				0.273	1188	121 45.0	2.69	900	209 0.254	0.647		0.299
1190	121 4			288		0.254				0.306	1204	122 45.0	2.72	958	210 0.254	0.681		0.316
982	128 4		2.67	305 311		0.254				0.320	2586	96 45.0		91	97 0.355	0.463	0,207	0.084
1010	130 4 121 4		2.71 2.68	287		0.254				0.318	2610	96 45.0	2.12	91	132 0.355	0.462	0.228	0.161
1112	122 4		2.71	296		0.254				0.294	2159	96 44.5	2.17	297	97 0.355	0.475	0.129	0.015
1165 1191	121 4		2.69	280		0.254				0.306	2160	96 44.5	2.17	302	97 0.355	0.482	0.123	0.016
1212	128 4		2.67	300		0.254				0.291	2167	96 44.5	2.17	303	98 0.355	0.477	0.135	0.016
1213	128 4		2.67	300		0.254				0.291	2587	96 45.0		303	97 0.355	0.454	0.175	0.071 0.082
1214	128 4		2.67	300		0.254				0.291	2593	94 45.0	2.08	304	97 0.355	0.421	0.100	0.085
1215	128 4	-	2.67	300		0.254				0.291	2594	94 45.0		302	97 0.355	0.422	0.106	0.033
1216	128 4		2.67	300		0.254				0.291	2600	96 45.0		296	98 0.355	0.420	0.155 0.160	0.091
1217	128 4		2.67	300		0.254				0.291	2601	96 45.0	2.12	304	99 0.355	0.501	0.161	0.045
983	128 4		2.67	303		0.254				0.326	2161	96 44.5		302	133 0.355	0.496	0.155	0.039
1011	130 4		2.71	305		0.254				0.333	2166	96 44.5	2.17	306	135 0.355 137 0.355	0.498	0.144	0.102
1111	121 4		2.68	280	208	0.254				0.341	2588	94 45.0	2.08	301 300	135 0.355	0.460	0.129	0.096
1164	122 4		2.71	296	210	0.254	0.663			0.316	2592	94 45.0	2.08	297	135 0.355	0.461	0.180	0,122
1192	121 4			285		0.254				0.330	2599	96 45.0	2.12 2.12	301	134 0.355	0.458	0.182	0.109
1098	121 4		2.68	500	82	0.254	0.513			0.204	2602	96 45.0	2.12	302	135 0.355	0.458	0.185	0,120
1133	122 4		2.70	520	82	0.254	0.441			0.174	2604	96 45.0 96 45.0	2.12	313	135 0.355	0.462	0.191	0.123
974	128 4		2.67	550	85	0.254	0.504			0.214	2606	96 44.5	2.17	300	178 0.355	0.525	0.183	0.064
984	128 4		2.67	500	87	0.254	0.469			0.199	2162 2165	96 44.5		301	180 0.355	0.533	0.186	0.074
1099	121 4		2.68	515		0.254				0.200	2589	94 45.0	2.08	303	185 0.355	0.533	0.170	0.137
1143	122 4		2.71	520		0.254				0.199	2603	96 45.0	=	300	181 0.355	0.497	0.205	0.141
1208	122 4	5.0	2.72	529		0.254				0.196	2605	96 45.0	2.12	302	181 0.355	0.496	0.205	0.132
975	128 4		2.67	547		0.254				0.278	2609	96 45.0		302	185 0.355	0.501	0.216	0.149
985			2.67	535	138	0.254	0.534			0.269	2007	,,, 4,,0						
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Ser.	Cal	B.W.	B _i	Vel	Load	Rad	In-nip	Reci	RecZ	Perm.	2319	116 44	.0	2.64	89	22 0.355			0,122
Hum.		g/= ²	cm ³ /g				-			strain		116 44		2.64	89	25 0.355			0.123
	•										2333	116 44	.0	2.64	89	22 0.355			0.117
2163	96	44.5	2.17	301	209	0.355	0.537	0.196		0.088	2334	116 44	.0	2.64	89	22 0.355			0.115
2164			2.17							0.090	2523	117 45	.0	2.60	92	26 0.355	0.381	0.163	0.141
2590			2.08	302		0.355				0.136	2524	117 45	.0	2.60	90	17 0.355	0.315	0.132	0.116
2591		45.0		303		0.355		-		0.133	2525	117 45	.0	2.60	91	25 0,355	0.379	0.167	0.141
2607			2.12			0.355				0.156	2022	121 45	.9	2.64	89	96 0.355	0.600	0.304	0.296
2608		45.0		302		0.355				0.141	2028	121 45	.9	2.64	91	96 0.355	0.574	0.315	0.304
2168	96	44.5	2.17	519		0.355		0.186		0.041	2030	121 45	.9	2.64	86	95 0.355	0.561	0.318	0.297
2174	96	44.5	2.17	525	96	0.355	0.561	0.159		0.031	2286	117 44		2.63	89	97 0.355	0.558		0.268
2169	96	44.5	2.17	516	130	0.355	0.559	0.203		0.060	2290	117 44		2.63	89	97 0.355			0.256
2173	96	44.5	2.17	523	133	0.355	0.593	0.184		0.067	2297	116 44		2.64	89	97 0.355			0.305
2175	96	44.5	2.17	517	129	0.355	0.578	0.177		0.061	2299	116 44	_	2.64	89	98 0.355	0.534		0.299
2611	96	45.0	2.12	530	131	0.355	0.452	0.177		0.142	2308	116 44		2.64	89	96 0.355			0.296
2616	96	45.0	2.12	524	135	0.355	0.450	0.157		0.143	2309	116 44	-	2.64	90	96 0.355			0.300
2617	96	45.0	2.12	519	135	0.355	0.451	0.157		0.134	2316	116 44		2.64	89	95 0.355			0.265
2170	96	44.5	2.17			0.355	0.636			0.096	2321	116 44		2.64	90	94 0.355			0.246
2172	96	44.5	2.17			0.355	0.617	0.217		0.081	2331	116 44		2.64	89	93 0.355			0.255
2176	96	44.5	2.17			0.355	0.595	0.189		0.089	2023	121 45		2.64	90	133 0.355			0.338
2612				516		0.355		0.186		0.149	2027	121 45		2.64	89	136 0.355		0.344	0.339
2615	96	45.0	2.12	523		0.355	0.492	0.185		0.160	2287			2.63	90	134 0.355	0.580		0.289
2171		44.5	2.17	519		0.355	0.640	0.219		0.106	2310	116 44	-	2.64	89	133 0.355			0.323
2177			2.17			0.355		0.207		0.109	2315	116 44		2.64	89	134 0.355	A 574	A 77A	0.302
2613			2.12			0.355	-			0.172	2469	118 45	-	2.63	90		0.571		0.302
2614			2.12			0.355				0.172	2512	120 45 117 45		2.66	93	129 0.355	0.585		0.328
2618				514		0.355				0.165	2522 2024	121 45		2.64	89	136 0.355 179 0.355	U.340	U.310	0.376
2619		45.0		521			0.507			0.155	2026	121 45		2.64	90	182 0.355	A 455	0.350	0.363
2654			2.53	91	_	0.355			0.138		2288	117 44	_	2.63		179 0.355		0.337	0.363
2655		45.0		91		0.355			0.174	0.074	2311	116 44		2.64	89	179 0.355	0.377		0.329
2656		45.0		92		0.355		0.211	0.217	0.126	2314	116 44		2.64	89	180 0.355			0.328
2657		45.0		91		0.355			-	0.147	2470	118 45		2.63	89	175 0.355	n 507	0.351	. 0.325
2653		45.0		91		0.355		0.341		0.269	2025	121 45		2.64	89	210 0.355	4.571	0.398	0.383
		45.0	2.53	92		0.355			0.354	0.276	2289	117 44		2.63	89	209 0.355	0.611		0.319
2664		45.0		174		0.355			0 153	0.250	2298	116 44		2.64	90	211 0.355	0.596	0.277	0.372
			2.53	302		0.355	-		0.152			116 44		2.64	89	210 0.355	01375		0.346
2660		45.0		301		0.355		0.137		0.063 0.101	_	116 44		2.64	89	210 0.355			0,339
		45.0		303		0.355		0.196	0.199	-		116 44		2.64	89	210 0.355			0.330
2661		45.0		301		0.355	-	U. IYI	0.187		2335	116 44		2.64	88	210 0.355			0.323
2666				922		0.355			0.132 0.083	0.112 0.072	2336	116 44		2.64	89	210 0.355			0.318
		45.0		921		0.355			0.096	0.072	2328	116 44		2.64	171	23 0.355			0.098
			2.53	921 919		0.355			0.136	0.077	2329	116 44	-	2.64	174	23 0.355			0.094
		45.0				0.355	0.397		0.135	0.070	2459	118 45	-	2.63	175	23 0.355	0.386	0.125	0.103
2670		45.0		923		0.355				0.219	2460	118 45		2.63	174	20 0.355	0.363	0.114	0.101
2665 2029		45.0 45.9	2.53 2.64	927 90	-	0.355		0.181	U.240	0.219	2461	118 45		2.63	175	20 0.355	0.361	0.116	0.094
		44.5		88		0.355		0.101		0.137	2462	118 45		2.63	175	23 0.355	0.385	0.128	0.095
2300		44.0	2.64	89		0.355				0.156	2527	117 45	-	2,60	175	25 0,355	0.368	0.149	0.122
2317			2.64	88		0.355	0.767			0.137		117 45	.0	2.60	176	18 0.355	0.306	0.117	0.088
		44.0		89		0.355				0.117	2529	117 45	.0	2.60	177	25 0.355	0.366	0.148	0.115
EJ 10	, 10	77.0	2.04	U7		4.333				v. 111									

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Ser.	Cal	8.V.	8.	Vel	Lcad	Rad	In-nip	Rec1	Rec2	Perm.	2483	116	45.0	2.63			0.509		0.198
Nus.		g/m²	cm ³ /g							strain	2513		45.0	2.66	536	93 0.355	0.538		0.195
		2 ,	, a			•					2520		45.0	2.66	516	91 0.355	0.534		0.185
2322	116	44.0	2.64	177	94	0.355				0.223	2066		45.9	2.64	519	136 0.355	0.580	0.476	0.276
2327		44.0	2.64	172		0.355				0,220	2070		45.9	2.64	527	139 0.355	0.571	0.468	0.270
2330		44.0	2.64	176		0.355				0.214	2473		45.0	2.63	527	127 0.355	0.542		0.250
2526		45.0	2.60	177		0.355	0.509	0.273		0.267	2067		45.9	2.64	520	182 0.355	0.598	0.487	0.295
2458		45.0	2.63	175		0.355		0.315		0.293	2069		45.9	2.64	526	186 0.355	0.595	0.482	0.303
2324		44.0	2.64	174		0.355				0.299	2068		45.9	2.64	521	211 0.355	0.604	0.492	0.310
2325		44.0	2.64	174		0.355				0.301	2303		44.0	2.64	521	210 0.355	0.586		0.306
2326		44.0	2.64	174		0.355				0.304	2305		44.0	2.64	522	211 0.355	0.566		0.343 0.106
2011		45.9	2.65	309		0.355	0.371			0.108	2095		45.9	2.63	937	23 0.355	0.475		0.100
2012		45.9	2.65	299	22	0.355	0.375			0.115	2088		45.9	2.63	928	96 0.355	0.569		0,251
2464		45.0	2.63	298		0.355	0.376			0.107	2094		45.9	2.63	930	98 0.355	0.551		0.239
2465		45.0	2.63	301	20	0.355	0.350			0.094	2096		45.9	2.63	912	97 0.355	0.559 0.555		0.225
2466		45.0	2.63	300	20	0.355	0.356			0.095	2097	_	45.9	2.63	921	97 0.355 97 0.355	0.552		0.213
2467		45.0	2.63	300	22	0.355	0.379			0.100	2098		45.9	2.63	926		0.592		0.213
2004		45.9	2.65	301	98	0.355	0.621			0.283	2089		45.9	2.63	924	137 0.355	0.592		0.269
2010	122	45.9	2.65	303	99	0.355	0.605			0.276	2093		45.9	2.63 2.63	924 913	179 0.355	0.614		V.LU/
2013	122	45.9	2.65	292	98	0.355	0.586			0.264	2090		45.9	2,63	931	183 0.355	0.612		0.314
2005	122	45.9	2.65	299	134	0.355	,	4		0.310	2092		45.9	2.63	919	210 0.355	0.619		0.330
2009	122	45.9	2.65	303	137	0.355	0.658	V.,		0.298	2091		45.9	2.73	91	11 0.355	0.302		0.067
2463	118	45.0	2.63	301	133	0.355		i i		0.293	2651 2049		45.9	2.67	88	23 0.355	0.465	0.196	0.150
2468	118	45.0	2.63	300		0.355	0.571	0.294		0.276	2124		44.5	2.69	92	23 0.355	0.468	0.153	0.150
2006	122	45.9	2.65	302	179	0.355				0.345	2236		45.0	2.73	95	22 0.355	0.404	0.125	0.134
2008	122	45.9	2.65	303		0.355				0.335	2043		45.9	2.67	90	97 0.355	0.538	0.328	0.300
2007	122	45.9	2.65	301	210	0.355				0.352	2048		45.9	2.67	91	97 0.355	0.519	0.335	0.313
2072	121	45.9	2.64	534	20	0.355	0.467	0.341		0.094	2117		44.5	2.69	89	98 0.355	0.534	0.305	0.324
2306		44.0	2.64	513	18	0.355	0.364			0.124	2123		44.5	2.69	90	98 0.355	0.516	0.290	0.300
2307		44.0	2.64	520	22	0.355	0.389			0.352 0.094 0.124 0.128 0.089 0.085 0.092 0.102	2133		44.5	2.69	69	96 0.355	0.513	0.279	0.282
2474	118	45.0	2.63	519	28	0.355	0.385			0.089	2134		44.5	2.69	89	96 0.355	0.510	0.276	0.274
2475			2.63	524	25	0.355	0.373			0.085	2229		45.0	2.73	90	94 0.355	0.547	0.262	0.278
2476		45.0	2.63	519	25	0.355	0.369			0.092	2235		45.0	2.73	92	95 0.355	0.539	0.267	0.271
2477		45.0	2.63	520							2237		45.0	2.73	86	94 0.355	0.533	0.271	0.280
2479		45.0	2.63	521		0.355				0.085	2244		45.0	2.73	90	92 0.355	0.631	0.272	0.280
2480		45.0	2.63	520		0.355				0.080	2340		44.0	2.73	89	97 0.355	0.552		0.299
2481		45.0	2.63	520		0.355				0.081 0.084	2343		44.0	2.73	90	97 0.355	0.551		
2482		45.0	2.63	520		0.355	0.382			0.086	2346		45.0	2.67	89	96 0.355	0.530	0.234	
2514		45.0	2.66	526		0.355	0.419			0.066	2347	120	45.0	2.67	89	96 0.355	0.609		0.314
2515		45.0	2.66	522		0.355	0.396	0.077		0.078	2348	120	45.0	2.67	89	97 0.355	0.486		0.316
2516		45.0	2.66	524		0.355	0.395	0.073		0.078	2352		45.0	2.67	89	95 0.355	0.470		0.299
2517	_	45.0	2.66	523		0.355	0.416	0.085		0.086	2356		45.0	2.70	90	96 0.355	0.512		0.290
2518		45.0	2.66	526		0.355		0.069		0.095	2367	122	45.0	2.72	89	98 0.355	0.566	0.412	0.338
2519		45.0	2.66	522		0.355	0.413	0.074 0.456		0.236	2368	122	45.0	2.72	89	97 0.355	0.563	0.404	0.314
2065		45.9	2.64	522		0.355				0.243	2396	115	44.0	2.71	89	96 0.355	0.539	0.437	0.269
2071		45.9	2.64	526		0.355		0.450		0.254	2398	119	44.0	2.71	89	95 0.355	0.540	0.431	0.268
2301		44.0	2.64	517	_	0.355				0.247	2401	115	44.0	2.71	89	94 0.355	0.530	0.425	0.246
2302		44.0	2.64	519		0.355	0.528			0.283	2431		45.0	2.76	90	95 0.355	0.543	0.308	0.286
2304		44.0	2.64	515	-	0.355				0.207	2622	122	45.0	2.71	91	98 0.355	0.504		
2478	118	45.0	2.63	521	71	درر. ی	0.501												

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Ser.	Cal	B.W.	Bi	Vel	Loed	Rad	In-nip	Rec1	Rec2	Perm.	2397	119	44.0	2.71	301	95 0.355	0.533	0.392	0.231
Num.		g/m²	cm ³ /g				-			strain	2399		44.0	2.71	300	94 0.355		0.386	0.242
	•	-	• •	-				•			2468	119	44.0	2.71	299	95 0.355	0.526	0.373	0.218
2623	122	45.0	2.71	91	98	0.355	0.497				2413	119	44.0	2.71	303	94 0.355	0.527	0.377	0.210
2647	123	45.0	2.73	92	98	0.355	0.565	0.346	0.267	0.257	2430	124	45.0	2.76	302	95 0,355	0.535	0.258	0.265
2044	123	45.9	2.67	89	133	0.355	0.559	0.354		0.339	2574		45.0	2.73	302	94 0.355	0.571	0.313	0.216
2118	120	44.5	2.69	88	134	0.355	0.557	0.332		0.352	2575		45.0		302	94 0.355	0.570	0.314	0.216
2122	120	44.5	2.69	91	137	0.355	0.554			0.356	2576		45.0	2.73	302	93 0.355	0.564	0.318	0.216
2230	123	45.0	2.73	89	130	0.355	0.584	0.294		0.311	2577		45.0	2.73	301	93 0.355	0.562	0.318	0.216
2234		45.0	2.73	92		0.355		0.292		0.299	2578		45.0	2.73	301		0.558	0.316	0.216
2648		45.0	2.73	91	_	0.355		0.375	0.297		2579		45.0	2.73	301	92 0.355	0.555	0.313	0.216
2045		45.9	2.67	89		0.355		0.379		0.368	2580		45.0	2.73	301	92 0.355	0.551	0.309	0.216
2119		44.5	2.69	67		0.355				0.379	2581		45.0		301		0.551		0.216
2121		44.5	2.69	91		0.355				0.389	2582 2583		45.0 45.0	2.73	301	91 0.355	0.549	0.301	0.216
2231		45.0	2.73	89		0.355				0.330	2051		45.9	2.73 2.67	301 295	91 0.355 132 0.355	0.548 0.530	0.295	0.216
2233		45.0	2.73	90		0.355		0.315		0.326	2130		44.5	2.69	295	131 0.355	0.536	0.318	0,315 0,296
2649		45.0	2.73	91		0.355			0.321		2394		44.0	2.71	300	134 0.355	0.572	0.424	0.256
2046			2.67	89		0.355		0.462		0.385	2400		44.0	2.71	301	131 0.355	0.568	0.417	0.271
2120			2.69	88			0.590	0.333		0.394 0.345	2402		44.0		301	130 0.355	0.565	0.417	0.252
2232		45.0 45.0		88 92		0.355	0.650		0.325		2407		44.0	2.71	301	133 0.355	0.564	0.410	0.256
2650 2125			2.69	174		0.355			0.323	0.289	2052			2.67	301	177 0.355	0.556	0.331	0.336
2349		45.0	2.67	174		0.355		0.213		0.207	2131		44.5	2.69	298	176 0.355	0.555		0.316
2357				175		0.355				0.267	2395	119	44.0	2.71	301	180 0.355		0.448	0,281
2126	. — .	44.5	2.69	175		0.355				0.312	2403	119	44.0	2.71	297	176 0.355	0.597	0.437	0.283
		45.0	2.76	175		0.355		0.300		0.297	2406	119	44.0	2.71	301	179 0.355	0.599	0.437	0.297
2127		44.5	2.69	173		0.355				0.345	2053	123	45.9	2.67	301	209 0.355	0.565		0.340
2128			2.69	173		0.355				0.357	2132		44.5	2.69	299	209 0.355	0.568		0.332
2409	119	44.0	2.71	298	26	0.355	0.390	0.232		0.097	2404		44.0	2.71	299	209 0.355	0.616		0.310
2410	119	44.0	2.71	300	23	0.355	0.367	0.219		0.085	2405		44.0	2.71	300	209 0.355	0.612	0.446	0.305
2411	119	44.0	2.71	299	23	0.355	0.367	0.218		0.086	2266		45.0	2.73	541	19 0.355	0.467		0.093
2412	119	44.0	2.71	299		0.355		0.235		0.091	2274		45.0	2.73	534	16 0.355			0.130
2050		45.9	2.67	312		0.355		0.267		0.272	2275		45.0 45.0	2.73	524 534	16 0.355 24 0.355	0.748	0 170	0.125
2129		44.5	2.69	306		0.355	0.514			0.267	2531 2532		45.0	2.69 2.69	521	19 0.355	0.368	0.120 0.099	0.135 0.112
2137		44.5	2.76	294		0.355		0.374		0.194	2533		45.0		525		0.323	0.097	0.112
2138		44.5	2.76	301		0.355		0.365		0.194	2534		45.0	2.69	523	24 0.355		0.106	0.123
2139			2.76	302		0.355	_	0.359		0.194 0.194	2255		45.0	2.73	527		0.601	000	0.215
2140		44.5	2.76	301 302		0.355 0.355	0.613 0.594	0.352 0.347		0.194	2055		45.9	2.67	532	96 0.355	0.496		0.247
2141		44.5		304		0.355		0.347		0.194	2114		44.5	2.69	533		0.549		0.275
2142 2143		44.5 44.5	2.76 2.76	301		0.355		0.336		0.194	2135		44.5	2.69	533	96 0.355	0.502		0.230
2145			2.76	300		0.355	0.551	0.322		0.194	2245		45.0	2.73	532		0.622		0.234
2146		44.5	2.76	301		0.355		0.315		0.194	2263	123	45.0	2.73	538	98 0.355	0.589		0.241
2147		44.5	2.76	300		0.355	0.529	0.309		0.194	2265	123	45.0	2.73	526	101 0.355	0.539		0.244
2344		44.0	2.73	296		0.355		0.507		••••	2267	123	45.0	2.73	505	99 0.355	0.543		0.242
2350		45.0	2.67	298		0.355				0.287	2273	123	45.0	2.73	537	100 0.355			0.280
2354		45.0	2.67	301		0.355	0.464			0.252	2276	123	45.0	2.73	503	99 0.355			0.260
2358		45.0	2.70	297		0.355				0.254	2277		45.0	2.73	543	98 0.355	0.581		
_		44.0	2.71	301		0.355		0.392		0.233	2278		45.0	2.73	537	98 0.355	0.598		
	119	44.0		302			0.538	0.396		0.234	2279	123	45.0	2.73	526	98 0.355	0.585		

Ser.	Cat B.V.	-			Rec1 Rec2			122 45.0				0.546		0.323
Num.	pm g/m²	cm³/g m/m	n kH/m m	strain	strain strain	strain	2637	122 45.0		918	176 0.355	0.540		0.293
							2640	122 45.0	2.71	926	180 0.355	U.340		0.295
2280	123 45.0	2.73 518	8 98 0.35	5 0.586	•		2363	121 45.0		916	198 0.355 207 0.355	0 500		0.292 0.295
2351	120 45.0	2.67 515	5 95 0.39	5 0.469	•	0.264	2435	124 45.0		917		0.598 0.598		0.292
2353	120 45.0	2.67 533	3 95 0.39	5 0.465		0.243	2436	124 45.0	2.76	923		0.566		0.334
2359	121 45.0	2.70 518	96 0.3 5	5 0.486		0.247	7626	122 45.0	2.71	918 923	209 0.355 209 0.355	0.560		0.329
2535	121 45.0	2.69 514	97 0.35	5 0.520	0.245	0,247	2627	122 45.0		921		0.563		0.312
2115	120 44.5	2.69 515	5 133 0.35	5 0.574		0.307	2638	122 45.0 122 45.0	2.71 2.71	922		0.561		0.310
2246	123 45.0	2.73 514	• 127 O.35	5 0.656		0.261	2639	126 45.0	2.79	90	22 0.355	0.413	0.186	0.143
2256	123 45.0	2.73 517				0.238	2448	126 45.0	2.79	89	19 0.355	0.394	0.176	0.139
2530	121 45.0					0.317	2449 2450	126 45.0		90	20 0.355	0.395	0.178	0.150
2536	121 45.0					0.293	2451	126 45.0	2.79	90	23 0.355	0.421	0.192	0.159
2537	121 45.0					0.284	2486	126 45.0	2.81	90	18 0.355	0.381	0.204	0.103
	123 45.0					0.289	2487	126 45.0	2.81	90		0.359	0.195	••••
		2.67 516				0.342	2488	126 45.0	2.81	89		0.360	0.194	0.088
_	123 45.0					0.307	2489	126 45.0	2.81	90	19 0.355	0.384	0.209	0.098
2249	123 45.0					0.299	2492	126 45.0		89	23 0.355	0.427	0.229	0.142
2264	123 45.0					0.316	2493	126 45.0		90	20 0.355	0.404	0.214	0.120
	123 45.0					0.299	2494	126 45.0	2.79	90	20 0.355	0.405	0.214	0.129
_	122 45.0			7 27222		0.094	2495	126 45.0	2.79	90	24 0.355	0.425	0.226	0.132
2633	122 45.0					0.080	2373	122 44.0	2.78	90	97 0.355	0.600		0.344
2441	124 45.0					0.105	2419	125 44.0	2.84	90	96 0.355	0.545	0.321	0.324
2442	124 45.0					0.102	2447	126 45.0	2.79	90	98 0.355	0.563	0.356	0.296
2631	122 45.0					0.134	2457		2.79	90	97 0.355	0.555	0.339	0.272
2634	122 45.0					0.121	2485	126 45.0	2.81	90	101 0.355	0.555	0.348	0.226
2643		2.71 927				0.114	2490	126 45.0	2.81	90	98 0.355	0.551	0.362	0.228
2644	122 45.0					0.117 0.223	2491	126 45.0	2.79	90	97 0.355	0.553	0.349	0.266
2238		2.73 935		-		0.225	2496	126 45.0	2.79	89	95 0.355	0.553	0.353	0.243
2360		2.70 926				0.216	2497	126 45.0	2.79	90	133 0.355	0.604	0.354	
	124 45.0	2.76 923 2.76 919				0.210	2499	126 45.0	2.79	90	127 0.355	0.603	0.355	0.323
2440 2443	124 45.0			-		0.212	2503	126 45.0	2.79	89	128 0.355	0.607	0.349	0.314
2445		2.76 923				0.206	2511	126 45.0	2.79	92	125 0.355	0.598	0.326	0.293
2630	122 45.0					0.263	2382	122 44.0	2.78	176	23 0.355	0.426	0.319	0.143
2635		2.71 916				0.234	2383	122 44.0	2.78	175	20 0.355	0.402	0.301	0.127
2642	122 45.0					0.224	2384	122 44.0	2.78	175	20 0.355	0.401	0.297	0.124
2645		2.71 914				0.209	2385	122 44.0		174	23 0.355	0.424	0.320	0.150
2361	121 45.0			-		0.264	2389	122 44.0	2.78	173	21 0.355	0.401	0.294	0.105
2433	124 45.0					0.258	2428	125 44.0	2.84	174	25 0.355	0.418	0.152	0.142
2438	124 45.0					0.249	2429	125 44.0	2.84	174	22 0.355		0.139	0.120
2624	122 45.0					0.288	2452	126 45.0		176	23 0.355	0.410	0.175	0.139
2629	122 45.0					0.287	2453	126 45.0	2.79	175	20 0.355	0.390	0.158	0.127 0.130
2636	122 45.0					0.269	2454	126 45.0	2.79	175	20 0.355	0.394	0.164	0.142
2641		2.71 926				0.265	2455	126 45.0	2.79	175	23 0.355		0.175	0.313
2362	121 45.0			_		0.274	2374	122 44.0	2.78	175	97 0.355	0.589		0.278
2364	121 45.0					0.267	2381	122 44.0	2.78	174	97 0.355	0.566	0.457	0.267
2434	124 45.0					0.280	2386	122 44.0	2.78	174	95 0.355		0.458	0.248
2437	124 45.0	2.76 923	5 176 0.39			0.278	2388	122 44.0		173	96 0.355 os n 355	0.554	0.452	0.237
2625	122 45.0	2.71 920	0 177 0.35	5 0.546		0.311	2390	122 44.0	2.78	176	95 0.355	U. J.74	V.7/L	

•	Ser.	Cal	0.V.	B.,	Vel	t oad	Rad	In-nip	Rec1	Rec2	Perm.
1	lun.	ji m	g/m²	cm ³ /g					strain	strain	strain
4	2420	125	44.0	2.84	174	04	0.355	0.542	0.308		0.290
	2427		44.0	2.84	175		0.355		0.300		0.256
	2456		45.0	2.79	176		0.355		0.323		0.254
	2375		44.0	2.78	175		0.355		0.263		0.330
	2380		44.0	2.78	174		0.355		0.499		0.300
			44.0				0.355		0.334		0.325
	2421			2.84	176						
	2426		44.0	2.84	174		0.355		0.321		0.303
	2500		45.0	2.79	173		0.355		0.323		0.310
	2504		45.0	2.79	175		0.355		0.314		0.292
	2376		44.0	2.78	175		0.355				0.349
	2379		44.0	2.78	175		0.355		_		0.329
1	2422	125	44.0	2.84	175	• • •	0.355		0.354		0.344
- 1	2425	125	44.0	2.84	175		0.355		0.350		0.333
1	2377	122	44.0	2.78	175	200	0.355	0.645	0.550		0.353
1	2378	122	44.0	2.78	175	200	0.355	0.643	0.543		0.348
- 1	2387	122	44.0	2.78	173	204	0.355	0.632	0.525		0.334
:	2423	125	44.0	2.84	176	210	0.355	0.617	0.368		0.359
1	2424	125	44.0	2.84	175	210	0.355	0.618	0.368		0.354
	2498	126	45.0	2.79	300	130	0.355	0.599	0.319		0.287
	2501	124	15 C	2.79	300	127	0.355	0.596	0.311		0.294
	2505	126	45.0	2.79	305	127	0.355	0.603	0.292		0.279
	2510	_	45.0	2.79	303	129	0.355	0.597	0.288		0.281
	2506		45.0	2.79	305		0.355		0.310		0.298
	2509		45.0	2.79	304		0.355				0.303
	2507		45.0	2.79	304		0.355		0.318		0.329
	2508		45.0	2.79	303		0.355		0.317		0.334

Appendix A2: Computer programs for data acquisition and control

```
File ADSUBS2.C
                             Global definitions
                              and subroutines
                                  for AQ2.C
                    Last modification: 27 February 1992
Modified for 2 A/D boards, May 1992
#define
           apply_toad_mask
raise_toad_mask
                                         /* digital output to apply load */
                                         /* digital output to relieve load */
#define
                              8
#define
                                         /* digital output to apply brake */
           brake_reel_mask
#define
                                         /* relieve load and apply brake */
           emerg_stop_mask 12
           num_chans 8
num_scans 2048
                                         /* 8 channels per board */
#define
                                         /* read each channel 2048 times */
/* continuous averaging parameter */
ddefine
#define
           num_avg 4
#define
           num_saved 512
                                         /* num_scans/num_avg */
                                         /* winder tension is board 1, channel 5 */
#define
           tenschan 5
                                         /* nip load is board 1, channel 6 */
#define
           Loadchan 6
#define contactload 120
                                         /* load cell output when roll contacts;
                                               525 for 711 rolls and 44 cylinder.
                                               -50 for 404 rolls and 4" cylinder,
                                         120 for 404 rolls and 2" cylinder*/
/* initiate A/D; 1 for hardware trigger */
#define
           trigger D
           board1 1
board2 2
#define
                                                 /* board addresses */
#define
           lift_time 1.0
#define
                                         /* time in sec to lift at end of exprent;
                                           longer for large rolls/small cylinder */
           motor_bits_per_vel 2.143033 /* 4.043256 bits/(m/min), 60:30 */
motor_bits_offset -20.45419 /* 14.19529 bits, 60:30 */
#define
#define
                                  44.045435 /* trim bits/(m/min) */
249.16481 /* trim bits offset */
#define
                                 44.045435
           trim_bits_per_vel
#define
           trim_bits_offset
                                  2235.597 /* bits/kN/m, loading, log fit */
/* for 711 rolls only */
#define
          load_bits_per_kN
                                              /* bits, loading, log fit */
/* bits/kN/m, lifting, log fit */
/* bits, lifting, log fit */
/* limits on analog output */
                                  -9574.39
#define
          load_bits_offset
#define
          relief_bits_per_kN -325.18
          relief_bits_offset 1578.596
min_load_bits 0
#define
#define
#define
           max_load_bits 4095
          load_sel_min 0
load_sel_max 3
#define
                                               /* indices for load select array */
#define
#define
                                               /* initial load selection */
          load_start 2
#define
           vel_sel_min 0
                                               /* indices for velocity select array */
#define
          vel_sel_max 5
#define
           min_trim_bits 0
                                               /* limits on analog output */
#define
          max_trim_bits 4095
#define
                                               /* vel increment, bits */
           d_t 110
                                               /* time increment, ms */
#define
#define
           minrevs 25
                                               /* min acq time in roll revs */
#define
          xerror 15
                                               /* x, y coords for error messages */
#define
           yerror 60
#define
           THIR 950
                                               /* target winder tension in bits */
#define
          smoothfact 0.4
                                               /* filtering for tension control */
#define gain_adjust 1.50
#define ctr1 1
#define ctr5 5
                                               /* factor to alter gains on the fly */
                                               /* counter addresses */
          source1 6
source5 10
#define
#define
#define NumEdgeIn 10
                                               /* number of rising edges */
#define NumEdgeOut 10
int huge *exp_buff1,
                                                /* raw data buffers */
huge *exp_buff2;
double outbuff1(num_saved)(num_chans), /* converted data buffers */
          outbuff2[num_saved] [num_chans],
          avg_data1 [num_chans] ,
          avg_data2(num_chans);
int
          motor_bits,
                                                /* main motor output, bits */
          trim bits,
                                               /* trim motor output, bits */
```

 $\cong c_{i,k_{1}}.$

```
load_bits,
                                              /* load control output, bits */
          LHid,
          vel_array(6) = (90,175,300,500,900,1100),
                                               /* index for velocities */
          vel_set = 0,
          load_array[6] = (10,20,40,65,15,50),
         /* for 404 rolls and 4" cylinder:

(70,60,95,135,175,175), "/

toad_array_bin[6] = (1450,600,1100,3000,1250,1825),
                                    /* for 404 rolls and 4" cylinder:
                                        (700,971,1253,1865,2346,2346),*/
1), /* 8 for relief, 1 for load */
          load_dir[6] = (8,8,1,1,8,1),
                                               /* index for loading */
          load_sel = 2,
          mask = 0,
                                               /* digital output mask */
                                    /* number of runs saved in a file */
          firstrum = 1,
          num_samples,
                                    /* num_scans * num_channels */
          endfile = 11111,
                                    /* flag at the end of a file */
          endblock = 10000,
                                     /* flag at the end of a block */
                                     /* end of acquisition flag */
          AOStatus,
                                     /* error handling */
          AgerrNum,
          tenserr[3],
                                     /* tension deviation from setpoint */
                                     /* averaging for tension control */
          avg_tens = 0,
          loaderr[3],
          monitor_buff[2*num_chans]: /* most recent block of data */
int buff_index, /* index to most recent block of data */
unsigned int buff_index,
                                     /* count from ingoing square wave */
               VinCount,
                                     /* count from outgoing square wave */
               VoutCount,
                                     /* count in ms from clock */
txt", /* parameter file name */
          ClockCount; /* c
par_file[12] = "info2.txt",
char
          calibr_file[12] = "calibr2.txt",
control_file[12] = "control.txt", /* PID gains */
          firstprn[50] = "c:\\phd\\calcs\\data\\", /* converted data files */
prn[5] = ".prn", /* converted data file: */
          dta[5] = ".dta",
                                     /* raw data files */
          sum[5] = ".sum";
                                     /* summary file */
                                     /* calibration constants */
flost
          cons [3] [16],
          Τp,
                                     /* PID controller coefficients */
          Τi,
           Td.
           Lp,
          Li,
           Ld,
           deltal.
                                                   /* PID control dt */
                                                   /* speed change due to stretch */
           deltaV,
           tensadj[3],
                                                   /* PID controller coefficients */
           loadadj [3],
           DRollin = 0.06335
                                                   /* idler roll diameters */
           DRollOut = 0.06334;
struct
           dosdate_t ddate;
struct
           dostime_t dtime;
struct
                                               /* laboratory ambient temp */
             float dry_bulb,
                    dew_pt,
                                               /* laboratory dew point */
                                               /* Laboratory barometric pressure */
                    baro,
                                               /* laboratory relative humidity */
                    rel_hum,
                                               /* basis weight */
                    bas_wt,
                                               /* sequence of supply voltages */
                    volt_seq[7],
                    calroll_d1,
                                               /* calender reel diameters in m */
                    calroll_d2,
                                               /* velocity in meters/min */
/* file identification number */
                  i_vel;
file_no,
             int
                    catroli_n1,
                    calroll_n2,
                                               /* calender roll id numbers */
                                               /* sequence channels read */
                    chan_seq1(8),
                                               /* sequence of gains */
                    gain_seq1[8],
                                                /* sequence channels read */
                    chan_seq2(81,
                                               /* sequence of gains */
                    gain_seq2[8],
```

```
Subroutines begin here
int motor()
                                 /* ramp motor up to desired speed */
{
  int dd_v,
      exitNum = 0,
      eltime,
                                 /* time spent in loop */
      over, old_time,
                                 /* time when loop starts */
                                 /* time when loop finishes */
      new_time,
                                 /* desired velocity in bits */
      new_motor_bits:
  new_motor_bits = (data.i_vel*motor_bits_per_vel+motor_bits_offset)*1;
if (new_motor_bits>4095) new_motor_bits=4095;
  if (motor_bits>new_motor_bits) dd_v = -d_v;
  else dd v = d v;
while (((motor bits <= (new_motor bits-d v))
|| (motor_bits >= (new_motor_bits+d v))) && (exitNum 1= 27))
    eltime=0;
    if ((AGerrNum = CTR_EvCount(board1,5,4,1))!=0) /* start timer */
      DAG_error("CTR_EvCount",1000);
    if ((AGerrNum = CTR_EvRead(board1,5,&over,&old_time))!=0)
      DAO_error("CTR_EvRead", 1010);
    while ((eltime<d_t) && (exitNum != 27))
    €
      if (kbhit()) exitNum = getch();
      if (exitNum != 27)
       get_last_data();
display_last_data();
trim_control(exitNum);
        load_control(exitNum);
        exitNum = 0;
        if ((AgerrNum = CTR_EvRead(board1,5,&over,&new_time))!=0)
         DAO_error("CTR_EvRead", 1020);
        eltime = new_time-old_time;
    motor_bits=motor_bits+dd_v;
    if ((AgerrNum = AO_Write(board1,0,motor_bits))!=0)
      DAG_error("AO_Write", 1030);
  if ((AGerrNum = CTR_Stop(board1,5))!=0)
                                                  /* stop the timer */
    DAO_error("CTR_Stop", 1040);
    if ((AgerrNum = A0_Write(board1,0,new_motor_bits))!=0)
     DAG_error("AO_Write",1050);
  return exithum:
int wait()
                   /* wait for a keypress while reading and controlling */
  int exitNum=0;
  _settextposition(2,1);
  printf(" \"A\" - begin ACQUISITION
                                                                    и):
```

```
_settextposition(20,1);
  printf("requested speed reached of %6.2f m/min
                                                                        ",data.i_vel);
  while ((exitNum!=27) && (exitNum!=97) && (exitNum!=65))
  ( if (kbhit()) exitNum = getch();
else exitNum = 0;
    get_last_data();
display_last_data();
trim_control(exitNum);
    load_control(exitHum);
  return exithum;
int express()
                             /* acquire the data and store in two arrays */
  int exitNum = 0,
      scan int;
  if ((AGerrNum = AI_Clear(board1))!=0)
  DAG_error("A1_Clear 1",1120);
if ((AGerrNum = A1_Clear(board2))!=0)
  DAQ_error("AI_Clear 2",1130);
if ((AQerrNum = DAQ_Config(board1,trigger,0))!=0)
  DAO_error("DAO_Config 1",1140);
if ((AQerrNum = DAO_Config(board2,trigger,0))!=0)
  DAO_error("DAO_Config 2",1150);
if ((AGerrNum = DAO_DB_Config(board1,0))!=0)
    DAG_error("DAG_DB_Config 1",1160);
  if ((AGerrNum = DAG_DB_Config(board2,0))!=0)
DAG_error("DAG_DB_Config_2",1170);
  buff_index = 1;
  _settextposition(20,1);
 printf("acquiring data . . . . scan_int = 1000000.0/data.scan_rate;
                                                                                   ");
  if ((AGerrNum = SCAN_Setup(board1, num_chans, data.chan_seq1,
         data.gain_seq1))(=0)
    DAQ_error("SCAN_Setup 1",1180);
  if ((AgerrNum = SCAN_Setup(board2,num_chans,data.chan_seq2,
         data.gain_seq2))!=0)
    DAQ_error("SCAN_Setup 2", 1190);
  if ((AGerrNum = SCAN_Start(board1,exp_buff1,num_samples,1,50,1,
         scan int)) (= 0)
  C DAQ_error("SCAN_Start 1",1200);
    exitNum=4;
    return exithum:
  if ((AgerrNum = SCAN_Start(board2,exp_buff2,num_samples,1,50,1,
         scan_int)) != 0)
  C DAQ_error("SCAN_Start 2",1210);
    exitNum=4;
    return exithum;
  while (buff_index < 2*num_chans)
    if ((AgerrNum = DAG_Check(board1, &AGStatus,&buff index))!=0)
      DAG_error("DAG_Check", 1220);
                 /* Check acquisition progress, display and control */
  C if((AgerrNum = DAG_Monitor(board),-1,0,num_chans,monitor_buff,
           &buff_index,&AQStatus))!=0)
      DAQ_error("DAQ_Monitor 1",1230);
    if((AGerrNum = DAG_Monitor(board2,-1,0,num_chans,
      &monitor_buff[num_chans],&buff_index,&AQStatus))!=0)
DAQ_error("DAQ_Monitor_2",1240);
    if (kbhit()) exitNum=getch();
    display_last_data();
trim_control(exitNum);
```

```
load control(exitNum);
  if (exitNum != 27) exitNum = 0;
) while ((!AQStatus) && (exitNum != 27));
   if ((AgerrNum = DAG_Clear(board1))1=0)
    DAG error("DAG Clear 1",1250);
   if ((AgerrNum = DAG Clear(board2))!=0)
    DAQ_error("DAQ_Clear 2",1260);
  return exithum;
int trim_control(Num) /* adjust trim as necessary */
int Num;
€
  int ii;
  if (Num > 0)
  { switch (Num)
    <
      case 56:
        Tp *= gain_adjust;
tensadj[0] = Tp + 0.5*Ti*deltaT + Td/deltaT;
tensadj[1] = -Tp + 0.5*Ti*deltaT - 2*Td/deltaT;
        tensad;[2] = Td*deltaT;
        break:
      case 53:
        Tp /= gain_adjust;
        tensadj[0] = Tp + 0.5*Ti*deltaT + Td/deltaT;
        tensadj[1] = -Tp + 0.5*Ti*deltaT - 2*Td/deltaT;
        tensadj[2] = Td*deltaT;
        break;
      case 57:
        Ti *= gain_adjust;
tensadj[0] = Tp + 0.5*Ti*deltaT + Td/deltaT;
tensadj[1] = -Tp + 0.5*Ti*deltaT - 2*Td/deltaT;
        tensad;[2] = Td*deltaT;
        break;
      case 54:
        Ti /= gain_adjust;
tensadj[0] = Tp + 0.5*Ti*deltaT + Td/deltaT;
        tensadj[1] = -Tp + 0.5*Ti*deltaT - 2*Td/deltaT;
        tensadj[2] = Td*deltaT;
        break;
     _settextposition (23,40);
    printf("Trim gains: Tp= %f, Ti = %f", Tp, Ti);
  avg_tens * smoothfact*avg_tens + (1-smoothfact)*monitor_buff[tenschan];
  tenserr[0] = TMid - avg_tens;
                                               /* compute deviation */
  for (ii=0;ii<3;ii++)
    trim_bits -= tensadj[ii]*tenserr[ii];
                                              /* compute corection */
  if (trim_bits > max_trim_bits) trim_bits = max_trim_bits;
  else if (trim_bits < min_trim_bits) trim_bits = min_trim_bits;
tenserr[2] = tenserr[1); /* save current deviation . . . */
                                          /* . . . for next loop */
  tenserr[1] = tenserr[0];
  if ((AgerrNum = Ag_Write(board2, 0, trim_bits))!=0)
    DAG_error("AO_Write", 1300);
                                           /* Output new value to trim */
   settextposition(22,62);
 printf("trimming %d bits
                              ",trim_bits);
/* adjust load as necessary */
int load_control(Num)
int Num;
 int li:
```

```
float load_corr = 0.0250;
 if (Num > 0 )
 ( switch (Num)
     case 43:
        if (load_dir[load_sel] == 1) /* adjust gains if neccessary */
          load_bits *= (1+load_corr); /* remove once controller fully */
        else
                                              /* debugged. */
          load_bits *= (1-load_corr);
        break;
      case 45:
        if (load_dir[load_sel] == 1)
          load_bits *= (1-load_corr);
          load_bits *= (1+load_corr);
        break;
    3
    _settextposition(22,40);
    printf("loading %d bits; ",load_bits);
  loaderr[0] = LMid - monitor_buff[loadchan];
 if (load_dir[load_sel] = 8)
( for (ii=0; ii<3; ii++)</pre>
      load_bits -= loadadj[ii]*loaderr[ii];
 if (load_bits > max_load_bits) load_bits = load_bits;
else if (load_bits < min_load_bits) load_bits = min_load_bits;</pre>
  loaderr[2] = Toaderr[1];
  loaderr[1] = loaderr[0];
  if ((AgerrNum = A0_Write(board1, 1, load_bits))!=0)
    DAO_error("AO_Write",1350);
get_last_data()
int ii:
  for (ii=0; ii<num_chans;ii++)</pre>
    if ((AGerrNum = Al_Read(board1, ii, data.gain_seq1[ii],
 Emonitor_buff[[i]]))!=0)
DAG_error("AI_Read 1",1420); /*
for (ii=num_chans; ii<(2*num_chans);ii++)
                                             /* read 8 channels from each board */
    if ((AGerrNum = Al_Read(board2,(ii-num_chans),
                       data.gain_seq2[ii-num_chans],&monitor_buff[ii]))!=0)
    DAG_error("AI_Read 2",1430);
 return;
                        /* convert and display latest data */
display_last_data()
int ii:
   settextposition(13,1);
  for (ii=0; ii<(2*num_chans);ii++)
    printf("%4.0f ",(cons[0][ii]+monitor_buff[ii]*(cons[i][ii]
          +cons(2)[ii]*monitor_buff(ii)));
  _settextposition(21,30);
  printf("%6u",buff_index);
  return;
```

```
<del>********************************</del>
write_data()
                                         /* update parameter file */
  FILE *x;
   int i.
        endblock < 10000:
  if ((x=fopen(par_file,"wt")) != NULL)
C fprintf(x,"X5.3f\n",data.dry_bulb);
fprintf(x,"X5.3f\n",data.dew_pt);
fprintf(x,"X5.3f\n",data.baro);
fprintf(x,"X5.3f\n",data.rel_hum);
fprintf(x,"X5.3f\n",data.rel_hum);
     fprintf(x, "%5.3f\n", data.calroli_d1);
fprintf(x, "%d\n", data.calroli_n1);
     fprintf(x, "X5.3f\n",data.calroll_d2);
fprintf(x, "Xd\n",data.calroll_n2);
    fprintf(x, "\n");
for (i=0; i<num_chans; i++)</pre>
       fprintf(x,"%3d\t",data.gain_seq1[i]);
     fprintf(x,"\n");
for (i=0; i<num_chans; i++)</pre>
       fprintf(x,"%1d\t",data.chan_seq2[i]);
     fprintf(x,"\n");
for (i=0; i<num_chans; i++)
  fprintf(x,"%3d\t",data.gain_seq2[i]);</pre>
     fprintf(x,"\n");
for (i=0; i<7; i++)</pre>
       fprintf(x,"%1d\t",data.sens_seq[i]);
    fprintf(x,"\n");
fprintf(x,"\n");
for (i=0; i<7; i++)
    fprintf(x,"\n\6.2f\t",data.volt_seq[i]);
fprintf(x,"\n\6.2f\data.i_vel);</pre>
     fprintf(x,"\n%.4d\n",data.file_no);
  eise
  { perror("write error: can't open INFO2.TXT for writing");
    exit(1):
  fclose(x):
  return;
/* display parameters */
data_screen()
  _clearscreen(_GCLEARSCREEN);
printf(" World's Narrowest Calender Data Acquisition Program\n");
  printf("C. THESE VALUES ARE CORRECT\n");
  printf("1. Dry bulb temperature
                                                                  \t\t\t%5.3f
                                            \n",data,dry bulb);
                                                                 \t\t\t%5.3f
  printf("2. Dew point temperature
                                            \n",data.dew_pt);
                                                                  \t\t\t%5.3f
  printf("3. Barometric pressure
                                            \n",data.baro);
                                                                 \t\t\t%5.3f
  printf("4. Relative humidity
                                            \n",data.rel_hum);
                                           diameter (cm): \t\t\t%5.3f
  printf("5. top calender reel -
                                            \n",data.calroli_d1);
```

```
printf("6.
                                 - id #:
                                                 \t\t\t %d
 \n",data.calroll n1);
printf("7. bottom calender real -diameter (cm): \t\t\t\5.3f
                                 printf("8.
                                 \n",data.calroll_n2);
\t\t\t%5.3f \n",data.bas_wt);
 printf("9. basis weight (Kg):
 printf("10. paper type:
                                                 \t\t\t %s\n",data.pap_type);
 printf("11. scan rate-one channel (scan/sec):
                                                 \t\t\t%8.2lf\n",data.s_rate);
 printf("12.
                      -all channels (scan/sec): \t\t\t%8.2if
                   \n", data.scan_rate);
Edit INFO.TXT to alter Gain vector
 printf(#\n--
                                                             --\n\n\n");
 printf("17. initial velocity (m/min):
                                                 \t\t\t%6.2f\n",data.i_vel);
 printf("18. file number:
                                                 \t\t\t %4d",data.file_no);
  _settextposition(22,1);
  return;
int change(ch1)
                              /* alter selected parameter */
                             /* first character of inputted choice */
char chi:
 int ascii,x,i;
char temp[100],y[3];
                             /* final choice to input data */
 stropy(y, Echi);
 ch1 = getche();
 ascii = toascii(ch1);
 if (ascii != 13)
 { stropy (y + 1, &ch1);
   ch1 = getch();
 x=atoi(y);
 if ((x>0) && (x<=18))
 clear((x+2),40,38);
_settextposition((x+2),66);
   if ((x!=11) & (x!=12))
     scanf("%s", temp );
    if (x==1)
     data.dry_bulb=atof( temp );
    else if (x==2)
      data.dew_pt = atof( temp ):
    else if (x==3)
     data.baro = atof( temp );
    else if (x==4)
      data.rel_hum = atof( temp );
    else if (x==5)
     data.cairoli_d1 = atof( temp );
    else if (x==6)
     data.calroll_n1 = atoi( temp );
   else if (x==7)
     data.cairoli_d2 = atof( temp );
   else if (x==8)
     data.calroll_n2 = atoi( temp );
   else if (x==9)
     data.bas_wt = atoi( temp );
   else if (x==10)
     strcpy(data.pap_type, temp);
   else if (x==11)
     scanf("%lf",&data.s_rate);
    else if (x==12)
     scanf("%lf",&data.scan_rate);
   else if (x==17)
    { data.i_vel = mtof( temp );
     data.scan_rate = 20*ceit(0.05*num_scans*data.i_vel/
                        (3.1416*60*data.calroll_d1*minrevs));
     clear1((14),40,38);
```

```
settextposition((14),66):
       printf("%8.2lf",data.scan_rate);
     else if (x==18)
      data.file_no = atoi( temp ):
  return;
int txconvert (filenum)
                                        /* convert raw binary file . . . */
                                     /* . . . to .PRN files */
int filenum:
  FILE *dtabin;
  int i,
end(8),
      temp,
trial = 0,
      ch = 96:
  char file_[10]:
   settextposition (21,1);
  printf("Reading file: %s
if ((dtabin = fopen(file_name, "rb")) == NULL)
                                                                 ",file_name);
  { printf("error opening file");
    exit(1);
  itoa(filenum,file_,10);
fread (end, sizeof(end), 1, dtabin);
  do
  { trial++;
    ch++;
    clear1(22,1,78);
    printf("Reading trial %d; ",trial);
    fread (&ddate, sizeof(ddate), 1, dtabin);
    fread (Edtime, sizeof(dtime), 1, dtabin);
    fread (&data, sizeof(data), 1, dtabin);
fread (cons, sizeof(cons), 1, dtabin);
    fread (&deltaY, sizeof(deltaY), 1, dtabin);
fread (&VinCount, sizeof(VinCount), 1, dtabin);
fread (&VoutCount, sizeof(VoutCount), 1, dtabin);
    fread (EClockCount, sizeof(ClockCount), 1, dtabin);
    for (i=0; i<4; i++)
fread (&temp, sizeof(temp), 1, dtabin);
    fread (&exp_buff1[0], sizeof(exp_buff1[0]), num_samples, dtabin);
fread (&exp_buff2[0], sizeof(exp_buff2[0]), num_samples, dtabin);
fread (end, sizeof(end), 1, dtabin);
    average();
  svetxt(file_,ch);
) while (end[3] != 11111);
  fclose(dtabin);
/* average and de-multiplex data */
average()
  int divisor.
      ch,
      SC,
  printf ("adding: ");
  sc = 0;
  for (ch=0;ch<num_chans;ch++) /* First channel scan */
  { outbuff1[sc][ch]=0.0;
```

```
outbuff2[sc][ch]=0.0:
  avg_data1 [ch] = 0.0;
  avg_data2[ch]=0.0;
  for (i=0; i<num_avg+1; i++) /* data from first num_avg+1 scans */
  { outbuff1[sc][ch] += exp_buff1[i*num_chans+ch];
 outbuff2[sc][ch] += exp_buff2[i*num_chans+ch];
    avg_data1[ch] += exp_buff1[i*num_chans+ch];
avg_data2[ch] += exp_buff2[i*num_chans+ch];
for (sc=1;sc<(num_saved-1);sc++) /* Scans 2 through 2047 */
( for (ch=0;ch<num_chans;ch++)
  { outbuff1[sc] [ch]=0.0;
    outbuff2[sc] [ch]=0.0;
    for (i=-num_avg;i<num_avg+1;i++) /* data near current one */
       outbuff1[sc][ch] += exp_buff1[(num_avg*sc+i)*num_chans+ch];
       outbuff2[sc][ch] += exp_buff2[(num_avg*sc+i)*num_chans+ch];
       avg_data1[ch] += exp_buff1[i*num_chans+ch];
avg_data2[ch] += exp_buff2[i*num_chans+ch];
  )
)
sc = num_saved-1;
for (ch=0;ch<num_chans;ch++)
                                           /* last channel scan */
{ outbuff1[sc][ch]=0;
  outbuff2[sc][ch]=0;
  for (i=-num_avg;i<num_avg;i++)</pre>
                                           /* add data from last scans */
    outbuff1[sc][ch] += exp_buff1[(num_avg*sc+i)*num_chans+ch];
    outbuff2[sc][ch] += exp_buff2[(num_avg*sc+i)*num_chans+ch];
avg_data1[ch] += exp_buff1[i*num_chans+ch];
avg_data2[ch] += exp_buff2[i*num_chans+ch];
3
printf ("averaging; ");
                                         /* divide sums by appropriate divisors */
divisor = num_avg+1;
sc = 0:
for (ch=0;ch<num_chans;ch++)
                                          /* and convert to real variables */
{ outbuff1[sc] [ch] = cons[0] [ch]
                     +outbuff1[sc][ch]*(cons[1][ch]+
                     cons[2] [ch] *outbuff1[sc] [ch] /divisor)/divisor;
  outbuff2[sc][ch] = cons[0][ch+num chans]
                     +outbuff2[sc][ch]*(cons[1][ch+num_chans]+
                     cons[2][ch+num_chans]*outbuff2[sc][ch]/divisor)/divisor;
  avg_data1[ch] = avg_data1[ch]/num_scans;
  avg_data2(ch) = avg_data2(ch)/num_scans;
divisor = 2*num_avg+1;
for (sc=1;sc<(num_saved-1);sc++)
 ( for (ch=0;ch<num_chans;ch++)
   ( outbuff1[sc][ch] = cons[0][ch]
                       +outbuff1[sc][ch]*(cons[1][ch]+
                       cons[2][ch]*outbuff1[sc][ch]/divisor)/divisor;
     outbuff2[se] [ch] = cons[0] [ch+num_chans]
+outbuff2[se] [ch]*(cons[1] [ch+num_chans]+
                        cons[2][ch+num_chans]*outbuff2[sc][ch]/divisor)/divisor;
  )
divisor = 2*num_avg;
 sc = num_saved-1;
 for (ch=0;ch<num_chans;ch++)
 { outbuff1(sc)(ch) = cons(0)(ch)
                     +outbuff1[sc][ch]*(cons[1][ch]+
                      cons[2] [ch] *outbuff1[sc] [ch]/divisor)/divisor;
   outbuff2[sc] [ch] = cons[0] [ch+num_chans]
```

```
+outbuff2(sc) [ch]*(cons[1] [ch+num_chans]+
                     cons[2] [ch+num_chans] *outbuff2[sc] [ch]/divisor)/divisor;
)
svetxt(file1,chext)
                             /* save converted data after averaging */
char file1[50];
char chext;
  FILE *dtatxt;
  int length,
      pdest,
      divisor,
      ch,
      SC,
      j, k;
  char ext[5]
       file[50];
  strcpy (firstprn,"c:\\phd\\calcs\\data\\");
  strcpy (prn, H.prn*);
strcpy(file, file1);
  pdest = strcpy( firstprn + 18, file );
  length = strlen(firstprn);
  pdest = strcpy( firstprn + length, &chext );
  length = strlen(firstprn);
pdest = strcpy( firstprn + length, prn );
  pdest = strcpy( file , firstprn);
  dtatxt = fopen(file, "wt");
  save_data(file,chext,dtatxt);
  for (sc=0;sc<num_saved;sc++)</pre>
  { for (ch=0;ch<num_chans;ch++)
      fprintf(dtatxt,"%4.8f, ",outbuff1[sc][ch]);
    for (ch=0;ch<num_chans;ch++)
  fprintf(dtatxt,"%4.8f, ",outbuff2[sc][ch]);</pre>
    fprintf (dtatxt,"\n");
  fclose(dtatxt);
save_data(file,ext,dtatxt)
                                /* save parameters with converted data */
char file[50],ext;
file *dtotxt;
int J, k;
  printf("writing file: %s\n",file);
  fprintf(dtatxt,"\"DATE: \"\"Xu/Xu/Xu\"\t\"TIME: \"\"Xu/Xu/Xu\"\n",ddate.day,
             fprintf(dtatxt,"\"Dry bulb \"
  fprintf(dtatxt,"\"Dew point \"
  fprintf(dtatxt,"\"Barometer \"
fprintf(dtatxt,"\"R.H. \"
                               \" X5.3f \n",data.calroll_d1);
\" Xd \n".data.calroll_n1\"
  fprintf(dtatxt,"\"TrollD
  fprintf(dtatxt,"\"id
fprintf(dtatxt,"\"BrollD
                               \" %d \n",data.calroll_n1);
\" %5.3f \n",data.calroll_d2);
  fprintf(dtatxt,"\"id
                               \" %d \n",da:a.catrott_n2);
  fprintf(dtatxt,"\"BW \" %5.3f \n",data.bas_wt);
fprintf(dtatxt,"\"paper type\"\" %s\"\n",data.pap_type);
  fprintf(dtatxt,"\"CRate
                               \" %8.21f\n",data.s_rate);
  fprintf(dtatxt, M\"SRate
                               \"%8.21f, %8.21f\n",data.scan_rate,
```

```
data.scan_rate/num_avg);
  fprintf(dtatxt, "\"channel v:\"");
  for (j=0; j<8; j++)
fprintf(dtatxt, =%1d
                               ".data.chan seq1[i]):
  for (j=0; j<8; j++)
    fprintf(dtatxt,"%1d
                                ",data.chan_seq2[j]);
  fprintf(dtatxt, "\n\"gain v: \"");
for (j=0; j<8; j++)</pre>
    fprintf(dtatxt,"23d
                             ",data.gain_seq1[j]):
  for (j=0; j<8; j++)
fprintf(dtatxt,"%3d
                             ",data.gain_seq2[j]);
  fprintf(dtatxt,"\n\"sensor v: \"");
  for (j=0; j<7; j++)
fprintf(dtatxt,"%1d</pre>
                                ",data.sens_seq[j]);
  fprintf(dtatxt, "\n\"Vin, Vout, Clock, DeltaV:\" %5u %5u %5u %6.6f",
                         VinCount, VoutCount, ClockCount, deltaV);
  fprintf(dtatxt,"\n\"\vel\" %6.2f %6.2f",data.i_vel,data.
fprintf(dtatxt,"\n\"file:\" %.4d \" %c \"\n",data.file_no, ext);
fprintf(dtatxt,"\n\"calibration \" \n");
                                       %6.2f %6.2f",data.i_vel,data.i_vel);
  for (j=0;j<3;j++)
  for (k=0;k<(2*num_chans);k++)</pre>
      fprintf(dtatxt, "%.8f,",cons[]][k]);
    fprintf(dtatxt, "\n");
  fprintf(dtatxt,"\n\"NO\"\t\"N1\"\t\"C2\"\t\"C3\"\t\"I4\"\t\"T5\"\t\"
                                                        L 6 \ " \ t \ " V 7 \ " \ t " ) ;
fprintf(dtatxt,"\"N10\"\t\"N11\"\t\"C12\"\t\"C13\"\t\\"X14\"\t\"X15\"\t\"
                   X16\"\t\"V17\"\t\n");
/*********************/
int DAQ_error(err_msg, location) /* general error message handler */
char err_msg[30];
int location;
   settextposition(xerror,yerror);
  printf("%s error # %d at %d",err msg, AQerrNum, location);
  return;
int eexit(int x)
                         /* emergency exit subroutine */
{
  time_t etstart, etstop;
  float emerg_time:
  if ((AgerrNum = DiG_Out_Port(boardi,1,emerg_stop_mask))!=0)
    DAG_error("DIG_Out_Port",3000);
  if ((AGerrNum = AO_Write(board1,0,0))!=0)
    DAG_error("AO_Write 1",3010);
  if ((Agerrhum = AO_Write(board1,1,4000))!=0)
    DAQ_error("AO_Write 1",3020);
  if ((AGerrNum = AG_Write(board2,0,0))!=0)
  DAR_error("AO_Write 2",3030);
if ((AGernNum = AO_Write(board2,1,4000))!=0)
    DA9_error("A0_Write 2",3040);
  motor_bits= 0:
  _settextposition(xerror,yerror);
  printf ("Emergency stop");
  _settextposition(xerror+1,yerror);
  printf ("error # %d",x);
  _settextposition(xerror+2,yerror);
if (x==1)
    printf("subroutine EXPRMNT");
  else if (x==2)
```

```
printf("subroutine WAIT");
  else if (x==3)
   printf("subroutine MOTOR");
  else if (x==4)
   printf("subroutine SCAN Start");
  else if (x==5)
   printf("subroutine MAIN");
  settextposition (xerror+3, yerror);
 time (&etstart);
                              /* Wait */
 do
  ( time (Letstop);
   emerg_time = difftime(etstop,etstart);
 while (emerg_time < lift_time);
if((AGerrNum = AO_Write(board1,1,brake_reel_mask))!=0)/* Load off, brake on */</pre>
   DAG_error("AO_Write Load",3050);
 printf("Press a key to exit:");
 return x;
int clear(row,column,num)
                              /* Clear n lines */
int row,
column,
   num;
 int i,j;
 _settextposition(row,column);
for (j=column;j<79;j++)
  printf(" ");
 printf("\n");
 for (i=1;i<num;i++)</pre>
 for (j=0;j<79;j++)
printf(" ");</pre>
   printf("\n");
 3
 _settextposition(row,column); return;
int clear((row,column,num))
                              /* Clear n spaces */
int row,
   column.
   num;
<
 int i;
  _settextposition(row,column);
 for (i*D;i<num;i++)
printf("");
  _settextposition(row.column);
 return;
/* Subroutines end here */
```

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```
/* AQ2.C: Main acquisition program;
/* Data Acquisition and control program for World's Warrowest Calender */
/* Written by Miles Sherman, August 1990 */
/* Modified for 2 A/D boards, May 1992, by Thomas Browne */
/* Automatic trim control installed, June/July 1992, by Thomas Browne */
/* Counters for draw installed July 1992, by Thomas Browne */
   Global variable definitions are in the file AGSUBS2.C
                                  /* header file for LABDRIVER routines */
#include <doslbdrv.h>
#include <comio.h>
#include <stdio.h>
#include <stdlib.h>
#include <dos.h>
#include <malloc.h>
#include <graph.h>
#include <ctype.h>
#include <float.h>
#include <math.h>
#include <time.h>
#include "aqsubs2.c"
main()
  FILE
         *outfbin,
                                   /* data file */
          *inftxt,
                                   /* parameter file */
          *outftxt:
  double ei_time;
  . int
          cont,
          escii.
                                   /* ascii value */
          i,j,
          press.
                                   /* value read from mip load cell */
          errNum,
          pdest.
          length,
          overflow,
                                    /* carriage return */
          cr;
                                    /* first character entered for vel */
   char
          quit,
          vel_char,
                                    /* next character entered for vel */
          vel_temp[100],
                                    /* temporary velocity as char string */
          firstdta[5] = "f:\\":
  time_t tstart,
          tstop:
   float DeltaVFact;
 Main program begins here
   _clearscreen(_GCLEARSCREEN);
num_samples = num_scans*num_chans;
                                              /* allocate data buffers */
   exp_buff1 = NULL:
   exp_buff2 = NULL;
   if (!(exp_buff1=(int huge *) halloc ((unsigned long)num_samples,
                            sizeof(int))))
   { perror ("error allocating memory for buffer 1");
     printf("ending program...");
     exit(1);
   if (!(exp_buff2=(int huge *) halloc ((unsigned long)num_samples,
                              sizeof(int))))
   { perror ("error allocating memory for buffer 2");
     printf("ending program...");
     exit(1);
   if ((inftxt=fopen(par_file,"rt")) (= NULL)
```

1

```
{ fscenf(inftxt, "Xf", &data.dry_bulb); /* Open and read INFO2.TXT */
   fscenf(inftxt, "%f", &data.dry_bulb);
fscenf(inftxt, "%f", &data.dew_pt);
fscenf(inftxt, "%f", &data.dew_pt);
fscenf(inftxt, "%f", &data.rel_hum);
fscenf(inftxt, "%f", &data.calroll_d1);
fscenf(inftxt, "%f", &data.calroll_n1);
fscenf(inftxt, "%f", &data.calroll_n2);
fscenf(inftxt, "%f", &data.calroll_n2);
fscenf(inftxt, "%f", &data.bes_ut);
fscenf(inftxt, "%f", &data.s_rate);
fscenf(inftxt, "%lf", &data.s_rate);
fscenf(inftxt, "%lf", &data.scen_rate);
for (i=0: i<num_chans: i++)</pre>
    for (i=0; i<num_chans; i++)
fscanf(inftxt,"%d",&data.chan_seq1[i]);</pre>
    for (i=0; i<num_chans; i++)
       fscanf(inftxt,"%d",&data.gain_seq1[i]);
    for (i=0; i<num_chans; i++)
fscanf(inftxt,"%d",&data.chan_seq2[i]);</pre>
    for (i=0; i<num_chans; i++)
fscanf(inftxt,"%d",&data.gain_seq2[i]);</pre>
    for (i=0; i<num_chans-1; i++)
       fscanf(inftxt,"%d",&data.sens_seq[i]);
    for (f=0; i<num_chans-1; i++)
    fscanf(inftxt,"Xf",&data.volt_seq[i]);
fscanf(inftxt,"\nXf",&data.i_vel);
fscanf(inftxt,"\nXd",&data.file_no);
    fclose(inftxt);
3
else
{ perror("error opening INFO2.TXT");
    exit(1):
if ((inftxt=fopen(calibr_file,"rt")) != NULL)
( for (i=0; i<3; i++)
   ( for (j=0;j<num_chans;j++)
  fscanf(inftxt,"%f",&cons[i][j]);</pre>
    for (i=0;i<3;i++)
    ( for (j=num_chans; j<(2*num_chans); j++)
fscanf(inftxt, "%f", &cons[i][j]);</pre>
3
else
{ perror("error opening CALIBR2.TXT");
   exit(1):
if ((inftxt=fopen(control_file,"rt")) != NULL)
if ((inftxt=fopen(control_file,
    fscanf(inftxt,"%f",&Tp);
    fscanf(inftxt,"%f",&Ti);
    fscanf(inftxt,"%f",&Id);
    fscanf(inftxt,"%f",&Lp);
    fscanf(inftxt,"%f",&Li);
    fscanf(inftxt,"%f",&Ld);
    fscanf(inftxt,"%f",&deltaT);
    fslossfinftxt,"%f",&deltaT);
                                                        /* Open and read CONTROL.TXT */
    fclose(inftxt);
)
else
{ perror("error opening CONTROL.TXT");
    exit(1):
                                                                                             /* PID constants */
tensadj[0] = Tp + 0.5*Ti*deltaT + Td/deltaT;
tensadj[1] = -Tp + 0.5*Ti*deltaT - 2*Td/deltaT;
tensadj[2] = Td*deltaT;
loadadj[0] = Lp + 0.5*Li*deltaT + Ld/deltaT;
 loadadj[1] = -Lp + 0.5*Li*deltaT - 2*Ld/deltaT;
 Loadedj(2) = Ld/deltaT;
```

```
for (i=0; i<3; i++)
{ tenserr[i] = 0:
  loaderr[i] = 0:
_dos_getdate(&ddate);
 dos gettime(&dtime):
if ((AgerrNum = AO_Config(board1,0,1,10.0,0))1=0)
DAQ error("AO Config 1, channel 0",10);
if ((AQerrNum = AO Config(board1,1,1,10.0,0)))!=0)
DAQ error("AO Config 1, channel 1",20);
if ((AQerrNum = AO Config(board2,0,1,10.0,0))!=0)
  DAO_error("AO_Config 2, channel 0",30);
if ((AGerrNum = AO_Config(board2,1,1,10.0,0))!=0)
   DAQ_error("AO_Config 2, channel 1",40);
if ((AGerrHum = AI_Config(board1,0,5,0))!=0)
  DAG_error("AI_Config 1",50);
if ((AGerrHum = AI_Config(board2,0,10,1))!=0)
DAQ_error("AI_Config 2",60);
if ((AGerrNum = DIG_Prt_Config(board1,0,0,0))!=0)
  DAG_error("DIG_Prt_config 1",70);
if ((AgerrNum = DIG_Prt_Config(board1,1,0,1))(=0)
DAG_error("DIG_Prt_Config 1",80);
if ((AGerrNum = DIG_Prt_Config(board2,0,0,0))1=0)
  DAG_error("DIG_Prt_Config 2",90);
if ((AgerrNum = DIG_Prt_Config(board2,1,0,1))!=0)
  DAG_error("DIG_Prt_Config 2",100);
motor_bits = 0;
trim bits = 0;
load_bits = 0;
DeltaVFact = DRollOut*NumEdgeIn/(DRollIn*NumEdgeOut);
if ((AGerrNum = AO_Write(board1,0,motor_bits))!=0)/* Motor off */
  DAG_error("AO_Write Main",110);
if ((AGerrNum = AO_Write(board1,1,load_bits))!=0) /* load off */
DAQ_error("AO_Write Load",120);
if ((AQerrNum = AO_Write(board2,0,trim_bits))!=0) /* Trim motor off */
  DAQ_error("AO_Write Trim",130);
mask = mask & 11;
                                                       /* leave load alone */
if ((AGerrNum = DIG_Out_Port(board1,1,mask))!=0) /* Release brake */
  DAQ_error("DIG_Out_Port",140);
data_screen();
                                         /* displays current parameters */
_settextposition(22,1);
printf("
                   Enti- choice:\n\t>");
cr = getche();
while ((cr != 'c') && (cr != 'C'))
{ change(cr);
                                           /* change selected parameter */
clear(23,10,2);
  cr = getche();
_settextposition(24,1);
printf ("Lowering roll . . .");
mask = raise_load_mask;
load_bits = 1575;
if ((AGerrNum = AO_Write(board1,1,load bits))!=0)
DAQ_error("AO_Write",150);
if ((AQerrNum = DIG_Out_Port(board1,1,mask))!=0)
                                                                /* pressure . . . */
  DAG_error("DIG_Dut_Port", 160);
  if ((AGerrNum = Al_Read (board1, loadchan, data.gain_seq1[loadchan],
                               &press))!=0)
      DAQ_error("AI Read",170);
while ((press > contactload) && (!kbhit())); /* until contact or keypress. */
mask = 0:
if ((AGerrNum = DIG_Out_Port(board1,1,mask))!=0)
                                                              /* then release */
  DAG_error("DIG_Out_Port",180);
if ((AgerrNum = Ag_Write(board1,1,0)))=0)
  DAG_error("AO_Write",190);
_clearscreen(_GCLEARSCREEN);
```

```
printf(" um um um um N/m N/m kN/m m/m um
                                                              um.
      m/m<sup>H</sup>);
do
                     /* experiment loop begins here */
cont = 1;
   settextposition(2,1);
  printf(" \"C\" - CONTINUE
                                                                     H);
  _settextposition (4,52);
  printf("Velocity: %8.2f m/min",data.i_vel);
  _settextposition (5,52);
  printf("Load: %6d kN/m", Load_array(load_sel));
  _settextposition(4,1);
  printf("Press V for Velocity, L for Load; ");
  _settextposition(5,1);
  printf("Press GREY + or - to increment or decrement:");
  quit = 0;
  while (quit != 'C') /* Get new sheet speed and load */
  ( while (!kbhit())
    { get_last_data();
                                     /* read latest data */
                                     /* display */
      display_last_data();
      if (firstrum > 1)
       trim_control(0);
                                    /* control */
                                     /* key has been hit */
                                     /* get the key */
/* convert to upper case */
    quit = getch();
    ascii * toupper(quit);
    quit = ascii;
    if ((ascii == 'L') || (ascii == 'V'))
{ _settextposition (5,45);
      printf("%1c",ascii);
      while (!kbhit())
      { get_last_data();
                                     /* read latest data */
                                     /* display */
        display_last_data();
                                     /* while waiting for second key */
        if (firstrum > 1)
          trim_control(0);
                                     /* control */
                                     /* key has been hit */
                                     /* get the key */
      quit = getch();
      if (quit == '+')
      { _settextposition (5,46);
        Printf ("+");
        if (ascil == 'L')
        { load sel++:
          if (load_sel > load_sel_max) load_sel = load_sel_max;
        else
        { vel_set++;
          if (vel_set > vel_set_max) vel_set = vel_set_max;
          data.i_vel = vel_array[vel_sel];
data.scan_rate = 20*ceil(0.05*num_scans*data.i_vel/
                            (3.1416*60*data.calroll_d1*minrevs));
       )
     )
      else if (quit = '-')
      { _settextposition (5,46):
        printf("-");
        if (ascii == 'L')
        { load_set--:
          if (load_set < toad_set_min) load_set = load_set_min;</pre>
        )
        else
        { vel_sel--;
          if (vel_set < vel_set_min) vel_set = vel_set_min;</pre>
          data.i_vel = vel_array(vel_sel);
data.scan_rate = 20*ceil(0.05*num_scans*data.i_vel/
                             (3.1416*60*data.calroll_d1*minrevs));
        )
```

```
settextposition (4,52);
    printf("Velocity: %8.2f m/min ",data.i_vel);
     settextposition (5,52);
    printf("Load: %6d kN/m ",load_array[load_sel]);
     settextposition (5,45);
    printf(" ");
  3
                     /* End get sheet speed and load*/
load_bits = load_array_bin[load_set];
clearl (4,1,45);
clearl (5,1,45);
_dos_gettime(&dtime):
settextposition (1,1);
printf("\"ESC\" - EMERGENCY EXIT");
settextposition(7,1);
printf("File number: %4d; Run: %4d printf("Asq. rate: %8.2if Hz
                                          \n",data.file_no,firstrum);
                                          \n", data.scan_rate);
settextposition (20,1);
printf("starting motor....");
errkum = 0:
mask = apply_load_mask;
                                                           /* apply load */
if ((AGerrNum = AO_Write(board1,1,load_array_bin[load_sel]))!=0)
  DAQ_error("AO_Write",280);
if ((AgerrNum = DIG_Out_Port(board1,1,mask))|=0)
DAO_error("DIG_Out_Port",290);
if ((cont = motor()) != 27) /* if motor() returns without ESC pressed */
( ii ((AgerrNum = A0_Write(board1,1,load_array_bin[load_sel]))!=0)
    DAG error("AD Write", 292);
  if ((AgerrNum = DIG_Out_Port(board1,1,load_dir[load_sel]))!=0)
  DAQ_error("DIG_Out_Port",294);
if ((cont = wait()) != 27)  /* if wait() returns without ESC pressed */
  { if ((AgerrNum = CTR_EvCount(board1,ctr5,4,1))!=0)
      DAQ_error("CTR_EvCount clk",300);
                                                    /* Start clock */
     if ((AGerrNum = CTR_EvCount(board2,ctr1,source1,1))1=0)
      DAG_error("CTR_EvCount Vin",310);
                                                  /* start in counter */
    if ((AgerrNum = CTR_EvCount(board2,ctr5,source5,1))!=0)
      DAQ_error("CTR_EvCount Vou",320);
                                                   /* Start out counter */
     if ((cont = exprmnt())==27)
      errNum = eexit(1);
                              /* if exprmnt() returns with ESC */
    if ((AGerrNum = CTR_EvRead(board1,ctr5,&overflow,&ClockCount))!=0)
    DAG_error("CTR_EvRead clk",330); /* Read clock */
if ((AGerrNum = CTR_EvRead(board2,ctr1,&overflow,&VinCount))!=0)
      DAO_error("CTR_EvRead Vin", 340);
                                                   /* Read pulses in */
     if ((AgerrNum = CTR_EvRead(board2,ctr5,&overflow,&VoutCount))!=0)
      DAQ_error("CTR_EvRead Vou",350);
                                                    /* Read pulses out */
    if ((AGerrNum = CTR_Stop(board1,ctr1))!=0)
    DAG_error("CTR_Stop clk",360);
if ((AGerrNum = CTR_Stop(board2,ctr1))!=0)
                                                    /* Stop clock */
      DAO_error("CTR_Stop Vin",370);
                                                    /* Stop in count */
    if ((AgerrNum = CTR_Stop(board2,ctr5))!=0)
      DAG_error("CTR_Stop Vou",380);
                                                    /* Stop out count */
    if (VinCount > 0)
      deltaV = 100*(VoutCount*DeltaVFact/VinCount-1);
    else deltaV = 9999;
    get_last_data();
                                     /* read latest data */
    display last_data();
trim_control(0);
                                     /* display */
                                          /* control */
  else errNum = eexit(2);
                               /* if wait() returns with ESC */
else errNum = eexit(3);
                               /* if motor() returns with ESC */
                             /* save whatever data was acquired */
if (errNum < 2)
{ itoa(data.file_no, file_name,10);
  pdest *strcpy( firstdta + 3, file_name );
  length=strlen(firstdta);
```

```
pdest =stropy( firstdta + length, dta );
    pdest=strcpy( file_name , firstdta);
     _{\text{settextposition}(2\overline{0},1)};
    printf("writing file Xs, data set # Xd", file_name, firstrum);
     if ((outfbin = fopen(file_name, Mab*)) |= NULL )
    ( for (i=0; i<8; i++)
         furite (&endblock, sizeof(endblock), 1, outfbin);
      fwrite (&ddate, sizeof(ddate), 1, outfbin);
fwrite (&dtime, sizeof(dtime), 1, outfbin);
fwrite (&data, sizeof(data), 1, outfbin);
fwrite (cons, sizeof(cons), 1, outfbin);
       furite (&deltaV, sizeof(deltaV), 1, outfbin);
       fwrite (&VinCount, sizeof(VinCount), 1, outfbin);
fwrite (&VoutCount, sizeof(VoutCount), 1, outfbin);
       furite (&ClockCount, sizeof(ClockCount), 1, outfbin);
       for (i=0; i<4; i++)
         furite (Sendblock, mizeof(endblock), 1, outfbin);
       furite (Eexp_buff1[0], sizeof(exp_buff1[0]), num_samples,
                                                           outfbin);
       furite (Eexp_buff2[0], sizeof(exp_buff2[0]), num_samples,
                                                           outfbin);
       fclose(outfbin);
    clear(20,1,1);
    firstrum++;
  if (terrNum)
  { _settextposition(1,1);
    Printf(" \"E\" - END \"")

printf(" \"C\" - CHANGE VELOCITY OR LOAD AND CONTINUE");
                                                                 \n");
    while ((cont != 69) && (cont != 67))
    ( get_last_data();
       trim_control(0);
       display_last_data();
       if (kbhit())
       { cont = getch();
         ascii = toascii(cont);
         if (ascii == 27)
                                    /* ASCII 69 = 'E', for exit */
         { cont = 69;
          errNum = eexit(5);
         else cont = toupper(cont);
      )
  ) else cont = 69;
                                  /* force exit if errNum */
                                  /* end experiment loop unless C pressed */
) while (cont == 67);
data.file_no++;
write_data();
                                  /* If this is not the first run */
if (firstrun>1)
{ outfbin = fopen(file_name,"ab");
  for (i=0; i<8; i++)
    furite(&endfile,sizeof(endfile),1,outfbin);
  fclose(outfbin);
if (errNum)
                                  /* If emergency exit . . . */
                                  /* don't release brake until keypressed */
( do () while ((kbhit());
  cont = getch();
                                  /* Otherwise normal exit, lift roll */
else
{ _settextposition(21,1);
  printf("Raising upper roll . . .
                                                      \n"):
  time (Etstart);
  if ((AgerrNum = DIG_Out_Port(board1,1,(raise_load_mask)))!=0)
    DAG_error("DIG_Out_Prt",480);
  if ((ADerrNum = AO_Write(board1,1,4000))!=0)
    DAG_error("AO_Write",490);
                                        /* Wait */
```

```
( time (&tstop);
        el_time = difftime(tstop,tstart);
    while (el_time < lift_time);
 if((AGerrNum = AO_Write(board1,0,0))1=0)
                                                                                              /* Main motor off */
    DAG_error("AO_Write Main",500);
if((AGerrNum = AO_Write(board1,1,0))!=0)

DAG_error("AO_Write(board2,0,0))!=0)

DAG_error("AO_Write(board2,0,0))!=0)

DAG_error("AO_Write(Tribm",520);

[6.4.]
                                                                                              /* Load off */
                                                                                              /* Trim motor off */
if ((AGerrNum = DIG_Out_Port(board1,1,0))!=0)
    DAG_error("DIG_Out_Prt_1",530);
 if ((AperrNum = DIG_Out_Port(board2,1,0))!=0)
DAG_error("DIG_Out_Prt 2",540);
if ((Agernum = Al_Clear(board1))!=0)
DAG_error("Al_Clear 1",570);
if ((Agernum = Al_Clear(board2))!=0)
 DAG_error("AI_Clear 2",580);
data.file_no--;
 txconvert(data.file_no);
txconvert(data.file_no);
if ((outftxt = fopen(control_file,"wt")) != NULL)
( fprintf(outftxt,"%10.8f\n",Tp);
  fprintf(outftxt,"%10.8f\n",Td);
  fprintf(outftxt,"%10.8f\n",Lp);
  fprintf(outftxt,"%10.8f\n",Lp);
  fprintf(outftxt,"%10.8f\n",Li);
  fprintf(outftxt,"%10.8f\n",Ld);
  fprintf(outftxt,"%10.8f\n",dd);
  fprintf(outftxt,"%10.8f\n",deltaT);
  fplose(outftxt);
     fclose(outftxt);
                                                                /* frees allocated memory */
 hfree(exp_buff1);
hfree(exp_buff2);
printf("\n");
                                             END MAIN PROGRAM
```

Appendix A3: Constitutive equations for linear viscoelastic models in a rolling nip

A3.1.	Nip geometry							
A3.2.	Model definitions							
A3.3.	Equations describing the standard linear liquid							
	A3.3.1.	Relaxation and retardation functions						
	A3.3.2.	Explicit expressions for stress and strain						
A3.4.	Equations of	describing Burger's model						
	A3.4.1.	Relaxation and retardation functions						
	A3.4.2.	Explicit expressions for stress and strain						

A3.1 Nip geometry

Refering to Figure A3.1, a cartesian coordinate system is defined with origin at the intersection of the plane of the undeformed sheet and the centreline joining the rolls. The x-axis is positive in the MD, while the z-axis is positive downwards. The sheet midplane is assumed to be a plane of symmetry. Time is measured from the point of first contact between sheet and roll: t=0 when x=-a. Distances z_i , z_n and z_f are the initial, in-nip and final recovered sheet half-thicknesses, z(t) is the downward deflection of a point P(x,z) on the surface of the paper, $z_0 = z_i - z_n$ is the maximum value of z(t), and z_e is the half-thickness at the nip exit. Distances a and b are the ingoing and outgoing nip lengths; b_{max} is the largest possible value of b. Calendering variables are the roll radius R, sheet velocity V and line load L. If the roll radius R is much larger than z_0 -z(t), then

$$z(t) = \frac{Vt}{2R}(2a - Vt)$$
 [A3.1]

Strain is the change in thickness divided by the initial thickness:

$$\varepsilon_1(t) = \frac{Vt}{2Rz_i} (2a - Vt)$$
 [A3.2]

In-nip and permanent strains are defined as follows:

$$\varepsilon_{n} = \frac{z_{i} - z_{n}}{z_{i}}$$
 [A3.3]

$$\varepsilon_{\rm p} = \frac{z_{\rm i} - z_{\rm f}}{z_{\rm i}}$$
 [A3.4]

The ingoing and maximum outgoing nip lengths a and b_{max} are then

$$a^2 = 2Rz_i \varepsilon_n$$
 [A3.5]

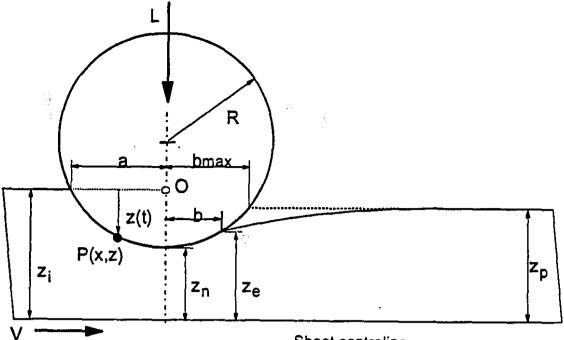


Figure A3.1 Geometry of a thin strip in a rolling nip.

$$b_{\text{max}}^2 = 2Rz_i (\varepsilon_n - \varepsilon_p)$$
 [A3.6]

The strain and strain rate while paper remains in contact with the rolls are then

$$\varepsilon_1(t) = \varepsilon_n \left(\frac{Vt}{a}\right) \left(2 - \frac{Vt}{a}\right)$$
 [A3.7]

$$\frac{\partial \varepsilon_1(t)}{\partial t} = \frac{2 \varepsilon_n V}{a} \left(1 - \frac{Vt}{a} \right)$$
 [A3.8]

The strain rate is thus a maximum at the nip entrance, when t=0:

$$\left(\frac{\partial \varepsilon_1(t)}{\partial t}\right)_{max} = \frac{2\varepsilon_n V}{a}$$
 [A3.9]

A3.2 Model definitions

The two models discussed here are the standard linear liquid and Burger's fourelement model. They are shown in Figures A3.2 and A3.3, and are composed of elastic elements G_M and G_K (with units Pa) and viscous elements η_M and η_K (with units Pa s).

The analysis is based on the following assumptions:

- 3. The web is a continuum in the z-direction.
- 4. Moduli G_i and viscosities η_j are material constants unaffected by stress, strain, or stress or strain rates.
- 5. MD and CD deformations are small; Poisson's ratio = 0.
- 6. Shear stresses are negligible.

Numerical solution methods for root-finding can fail when the order of the coefficients is significantly different from 1. Although the analysis here will not use these substitutions, the analysis can proceed using non-dimensional model parameters:

$$\eta_1 = \frac{V}{L} \eta_M \qquad \qquad \eta_2 = \frac{V}{L} \eta_K$$

$$G_1 = \frac{a}{L} G_M \qquad \qquad G_2 = \frac{a}{L} G_K$$
[A3.10]

Two other non-dimensional numbers are used. The Devorah number is the ratio of a material time to a typical processing time:

$$\xi = \frac{\tau}{a/V}$$
 [A3.11]

where the time constant τ is defined in the analysis. The nip length ratio β is

$$\beta = \frac{a+b}{a}$$
 [A3.12]

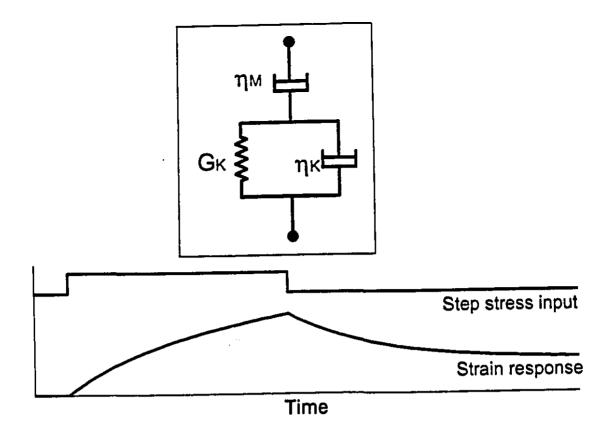


Figure A3.2 A linear liquid showing partial time-delayed recovery.

When b is small, β approaches 1; as b approaches b_{max} , β approaches β_{max} :

$$\beta_{\text{max}} = 1 + \sqrt{1 - \frac{\varepsilon_p}{\varepsilon_n}}$$
 [A3.13]

As the permanent strain ε_p approaches zero, β_{max} approaches 2. For linear materials the Boltzmann superposition principle states

$$\sigma(t) = -\int_{0}^{t} \psi(t-t') \frac{\partial \varepsilon_{1}(t')}{\partial t'} dt'$$
 [A3.14]

When t_e is the elapsed time at the nip exit, the strain recovery for $t > t_e$ is

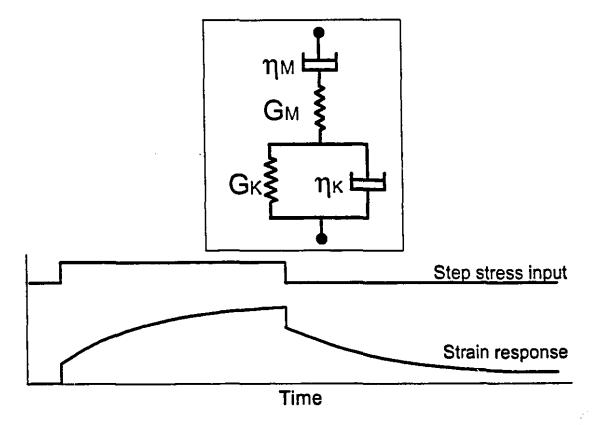


Figure A3.3 Burger's model showing instantaneous elasticity and delayed partial recovery.

$$\varepsilon_2(t) = \int_0^{tc} \phi(t-t') \frac{\partial \sigma(t')}{\partial t'} dt'$$
 [A3.15]

where

$$t_e = \frac{a+b}{V}$$
 [A3.16]

The relaxation function $\psi(t)$ and the retardation function $\phi(t)$ are functions of the model parameters, as detailed next.

A3.3 Equations describing the standard linear liquid

A3.3.1 Relaxation and retardation functions

The relaxation function $\psi(t)$ is derived following Findley et al. [27]. The total strain is the sum of the strains in the Kelvin element ε_K and in the viscous element ε_M :

$$\varepsilon = \varepsilon_{\rm M} + \varepsilon_{\rm K}$$
 [A3.17]

where the strains are related to the overall stress σ by

$$\sigma = \eta_M \dot{\varepsilon}_M$$

$$\sigma = \eta_K \dot{\varepsilon}_K + G_K \varepsilon_K$$
[A3.18]

Substitute A3.18 into A3.17, rearrange and transform to the Laplace domain:

$$\varepsilon(s) = \left[\frac{1}{\eta_M s} + \frac{1}{\eta_2 (s + G_K/\eta_K)}\right] \sigma(s)$$
 [A3.19]

Multiply through by $\eta_M \eta_K(s + G_K/\eta_M)$, solve for $\sigma(s)$, and transform back to the time domain:

$$\sigma(t) + \left(\frac{\eta_M + \eta_K}{G_K}\right) \dot{\sigma}(t) = \eta_M \dot{\varepsilon}(t) + \left(\frac{\eta_M \eta_K}{G_K}\right) \ddot{\varepsilon}(t)$$
 [A3.20]

Define the relaxation time and two temporary variables q_1 and q_2 :

$$\tau = \frac{\eta_M + \eta_K}{G_K}$$
 [A3.21]

$$q_1 = \eta_M \qquad q_2 = \frac{\eta_M \eta_K}{G_K} \qquad [A3.22]$$

Rewrite Equation A3.20 with a step strain input of magnitude ε_0 , then retransform (Tschoegl [79], p.569):

$$\sigma(t) + \tau \dot{\sigma}(t) = q_1 \varepsilon_0 \delta(t) + q_2 \varepsilon_0 \frac{\partial}{\partial t} \delta(t)$$
 [A3.23]

$$\sigma(s) = \varepsilon_{O} \left[\frac{q_{1} + q_{2}s}{1 + \tau s} \right]$$

$$= \varepsilon_{O} \left[\frac{q_{1}}{\tau} \left(\frac{\tau}{1 + \tau s} \right) + \frac{q_{2}}{\tau} \left(\frac{\tau s}{1 + ts} \right) \right]$$
[A3.24]

$$\sigma(t) = \varepsilon_0 \frac{q_2}{\tau} \left[\left(\frac{q_1}{q_2} - \frac{1}{\tau} \right) e^{-t/\tau} + \delta(t) \right]$$
 [A3.25]

Define the equivalent viscosity η_e and equivalent time constant τ_e :

$$\frac{q_2}{\tau} = \frac{\eta_M \eta_K}{\eta_M + \eta_K}$$

$$= \eta_E$$
[A3.26]

$$\frac{q_1}{q_2} - \frac{1}{\tau} = \frac{1}{\tau_K} - \frac{1}{\tau}$$

$$= \frac{1}{\tau_E}$$
[A3.27]

where the Kelvin time constant $\tau_K = \eta_K/G_K$. Then the relaxation function is the response to the step strain ε_0 :

$$\psi(t) = \eta_e \left[\frac{e^{-t/\tau}}{\tau_e} + \delta(t) \right]$$
 [A3.28]

The retardation function for this model is given by Findley et al. [27]:

$$\phi(t) = \frac{t}{\eta_M} + \frac{1}{G_K} (1 - e^{-t/\tau_k})$$
 [A3.29]

A3.3.2 Standard linear liquid: explicit expressions for stress and strain

Substitute the relaxation function, Equation A3.28, and the strain rate, Equation A3.8, into the Boltzmann integral, Equation A3.14, and integrate:

$$\sigma(t) = \int_{0}^{t} \eta_{e} \left[\frac{1}{\tau_{e}} \exp\left(\frac{-(t-t')}{\tau}\right) + \delta(t-t') \right] \frac{2 \epsilon_{n} V}{a} \left(1 - \frac{Vt'}{a} \right) dt'$$

$$= \frac{2 \epsilon_{n} V}{a} \frac{\eta_{e}}{\tau_{e}} e^{-t/\tau} \left[\int_{0}^{t} e^{t'/\tau} dt' - \frac{V}{a} \int_{0}^{t} e^{t'/\tau} t' dt' \right]$$

$$+ \frac{2 \epsilon_{n} V}{a} \eta_{e} \int_{0}^{t} \delta(t-t') \left(1 - \frac{Vt'}{a} \right) dt'$$
[A3.30]

A solution for the second integral can be found in published tables of integrals:

$$\int u e^{\alpha u} du = \frac{(\alpha u - 1)}{\alpha^2} e^{\alpha u} + C$$
 [A3.31]

The third integral can be solved by noting that $\delta(t-t')$ is different from 0 only when t = t'. Therefore

$$\int_{0}^{\pi} \delta(t-t') f(t') dt' = f(t) \int_{0}^{\pi} \delta(t-t') dt'$$

$$= f(t)$$
[A3.32]

and

$$I_3 = \left(1 - \frac{Vt}{a}\right)$$
 [A3.33]

Performing the other two integrations gives

$$\sigma(t) = \frac{2\varepsilon_{n}V}{a} \frac{\eta_{M}^{2}}{\eta_{M} + \eta_{K}} \left[(1+\xi)(1-e^{-t/\tau}) - \frac{Vt}{a} \right] + \frac{2\varepsilon_{n}V}{a} \left(1 - \frac{Vt}{a}\right) \frac{\eta_{M}\eta_{K}}{\eta_{M} + \eta_{K}}$$
[A3.34]

where the Devorah number ξ is as defined in Equation A3.11. The stress rate is therefore

$$\frac{\partial \sigma(t)}{\partial t} = \frac{2 \varepsilon_n V}{a} \left[\frac{\eta_e^2}{\eta_K \tau_K} (1 + \xi) e^{-t/\tau} - \eta_M \frac{V}{a} \right]$$
 [A3.35]

Substitute A3.29 and the stress rate in the nip, A3.35, into A3.15 and integrate:

$$\varepsilon_{2}(t) = \int_{0}^{t_{c}} \left[\frac{t - t'}{\eta_{M}} + \frac{1 - e^{-(t - t')/\tau_{K}}}{G_{K}} \right] \frac{\partial \sigma(t')}{\partial t'} dt'$$
 [A3.36]

Collect terms:

$$\varepsilon_{2}(t) = \left[\frac{t}{\eta_{M}} + \frac{1}{G_{K}}\right] \int_{0}^{t_{k}} \frac{\partial \sigma(t')}{\partial t'} dt' - \int_{0}^{t_{k}} \left[\frac{t'}{\eta_{M}} + \frac{e^{-t/\tau_{k}} e^{t'/\tau_{K}}}{G_{K}}\right] \frac{\partial \sigma(t')}{\partial t'} dt'$$
[A3.37]

The first integral is zero, since

$$\int_{0}^{t_{e}} \frac{\partial \sigma(t)}{\partial t} dt = \sigma(t_{e}) - \sigma(0)$$
 [A3.38]

Evaluating the second integral gives an expression with a time-dependent term and a constant term:

$$\varepsilon_2(t) = C_0 - C_1 e^{-V^{\tau_k}}$$
 [A3.39]

where the constant term is equal to the fully recovered strain ε_p :

$$C_{0} = \varepsilon_{p}$$

$$= \frac{2\varepsilon_{n}V}{a} \frac{\eta_{M}}{G_{K}} \left[(1+\xi) - (1+\xi) \left(\frac{\beta}{\xi} + 1 \right) e^{\beta/\xi} - \frac{a\eta_{M}\beta^{2}}{2VG_{K}} - \xi_{K} (e^{\beta/\xi_{K}} - 1) \right]$$
[A3.40]

and where

$$C_1 = 2\varepsilon_n \left(\frac{\eta_M}{\eta_M + \eta_K}\right)^2 \frac{\xi(1+\xi)}{1-\xi/\xi_K} \left(e^{-\frac{\beta}{\xi}\left(1-\frac{\xi}{\xi_K}\right)} - 1\right)$$
 [A3.41]

A3.4 Equations describing Burger's model

A3.4.1 Relaxation and retardation functions

The relaxation function is derived by Findley et al. [27]:

$$\psi(t) = K_1 e^{-\psi \tau_1} - K_2 e^{-\psi \tau_2}$$
 [A3.42]

where K_1 , K_2 , τ_1 and τ_2 are functions of the model parameters G_M , G_K , η_M , and η_K :

$$p_{1} = \frac{\eta_{M}}{G_{M}} + \frac{\eta_{K}}{G_{K}} + \frac{\eta_{M}}{G_{K}} \qquad (r_{1}, r_{2}) = \frac{p_{1} \pm A}{2p_{2}}$$

$$p_{2} = \frac{\eta_{M} \eta_{K}}{G_{M} G_{K}} \qquad (\tau_{1}, \tau_{2}) = \left(\frac{1}{r_{1}}, \frac{1}{r_{2}}\right)$$

$$q_{1} = \eta_{M} \qquad (\xi_{1}, \xi_{2}) = \left(\frac{V \tau_{1}}{a}, \frac{V \tau_{2}}{a}\right) \qquad [A3.43]$$

$$q_{2} = \frac{\eta_{M} \eta_{K}}{G_{K}} \qquad K_{1} = \frac{q_{1} - q_{2} r_{1}}{A}$$

$$A = \sqrt{p_{1}^{2} - 4p_{2}} \qquad K_{2} = \frac{q_{1} - q_{2} r_{2}}{A}$$

The retardation function $\phi(t)$ is also derived by Findley et al [27]:

$$\phi(t) = \frac{1}{G_{M}} + \frac{t}{\eta_{M}} + \frac{1 - e^{-t/\tau_{E}}}{G_{K}}$$

$$= \frac{1}{G_{M}} \left[1 + \frac{t}{\tau_{M}} + \frac{G_{M}}{G_{K}} \left(1 - e^{-t/\tau_{E}} \right) \right]$$
[A3.44]

where the Maxwell and Kelvin time constants au_{M} and au_{K} are given by

$$\tau_{M} = \frac{\eta_{M}}{G_{M}} \qquad \qquad \tau_{K} = \frac{\eta_{K}}{G_{K}} \qquad [A3.45]$$

A3.4.2 Burger's model: explicit expressions for stress and strain

The derivation proceeds as for the standard linear liquid, Section A3.3.2, with the appropriate response functions substituted. The stress profile in the nip is obtained by substituting the relaxation function, Equation A3.42, and the strain rate, Equation A3.8, into the Boltzmann equation, A3.14:

$$\sigma(t) = -2\varepsilon_n \left[K_1 \xi_1 (1 + \xi_1) (1 - e^{-t/\tau_1}) - K_2 \xi_2 (1 + \xi_2) (1 - e^{-t/\tau_2}) - (K_1 \xi_1 - K_2 \xi_2) \frac{Vt}{a} \right]$$
[A3.46]

Differentiating the stress profile, Equation A3.46, and substituting into A3.15 along with the retardation equation, A3.44, gives the strain after the nip exit:

$$\varepsilon_{2}(t) = \frac{1}{G_{M}} \int_{0}^{t_{c}} \left[1 + \frac{t-t'}{\tau_{M}} + \frac{G_{M}}{G_{K}} \left(1 - e^{-(t-t')/\tau_{K}} \right) \right] \frac{\partial \sigma(t')}{\partial t'} dt' \quad [A3.47]$$

Separating terms in t from terms in t' gives

$$\epsilon_{2}(t) = \frac{1}{G_{M}} \left[1 + \frac{t}{\tau_{M}} + \frac{G_{M}}{G_{K}} \right] \int_{0}^{t} \frac{\partial \sigma(t')}{\partial t'} dt'$$

$$- \frac{1}{G_{M}} \int_{0}^{t_{E}} \left[\frac{t'}{\tau} + e^{-t/\tau_{K}} e^{t'/\tau_{K}} \right] \frac{\partial \sigma(t')}{\partial t'} dt'$$
[A3.48]

As for the standard linear liquid, the first integral is zero. Evaluating the second integral gives

$$\varepsilon_2(t) = C_0 - C_1 e^{-t/\tau_E}$$
 [A3.49]

where the constant term is equal to the fully recovered strain $\varepsilon_{\rm p}$:

$$C_{0} = \varepsilon_{p}$$

$$= \frac{2\varepsilon_{n}a}{V\eta_{M}} \left[K_{2}\xi_{2}(1+\xi_{2})(\beta+\xi_{2})e^{-\beta/\xi_{2}} - K_{1}\xi_{1}(1+\xi_{1})(\beta+\xi_{1})e^{-\beta/\xi_{1}} \right]$$

$$+ \left(K_{1}\xi_{1}^{2}(1+\xi_{1}) - K_{2}\xi_{2}^{2}(1+\xi_{2}) - (K_{1}\xi_{1} - K_{2}\xi_{2}) \frac{\beta^{2}}{2} \right]$$

and where

$$C_{1} = \frac{2\varepsilon_{n}}{G_{K}} \left[\frac{K_{1}\xi_{1}(1+\xi_{1})}{1-\xi_{1}/\xi_{K}} \left(e^{-\frac{\delta}{\xi_{1}}\left(1-\frac{\xi_{1}}{\xi_{k}}\right)} - 1 \right) - \frac{K_{2}\xi_{2}(1+\xi_{2})}{1-\xi_{2}/\xi_{K}} \left(e^{-\frac{\beta}{\xi_{2}}\left(1-\frac{\xi_{2}}{\xi_{k}}\right)} - 1 \right) - \xi_{k} \left(K_{1}\xi_{1} - K_{2}\xi_{2} \right) \left(1 - e^{\beta/\xi_{k}} \right) \right]$$
[A3.51]